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CRUSH1:

A SIMPLIFIED COMPUTER PROGRAM FOR IMPACT ANALYSIS
OF RADIOACTIVE MATERIAL TRANSPORT CASKS

July 1996

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CRUSH1: A Simplified Computer Program for Impact
Analysis of Radioactive Material Transport Casks

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In drop impact analyses for radioactive transport casks, it has become possible to perform them in detail by using interaction evaluation, computer programs, such as DYNA2D, DYNA3D, PISCES and HONDO. However, the considerable cost and computer time are necessitated to perform analyses by these programs. To meet the above requirements, a simplified computer program CRUSH1 has been developed. The CRUSH1 is a static calculation computer program capable of evaluating the maximum acceleration of cask bodies and the maximum deformation of shock absorbers using an Uniaxial Displacement Method (UDM).

The CRUSH1 is a revised version of the CRUSH. Main revisions of the computer program are as follows;

- (1) not only main frame computer but also work stations (OS UNIX) and personal computer (OS Windows 3.1 or Windows NT) are available for use of the CRUSH1 and
- (2) input data set are revised.

In the paper, brief illustration of calculation method using UDM is presented. The second section presents comparisons between UDM and the detailed method. The third section provides a use's guide for CRUSH1.

Keywords: Computer Program, Impact Analysis, Cask, Drop Impact, UDM, Transport Cask, Structural Analysis, Static Analysis

CRUSH1：放射性物質輸送容器の簡易衝突計算プログラム

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(1996年6月14日受理)

放射性物質輸送容器の落下衝突解析では、DYNA2D、DYNA3D、PISCESおよびHONDOのような詳細計算プログラムを用いて計算されている。しかし、これらの計算プログラムによる計算は、多くの計算費用と計算時間が必要とされる。このような背景から、簡易計算プログラムCRUSH1を開発した。CRUSH1は1次元変形法(UDM法)を用いた静的計算プログラムであり、輸送容器本体の最大加速度およびショックアブソーバの最大変形量を計算するものである。

CRUSH1はCRUSHの改良版であり、主要な改良点は下記の通りである。

- (1) 大型計算機以外にもワークステーション(OS UNIX)およびパーソナルコンピュータ(OS Windows 3.1またはWindows NT)によって使用できるプログラムが用意されている。
- (2) 入力データの一部が変更された。

本報告書はUDM法の説明、UDM法と詳細計算法とによる計算結果の比較およびCRUSH1の使用のための、入力データ等のユーザーガイドについても記述した。

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1. Introduction

In the drop impact analyses for radioactive transport casks, it has become possible to perform them in detail by using interaction evaluation computer programs, such as DYNA2D, DYNA3D, PISCES and HONDO. However, the considerable cost and computer time are necessitated to perform analyses by these programs. To meet the above requirements, a simplified computer program CRUSH1 as shown in Fig. 1.1 has been developed. The CRUSH1 is a static calculation computer program capable of evaluating the maximum acceleration of cask bodies and the maximum deformation of shock absorbers using a Uniaxial Displacement Method(UDM)⁽¹⁾ as shown in Fig. 1.2.

CRUSH1 is a revised version of CRUSH⁽²⁾. Main revisions of the computer program are as follows;

(1) not only main frame computer but also work stations(OS UNIX) and personal computer(OS DOS or Windows) are available for use of CRUSH1 and

(2) input data set are revised.

In the paper, brief illustration of calculation method using UDM is presented. The second section presents comparisons between UDM and the detailed method. The third section provides a use's guide for CRUSH1.

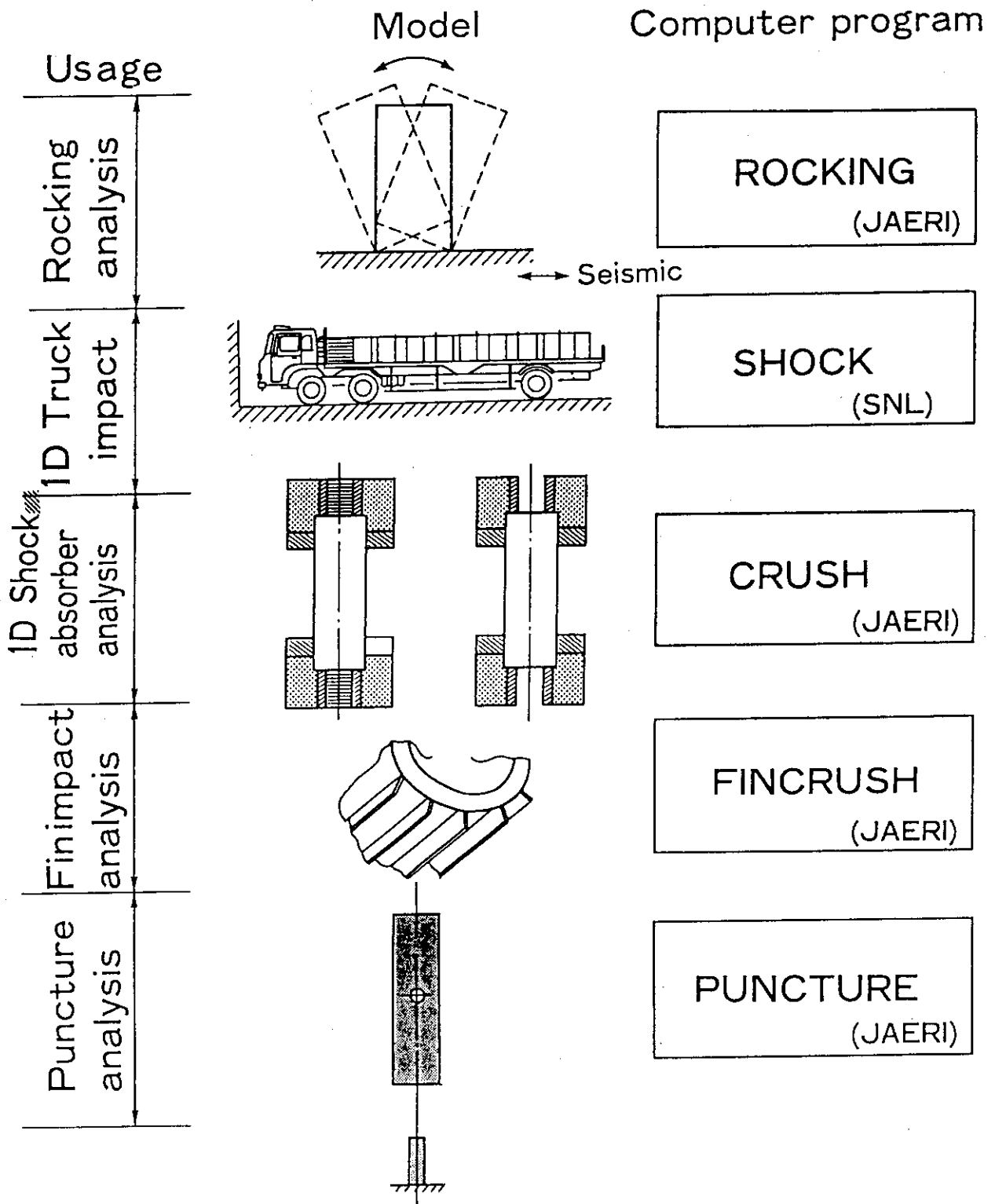


Fig.1.1 Simplified analysis computer programs

Strain : ϵ

$$\epsilon_i = \frac{\delta_i}{l_i}$$

Force : F_i

$$F_i = K_i \sigma_i(\epsilon_i) A_i$$

Total force : F

$$F = \sum_i F_i$$

Acceleration : α

$$\alpha = \frac{F}{M}$$

M : Mass, A : Area
 σ : Stress, K : Boundary condition constant

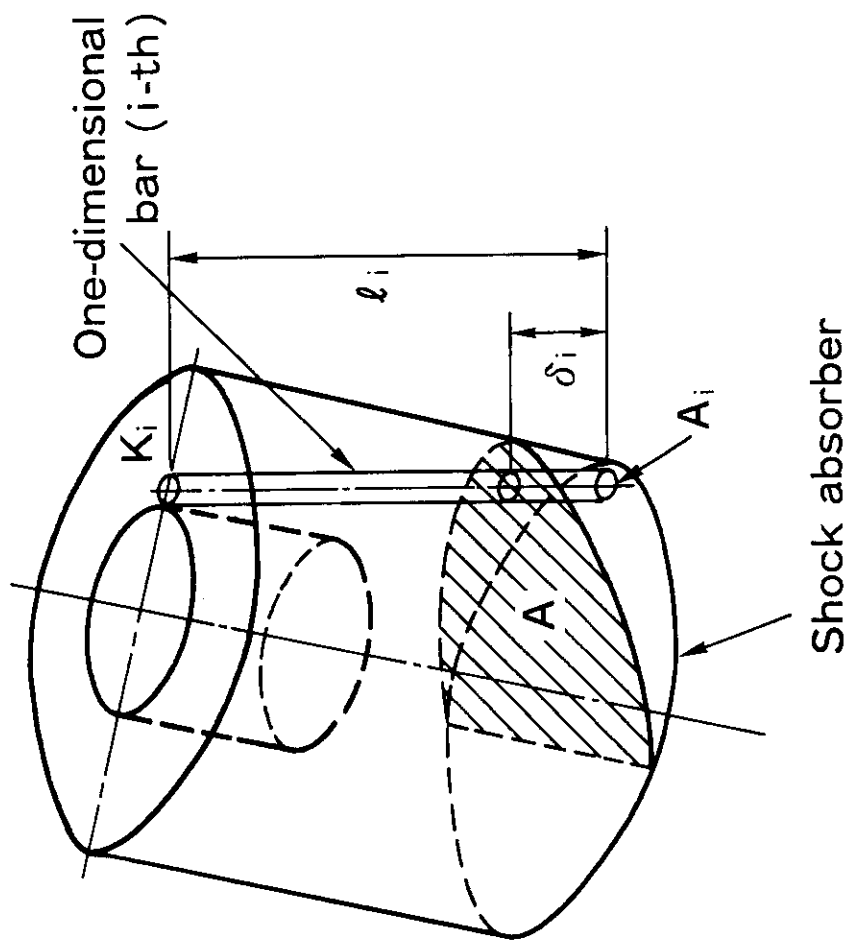


Fig.1.2 Uni-axial displacement method

2. Calculation equation

2.1 Calculation model

In the modeling of shock absorber, it is assumed that the shock absorber consists of three or four species of material as shown in Figs. 2.1 to 2.3. In the figures, the symbols of K_i ($i=1\sim 5$) indicates boundary condition constants which are estimated by an overpack stiffness and boundary condition of the shock absorber.

When the shock absorber deforms by the displacement $\Delta \ell$ in a corner drop as shown in Fig.2.4, the strain ε of an one-dimensional bar is

$$\varepsilon = \frac{\Delta \ell}{\ell} . \quad (2.1)$$

The force f of the one-dimensional bar is

$$f = K \sigma(\varepsilon) \Delta A . \quad (2.2)$$

where ℓ , σ and ΔA are the length, stress and area of the one-dimensional bar. K is the boundary condition constant. The total force F of the shock absorber is

$$F = \sum_i f_i . \quad (2.3)$$

The dissipated energy $E(\delta)$ can also obtained using equation similar to above Eq.(2.3)

$$E(\delta) = \int_0^\delta F d\ell . \quad (2.4)$$

Therefore, when a cask whose weight of W is dropped from a height H with an oblique angle θ , the maximum displacement of the shock absorber δ and the maximum acceleration of the cask body α are given as follows.

$$E(\delta) = \gamma \cdot W \cdot H \quad , \quad (2.5)$$

$$\alpha = \frac{F(\delta)}{M} \quad , \quad (2.6)$$

where γ is the ratio of the energy absorbed in the primary impact to the total energy absorbed in the primary and secondary impacts. M is mass of the cask.

2.2 Vertical drop impact

In the modeling a shock absorber in CRUSH, it is assumed that the shock absorber consists of three or four species of shock absorber materials as shown in Fig. 2.5. In the figure, the symbols of σ_A , σ_B , σ_C , σ_D indicate their materials determined by the geometries of the cask and the shock absorber.

When the shock absorber deforms by a displacement Z in a vertical drop, the impact load F and the dissipated energy E are given by following equations.

$$F = F_{01} + F_{12} + F_{23} + F_{34} + F_{45} \quad , \quad (2.7)$$

$$E = \int_0^Z F \, dZ \quad , \quad (2.8)$$

where, for the region $0 \sim R_1$

$$F_{01} = K_1 \pi \sigma_D \left(\frac{Z}{Z_2} \right) R_1^2 \quad , \quad (2.9)$$

for the region $R_1 \sim R_2$

$$F_{12} = K_2 \pi \sigma_A \left(\frac{Z}{Z_2} \right) (R_2^2 - R_1^2) \quad , \quad (2.10)$$

for the region $R_2 \sim R_3$

$$F_{23} = K_3 \pi \sigma_B \left(\frac{Z}{Z_2} \right) (R_3^2 - R_1^2) \quad , \quad (2.11)$$

for the region $R_3 \sim R_4$

$$\left. \begin{aligned} F_{34} &= K_4 \pi \sigma_B \left(\frac{\Delta Z_B}{Z_3} \right) (R_4^2 - R_3^2); \text{ (for } \sigma_B \text{ material) ,} \\ F_{34} &= K_4 \pi \sigma_C \left(\frac{\Delta Z_C}{Z_4 - Z_3} \right) (R_4^2 - R_3^2); \text{ (for } \sigma_C \text{ material) ,} \end{aligned} \right\} \quad (2.12)$$

$$\left. \begin{aligned} Z &= \Delta Z_B + \Delta Z_C , \\ \sigma_B \left(\frac{\Delta Z_B}{Z_3} \right) &= \sigma_C \left(\frac{\Delta Z_C}{Z_4 - Z_3} \right) , \end{aligned} \right\} \quad (2.13)$$

for the region $R_4 \sim R_5$

$$\left. \begin{aligned} F_{45} &= K_5 \pi \sigma_B \left(\frac{\Delta Z_B}{Z_3} \right) (R_5^2 - R_4^2); \text{ (for } \sigma_B \text{ material) ,} \\ F_{45} &= K_5 \pi \sigma_C \left(\frac{\Delta Z_C}{Z_4 - Z_3} \right) (R_5^2 - R_4^2); \text{ (for } \sigma_C \text{ material) ,} \end{aligned} \right\} \quad (2.14)$$

$$\left. \begin{aligned} Z &= \Delta Z_B + \Delta Z_C , \\ \sigma_B \left(\frac{\Delta Z_B}{Z_3} \right) &= \sigma_C \left(\frac{\Delta Z_C}{Z_4 - Z_3} \right) , \end{aligned} \right\} \quad (2.15)$$

$$0 \leq K_i \leq 1 .$$

σ_i ($i = A, B, C, D$) are stress in these material regions.

2.3 Horizontal drop impact

When the shock absorber deforms by the displacement Z in a horizontal drop as shown in Figs. 2.6 to 2.8, the impact load and the dissipated energy are given by following equations.

$$F = \sum F_{ij} = F_{01} + F_{12} + F_{23} + F_{34} + F_{45} + F_{56} + F_{67} + F_{78} + F_{89} , \quad (2.16)$$

$$F_{ij} = 2 \int_0^{R_i} \Delta F \, dx , \quad (2.17)$$

$$E = \int_0^y F \, dy , \quad (2.18)$$

where, for the region $0 \sim Z_1$ and $0 \sim R_1$

$$\left. \begin{aligned} \Delta F &= K_1 \sigma_D \left(\frac{\Delta y_1}{y_1} \right) dx \cdot Z_1; \text{ (for } \sigma_D \text{ material) ,} \\ \Delta F &= K_1 \sigma_A \left(\frac{\Delta y_2}{y_2} \right) dx \cdot Z_1; \text{ (for } \sigma_A \text{ material) ,} \\ \Delta F &= K_1 \sigma_B \left(\frac{\Delta y_3}{y_3} \right) dx \cdot Z_1; \text{ (for } \sigma_B \text{ material) ,} \end{aligned} \right\} \quad (2.19)$$

$$\left. \begin{aligned} y_1 &= 2\sqrt{R_1^2 - x^2} \quad , \\ y_2 &= 2(\sqrt{R_2^2 - x^2} - \sqrt{R_1^2 - x^2}) \quad , \\ y_3 &= 2(\sqrt{R_5^2 - x^2} - \sqrt{R_2^2 - x^2}) \quad . \end{aligned} \right\} \quad (2.20)$$

$$\left. \begin{aligned} U &= \Delta y_1 + \Delta y_2 + \Delta y_3 \quad , \\ \sigma_D \left(\frac{\Delta y_1}{y_1} \right) &= \sigma_A \left(\frac{\Delta y_2}{y_2} \right) = \sigma_B \left(\frac{\Delta y_3}{y_3} \right) \quad . \end{aligned} \right\} \quad (2.21)$$

$$\left. \begin{aligned} U &= y - h; \left\{ \begin{array}{l} U \leq 0, \text{ for } \Delta F = 0 \\ U > 0, \text{ for } \Delta F \neq 0 \end{array} \right\} , \\ h &= R_5 - \sqrt{R_5^2 - x^2} \quad . \end{aligned} \right\} \quad (2.22)$$

For the region $0 \sim Z_1$ and $R_1 \sim R_2$

$$\left. \begin{aligned} \Delta F &= K_1 \sigma_A \left(\frac{\Delta y_1}{y_1} \right) dx \cdot Z_1; \text{ (for } \sigma_A \text{ material) ,} \\ \Delta F &= K_1 \sigma_B \left(\frac{\Delta y_2}{y_2} \right) dx \cdot Z_1; \text{ (for } \sigma_B \text{ material) .} \end{aligned} \right\} \quad (2.23)$$

$$\begin{aligned} y_1 &= 2\sqrt{R_2^2 - x^2} \\ y &= 2(\sqrt{R_5^2 - x^2} - \sqrt{R_2^2 - x^2}) \quad . \\ U &= \Delta y_1 + \Delta y_2 \quad , \\ \sigma_A \left(\frac{\Delta y_1}{y_1} \right) &= \sigma_B \left(\frac{\Delta y_2}{y_2} \right) \quad . \end{aligned} \quad (2.25)$$

For the region $0 \sim Z_1$ and $R_2 \sim R_3$

$$\Delta F = K_1 \sigma_B \left(\frac{U}{y} \right) dx \cdot Z_1 \quad , \quad (2.26)$$

$$y = 2\sqrt{R_5^2 - x^2} \quad . \quad (2.27)$$

For the region $Z_1 \sim Z_2$ and $0 \sim R_5$ equations are similar as eqs.(2.9) through (2.25) when Z_1 changes to $(Z_2 - Z_1)$ and K_1 to K_4 .

For the region $Z_2 \sim Z_3$ and $0 \sim R_3$

$$\Delta F = K_3 \sigma_B \left(\frac{U}{y} \right) dx \cdot (Z_3 - Z_2), \quad (2.28)$$

where

$$y = \sqrt{R_5^2 - x^2} - \sqrt{R_3^2 - x^2}. \quad (2.29)$$

For the region $Z_2 \sim Z_3$ and $R_3 \sim R_5$

$$\Delta F = K_3 \sigma_B \left(\frac{U}{y} \right) dx \cdot (Z_3 - Z_2), \quad (2.30)$$

where

$$y = 2\sqrt{R_5^2 - x^2}. \quad (2.31)$$

For the region $Z_3 \sim Z_4$ and $0 \sim R_3$

$$\Delta F = K_4 \sigma_C \left(\frac{U}{y} \right) dx \cdot (Z_3 - Z_2), \quad (2.32)$$

where

$$y = \sqrt{R_5^2 - x^2} - \sqrt{R_3^2 - x^2}. \quad (2.33)$$

For the region $Z_3 \sim Z_4$ and $R_3 \sim R_5$

$$\Delta F = K_4 \sigma_C \left(\frac{U}{y} \right) dx \cdot (Z_4 - Z_3), \quad (2.34)$$

where

$$y = 2\sqrt{R_5^2 - x^2}. \quad (2.35)$$

For the region $Z_5 \sim Z_6$ equations are same as the region $Z_3 \sim Z_4$, the region $Z_6 \sim Z_7$ same as the region $Z_2 \sim Z_3$, the region $Z_7 \sim Z_8$ same as the region $Z_1 \sim Z_2$ and the region $Z_8 \sim Z_9$ same as the region $0 \sim Z_1$.

2.4 Oblique drop impact

In the modeling a shock absorber in CRUSH at a oblique drop impact, it is assumed that shock absorber consists of one specy of shock absorber material as shown in Fig. 2.3. The stress of a wooden

shock absorber can be written as following, including the effect of the wood grain angle θ .

$$\sigma_X = \sigma_A \cos^2\theta + \sigma_B \sin^2\theta \quad , \quad (2.36)$$

where σ_A , σ_B and σ_X are stresses in the wood whose grain direction is parallel, perpendicular and angle θ degree to the drop direction, respectively.

Let consider a cutway section of a shock absorber as shown in Figs. 2.9, 2.10 and 2.11. The impact load and the dissipated energy are given by following equations (see Fig. 2.11).

$$F_i = 2 \int_0^{XM} \Delta F dx \quad , \quad (2.37)$$

$$F = \int_S F_i dS \quad , \quad (2.38)$$

$$E = \int_0^y F dy \quad , \quad (2.39)$$

$$U = y-h_0-h; \quad \left\{ \begin{array}{l} U \leq 0, \text{ for } \Delta F = 0 \\ U > 0, \text{ for } \Delta F \neq 0 \end{array} \right\} \quad . \quad (2.40)$$

Sectional figures of the shock absorber in the case of oblique drop are shown in Figs. 2.12 and 2.13.

Let consider S-T coordinate as shown in Fig. 2.14.

The coordinates of points $P_1 \sim P_{12}$ and $Q_1 \sim Q_4$ are as follows.

$$\left. \begin{array}{l} P_1 \{ -R_5 \cos\theta \quad , \quad R_5 \sin\theta \} \\ P_2 \{ -R_1 \cos\theta \quad , \quad R_1 \sin\theta \} \\ P_3 \{ R_1 \cos\theta \quad , \quad -R_1 \sin\theta \} \\ P_4 \{ R_5 \cos\theta \quad , \quad -R_5 \sin\theta \} \\ P_5 \{ -R_3 \cos\theta + Z_2 \sin\theta, \quad R_3 \sin\theta + Z_2 \cos\theta \} \\ P_6 \{ -R_1 \cos\theta + Z_2 \sin\theta, \quad R_1 \sin\theta + Z_2 \cos\theta \} \\ P_7 \{ R_1 \cos\theta + Z_2 \sin\theta, \quad -R_1 \sin\theta + Z_2 \cos\theta \} \\ P_8 \{ R_3 \cos\theta + Z_2 \sin\theta, \quad -R_3 \sin\theta + Z_2 \cos\theta \} \\ P_9 \{ -R_5 \cos\theta + Z_4 \sin\theta, \quad R_5 \sin\theta + Z_4 \cos\theta \} \\ P_{10} \{ -R_3 \cos\theta + Z_4 \sin\theta, \quad R_3 \sin\theta + Z_4 \cos\theta \} \end{array} \right\} \quad (2.41)$$

$$\left. \begin{aligned} P_{11} & \{ R_3 \cos \theta + Z_4 \sin \theta, \quad -R_3 \sin \theta + Z_4 \cos \theta \} \\ P_{12} & \{ R_5 \cos \theta + Z_4 \sin \theta, \quad -R_5 \sin \theta + Z_4 \cos \theta \} \\ Q_2 & \{ -R_1 \cos \theta + Z_1 \sin \theta, \quad R_1 \sin \theta + Z_1 \cos \theta \} \\ Q_3 & \{ R_1 \cos \theta + Z_1 \sin \theta, \quad -R_1 \sin \theta + Z_1 \cos \theta \} \\ Q_0 & \{ \quad Z_2 \sin \theta \quad , \quad Z_2 \cos \theta \quad \} \\ Q_4 & \{ R_4 \cos \theta + Z_4 \sin \theta, \quad -R_4 \sin \theta + Z_4 \cos \theta \} \end{aligned} \right\}$$

The length of h_0 as shown in Figs. 2.12 and 2.13 are as follows.

When $\sigma_D \neq 0$

$$\left. \begin{aligned} h_0 &= R_5 \sin \theta - \frac{\sin \theta}{\cos \theta} S ; \text{ (region } P_1 \sim P_2 \text{ and } P_3 \sim P_4) , \\ h_0 &= \frac{\cos \theta}{\sin \theta} (S - R_5 \cos \theta); \text{ (region } P_4 \sim P_{12}) . \end{aligned} \right\} \quad (2.42)$$

When $\sigma_D = 0$

$$\left. \begin{aligned} h_0 &= R_5 \sin \theta - \frac{\sin \theta}{\cos \theta} S ; \text{ (region } P_1 \sim P_2 \text{ and } P_3 \sim P_4) , \\ h_0 &= \frac{\cos \theta}{\sin \theta} (S - R_5 \cos \theta); \text{ (region } P_4 \sim P_{12}) , \\ h_0 &= \frac{\cos \theta}{\sin \theta} (S + R_1 \cos \theta) + (R_1 + R_5) \sin \theta; \\ & \quad \text{(region } (P_2 \sim Q_2 \sim P_6) , \\ h_0 &= \infty ; \text{ (region } P_6 \sim Q_0 \sim P_7) . \end{aligned} \right\} \quad (2.43)$$

The lengths of h_1 , l_1 and l_2 are as follows.

(a) If l_2 is known ($S < P_5$, $S \leq 0$, $l_1 = 0$ and $K_h = 0$), h_0 , h_1 and l_2 are shown in Fig. 2.15(a).

$$\left. \begin{aligned} l_2 &= \frac{R_5}{\sin \theta} ; \text{ (region } P_1 \sim P_9) , \\ l_2 &= \frac{1}{\sin \theta \cos \theta} (Z_4 \sin \theta - S); \text{ (region } P_9 \sim P_{10}). \end{aligned} \right\} \quad (2.44)$$

When $\sigma_D \neq 0$

$$h_1 = - \frac{S}{\sin \theta \cos \theta} . \quad (2.45)$$

When $\sigma_D = 0$

$$\left. \begin{aligned} h_1 &= -\frac{S}{\sin\theta\cos\theta}; \text{ (region } P_1\sim P_2) , \\ h_1 &= \frac{R_1}{\sin\theta} \quad ; \text{ (Region } P_2\sim P_6) . \end{aligned} \right\} \quad (2.46)$$

(b) If ℓ_1 and ℓ_2 are known ($S < P_5$, $S > 0$, $\sigma_D \neq 0$, $h_1 = 0$ and $K_h = 0$), h_0 , ℓ_1 and ℓ_2 are shown in Fig. 2.15(b).

$$\left. \begin{aligned} \ell_2 &= \frac{R_5}{\sin\theta} \quad ; \text{ (region } P_1\sim P_9) , \\ \ell_2 &= \frac{1}{\sin\theta\cos\theta} (Z_4\sin\theta - S); \text{ (region } P_9\sim P_{10}) , \end{aligned} \right\} \quad (2.47)$$

$$\left. \begin{aligned} \ell_1 &= \frac{S}{\sin\theta\cos\theta} \quad ; \text{ (} S \leq P_4, \text{ region } P_2\sim P_4\sim P_{12}), \\ \ell_1 &= \frac{R_5}{\sin\theta} \quad ; \text{ (} S > P_4, \text{ region } P_2\sim P_4\sim P_{12}), \end{aligned} \right\} \quad (2.48)$$

(c) If ℓ_2 is known ($S < P_5$, $S > 0$ and $\sigma_D = 0$), h_0 , h_1 and ℓ_1 are shown in Fig. 2.15(c) and are same as Eqs. (2.42), (2.43) and (2.43).

(d) If ℓ_2 is known ($P_5 \leq S \leq Q_0$ and $K_h = 1$), h_0 , h_1 , ℓ_1 and ℓ_2 are shown in Fig. 2.16(d).

$$\ell_2 = \frac{1}{\sin\theta\cos\theta} (Z_1\sin\theta - S) \quad . \quad (2.49)$$

When $\sigma_D \neq 0$

$$\left. \begin{aligned} h_1 &= -\frac{S}{\sin\theta\cos\theta} , \\ \ell_1 &= 0 , \end{aligned} \right\} ; \text{ (for } S \leq 0) , \quad (2.50)$$

$$\left. \begin{aligned} h_1 &= 0 , \\ \ell_1 &= \frac{S}{\sin\theta\cos\theta} , \end{aligned} \right\} ; \text{ (for } 0 < S \leq P_4) , \quad (2.51)$$

$$\left. \begin{aligned} h_1 &= 0 , \\ \ell_1 &= \frac{R_5}{\sin\theta} , \end{aligned} \right\} ; \text{ (for } S > P_4) . \quad (2.52)$$

When $\sigma_D = 0$

$$\left. \begin{aligned} h_1 &= \frac{R_1}{\sin\theta} , \\ \ell_1 &= 0 , \end{aligned} \right\} ; \text{ (for } S \leq P_6) , \quad (2.53)$$

$$\left. \begin{aligned} h_1 &= \frac{1}{\sin\theta\cos\theta} (Z_2\sin\theta - S) , \\ \ell_1 &= 0 , \end{aligned} \right\} ; \text{ (for } S > P_6) . \quad (2.54)$$

(e) If ℓ_1 is known ($Q_0 \leq S \leq P_1$, $\ell_2=0$ and $K_h=1$), h_1 and ℓ_1 are shown in Fig. 2.16(e).

For region $Q_0 \sim P_8$

$$h_1 = \frac{1}{\sin\theta\cos\theta} (S - Z_2\sin\theta) , \quad (2.55)$$

$$\left. \begin{aligned} \ell_1 &= \frac{S}{\sin\theta\cos\theta} ; \text{ (for } S < P_4) , \\ \ell_1 &= \frac{R_5}{\sin\theta} ; \text{ (for } S \geq P_4) . \end{aligned} \right\} \quad (2.56)$$

For region $P_8 \sim P_1$

$$h_1 = \frac{R_3}{\sin\theta} , \quad (2.57)$$

$$\left. \begin{aligned} \ell_1 &= \frac{S}{\sin\theta\cos\theta} ; \text{ (for } S < P_4) , \\ \ell_1 &= \frac{R_5}{\sin\theta} ; \text{ (for } S \geq P_4) . \end{aligned} \right\} \quad (2.58)$$

(f) If ℓ_1 is known ($S > P_1$, $\ell_2=0$ and $K_h=1$), h_1 and ℓ_1 are shown in Fig. 2.16(f).

$$h_1 = \frac{1}{\sin\theta\cos\theta} (S - Z_4\sin\theta) , \quad (2.59)$$

$$\left. \begin{aligned} \ell_1 &= \frac{S}{\sin\theta\cos\theta} ; \text{ (for } S < P_4) , \\ \ell_1 &= \frac{R_5}{\sin\theta} ; \text{ (for } S \geq P_4) . \end{aligned} \right\} \quad (2.60)$$

The sectional figure of the shock absorber in the case of the oblique drop impact as shown in Fig. 2.17, is an ellipsoid. The

equation of the ellipsoid is as follows.

$$\frac{x^2}{R_5^2} + \frac{y^2}{b^2} = 1 \quad , \quad (2.61)$$

where

$$b = \frac{R_5}{\sin\theta} \quad . \quad (6.62)$$

Rearranged eq. (2.61), using $y=h(x_M)$

$$x_M = \sqrt{b^2 - h^2} \cdot \sin\theta \quad , \quad (2.63)$$

$$y = \frac{1}{\sin\theta} \sqrt{R_5^2 - x^2} \quad . \quad (2.64)$$

Force ΔF at the sectional area (ΔS , Δx) is as follows.

$$\Delta F = K \cdot \sigma \left(\frac{U}{\ell} \right) \Delta S \cdot \Delta x \quad , \quad (2.65)$$

where K is the boundary condition constant as shown in Fig. 2.18.

$$0 \leq K_i \leq 1 \quad (2.66)$$

where

K_1 : for $\sigma_D=0$; region $P_3 \sim Q_3$,

for $\sigma_D \neq 0$; region $P_1 \sim X_1$,

K_2 : for $\sigma_D=0$; region $Q_3 \sim P_7$,

for $\sigma_D \neq 0$; region $X_1 \sim X_2$,

K_3 : for region $P_{11} \sim Q_4$,

K_4 : for region $Q_4 \sim P_{12}$.

2.5 Convergence method

The convergence methods for two or three materials in one dimensional bar in the case of the oblique drop impact, are as follows.

(1) Two materials

According to the stress-strain relation as shown in Fig. 2.19, the following equations are derived.

$$\left. \begin{aligned} \sigma_A &= \sigma_A^0 + K_A \varepsilon_A = \sigma_A^0 + \frac{K_A}{\ell_A} U_A, \\ \sigma_B &= \sigma_B^0 + K_B \varepsilon_B = \sigma_B^0 + \frac{K_B}{\ell_B} U_B, \end{aligned} \right\} \quad (2.67)$$

and

$$\left. \begin{aligned} U_A + U_B &= U, \\ -\sigma_A + \sigma_B &= 0, \end{aligned} \right\} \quad (2.68)$$

from above two equations

$$\begin{bmatrix} 1 & 1 \\ -\frac{K_A}{\ell_A} & \frac{K_B}{\ell_B} \end{bmatrix} \begin{Bmatrix} U_A \\ U_B \end{Bmatrix} = \begin{Bmatrix} U \\ \sigma_A^0 - \sigma_B^0 \end{Bmatrix} . \quad (2.69)$$

solve Eq. (2.69), we obtain U_A and U_B .

(2) Three materials

$$\begin{bmatrix} 1 & 1 & 1 \\ -\frac{K_A}{\ell_A} & \frac{K_B}{\ell_B} & 0 \\ -\frac{K_A}{\ell_A} & 0 & \frac{K_K}{\ell_K} \end{bmatrix} \begin{Bmatrix} U_A \\ U_B \\ U_K \end{Bmatrix} = \begin{Bmatrix} U \\ \sigma_A^0 - \sigma_B^0 \\ \sigma_A^0 - \sigma_K^0 \end{Bmatrix} . \quad (2.70)$$

(3) Strain, stiffness and stress

Strain

$$\varepsilon_i = \frac{U_i}{\ell_i} . \quad (2.71)$$

Stiffness (see Fig. 2.20)

$$K_i = \frac{\sigma_2 - \sigma_1}{\varepsilon_2 - \varepsilon_1} . \quad (2.72)$$

Stress (see Fig. 2.20)

$$\sigma_i = \sigma_1 - K_i \cdot \varepsilon_i . \quad (2.73)$$

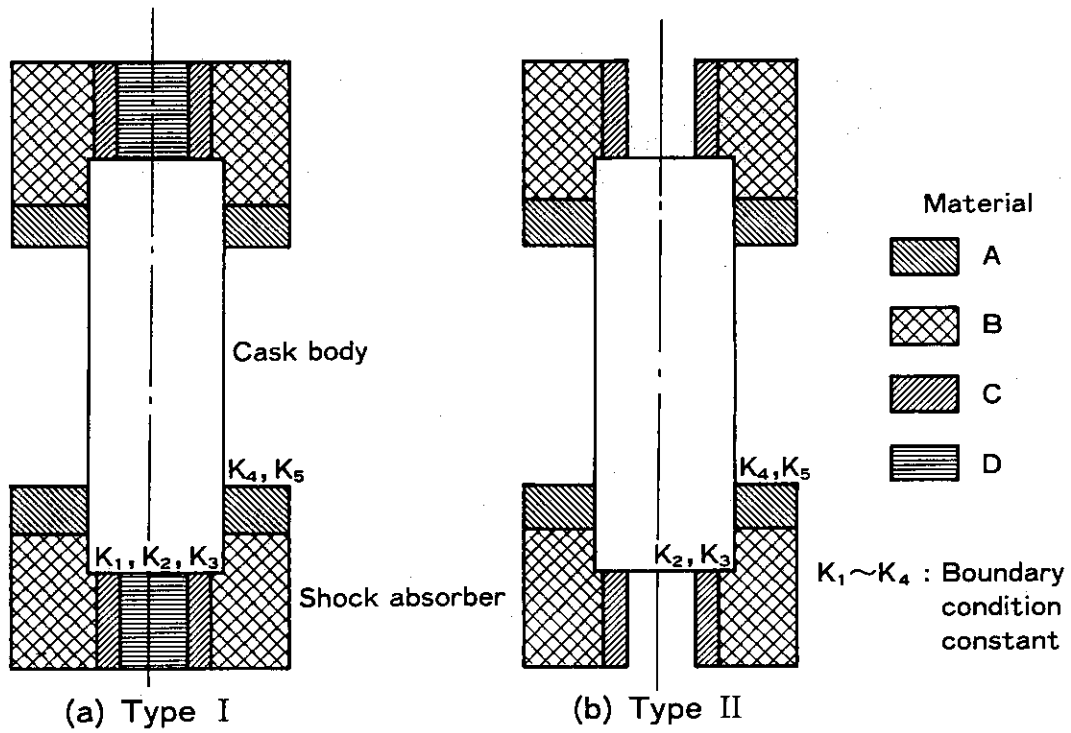


Fig.2.1 Vertical drop model

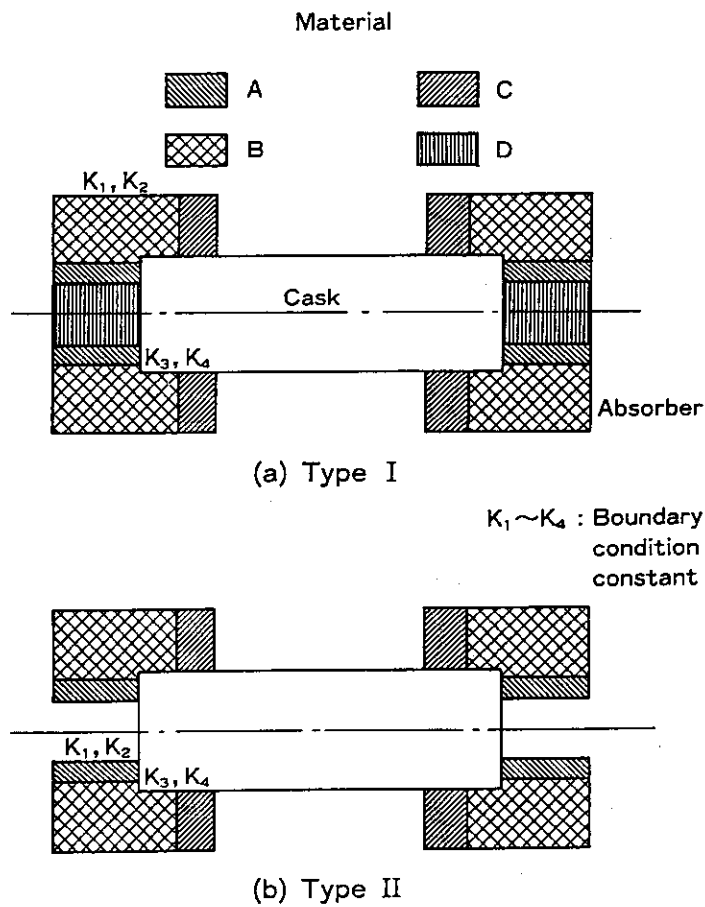


Fig.2.2 Horizontal drop model

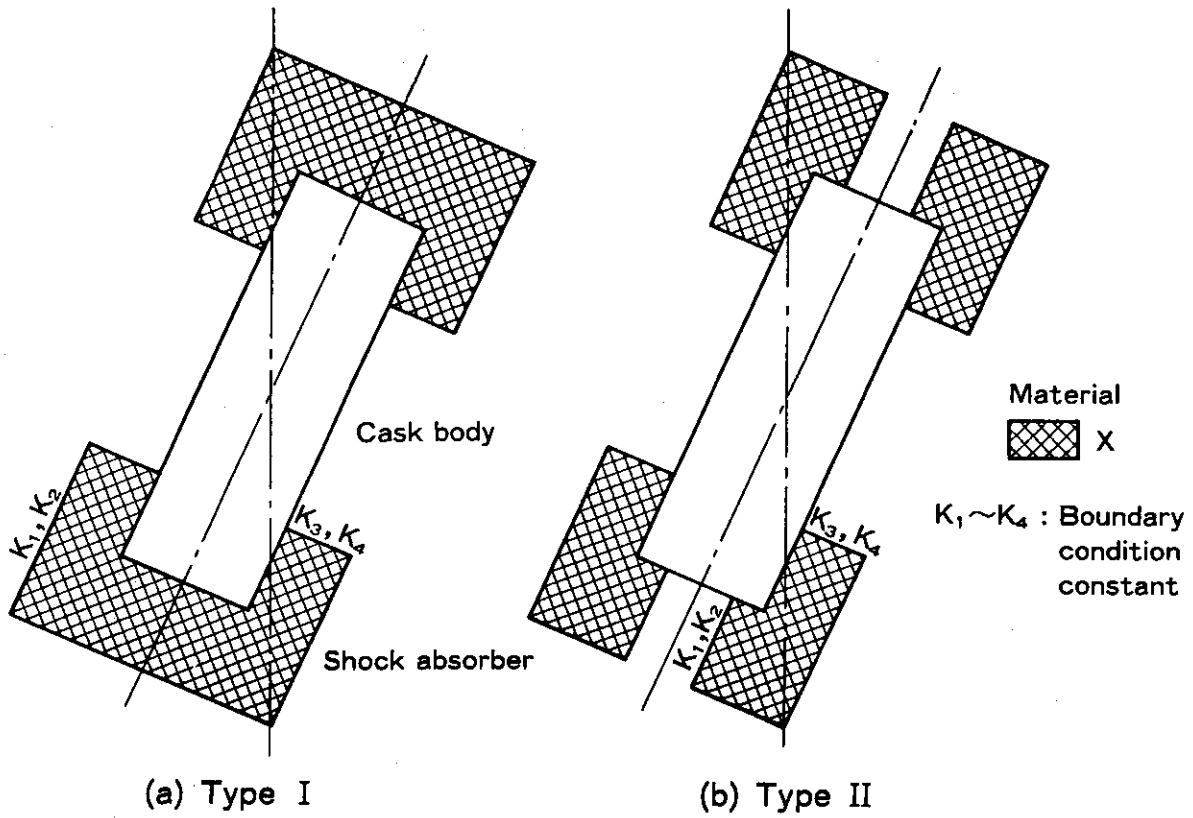
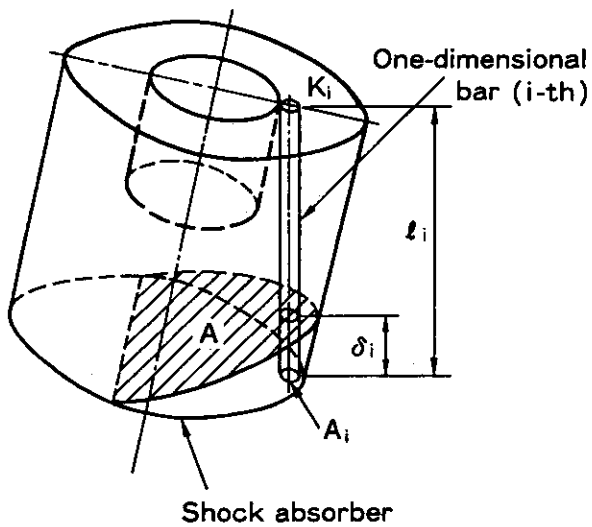


Fig.2.3 Oblique drop model



Strain : ϵ

$$\epsilon_i = \frac{\delta_i}{l_i}$$

Force : F_i

$$F_i = K_i \sigma_i(\epsilon_i) A_i$$

Total force : F

$$F = \sum_i F_i$$

Acceleration : α

$$\alpha = \frac{F}{M}$$

M : Mass, A : Area

σ : Stress, K : Boundary condition constant

Fig.2.4 Uni-axial displacement method

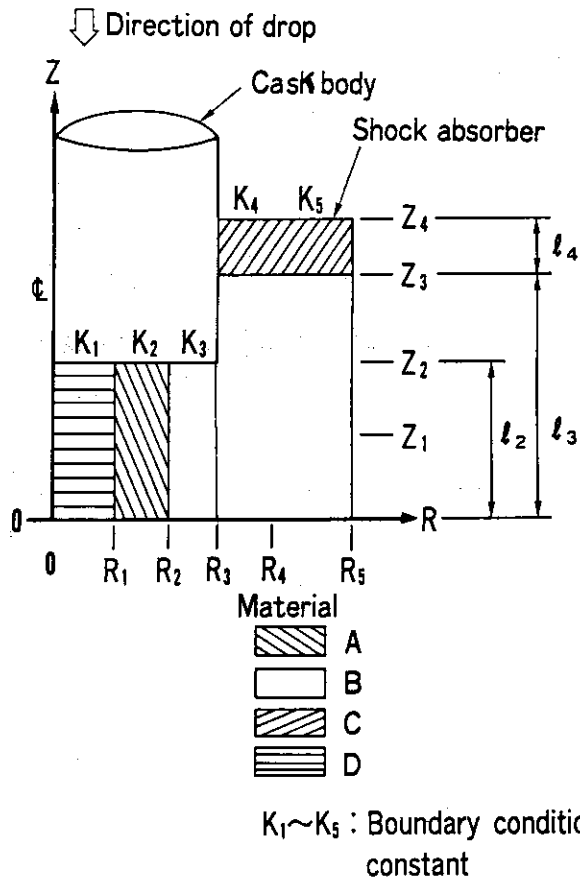


Fig.2.5 Geometry and material in the case of vertical drop model

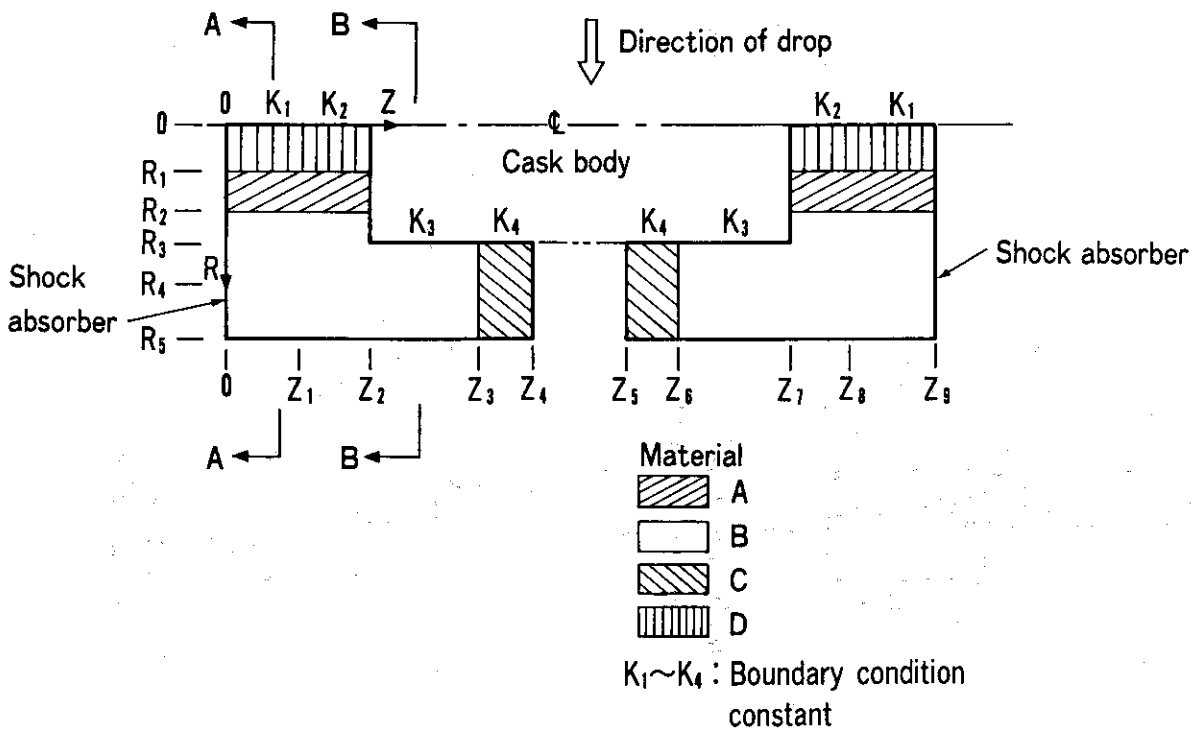


Fig.2.6 Geometry and material in the case of horizontal drop model

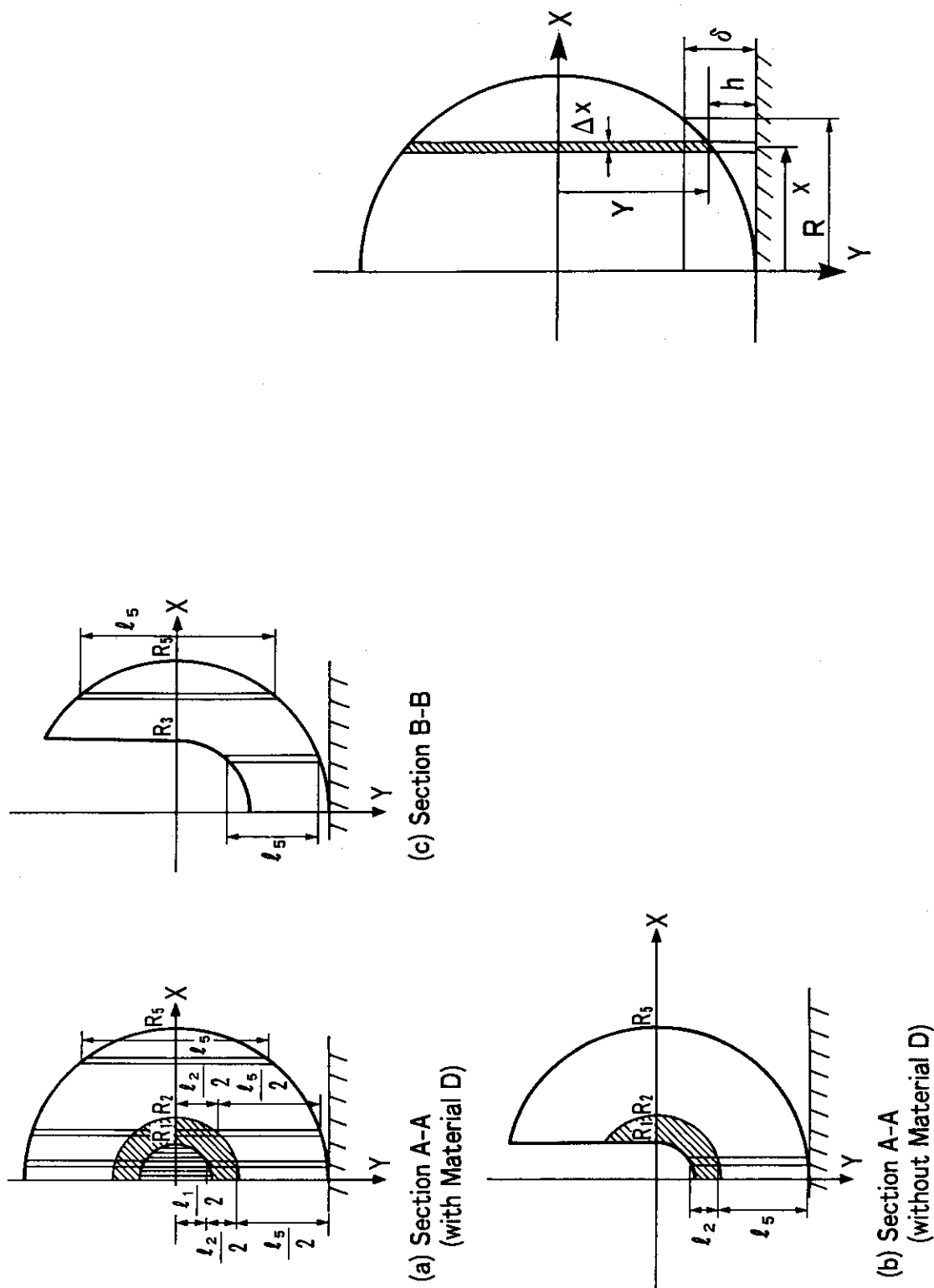


Fig.2.7 Section view of horizontal drop model

Fig.2.8 Sectional area of shock absorber

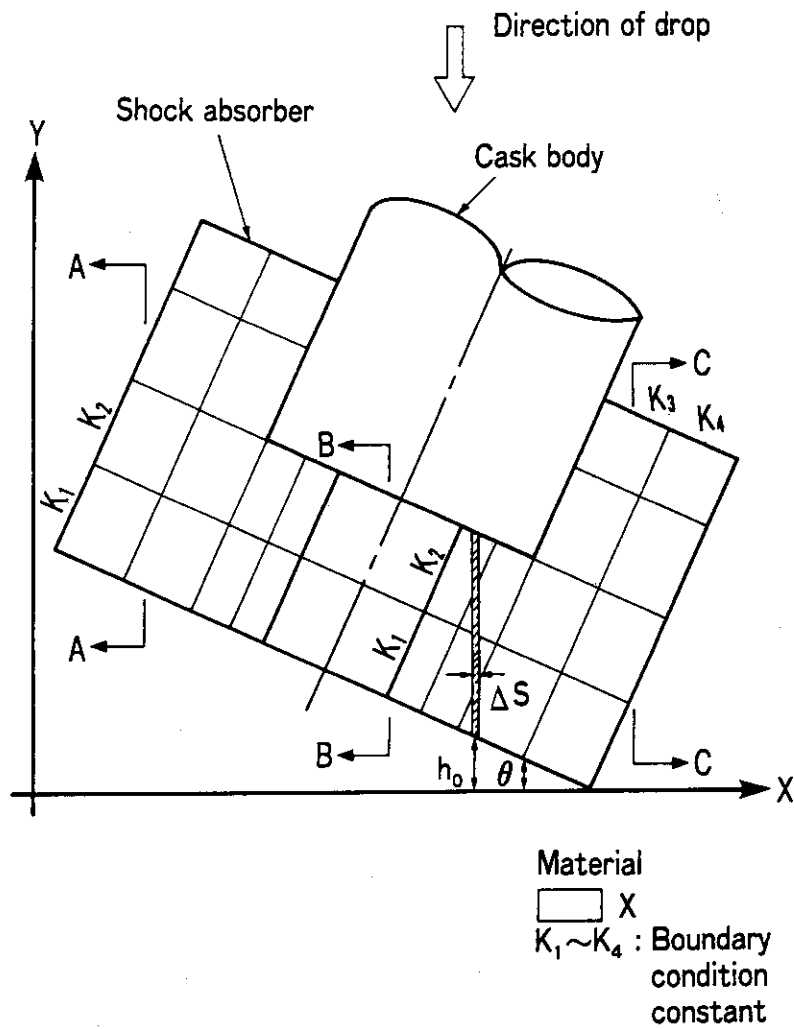


Fig.2.9 Geometry and material in the case of oblique drop model

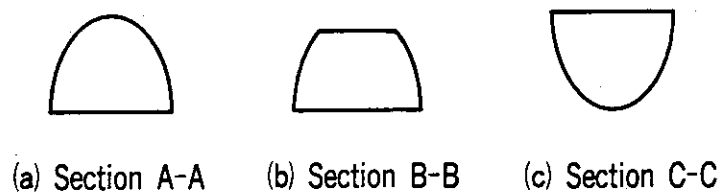


Fig.2.10 Sectional figure of shock absorber in the case of oblique drop (I)

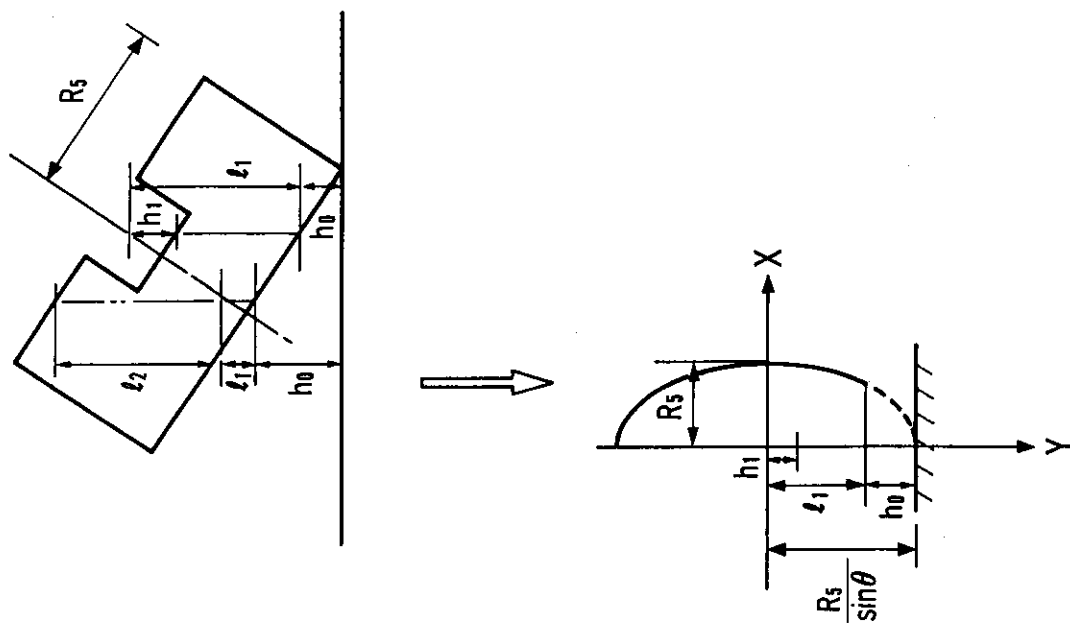


Fig.2.11 Sectional figure of shock absorber in the case of oblique drop (II)

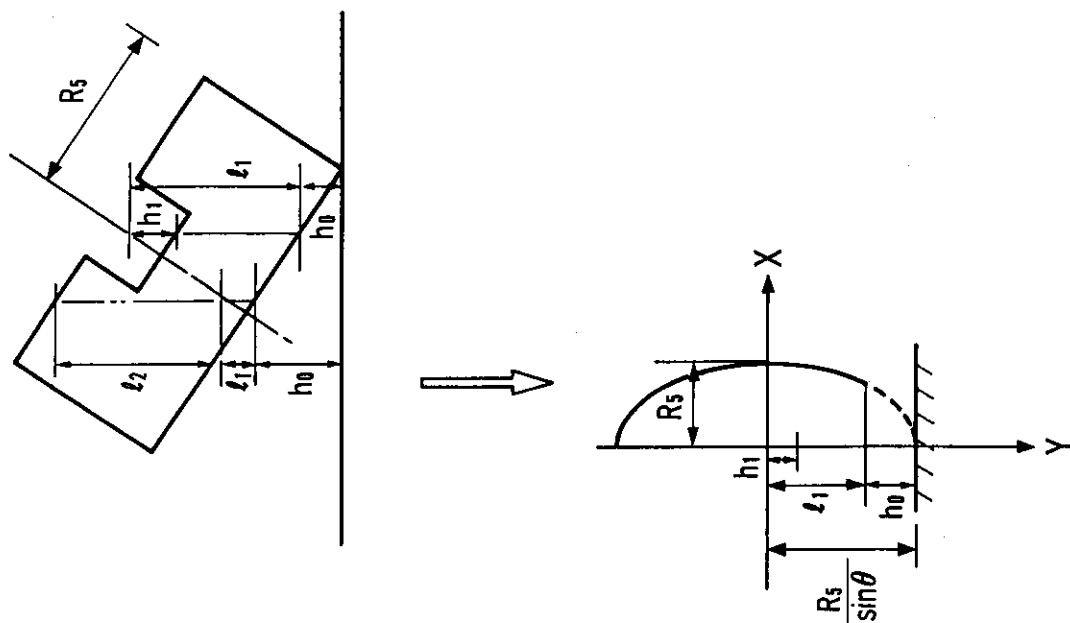


Fig.2.12 Sectional figure of shock absorber in the case of oblique drop (III)

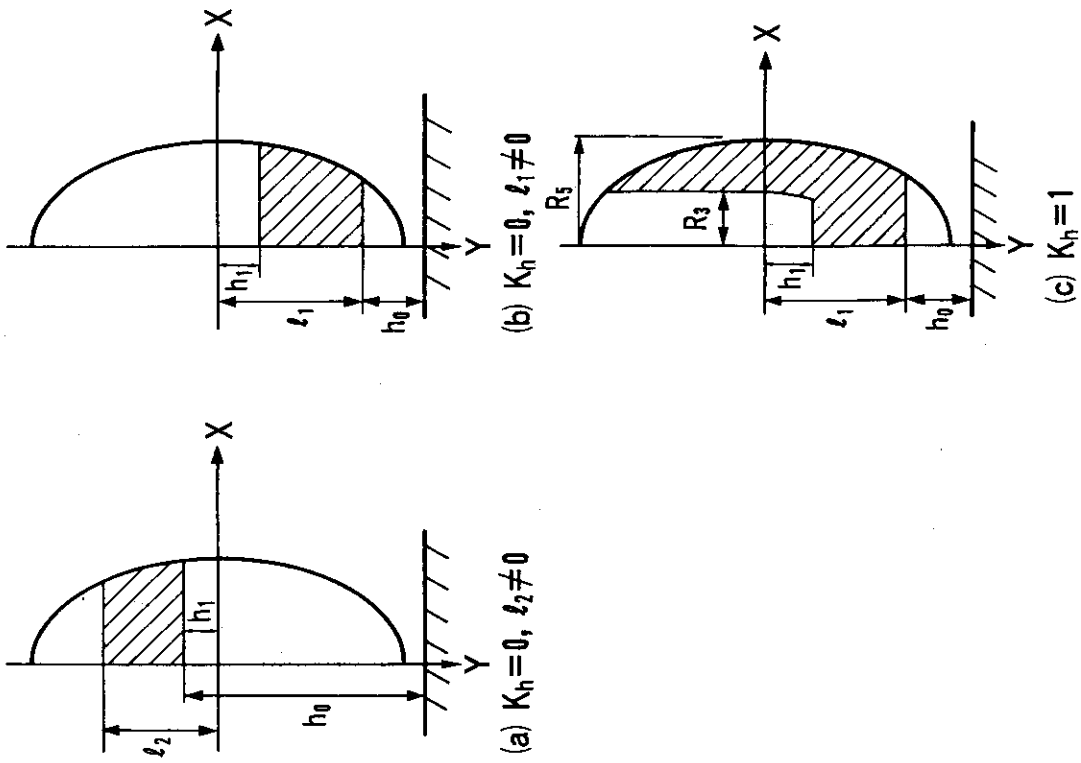


Fig.2.13 Sectional figure of shock absorber in the case of oblique drop (IV)

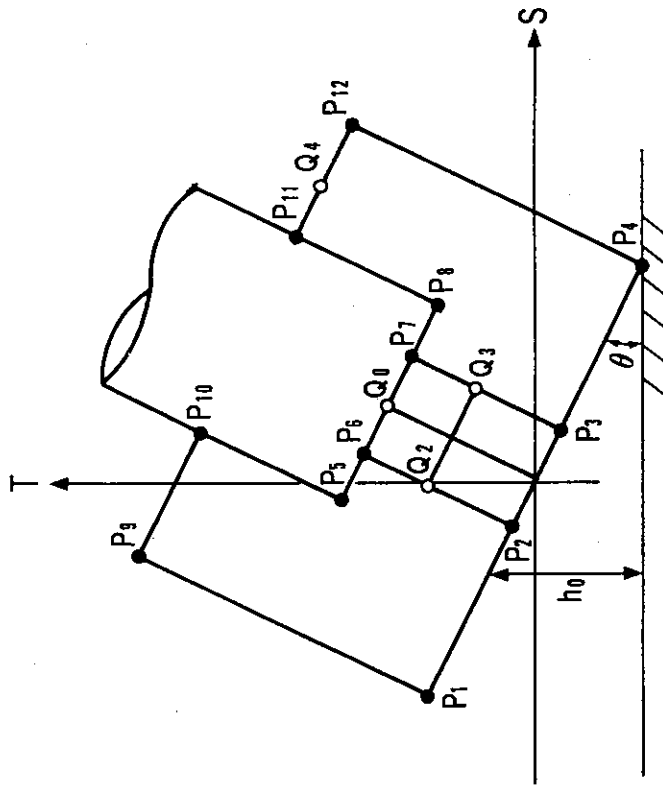


Fig.2.14 Geometry of shock absorber in the case of oblique drop and S-T coordinate

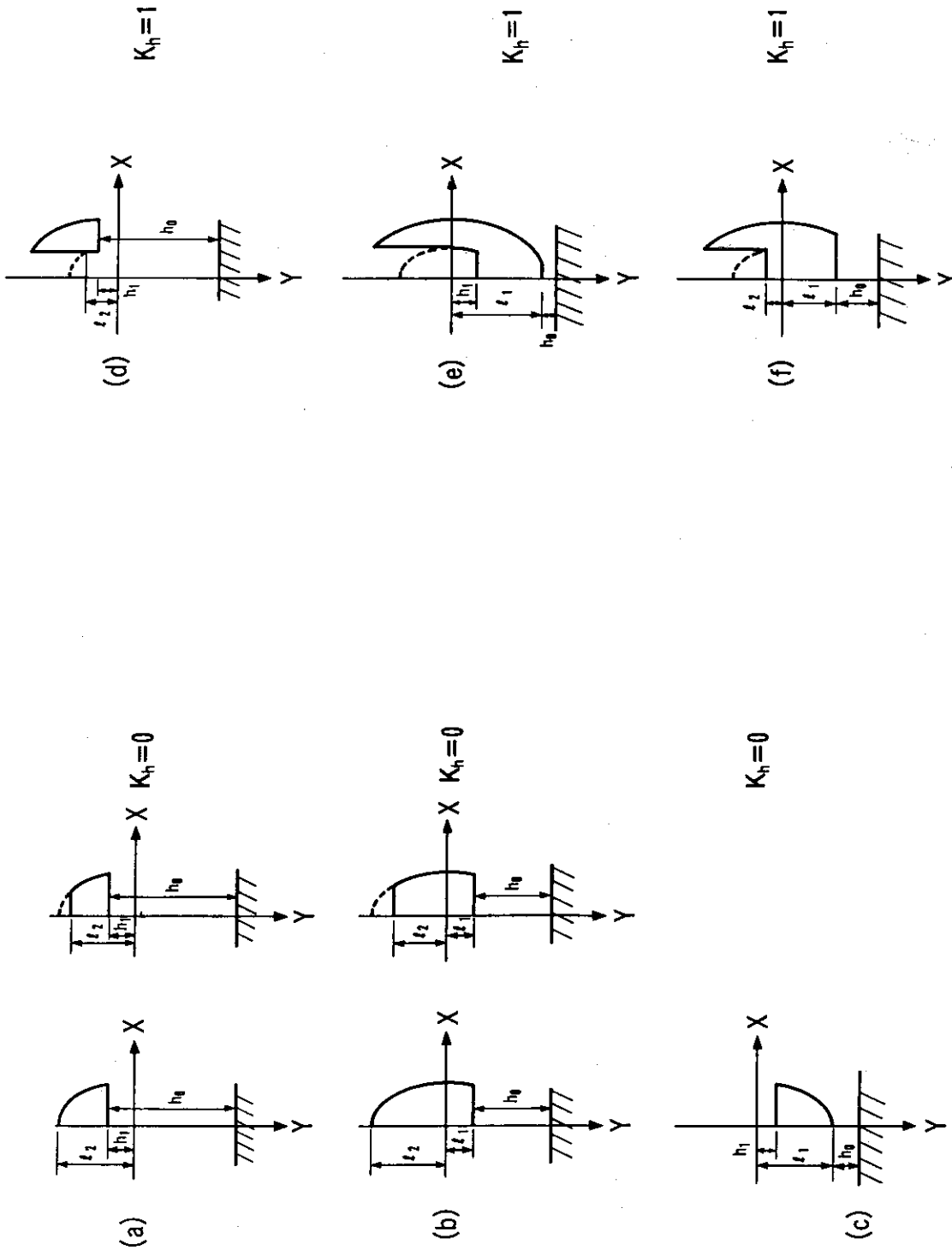


Fig.2.15 Various sectional figure of shock absorber in the case of oblique drop (I)

Fig.2.16 Various sectional figure of shock absorber in the case of oblique drop (II)

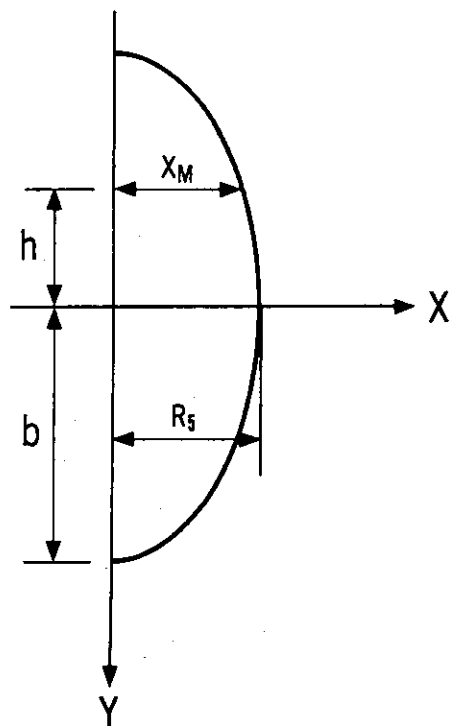


Fig.2.17 Ellipsoid

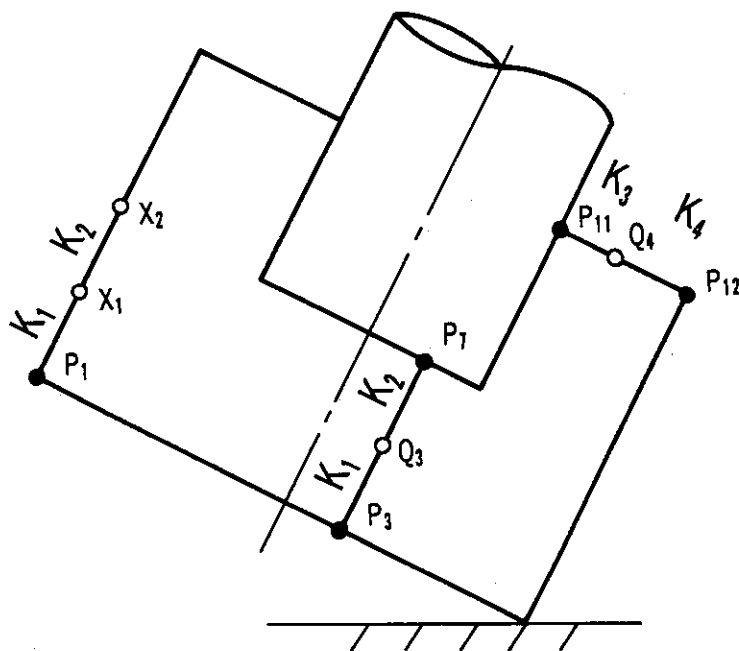


Fig.2.18 Boundary condition constant in the case of oblique drop

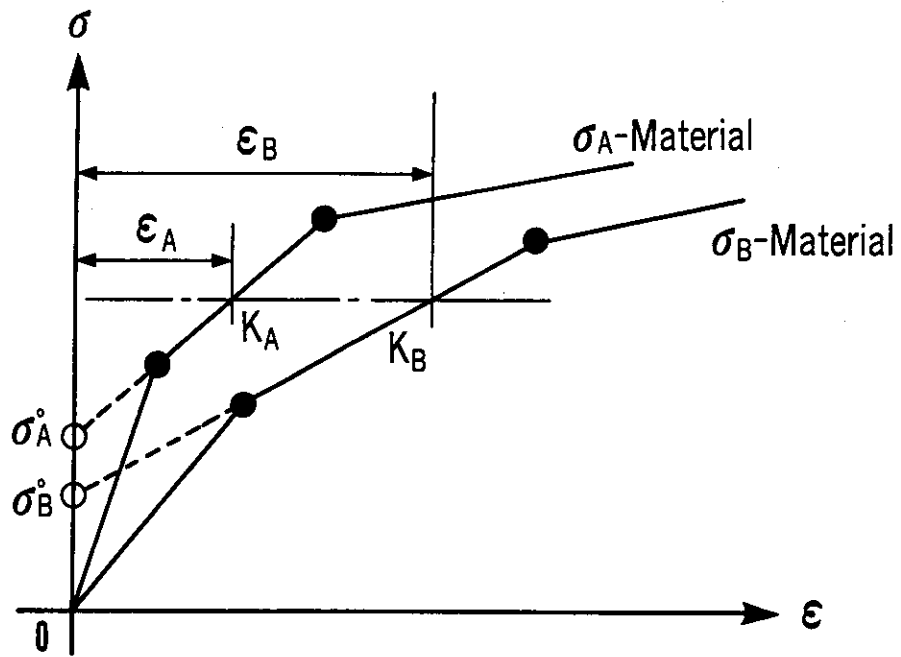


Fig.2.19 Stress-strain curves of shock absorber materials

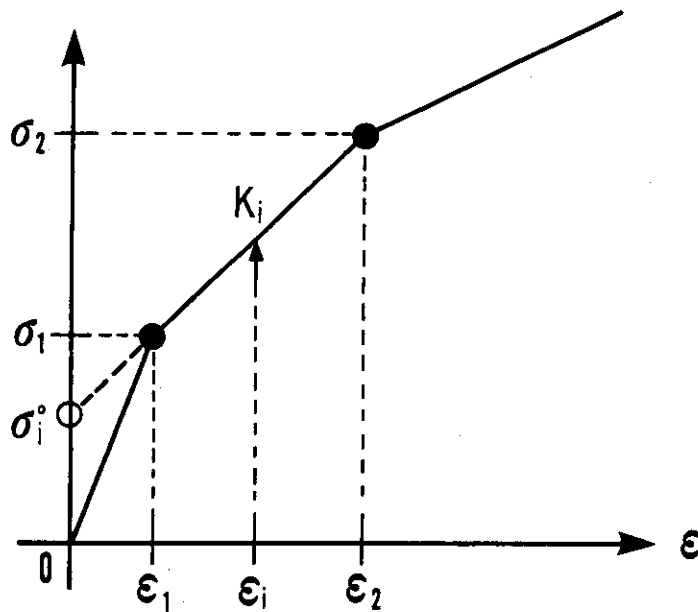


Fig.2.20 Stress-strain curve of shode absorber material

3. Benchmark Calculation

In order to demonstrate the adequacy of the simplified computer program CRUSH1, the benchmark calculations using experimental results of the 1/4 scale model of NUPAC 125B cask as shown in Fig. 3.1 have been performed.

Figure 3.2 is the deformed shapes of the cask after 9 meters drop impact to obtained by the detailed computer program DYNA3D. The comparison among the results obtained by the experiments, the simplified computer program CRUSH1 and the detailed computer program DYNA3D is shown in Table 3.1. The relation among the oblique angles of cask drop attitude and the maximum accelerations and the maximum deformations obtained by CRUSH1 are shown in Figs. 3.3 and 3.4. According to Table 3.1 and Figs. 3.3 and 3.4, results by the simplified computer program CRUSH1 agree with both the experimental results and that of the results of the detailed program DYNA3D.

Table 3.1 Comparison between simplified and detailed analyses and experiment

Attitude	Acceleration (G)			Deformation (mm)		
	Experiment	Simplified analysis CRUSH	Detailed analysis DYNA3D	Experiment	Simplified analysis CRUSH	Detailed analysis DYNA3D
Vertical	200	179	271 * (200)**	51	52	50
Corner	106	125	130 (85)	127	151	132
Horizontal	180	183	216 (160)	61	63	64

* Value of low pass filter is 600 Hz.

** Mean value = $\frac{\text{Impact velocity}}{\text{Rebound time}}$
(NUPAC 125 B cask 1/4 scale model).

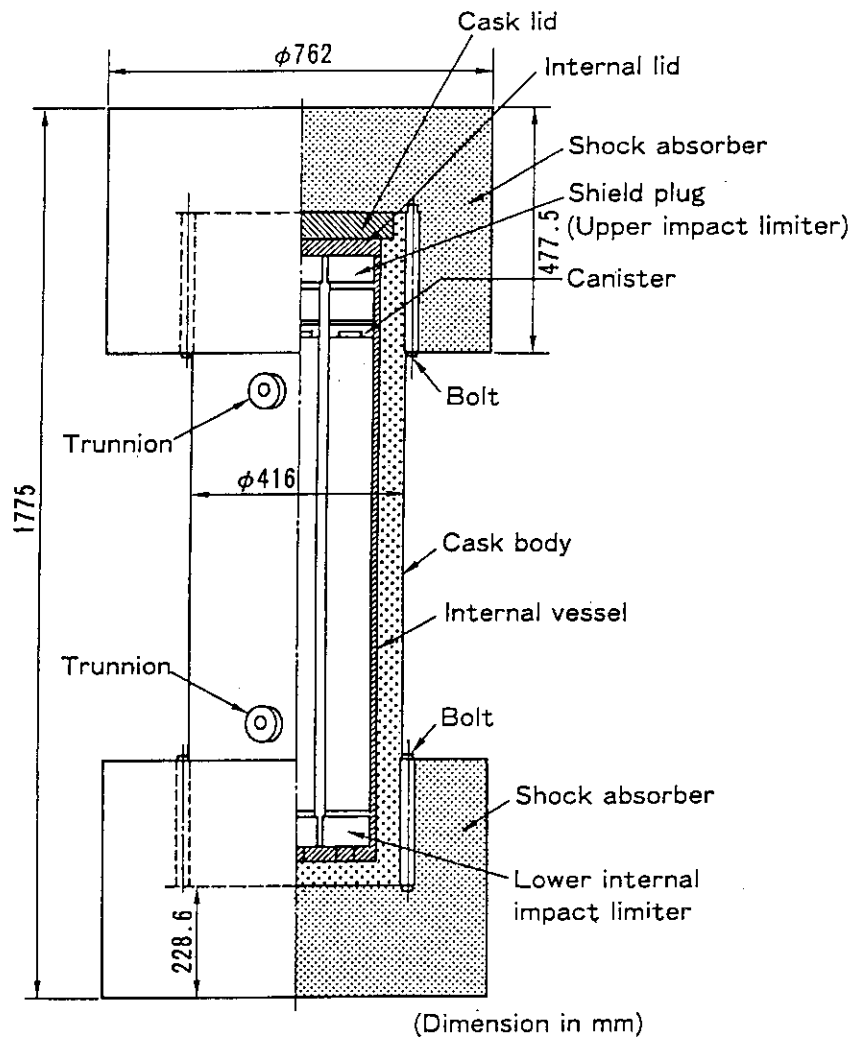
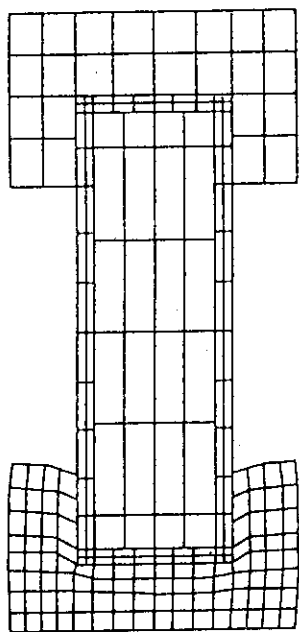
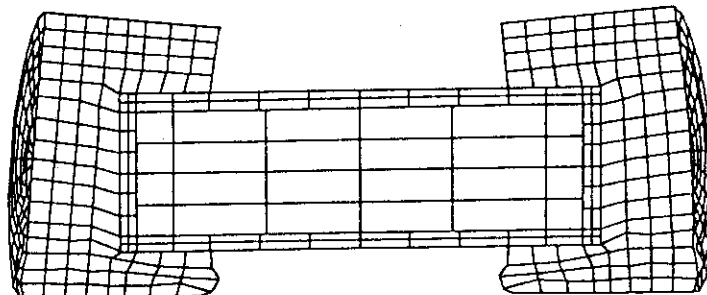


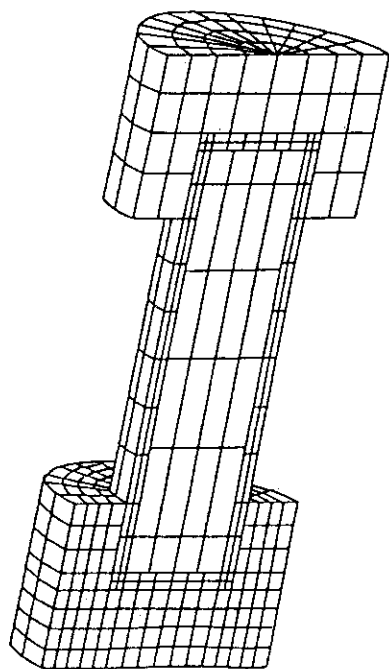
Fig.3.1 Shipping cask-NUPAC 125B (1/4 scale model)



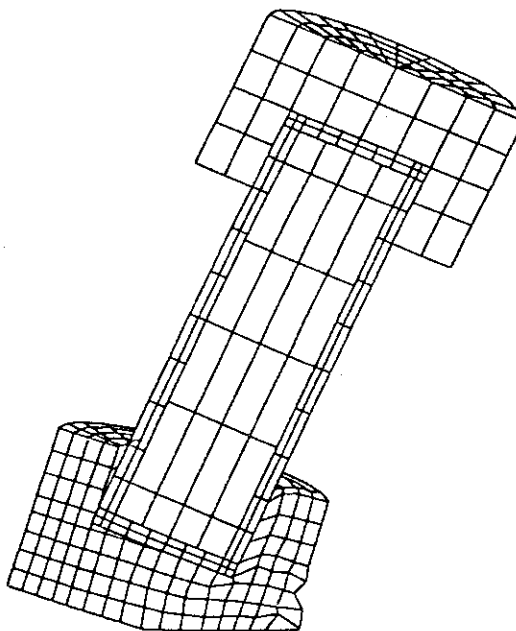
(a) Vertical drop



(b) Horizontal drop



(c) Underformed mesh



(d) Corner drop

Fig.3.2 Deformed shape after 9m drop impact
(NUPAC 125B cask 1/4 scale model)

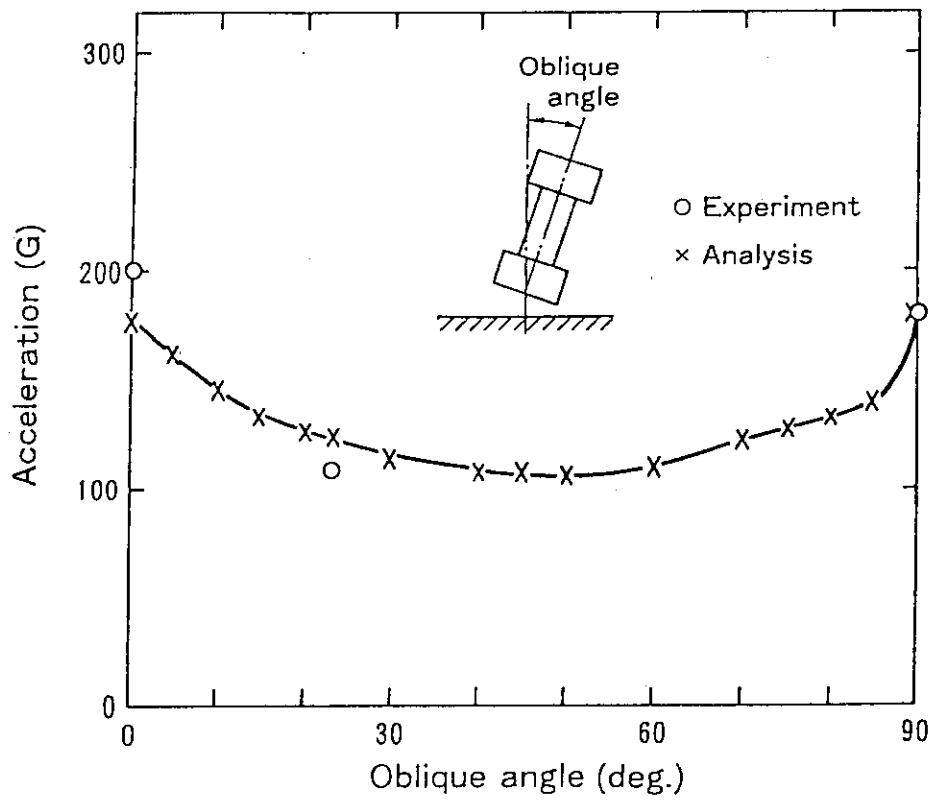


Fig.3.3 Comparison between simplified analysis and experiment on acceleration

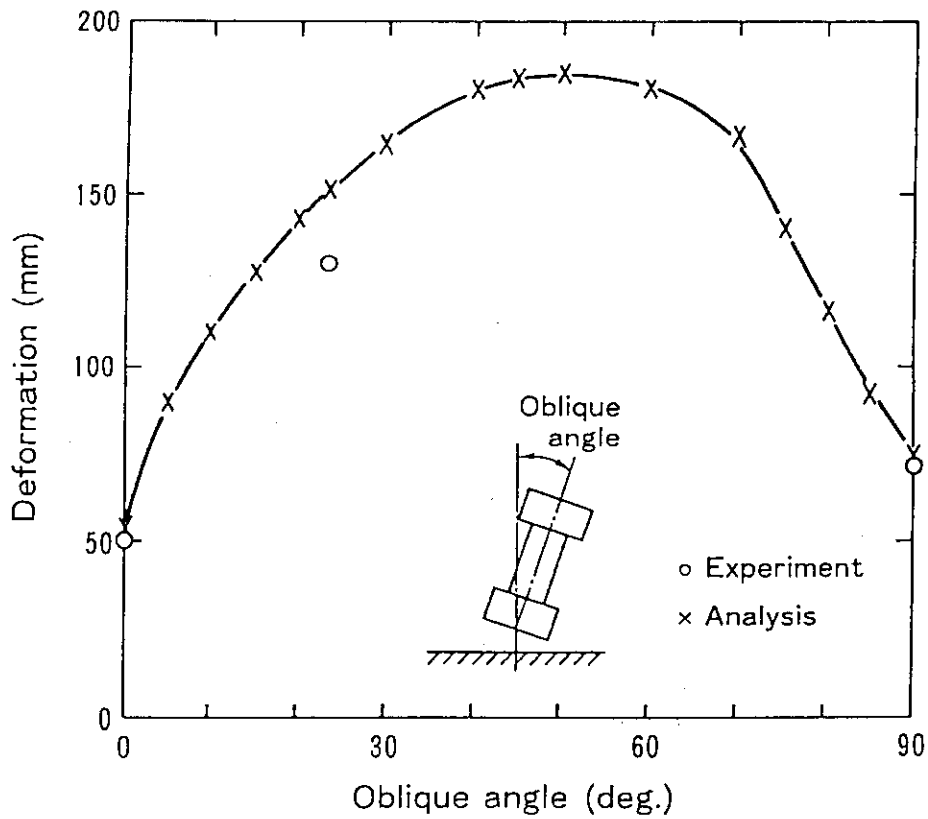


Fig.3.4 Comparison between simplified analysis and experiment on deformation

4. Computer Program

4.1 Program Description

The computer program CRUSH1 is static calculation program capable of evaluating the maximum acceleration of the cask body and the maximum deformation of the shock absorber using the uniaxial displacement method.

The computer program CRUSH1 consists of a main routine and thirteen subroutines that are MAIN, CARDIN, VERT, HORIZ, CORNER, MULTI, GETMAT, EQSOLV, GETSIG, FITSTP, MPLOT, DPLOT and UPLOT. Overall structure of CRUSH1 is shown in Fig. 4.1. Functions of subroutines are as follows:

MAIN : initiarizes start of run,
CARDIN : reads input data,
VERT : computes acceleration and deformation in the case
of vertical(head-on or bottom-on) drop ompact,
HORIZ : computes acceleration and deformation in the case
of horizontal (side-on) drop impact,
CORNER : computes acceleration and deformation in the case
of oblique drop impact,
MULTI : computes relation of stress-strain data for multi-
material layers,
GETMAT : computes relation of stress-strain data for single
material,
EQSOLV : computes matrix data,
GETSIG : computes stress,
FITSTP : curve fitting for stress-strain data,
MPLOT : plots data,
DPLOT : plots data,
UPLOT : plots data.

A macroscopic flow chart of CRUSH1 is shown in Fig. 4.2.

4.2 Description of Input Data

This section describes the input data required by CRUSH1. The input data consists of the job description, the analysis type such as drop attitude, geometry, the cask weight, the initial condition, the boundary condition constants, the geometry unit selection and options for output plotting. The input instruction is simple and easy to follow. The input data forms are presented in Table 4.1.

4.3 Description of Output Data

This section describes the output data forms of CRUSH1. The contents of these various quantities are describes in the followings.

(1) Input data

The input data are printed in two formats. The first print format is exactly the same as they were read. Second, the computer program lists the input data as interpreted by CRUSH1.

(2) Calculation data

The calculation data; the deformation, the force, the energy and acceleration are printed at every interval steps.

(3) Graphical output

CRUSH1 provides users with graphical output of the deformation, relationship between acceleration and deformation, dissipation energy and deformation, acceleration and dissipation, energy and so forth.

Table 4.1 Input data for CRUSH1

Columns	Format	Variables	Descriptions
Data set No.1:Job description.			
1 - 8	2A4	NAME	Flag for data type. 'TITLE '.
9 - 10	2X	-	Blank,
11 - 78	70A1	NTITLE	Job description.
Data set No.2:Calculation model data.			
1 - 8	2A4	NAME	Flag for type of drop. 'VERTICAL':Vertical drop(Head-on drop or Bottom-on drop). 'HORIZONTAL':Horizontal drop(Side-on drop). 'CORNER ':Oblique drop.
9 - 10	2X	-	Blank.
11 - 20	F10.0	WEIGHT	Weight of cask(kg).
21 - 30	F10.0	HEIGHT	Height of cask drop(mm or cm).
31 - 40	F10.0	ANGLE	Oblique angle in the case of oblique drop (deg.), (ANGLE is 5~85deg.).
41 - 50	F10.0	DISP	Incremental deformation for calculation(mm or cm), if DISP is blank or zero, DISP equal to 1.0.
51 - 60	F10.0	CONV	Truncation error, if CONV is blank or zero, CONV equal to 1.0×10^{-5} .
61 - 64	A4	NDATA	Flag for unit. 'MM ':unit is mm. 'CM ':unit is cm.
65 - 70	6X	-	Blank.
71 - 80	F10.0	EPMAX	Limit strain for calculation, $0.0 < EPMAX < 1.0$. If EPMAX is blank or zero, EPMAX equal to 1.0,

Table 4.1 (Continued)

Columns	Format	Variables	Descriptions
Data set No.3:Geometry data.			
1 - 8	2A4	NAME	Flag for data type. 'LENGTH'
9 - 10	2X	-	Blank,
11 - 20	F10.0	CLENGTH	Length of cask body.
21 - 30	F10.0	RADIUS	Radius of cask body.
31 - 35	5X	-	Blank.
36 - 40	A4	KDTYPE	Flag of material data. 'SIGD':material σ_D exist. 'NONE':material σ_D do not exist.
Data set No.4:Options for input data check and plotting.			
1 - 8	2A4	NAME	Flag for input data check or execution. 'MODEL':input data check. 'RUN':execution.
9 - 10	2X	-	Blank.
11 - 14	A4	NAME1	Plotting of calculation results. 'PLOT':plotting of calculation results. '':no plotting.
15	1X	-	Blank.
16 - 19	A4	NDATA	Flag for debugging of program. 'CHEC':debugging of program. '':no debugging.
Data set No.5:Geometry of shock absorber(I).			
1 - 8	2A4	NAME	Flag for coordinate of radial direction. 'RCOOR'
9 - 10	2X	-	Blank.
11 - 20	F10.0	R1	Radius of shock absorber(see Fig.2.5, 2.6 and 2.9).
21 - 30	F10.0	R2	Same as above.

Table 4.1 (Continued)

Columns	Format	Variables	Descriptions
31 - 40	F10.0	R3	Same as above.
41 - 50	F10.0	R4	Same as above.
51 - 60	F10.0	R5	Same as above.
Data set No.6A:Geometry of shock absorber(IIA).			
1 - 8	2A4	NAME	Flag for coordinate of axial direction. 'ZCOOR'.
9 - 10	2X	-	Blank.
11 - 20	F10.0	Z1	Axial length of shock absorber(see Fig.2.5 and 2.6).
21 - 30	F10.0	Z2	Same as above.
31 - 40	F10.0	Z3	Same as above.
41 - 50	F10.0	Z4	Same as above.
51 - 60	F10.0	Z5	Same as above.
61 - 70	F10.0	Z6	Same as above.
71 - 80	F10.0	Z7	Same as above.
Data set No.6B:Geometry of shock absorber(IIB).			
1 - 8	2A4	NAME	Flag for coordinate of axial direction. 'ZCOOR'.
9 - 10	2X	-	Blank.
11 - 20	F10.0	Z8	Axial length of shock absorber(see Fig.2.5 and 2.6).
21 - 30	F10.0	Z9	Same as above.
Data set No.7:Boundary condition constant.			
1 - 4	2A4	NAME	Flag for data type. 'MESH'.
9 - 10	2X	-	Blank.
11 - 15	I5	NPART	Number of partitions in the X-Y plain of horizontal or oblique drop case.

Table 4.1 (Continued)

Columns	Format	Variables	Descriptions
15 - 20	I5	KPART	Maximum NPART is 300. If NPART is blank or zero, NPART equal to 100. Number of partitions in the R-Z plain of oblique drop case. Maximum KPART is 400. If KPART is blank or zero, KPART equal to 100.
21 - 30	F10.0	K1	Boundary condition constant(see Fig.2.5, 2.6 and 2.9).
31 - 40	F10.0	K2	Same as above.
41 - 50	F10.0	K3	Same as above.
51 - 60	F10.0	K4	Same as above.
61 - 70	F10.0	K5	Same as above.
Data set No.8A:Calculation model data.			
1 - 8	2A4	NAME	Flag for material data identification. 'SIGA': σ_A stress-strain data(see Fig. 2.5, 2.6 and 2.9). 'SIGB': σ_B stress-strain data. 'SIGC': σ_C stress-strain data. 'SIGD': σ_D stress-strain data. 'SIGX': σ_X stress-strain data.
9 - 10	2X	-	Blank.
11 - 15	I5	NN	Number of stress-strain data.
16 - 20	5X	-	Blank.
21 - 30	F10.0	FACT	Multiplication factor of stress data.
Data set No.8B:Stress-strain-data.			
1 - 10	F10.0	STRAIN(1)	Strain(see Fig.2.19 and 2.20).
11 - 20	F10.0	STRESS(1)	Stress(see Fig.2.19 and 2.20).
21 - 30	F10.0	STRAIN(2)	Strain.
31 - 40	F10.0	STRESS(2)	Stress.

Table 4.1 (Continued)

Columns	Format	Variables	Descriptions
41 - 50	F10.0	STRAIN(3)	Strain.
51 - 60	F10.0	STRESS(3)	Stress.
61 - 70	F10.0	STRAIN(4)	Strain.
71 - 80	F10.0	STRESS(4)	Stress.
Repeat 8B data set for NN data.			
Repeat 8A and 8B data set for SIGA, SIGB, SIGD and SIGX stress-strain data.			

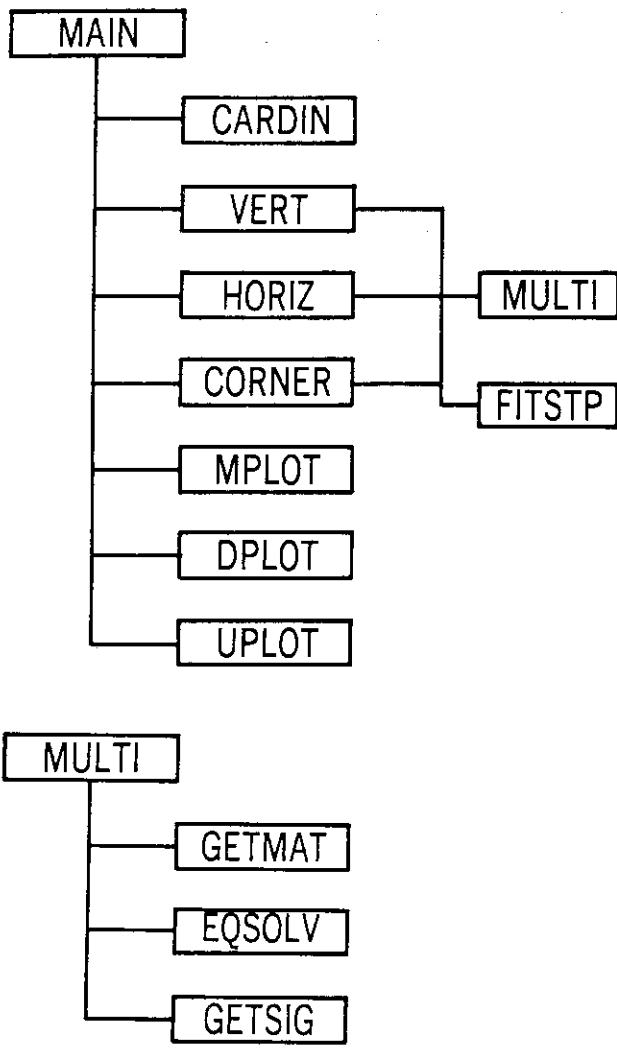


Fig.4.1 Structure of computer program CRUSH1

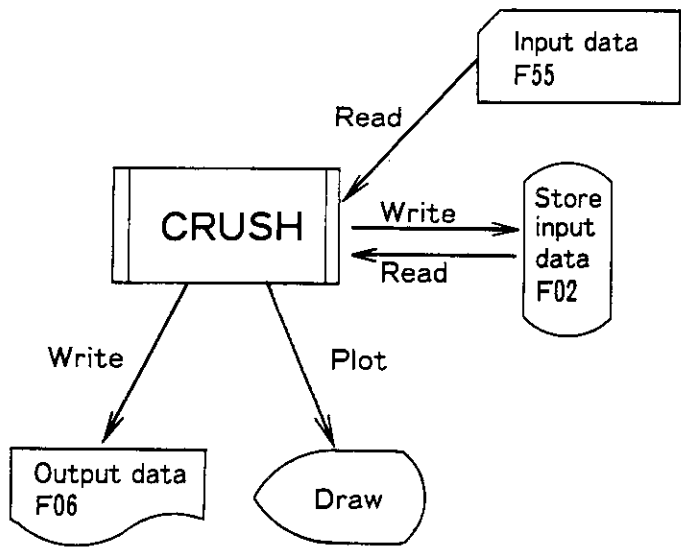


Fig.4.2 Program flow

5. Conclusions

In regard to the evaluation of the maximum acceleration of the cask body and the maximum deformation of the shock absorber on the drop impact, a simplified computer program CRUSH1 was made to analyze economical and by shortening input and computer time to about 1/20 or less as compared with other detailed computer programs to analyze dynamic interactions. The results obtained by the simplified computer program CRUSH1 has an enough adequacy for its practical use. CRUSH1 is further being utilized satisfactorily in safety analysis and designing not only spent fuel transport casks but also those for various radioactive transport casks.

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References

- (1) Asada, K. et al., "Development of Simplified Analysis Codes for Nine Meter Drop and One Meter Puncture Tests for a Radioactive Material Transport Cask", Waste Management '88 (Tuscon) (1988).
- (2) Ikushima, T., "CRUSH: A Simplified Computer Program for Impact Analysis of Radioactive Material Transport Casks", JAERI-M 90-004 (1990).

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References

- (1) Asada, K. et al., "Development of Simplified Analysis Codes for Nine Meter Drop and One Meter Puncture Tests for a Radioactive Material Transport Cask", Waste Management '88 (Tuscon) (1988).
- (2) Ikushima, T., "CRUSH: A Simplified Computer Program for Impact Analysis of Radioactive Material Transport Casks", JAERI-M 90-004 (1990).

Appendix A Sample Problem Input

C R U S H 1 M O D E L D A T A

	1	2	3	4	5	6	7	8	
1- TITLE	NUPAC-125B CASK VERTICAL IMPACT MODEL								
2- VERTICAL	1283.6	9000.0	0.0	5.0	0.0	MM			
3- LENGTH	1775.0	208.0	SIGD						
4- RUN	PLOT								
5- RCOORD	0.0	220.0	247.0	350.0	380.0				
6- ZCOORD	224.0	224.0	350.0	474.0					
7- ZCOORD									
8- MESH		1.0	1.0	1.0	1.0	1.0			
9- SIGA	24	1.0							
10- 0.0	0.0	0.02	0.2	0.03	0.27	0.04	0.29		
11- 0.05	0.3	0.25	0.3	0.33	0.33	0.4	0.36		
12- 0.45	0.4	0.5	0.44	0.55	0.52	0.58	0.57		
13- 0.6	0.61	0.62	0.68	0.64	0.74	0.66	0.82		
14- 0.68	0.91	0.7	1.06	0.72	1.20	0.74	1.42		
15- 0.76	1.75	0.78	2.18	0.8	2.78	0.82	3.5		
16- SIGB	24	1.0							
17- 0.0	0.0	0.02	0.2	0.03	0.27	0.04	0.29		
18- 0.05	0.3	0.25	0.3	0.33	0.33	0.4	0.36		
19- 0.45	0.4	0.5	0.44	0.55	0.52	0.58	0.57		
20- 0.6	0.61	0.62	0.68	0.64	0.74	0.66	0.82		
21- 0.68	0.91	0.7	1.06	0.72	1.20	0.74	1.42		
22- 0.76	1.75	0.78	2.18	0.8	2.78	0.82	3.5		
23- SIGC	24	1.0							
24- 0.0	0.0	0.02	0.2	0.03	0.27	0.04	0.29		
25- 0.05	0.3	0.25	0.3	0.33	0.33	0.4	0.36		
26- 0.45	0.4	0.5	0.44	0.55	0.52	0.58	0.57		
27- 0.6	0.61	0.62	0.68	0.64	0.74	0.66	0.82		
28- 0.68	0.91	0.7	1.06	0.72	1.20	0.74	1.42		
29- 0.76	1.75	0.78	2.18	0.8	2.78	0.82	3.5		
30- SIGD	24	1.0							
31- 0.0	0.0	0.02	0.2	0.03	0.27	0.04	0.29		
32- 0.05	0.3	0.25	0.3	0.33	0.33	0.4	0.36		
33- 0.45	0.4	0.5	0.44	0.55	0.52	0.58	0.57		
34- 0.6	0.61	0.62	0.68	0.64	0.74	0.66	0.82		
35- 0.68	0.91	0.7	1.06	0.72	1.20	0.74	1.42		
36- 0.76	1.75	0.78	2.18	0.8	2.78	0.82	3.5		

..... MODEL WEIGHT = 1283.60 (KG)
 MODEL HEIGHT = 9000.00 (MM)
 CORNER ANGLE = 0.00 (DEG.)
 INCREMENT DISP = 5.00 (MM)

..... K-FACTOR = 1.0000 1.0000 1.0000 1.0000 1.0000

Appendix B Sample Problem Output

I N P U T D A T A E C H O

DATA SEQ. NO.	1	2	3	4	5	6	7	8	
1	TITLE NUPAC-125B CASK VERTICAL IMPACT MODEL								
2	VERTICAL	1283.6	9000.0	0.0	5.0	0.0	MM		
3	LENGTH	1775.0	208.0	SIGD					
4	RUN	PLOT							
5	RCOORD	0.0	220.0	247.0	350.0	380.0			
6	ZCOORD	224.0	224.0	350.0	474.0				
7	ZCOORD								
8	MESH		1.0	1.0	1.0	1.0	1.0		
9	SIGA	24	1.0						
10	0.0	0.0	0.02	0.2	0.03	0.27	0.04	0.29	
11	0.05	0.3	0.25	0.3	0.33	0.33	0.4	0.36	
12	0.45	0.4	0.5	0.44	0.55	0.52	0.58	0.57	
13	0.6	0.61	0.62	0.68	0.64	0.74	0.66	0.82	
14	0.68	0.91	0.7	1.06	0.72	1.20	0.74	1.42	
15	0.76	1.75	0.78	2.18	0.8	2.78	0.82	3.5	
16	SIGB	24	1.0						
17	0.0	0.0	0.02	0.2	0.03	0.27	0.04	0.29	
18	0.05	0.3	0.25	0.3	0.33	0.33	0.4	0.36	
19	0.45	0.4	0.5	0.44	0.55	0.52	0.58	0.57	
20	0.6	0.61	0.62	0.68	0.64	0.74	0.66	0.82	
21	0.68	0.91	0.7	1.06	0.72	1.20	0.74	1.42	
22	0.76	1.75	0.78	2.18	0.8	2.78	0.82	3.5	
23	SIGC	24	1.0						
24	0.0	0.0	0.02	0.2	0.03	0.27	0.04	0.29	
25	0.05	0.3	0.25	0.3	0.33	0.33	0.4	0.36	
26	0.45	0.4	0.5	0.44	0.55	0.52	0.58	0.57	
27	0.6	0.61	0.62	0.68	0.64	0.74	0.66	0.82	
28	0.68	0.91	0.7	1.06	0.72	1.20	0.74	1.42	
29	0.76	1.75	0.78	2.18	0.8	2.78	0.82	3.5	
30	SIGD	24	1.0						
31	0.0	0.0	0.02	0.2	0.03	0.27	0.04	0.29	
32	0.05	0.3	0.25	0.3	0.33	0.33	0.4	0.36	
33	0.45	0.4	0.5	0.44	0.55	0.52	0.58	0.57	
34	0.6	0.61	0.62	0.68	0.64	0.74	0.66	0.82	
35	0.68	0.91	0.7	1.06	0.72	1.20	0.74	1.42	
36	0.76	1.75	0.78	2.18	0.8	2.78	0.82	3.5	

* * * I N P U T D A T A E N D * * *

Appendix B (Continued)

Appendix B (Continued)

NUPAC-125B CASK VERTICAL IMPACT MODEL

MODEL TYPE = VERTICAL

TOTAL ENERGY = 11552.40 (KG-M)

STEP	DEPTH (MM)	FORCE (KG)	ENERGY (KG-M)	ACCELERATION (G)
1	5.00	69082.74	345.41	53.82
2	10.00	110880.80	899.82	86.38
3	15.00	129096.61	1545.30	100.57
4	20.00	134048.81	2215.54	104.43
5	25.00	136093.81	2896.01	106.03
6	30.00	136093.81	3576.48	106.03
7	35.00	136093.81	4256.95	106.03
8	40.00	136093.81	4937.42	106.03
9	45.00	136093.81	5617.89	106.03
10	50.00	136093.81	6298.36	106.03
11	55.00	136093.81	6978.83	106.03
12	60.00	137377.28	7665.71	107.03
13	65.00	138981.62	8360.62	108.27
14	70.00	140585.97	9063.55	109.52
15	75.00	142239.82	9774.75	110.81
16	80.00	144073.35	10495.12	112.24
17	85.00	145906.88	11224.65	113.67
18	85.50	146090.24	11297.70	113.81
19	86.00	146273.59	11370.83	113.96
20	86.50	146456.94	11444.06	114.10
21	87.00	146640.30	11517.38	114.24
22	87.24	146727.76	11552.40	114.31

Appendix C Graphical Output

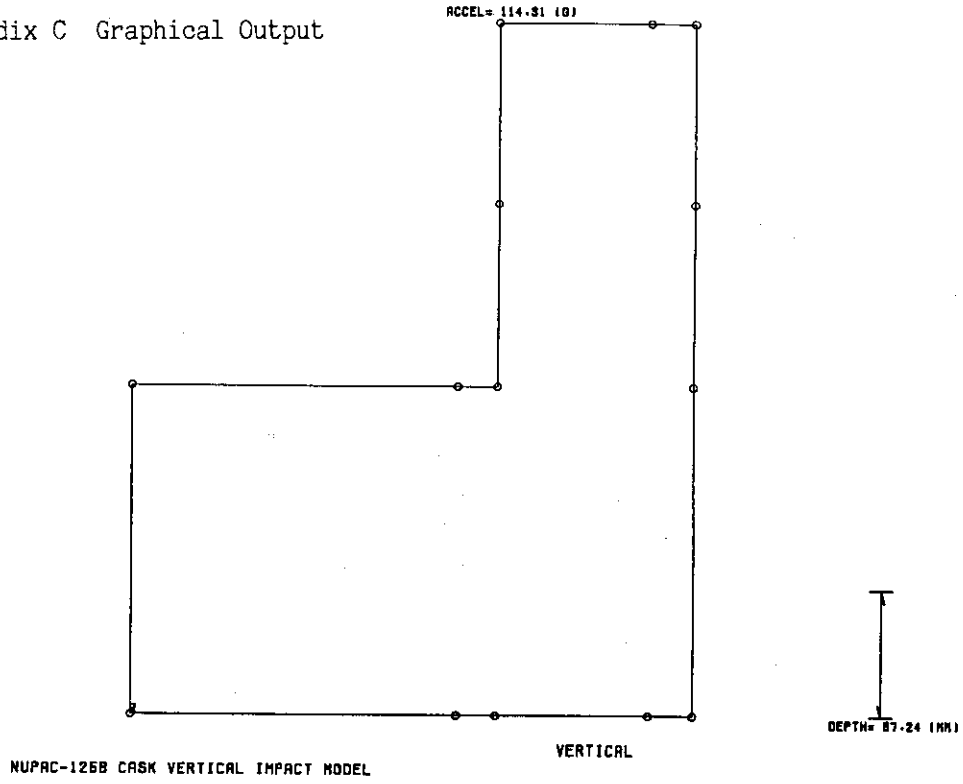


Fig.C.1 Graphical Output of CRUSH1(1)

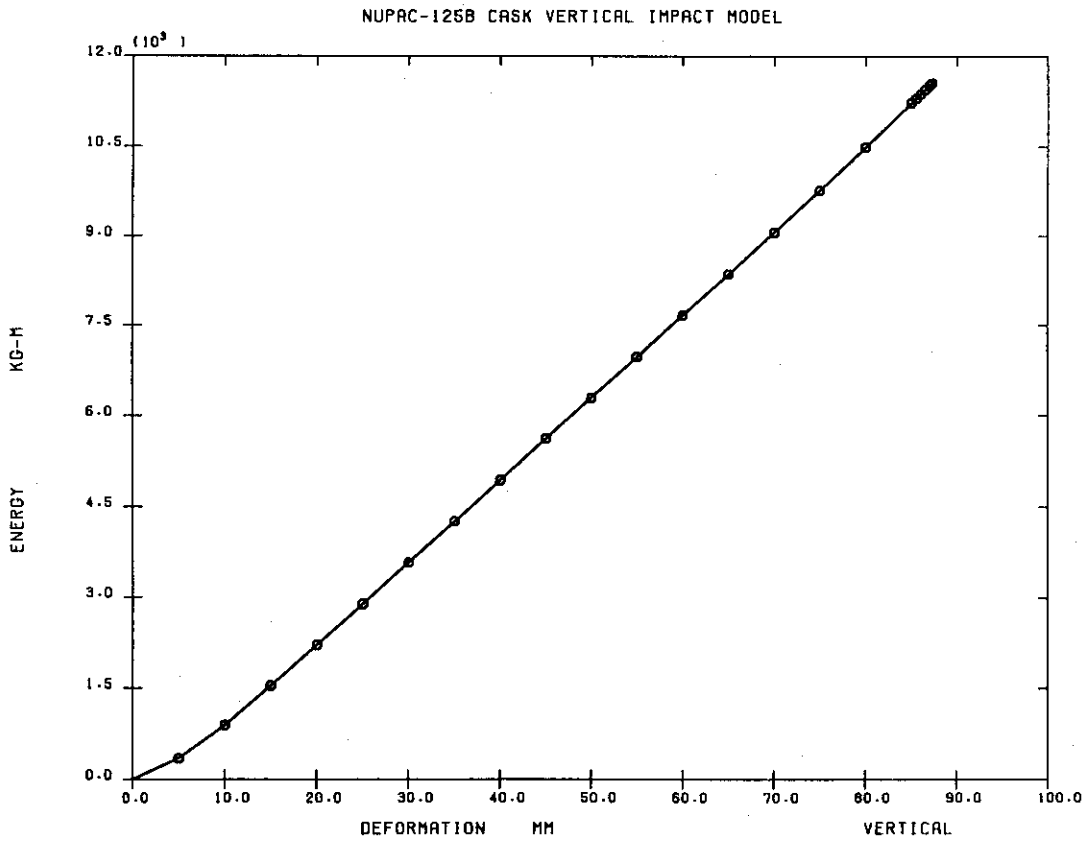


Fig.C.2 Graphical Output of CRUSH1(2)

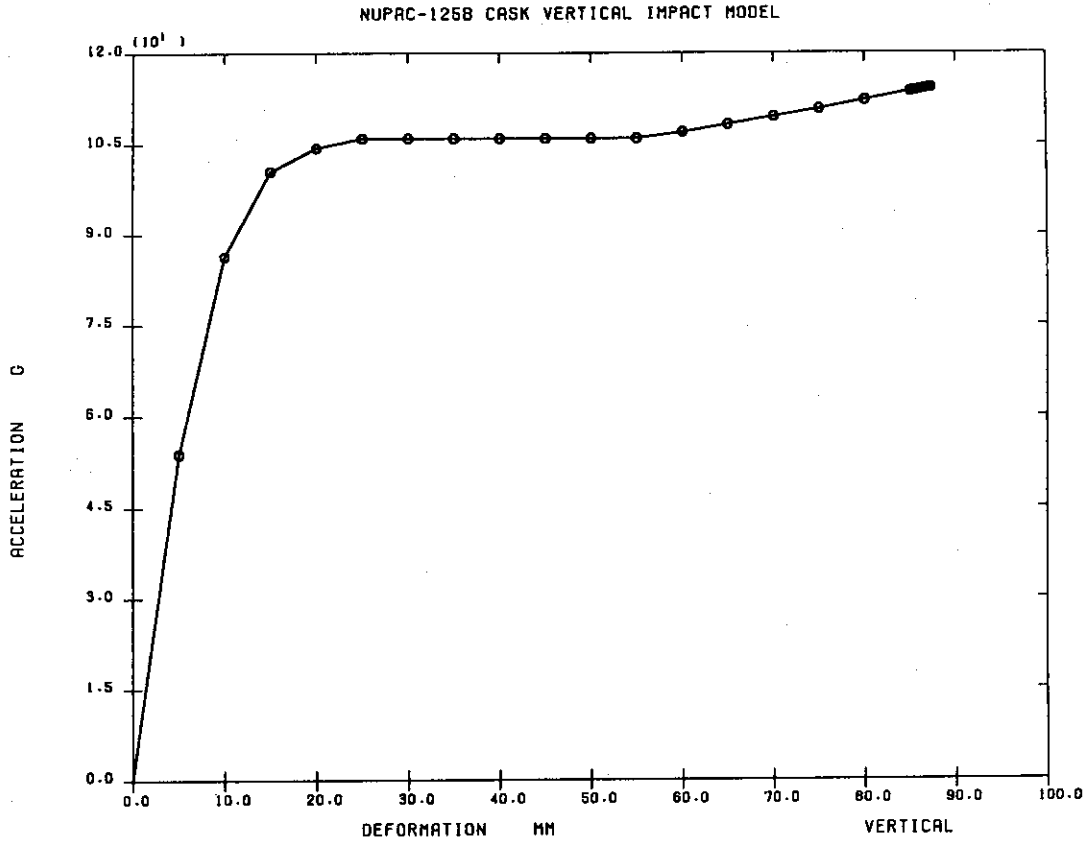


Fig.C.3 Graphical Output of CRUSH1(3)

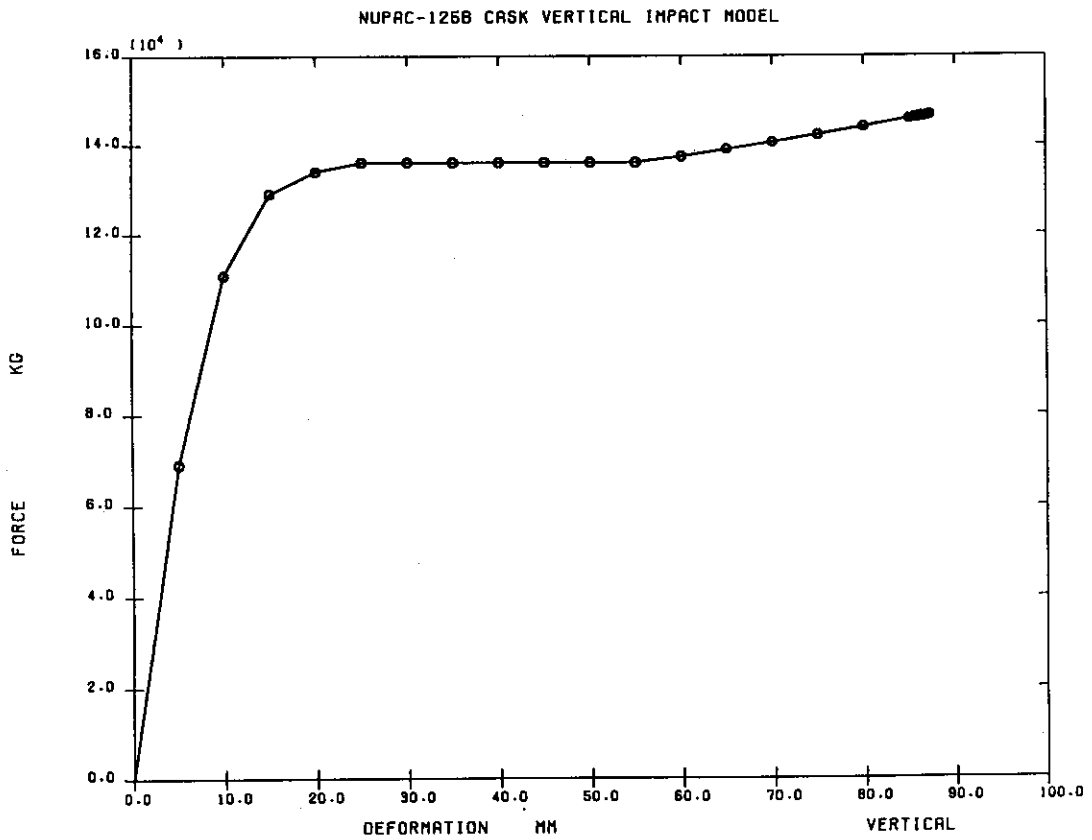


Fig.C.4 Graphical Output of CRUSH1(4)

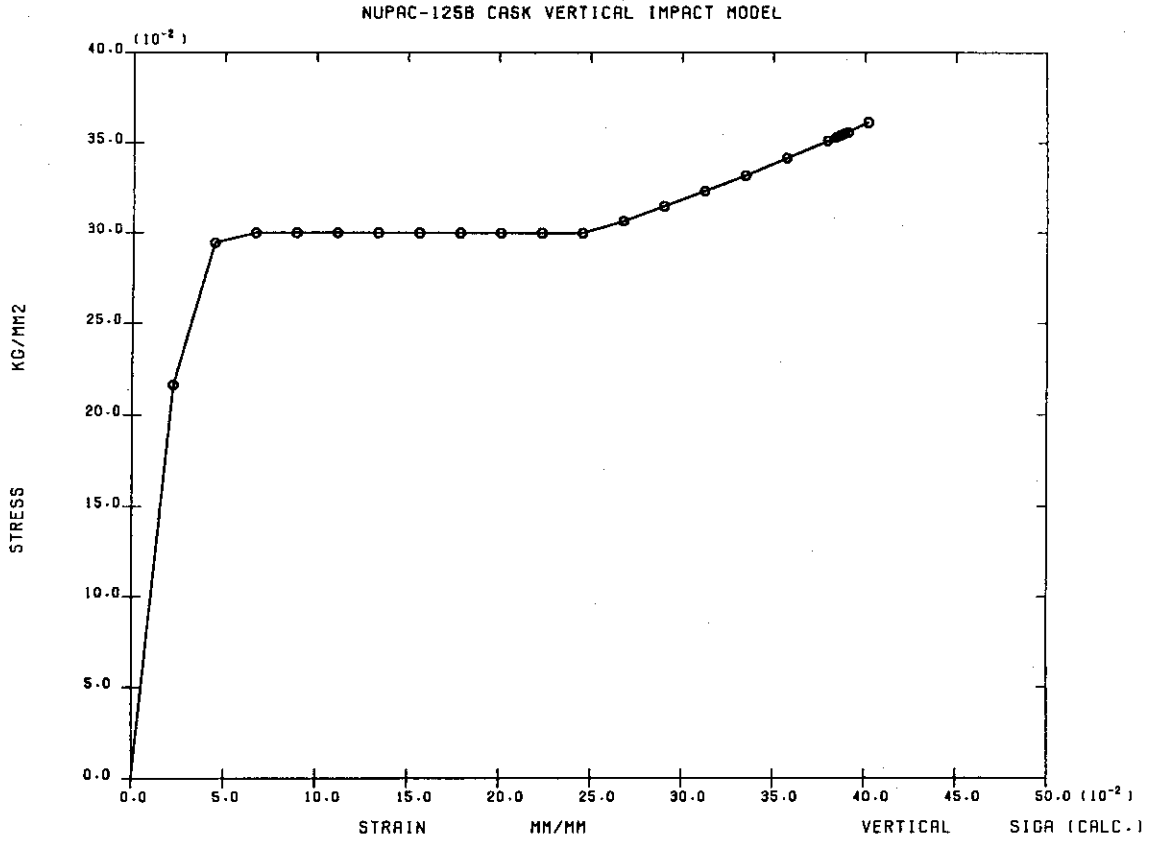


Fig.C.5 Graphical Output of CRUSH1(5)

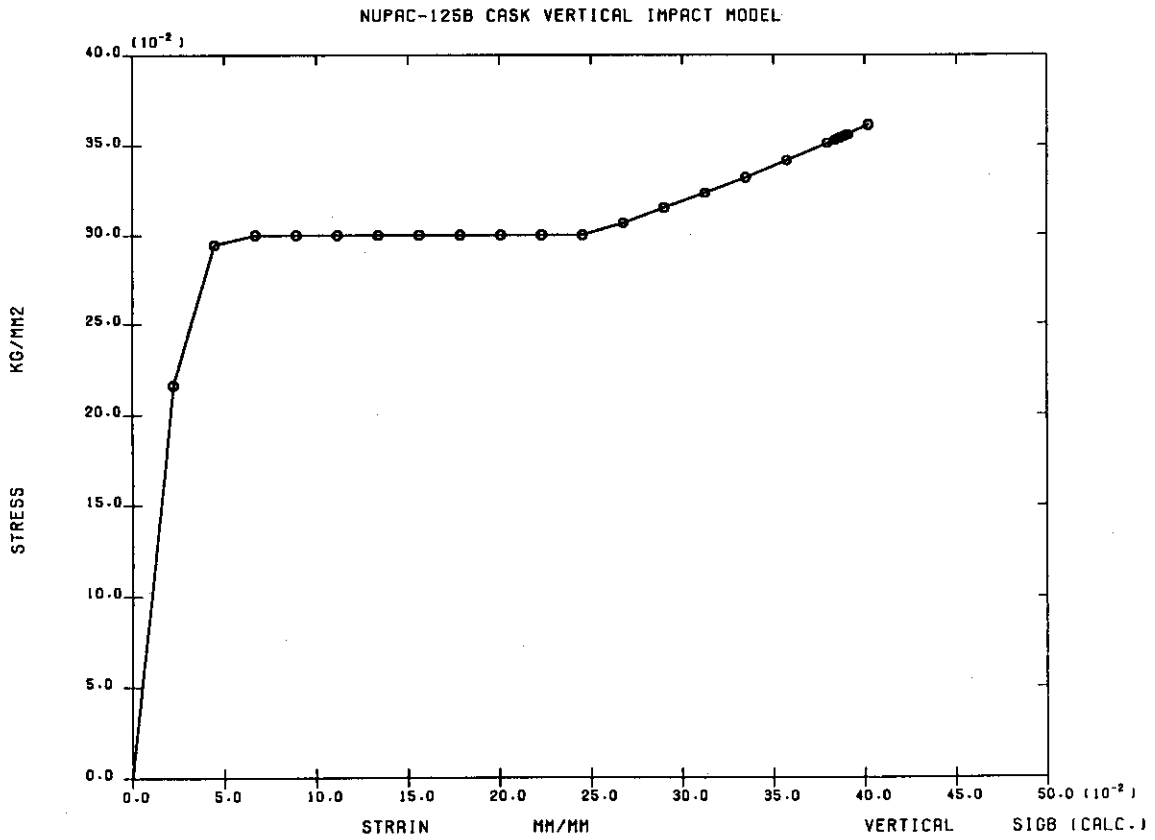


Fig.C.6 Graphical Output of CRUSH1(6)

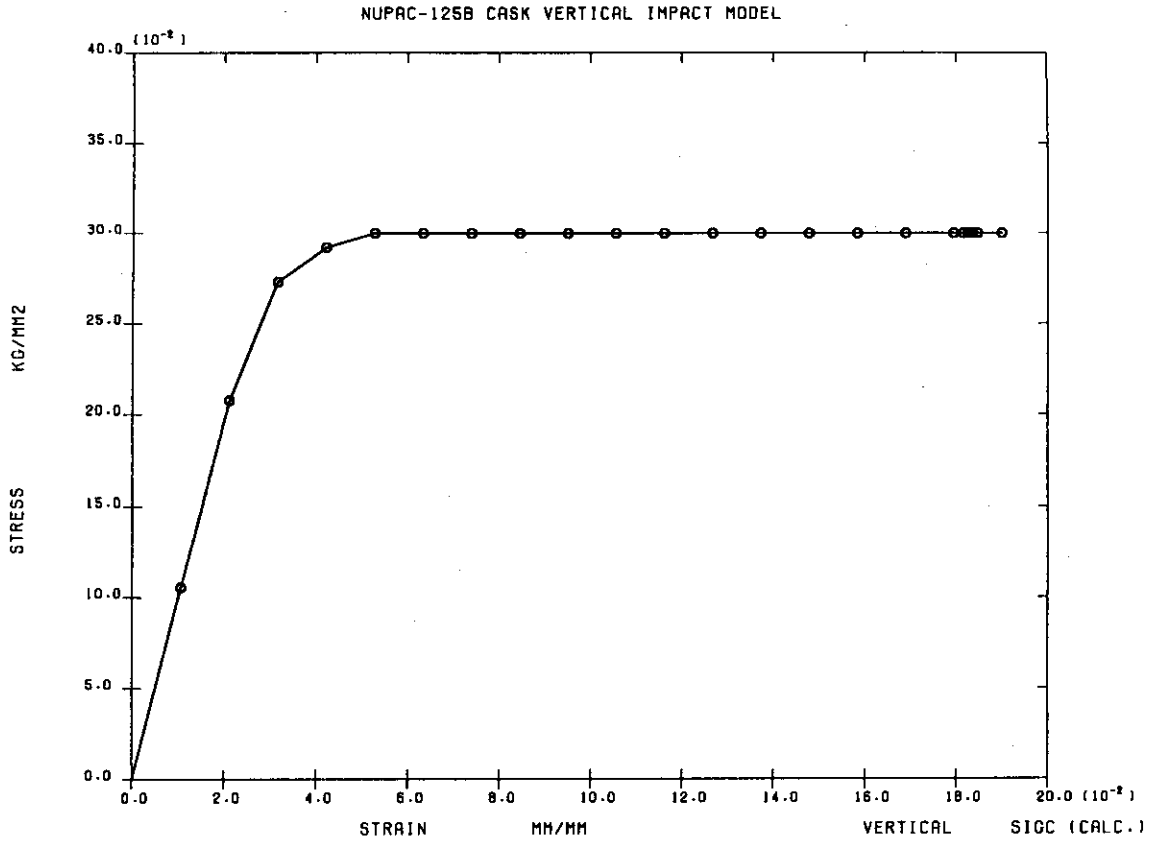


Fig.C.7 Graphical Output of CRUSH1(7)

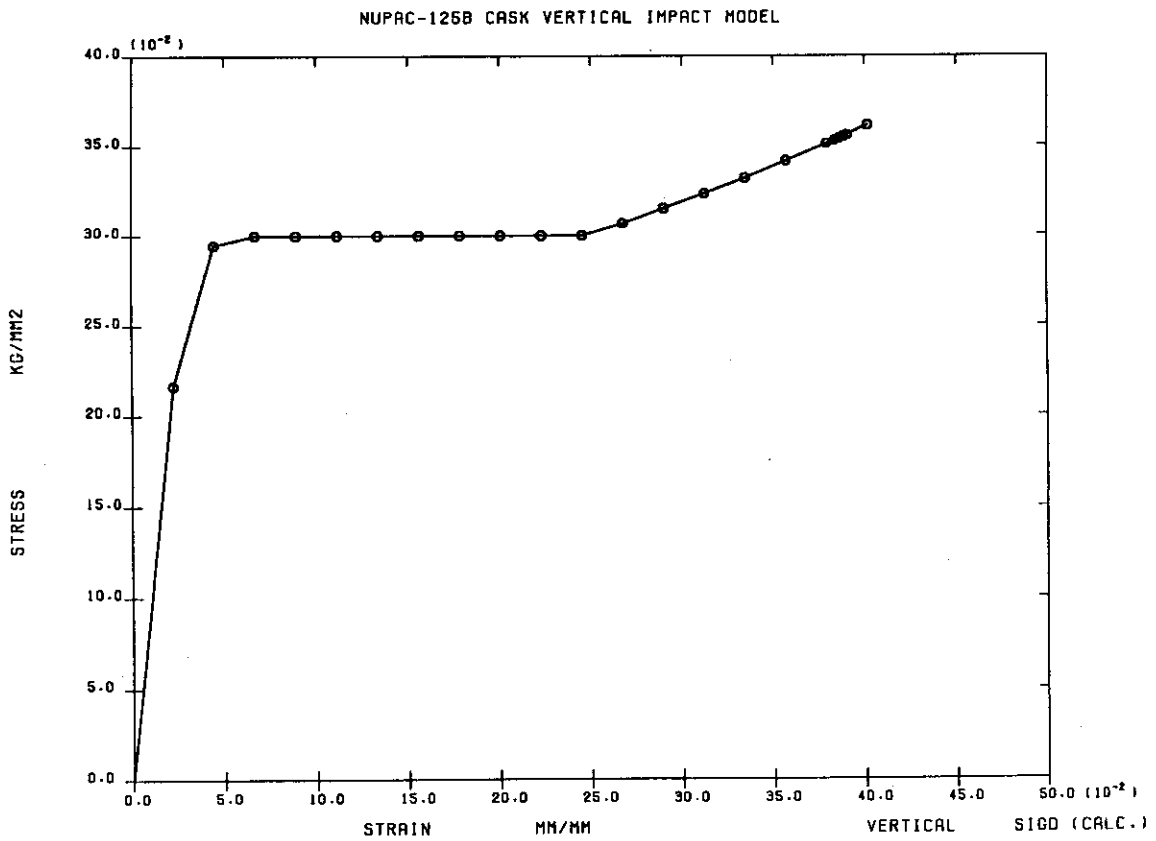


Fig.C.8 Graphical Output of CRUSH1(8)

Appendix D Job Control Data

The job control data for CRUSH1 execution on the computer FACOM M-780 in JAERI is as follows:

```
//JCLG JOB
// EXEC JCLG
//SYSIN DD DATA,DLM='++'
// JUSER XXXXXXXX.XX,XXXXXXXX,XXXX.XX,CRUSH1
// T.03 C.02 W.01 I.02 CLS GRP
// OPTP MSGCLASS=A,MSGLEVEL=(2,0,1),CLASS=B,NOTIFY=JXXXX
// OPTP PASSWORD=XXXXXXXX
// EXEC LMG0EX,LM=J2322.LMCRUSH1,PNM=CRUSH1
// EXPAND GRNLP
//FT55F001 DD DSN=JXXXX.DTCRUSH1.DATA,DISP=SHR
//FT02F001 DD DSN=SPACE=(TRK,(5,5)),UNIT=TSSWK
++
//
```

Appendix E Program Abstract

1. Name :
CRUSH1.
2. Computer for which the program is designed and others upon which it is possible:
FACOM M-780, SUN4 or IBM-PC.
3. Nature of physical problem solved:
Drop impact analysis of radioactive material transport casks.
4. Method of solutions:
One dimensional static analysis.
5. Restrictions on the complexity of the problem:
None.
6. Typical running time:
FACOM-M780 : 2 seconds.
SUN4 : 5 seconds.
IBM-PC : 10 seconds.
7. Unusual features of the program:
None.
8. Related and auxiliary program:
None.
9. Status:
-
10. References:
"Simplified Analysis Computer Program and Their Adequacy for Radioactive Materials Shipping casks", PATRAM'89 (Washington) DC, USA June 11-16, 1989).
11. Machine requirement:
Required 1100 k bytes of core memory.
12. Program language used:
FORTRAN-77.

13. Operating system or monitor under which the program is executed:

FACOM M-780 : MSP.

SUN4 : Solaris 2.1.

IBM PC : Windows 3.1, Windows 95 or Windows NT.

14. Any other programming or operating information or restrictions:

The program is approximately 2800 source steps (include comment lines). The graphical programs are used as follows:

FACOM M-780 : CALCOMP pltter or the compatible ones.

SUN4 : X-windows.

IBM PC : windows 3.1, Windows 95 or Windows NT.

15. Name and establishment of author:

T. Ikushima

Japan Atomic Energy Research Institute,

Tokai Research Establishment,

Department of Fuel Cycle Safety Research,

Tokai-mura, Naka-gun, Ibaraki-ken, 319-11

Japan

16. Material available:

Source.

Appendix F (Continued)

Appendix F Program Source List

```

C.....
C C R U S H I C O D E (1989-01-05)
C.....
C... /PROGRAMMED BY JAERI 1988.01.29
C... MODIFIED BY JAERI 1988.09.19
C... MODIFIED BY JAERI 1988.10.31
C... MODIFIED BY JAERI 1989.01.05
C... MODIFIED BY JAERI 1994.06.08
C
C IMPLICIT REAL*8 (A-H,O-Z)
COMMON /MODELS/
+ RCOORD(5), ZCOORD(9), WEIGHT, HEIGHT, ANGLE,
+ DISP, FACTK(5), ENERGO, GRAVIT, EPMAX,
+ MTINN, MTOUT, MODEL, IFLAG, KSOLV,
+ KOUT, NPART, KPART, NTITLE(18)
COMMON /MATDAT/ TABLE(100,2,5),
COMMON /RESLT/
+ DEPTH(500), FORCE(500), ACCEL(500),
+ ENERGO(500), ENERGY, TODISP,
+ EPSU(500,6), SIGU(500,6),
+ MSUSE(6)
COMMON /SWITCH/ KEOF, KKTEST
C
C DIMENSION LABEL(3,3), NDIMM(2)
C
C DATA LABEL/ 4HVERT, 4HCAL, 4H / 4HHORI, 4HZONT, 4HAL /
+ 4HCORN, 4HER, 4H /
C
C DATA NDIMM/ 4HMM / 4HCM /
C DATA NDIM / 4H /
C
C... GET MODEL
C
C KSWPLT = 0
C KEOF = 0
C
C 100 CONTINUE
C CALL CARDIN( MEOF )
C IF( MEOF .NE. 0 ) GO TO 8000
C
C... PLOT OR ANALYSIS
C
C IF( KSOLV.EQ.0 )
C IF( KOUT.EQ.0 )
C KSWPLT = KSWPLT + 1
C IF ( KSWPLT .EQ. 1 ) CALL PLTBGN
C CALL MPLOT ( LABEL(1,MODEL) )
C DO 200 J= 1, 5
C IF( MLENG(J).EQ.0 ) GO TO 200
C CALL DPLOT ( LABEL(1,MODEL), NTITLE, TABLE(1,1,J), TABLE(1,2,J),
+ MLENG(J), 3, IFLAG, J )
C 200 CONTINUE
C
C GO TO 8000
C
C... CALCULATION
C
C 300 CONTINUE
C GO TO ( 1000, 2000, 3000 ) , MODEL
C
C... VERTICAL
C 1000 CONTINUE
C CALL VERT
C GO TO 5000
C
C... HORIZONTAL
C 2000 CONTINUE
C CALL HORIZ
C GO TO 5000
C
C... CORNER
C 3000 CONTINUE
C CALL CONER
C
C... END OF ANALYSIS
C
C 5000 CONTINUE
C IF ( IFLAG .EQ. 0 ) NDIM = NDIMM(1)
C IF ( IFLAG .EQ. 1 ) NDIM = NDIMM(2)
C IF ( IFLAG .EQ. 0 ) FMETER = 0.001
C IF ( IFLAG .EQ. 1 ) FMETER = 0.01
C ENERGO = ENERGO * FMETER
C WRITE(MTOUT,9010) NTITLE
C 9010 FORMAT(1H1/5X,18A4//)
C WRITE(MTOUT,9020)
C 9020 FORMAT(5X,13HMODEL TYPE = ,3A4,10X,14HTOTAL ENERGY =, F12.2,
+ 1X, 6H(KG-M) // )
C
C WRITE(MTOUT,9030)
C 9030 FORMAT(10X,4HSTEP,12H DEPTH ,4X,11H FORCE,9X,6HENERGY,
+ 3X,12HACCELERATION )
C WRITE(MTOUT,9035) NDIM
C 9035 FORMAT(22X, 1H(, A2, 1H), 11X, 4H(KG), 9X, 6H(KG-M), 12X, 3H(G//)
C DO 5200 J= 1, MXSTEP
C ENERGO(J) = ENERGO(J) * FMETER
C WRITE(MTOUT,9040) J, DEPTH(J), FORCE(J), ENERGO(J), ACCEL(J)
C 9040 FORMAT(10X,14,F12.2,3X,F12.2,3X,F12.2,3X,F12.2)
C IF ( MOD( J , 50 ) .NE. 0 ) GO TO 5200
C WRITE(MTOUT,9050) NTITLE
C 9050 FORMAT(1H1/5X,18A4//
+ 10X,4HSTEP,12H DEPTH ,4X,11H FORCE,9X,6HENERGY,
+ 3X,12HACCELERATION )
C WRITE(MTOUT,9035) NDIM
C 5200 CONTINUE
C
C... PLOT RESULT
C
C 5500 CONTINUE
C IF( KOUT.EQ.0 )
C KSWPLT = KSWPLT + 1
C GO TO 100

```

Appendix F (Continued)

```

C
IF ( KSWPLT .EQ. 1 ) CALL PLTBGN
CALL UPLOT ( LABEL(1,MODEL), DEPTH(MXSTEP), ACCEL(MXSTEP) )
+
CALL DPLOT ( LABEL(1,MODEL), NTITLE, DEPTH, ENERG, MXSTEP, 1,
+ IFLAG, 0 )
+
CALL DPLOT ( LABEL(1,MODEL), NTITLE, DEPTH, ACCEL, MXSTEP, 2,
+ IFLAG, 0 )
+
CALL DPLOT ( LABEL(1,MODEL), NTITLE, DEPTH, FORCE, MXSTEP, 5,
+ IFLAG, 0 )
C
DO 5700 J= 1, 6
IF ( MSUSE(J).EQ.0 ) GO TO 5700
CALL DPLOT ( LABEL(1,MODEL), NTITLE, EPSU(1,J), SIGU(1,J),
+ MXSTEP, 4, IFLAG, J )
5700 CONTINUE
GO TO 100
C
8000 CONTINUE
IF ( KSWPLT .GT. 0 ) CALL PLTEND
STOP
END

```

SUBROUTINE CARDIN (MEOF)

```

C
C**** /PURPOSE/
READ MODEL DATA
C
C**** /OUTPUT/
MEOF = 0 :NORMAL RUN
MEOF = 1 :END OF INPUT DATA
C
C**** PROGRAMMED BY JAERI 1988.01.29
C**** MODIFIED BY JAERI 1988.09.19
C**** MODIFIED BY JAERI 1989.01.05
C
IMPLICIT REAL*8 (A-H,O-Z)
COMMON /MODELS/
RCOOR(5), ZCOOR(9), WEIGHT, HEIGHT, ANGLE,
+ DISP, FACTK(5), ENERGO, GRAVIT, EPMAX,
+ MTINN, MTOUT, MODEL, IFLAG, KSOLV,
+ KOUT, NPART, KPART, NTITLE(18)
COMMON /MODEL2/
CLENRG, RADIUS
COMMON /MATDAT/ TABLE(100,2,5), MLENG(5)
COMMON /RESLT/
DEPTH(500), FORCE(500), ACCEL(500),
+ ENERG(500), ENERGY, TODISP,
+ EPSU(500,6), SIGU(500,6), MXSTEP,
+ MSUSE(6)
COMMON /SIGCNV/ CONV, MXITER
COMMON /MODEL1/
S, S2, C2
COMMON /SWITCH/ KEOF, KKTEST
C
C

```

Appendix F (Continued)

```

+ DIMENSION LABEL(2,7), KOPT(2,3), KTYPE(2,5),
+ NAME(2), KEY(5), NIVAL(2), NCARD(20), NDIHM(2)
C
DATA NCARD/ 20*4H /
DATA LABEL/ 4HTITL, 4HE , 4HRCOO, 4HR ,
+ 4HTRUS, 4HS , 4HMODE, 4HL ,
+ 4HRUN, 4H , 4HMESH, 4H /
DATA KOPT/ 4HVERT, 4HICAL, 4HHORI, 4HZONT,
+ 4HCORN, 4HER /
DATA KTYPE/ 4HSIGA, 1, 4HSIGB, 2, 4HSIGC, 3, 4HSIGD, 4,
+ 4HSIGX, 5 /
DATA KEY/ 4HMM , 4HCM , 4HPLOT, 4H , 4HCHEC /
DATA LCOMM/ 4H*** /
DATA MFIN/ 4HFIN /
DATA RAD/ 0.01745329 /
DATA EPSS/ 1.0E-8 /
DATA NDIHM / 4HMM , 4HCM /
DATA NDIHM / 4H /
DATA KSIGN/ 4HNONE /
C
C**** FUNCTION
ABS(Z) = DABS(Z)
SIN(Z) = DSIN(Z)
COS(Z) = DCOS(Z)
C
C**** INITIAL VALUES
MTINN = 2
MTOUT = 6
MXITER = 20
DO 80 J= 1, 5
MSUSE(J) = 0
80 MLENG(J) = 0
MSUSE(6) = 0
C
DO 90 J= 1, 6
DO 90 I= 1, 100
EPSU(I,J)= 0.0
90 SIGU(I,J)= 0.0
C
C**** CARD IMAGE
LINE = 0
REWIND MTINN
IF ( KEOF.EQ.0 ) GO TO 110
MEOF = 1
RETURN
C
110 CONTINUE
READ(55,120,END=150) NCARD
120 FORMAT(20A4)
IF ( NCARD(1).EQ.MFIN ) GO TO 160
LINE = LINE+1
IF ( MOD(LINE,50).NE.1 ) GO TO 130

```


Appendix F (Continued)

Appendix F (Continued)

```

WRITE(MTOUT,100) (I,I=1,8)
100 FORMAT(1H1 //34X,34HC R U S H 1 M O D E L D A T A //
+ 10X,8I10/10X,10H.....1.....0,7(10H.....0) )
C
130 WRITE(MTOUT,140) LINE, NCARD
140 FORMAT(1X,14,5H-
,20A4)
C
IF( NCARD(1).EQ.1COMM ) GO TO 110
WRITE(MTINN,120) NCARD
GO TO 110
C... GET A COMMAND.
150 KEOF = 1
160 IF( LINE.GT.0 ) GO TO 170
MEOF = 1
RETURN
C. (1)
170 REWIND MTINN
READ(MTINN,211) NAME, NTITLE
211 FORMAT(2A4,2X,17A6,A2)
LINE = 1
IF( NAME(1).NE.LABEL(1,1) ) GO TO 999
IF( NAME(2).NE.LABEL(2,1) ) GO TO 999
C
C. (2)
READ(MTINN,221) NAME, WEIGHT, HEIGHT, ANGLE, DISP, CONV,
+ NDATA, EPMAX
221 FORMAT(2A4,2X,5F10.0,A6,6X,F10.0)
LINE = LINE+1
DO 222 J= 1, 3
IF( NAME(1).NE.KOPT(1,J) ) GO TO 222
IF( NAME(2).EQ.KOPT(2,J) ) GO TO 224
222 CONTINUE
GO TO 999
C
224 MODEL = J
IF( NDATA.EQ.KEY(1) ) IFLAG = 0
IF( NDATA.EQ.KEY(2) ) IFLAG = 1
IF( NDATA.EQ.KEY(4) ) IFLAG = 0
IF( DISP.EQ.0.0 ) DISP = 1.0
IF( CONV.EQ.0.0 ) CONV = 1.0D-5
EMERGO = WEIGHT*HEIGHT
IF( IFLAG.EQ.0 ) GRAVIT = 9800.0
IF( IFLAG.EQ.1 ) GRAVIT = 980.0
IF( IFLAG.EQ.0 ) NDIM = NDIMM(1)
IF( IFLAG.EQ.1 ) NDIM = NDIMM(2)
IF( EPMAX.EQ.0.0 ) EPMAX = 1.0
IF( EPMAX.LE.0.0 .OR. EPMAX.GT.1.0 ) GO TO 996
C
C. (2A)
READ(MTINN,225) CLENGT, RADIUS, KTYPED
225 FORMAT( 10X, 2F10.0,5X,A4)
LINE = LINE+1
IF( CLENGT.LT.0.0 ) GO TO 992

```

```

IF( RADIUS.LT.0.0 ) GO TO 994
KSIGDD = 0
IF( KTYPED.EQ.KTYPE(1,4) ) KSIGDD = 0
IF( KTYPED.EQ.KSIGN ) KSIGDD = 1
C. (3)
READ(MTINN,227) NAME, NAME1, NDATA
227 FORMAT(2A4,2X,A4,1X,A4)
LINE = LINE+1
DO 228 J= 5, 6
IF( NAME(1).NE.LABEL(1,J) ) GO TO 228
IF( NAME(2).EQ.LABEL(2,J) ) GO TO 229
228 CONTINUE
GO TO 999
C
229 KSOLV = J-5
KOUT = 0
IF( NAME1.EQ.KEY(3) ) KOUT = 1
KKTEST = 0
IF( NDATA.EQ.KEY(5) ) KKTEST = 1
C
C. (4)
READ(MTINN,231) NAME, (RCOOR(I),I=1,5)
231 FORMAT(2A4,2X,7F10.0/10X,7F10.0)
LINE = LINE+1
IF( NAME(1).NE.LABEL(1,2) ) GO TO 999
IF( NAME(2).NE.LABEL(2,2) ) GO TO 999
C
C. (5)
READ(MTINN,231) NAME, (ZCOOR(I),I=1,9)
LINE = LINE+2
IF( NAME(1).NE.LABEL(1,3) ) GO TO 999
IF( NAME(2).NE.LABEL(2,3) ) GO TO 999
DO 233 J= 2, 4
IF( RCOOR(J).LT.RCOOR(J-1) ) GO TO 999
IF( ZCOOR(J).LT.ZCOOR(J-1) ) GO TO 999
233 CONTINUE
IF( RCOOR(5).LT.RCOOR(4) ) GO TO 999
C
C. (6)
READ(MTINN,251) NAME, NNVAL, FACTK
251 FORMAT(2A4,2X,2I5,5F10.0)
LINE = LINE+1
IF( NAME(1).NE.LABEL(1,7) ) GO TO 999
IF( NAME(2).NE.LABEL(2,7) ) GO TO 999
IF( MODEL.NE.1 ) GO TO 253
DO 252 J= 1, 5
IF( FACTK(J).LT.0.0 ) GO TO 997
IF( FACTK(J).GT.1.0 ) GO TO 997
IF( FACTK(J).EQ.0.0 ) FACTK(J) = 1.0
252 CONTINUE
MXFACTK = 5
GO TO 260
C
253 IF( MODEL.NE.2 ) GO TO 255
NPART = NNVAL(1)
IF( NPART.EQ.0 ) NPART = 100

```

Appendix F (Continued)

```

IF( NPART.GT.300 )
DO 254 J= 1, 4
IF( FACTK(J).LT.0.0 )
IF( FACTK(J).GT.1.0 )
IF( FACTK(J).EQ.0.0 )
254 CONTINUE
MXFCTK = 4
GO TO 260

C
255 NPART = NVAL(1)
KPART = NVAL(2)
IF( NPART.EQ.0 )
IF( NPART.GT.300 )
IF( KPART.EQ.0 )
IF( KPART.GT.400 )
DO 256 J= 1, 4
IF( FACTK(J).LT.0.0 )
IF( FACTK(J).GT.1.0 )
IF( FACTK(J).EQ.0.0 )
256 CONTINUE
FACTK(S) = 1.0
MXFCTK = 4

```

C
C. (7)

```

260 CONTINUE
READ(MTINN,271,END=280) NAME, NN, FACT
271 FORMAT(2A4,2X,15,F10.0)
LINE = LINE+1
DO 272 J= 1, 5
IF( NAME(1).EQ.KTYPE(1,J) )
GO TO 274
272 CONTINUE
GO TO 999

C
274 ID = KTYPE(2,J)
IF( NN.EQ.0 )
IF( FACT.LT.0.0 )
IF( FACT.EQ.0.0 )
FACT = 1.0
MLENG(ID) = NN

```

C
C. DATA OF TABLE

```

READ(MTINN,275) (TABLE(I,1,1),TABLE(I,2,1), I=1,NN)
275 FORMAT(8F10.0)
LINE = LINE+1+(NN-1)/4
DO 276 I= 1, NN
TABLE(I,2,1) = FACT*TABLE(I,2,1)
276 CONTINUE
DO 277 I= 2, NN
IF( TABLE(I,1,1).LE.TABLE(I-1,1,1) )
GO TO 998
277 CONTINUE
DO 278 I= 2, NN
DS = ABS(TABLE(I,2,1)-TABLE(I-1,2,1))
IF( DS.GT.EPSS )
GO TO 278
TABLE(I,2,1) = TABLE(I,2,1)+TABLE(I-1,2,1)*EPSS
278 CONTINUE
IF( ID.NE.4 )
GO TO 260
IF( KSIIGD.EQ.0 )
MLENG(ID) = 0

```

Appendix F (Continued)

```

GO TO 260
C
C...
280 IF( MODEL.EQ.3 )
IF( MLENG(1).EQ.0 )
IF( MLENG(2).EQ.0 )
IF( MLENG(3).EQ.0 )
GO TO 900
GO TO 281
GO TO 999
GO TO 999
GO TO 999

C
281 CONTINUE
S = SIN(RAD*ANGLE)
C = COS(RAD*ANGLE)
S2 = S*S
C2 = C*C
NEWVVV = 0
IF( MLENG(5).GT.0 )
NEWVVV = 1
N1 = MLENG(1)
N2 = MLENG(2)
TABLE(N1+1,1,1) = TABLE(N2,1,2)+1.0
TABLE(N2+1,1,2) = TABLE(N1,1,1)+1.0
NN = 0
N1 = 1
N2 = 1

```

C...

```

285 IF( N1.GT.MLENG(1) .AND. N2.GT.MLENG(2) )
NN = NN+1
EPS1 = TABLE(N1,1,1)
SIG1 = TABLE(N1,2,1)
EPS2 = TABLE(N2,1,2)
SIG2 = TABLE(N2,2,2)
IF( ABS(EPS1-EPS2).GT.EPSS )
TABLE(NN,1,5) = EPS1
TABLE(NN,2,5) = C2*SIG1 + S2*SIG2
N1 = N1+1
N2 = N2+1
GO TO 285
287 IF( EPS2.GT.EPS1 )
K1 = N1
IF( N1.GT.MLENG(1) )
K1 = MLENG(1)
K0 = K1-1
DX = TABLE(K1,1,1)-TABLE(K0,1,1)
DY = TABLE(K1,2,1)-TABLE(K0,2,1)
SIG1 = TABLE(K0,2,1) + DY/DX*(EPS2-TABLE(K0,1,1))
TABLE(NN,1,5) = EPS2
TABLE(NN,2,5) = C2*SIG1 + S2*SIG2
N2 = N2+1
GO TO 285
288 K1 = N2
IF( N2.GT.MLENG(2) )
K1 = MLENG(2)
K0 = K1-1
DX = TABLE(K1,1,2)-TABLE(K0,1,2)
DY = TABLE(K1,2,2)-TABLE(K0,2,2)
SIG2 = TABLE(K0,2,2) + DY/DX*(EPS1-TABLE(K0,1,2))
TABLE(NN,1,5) = EPS1
TABLE(NN,2,5) = C2*SIG1 + S2*SIG2
N1 = N1+1
GO TO 285

```

Appendix F (Continued)

```

290 MLENG(5) = NN
C
C... END OF INPUT CARD.
C
900 CONTINUE
WRITE(MTOUT,910) WEIGHT, HEIGHT, NDM, ANGLE, DISP, NDM
910 FORMAT(////11X,22H..... MODEL WEIGHT =,F10.2,2X,4H(KG) /
+ 11X,22H..... MODEL HEIGHT =,F10.2,2X,4H(A2.1H) /
+ 11X,22H..... CORNER ANGLE =,F10.2,2X,6H(DEG.) /
+ 11X,22H..... INCREMENT DISP =,F10.2,2X,4H(A2.1H) /)
WRITE(MTOUT,920) (FACTK(J),J=1,NXFACTK)
920 FORMAT(/ 11X,22H..... K-FACTOR RETURN =,5F10.4)
IF( MODEL.NE.3 ) RETURN
IF( NEWVVV.EQ.0 ) RETURN
WRITE(MTOUT,930) (TABLE(I,1.5),TABLE(I,2.5),I=1,NN)
WRITE(MTOUT,940) (TABLE(I,1.5),TABLE(I,2.5),I=1,NN)
930 FORMAT(//)
CJ940 FORMAT(12H (EPS,SIG) =,2F12.5,3X,2F12.5,3X,2F12.5)
940 FORMAT(12H (EPS,SIG) =,2F12.5,3X,2F12.5,3X,2F12.5)
RETURN
C
C... ERROR LENGTH
992 WRITE(MTOUT,9921) CLENGT LENGTH VALUE.
9921 CALL EXIT
RETURN
C
C... ERROR RADIUS
994 WRITE(MTOUT,9924) RADIUS RADIUS VALUE.
9924 CALL EXIT
RETURN
C
C... ERROR EPMAK
996 WRITE(MTOUT,9961) EPMAK EPMAK VALUE.
9961 CALL EXIT
RETURN
C
C... ERROR K-FACTOR
997 WRITE(MTOUT,9971) J, FACTK(J)
9971 CALL EXIT
RETURN
C
C... ERROR TABLE
998 WRITE(MTOUT,9981) KTYPE(1,J), I
9981 CALL EXIT
RETURN
C
C... ERROR FORMAT CARD.
999 WRITE(MTOUT,9991) LINE
9991 CALL EXIT
RETURN
END

```

Appendix F (Continued)

```

SUBROUTINE VERT
C
C**** /PURPOSE/
CALCULATION ENERGY OF VERTICAL SHPAE
C
C*** PROGRAMMED BY JAERI 1988.01.29
C*** MODIFIED BY JAERI 1988.09.19
C
IMPLICIT REAL*8 (A-H,O-Z)
COMMON /MODELS/
RCOOR(5), ZCOOR(9), WEIGHT, HEIGHT, ANGLE,
DISP, FACTK(5), ENERGO, GRAVIT, EPMAK,
MTINN, MTOUT, MODEL, IFLAG, KSOLV,
KOUT, NPART, KPART, NTITLE(18)
COMMON /MATDAT/
TABLE(100,2,5), MLENG(5)
COMMON /RESLT/
DEPTH(500), FORCE(500), ACCEL(500),
ENERG(500), ENERGY, TODISP,
EPSU(500,6), SIGU(500,6), MXSTEP,
MSUSE(6)
C
C DIMENSION P(5), DL(2), A(5), ID(2)
DATA PAI/ 3.14159265 /
C... INITIAL STATUS
C
MXSTEP = 0
TODISP = 0.0
MINI = 0
DISPX = DISP
A(1) = PAI*RCOOR(1)**2
A(2) = PAI*(RCOOR(2)**2-RCOOR(1)**2)
A(3) = PAI*(RCOOR(3)**2-RCOOR(2)**2)
A(4) = PAI*(RCOOR(4)**2-RCOOR(3)**2)
A(5) = PAI*(RCOOR(5)**2-RCOOR(4)**2)
CRC WRITE(6,1) A
CRC 1 FORMAT(6H AREA=,F12.2)
C
DL(1) = ZCOOR(3)
DL(2) = ZCOOR(4)-ZCOOR(3)
ID(1) = 2
ID(2) = 3
KSS = 2
C... GET CURRENT FORCE
C
100 CONTINUE
MXSTEP = MXSTEP+1
IF( MXSTEP.GT.500 ) GO TO 1000
TODISP = TODISP + DISPX
C
C. (O-R1)
EPS = TODISP/ZCOOR(2)
CALL GETSIG ( EPS, 4, SIGM )
P(1) = A(1)*SIGM
C

```

Appendix F (Continued)

```

C. (R1-R2)
EPS = TODISP/ZCOOR(2)
CALL GETSIG ( EPS, 1, SIGM )
P(2) = A(2)*SIGM
C
C. (R2-R3)
CALL GETSIG ( EPS, 2, SIGM )
P(3) = A(3)*SIGM
C
C. (R3-R4)
CALL MULTI ( TODISP, DL, ID, KS3, SIGM, IERR )
IF( IERR.NE.0 )
P(4) = A(4)*SIGM
C
C. (R4-R5)
P(5) = A(5)*SIGM
C
C.... SUM OF EACH FORCES
SUM = 0.0
DO 300 J = 1, 5
300 SUM = SUM + P(J)*FACTK(J)
ENERGY = SUM*DISPX
C
DEPTH(MXSTEP) = TODISP
FORCE(MXSTEP) = SUM
ACCEL(MXSTEP) = SUM/WEIGHT
ENERG(MXSTEP) = ENERGY
IF( MXSTEP.NE.1 )
ENERG(MXSTEP) = ENERG(MXSTEP-1)+ENERGY
C
IF( ENERG(MXSTEP).LT.ENERGO )GO TO 100
IF( MINI.NE.0 )
GO TO 800
MINI = 1
TODISP = TODISP - DISPX
MXSTEP = MXSTEP-1
DISPX = 0.1*DISPX
GO TO 100
C
C.... INTERPOLATION OF LAST STEP
800 CONTINUE
CALL FITSTP
C
C.... END OF VERTICAL
1000 CONTINUE
IF( MXSTEP.GT.500 )
MXSTEP = 500
RETURN
C
C*** ERROR
2000 CONTINUE
WRITE(MTOUT,2099)
2099 FORMAT(// 29H *** MXSTEP, TODISP, KS3, ID
+ 5X,7HDEPTH =,F10.4,5X,11HNO. OF MAT=,I3,5X,5I3)
CALL EXIT
RETURN
END
    
```

Appendix F (Continued)

```

SUBROUTINE HORIZ
C**** /PURPOSE/
C CALCULATION ENERGY OF HORIZONTAL SHAPE
C
C*** PROGRAMMED BY JAERI 1988.02.05
C*** MODIFIED BY JAERI 1988.09.19
C*** MODIFIED BY JAERI 1994.08.01
C
IMPLICIT REAL*8 (A-H,O-Z)
COMMON /MODELS/
RCOORD(5), ZCOORD(9), WEIGHT, HEIGHT, ANGLE,
DISP, FACTK(5), ENERGO, GRAVIT, EPMAX,
+ MTINN, MOUT, MODEL, IFLAG, KSOVLV,
+ KOUT, NPART, KPART, NTITLE(18),
COMMON /MATDAT/
TABLE(100,2,5), MLENG(5)
COMMON /SWITCH/
KKEF, KKTEST
+ DEPTH(500), FORCE(500), ACCEL(500),
+ ENERG(500), ENERGO, TODISP,
+ EPSU(500,6), SIGU(500,6), MXSTEP,
+ MSUSE(6)
C
DIMENSION P(4), E(4), DZ(4), R(5), R0(3),
ID(3), MESH1(3), MESH2(2), XLENG(5,300,4)
C
FUNCTION
SQR(Z) = DSQRT(Z)
C.... INITIAL STATUS
MXSTEP = 0
TODISP = 0.0
MINI = 0
DISPX = DISP
C
C... GET (Z) LENGTH
DZ(1) = ZCOORD(1)
DZ(2) = ZCOORD(2)-ZCOORD(1)
DZ(3) = ZCOORD(3)-ZCOORD(2)
DZ(4) = ZCOORD(4)-ZCOORD(3)
IF( ZCOORD(5).GT.ZCOORD(4) ) GO TO 30
DO 20 I = 1, 4
20 DZ(I) = 2.0*DZ(I)
GO TO 50
30 DZ(1) = DZ(1) + ZCOORD(9)-ZCOORD(8)
DZ(2) = DZ(2) + ZCOORD(8)-ZCOORD(7)
DZ(3) = DZ(3) + ZCOORD(7)-ZCOORD(6)
DZ(4) = DZ(4) + ZCOORD(6)-ZCOORD(5)
C
C... GET (R) LENGTH
50 R(1) = RCOORD(1)
R(2) = RCOORD(2)-RCOORD(1)
R(3) = RCOORD(3)-RCOORD(2)
R(4) = RCOORD(4)-RCOORD(3)
R(5) = RCOORD(5)-RCOORD(4)
DPART = NPART
    
```

Appendix F (Continued)

```

C
DX = RCOOR(4)/DPART
MESH1(1) = R(1)/DX + 1
MESH1(2) = R(2)/DX + 1
MESH1(3) = NPART - (MESH1(1)+MESH1(2))
MESH2(1) = R(4)/DX + 1
MESH2(2) = NPART - MESH2(1)
IF( KKTEST.NE.0 ) WRITE(6,1)
1 FORMAT(6H MESH=,3I5,5X,2I5,5X,4F10.3)
MESH1, MESH2, DZ
C
C... (C-C) MESH DATA
MLOOP = 0
RO(1) = 0.0
RO(2) = RCOOR(1)
RO(3) = RCOOR(2)
C
DO 150 J= 1, 3
LOOP = MESH1(J)
DPART = LOOP
DX = R(J)/DPART
XX = RO(J) + 0.5*DX
DO 100 I= 1, LOOP
XR5 = SQRT( RCOOR(5)**2-XX**2 )
H = RCOOR(5)-XR5
MLOOP = MLOOP+1
XLENG(1,MLOOP,1) = DX
XLENG(2,MLOOP,1) = H
XLENG(1,MLOOP,2) = DX
XLENG(2,MLOOP,2) = H
C
GO TO ( 91, 92, 93) , J
91 XR1 = SQRT( RCOOR(1)**2-XX**2 )
XR2 = SQRT( RCOOR(2)**2-XX**2 )
IF( MLENG(4).EQ.0 ) GO TO 99
XLENG(3,MLOOP,1) = 2.0*(XR5-XR2)
XLENG(4,MLOOP,1) = 2.0*(XR2-XR1)
XLENG(5,MLOOP,1) = 2.0*XR1
XLENG(3,MLOOP,2) = 2.0*(XR5-XR2)
XLENG(4,MLOOP,2) = 2.0*(XR2-XR1)
XLENG(5,MLOOP,2) = 2.0*XR1
GO TO 100
99 XLENG(3,MLOOP,1) = XR5-XR2
XLENG(4,MLOOP,1) = XR2-XR1
XLENG(5,MLOOP,1) = 0.0
XLENG(3,MLOOP,2) = XR5-XR2
XLENG(4,MLOOP,2) = XR2-XR1
XLENG(5,MLOOP,2) = 0.0
GO TO 100
92 XR2 = SQRT( RCOOR(2)**2-XX**2 )
XLENG(3,MLOOP,1) = 2.0*(XR5-XR2)
XLENG(4,MLOOP,1) = 2.0*XR2
XLENG(5,MLOOP,1) = 0.0
XLENG(3,MLOOP,2) = 2.0*(XR5-XR2)
XLENG(4,MLOOP,2) = 2.0*XR2
XLENG(5,MLOOP,2) = 0.0
GO TO 100
93 XLENG(3,MLOOP,1) = 2.0*XR5
XLENG(4,MLOOP,1) = 0.0

```

Appendix F (Continued)

```

XLENG(5,MLOOP,1) = 0.0
XLENG(3,MLOOP,2) = 2.0*XR5
XLENG(4,MLOOP,2) = 0.0
XLENG(5,MLOOP,2) = 0.0
C
100 XX = XX + DX
150 CONTINUE
C
C... (D-D) MESH
MLOOP = 0
RO(2) = RCOOR(3)
C
DO 250 J= 1, 2
LOOP = MESH2(J)
DPART = LOOP
DX = R(J+3)/DPART
XX = RO(J) + 0.5*DX
DO 200 I= 1, LOOP
XR5 = SQRT( RCOOR(5)**2-XX**2 )
H = RCOOR(5)-XR5
MLOOP = MLOOP+1
XLENG(1,MLOOP,3) = DX
XLENG(2,MLOOP,3) = H
XLENG(1,MLOOP,4) = DX
XLENG(2,MLOOP,4) = H
C
IF( J.NE.1) GO TO 190
XR3 = SQRT( RCOOR(3)**2-XX**2 )
XLENG(3,MLOOP,3) = XR5-XR3
XLENG(4,MLOOP,3) = 0.0
XLENG(5,MLOOP,3) = 0.0
XLENG(3,MLOOP,4) = XR5-XR3
XLENG(4,MLOOP,4) = 0.0
XLENG(5,MLOOP,4) = 0.0
GO TO 200
190 XLENG(3,MLOOP,3) = 2.0*XR5
XLENG(4,MLOOP,3) = 0.0
XLENG(5,MLOOP,3) = 0.0
XLENG(3,MLOOP,4) = 2.0*XR5
XLENG(4,MLOOP,4) = 0.0
XLENG(5,MLOOP,4) = 0.0
C
200 XX = XX + DX
250 CONTINUE
C
C... GET CURRENT FORCE
1000 CONTINUE
MXSTEP = MXSTEP+1
IF( MXSTEP.GT.500 ) GO TO 2000
TODISP = TODISP + DISPX
C
C... Z = 0.0 - Z1
P(1) = 0.0
E(1) = 0.0
MLOOP = 0

```

Appendix F (Continued)

```

DO 600 J=1,3
LOOP = MESH1(J)
GO TO ( 571, 581, 591 ) , J
C
571 ID(1) = 2
ID(2) = 1
ID(3) = 4
KS3 = 3
IF( MLENG(4).EQ.0 ) KS3 = 2
DO 572 I=1, LOOP
MLOOP = MLOOP+1
U = TODISP-XLENG(2,MLOOP,1)
IF( U.LE.0.0 ) GO TO 572
CALL MULTI ( U, XLENG(3,MLOOP,1), ID, KS3, SIGM, IERR )
IF( IERR.NE.0 ) GO TO 3000
UO = DISPX
IF( U.LT.DISPX ) UO = U
P(1) = P(1) + SIGM*XLENG(1,MLOOP,1)
E(1) = E(1) + SIGM*XLENG(1,MLOOP,1)*UO
572 CONTINUE
GO TO 600
581 ID(1) = 2
ID(2) = 1
ID(3) = 2
KS3 = 2
DO 582 I=1, LOOP
MLOOP = MLOOP+1
U = TODISP-XLENG(2,MLOOP,1)
IF( U.LE.0.0 ) GO TO 582
CALL MULTI ( U, XLENG(3,MLOOP,1), ID, KS3, SIGM, IERR )
IF( IERR.NE.0 ) GO TO 3000
UO = DISPX
IF( U.LT.DISPX ) UO = U
P(1) = P(1) + SIGM*XLENG(1,MLOOP,1)
E(1) = E(1) + SIGM*XLENG(1,MLOOP,1)*UO
582 CONTINUE
GO TO 600
591 CONTINUE
MLOOP = MLOOP+1
IF( U.LE.0.0 ) GO TO 592
EPS = U/XLENG(3,MLOOP,1)
CALL GETSIG ( EPS, 2, SIGM )
UO = DISPX
IF( U.LT.DISPX ) UO = U
P(1) = P(1) + SIGM*XLENG(1,MLOOP,1)
E(1) = E(1) + SIGM*XLENG(1,MLOOP,1)*UO
592 CONTINUE
C
600 CONTINUE
C
C
C
C
Z = Z1- Z2
P(2) = 0.0
E(2) = 0.0
MLOOP = 0
DO 700 J=1,3
LOOP = MESH1(J)

```

Appendix F (Continued)

```

GO TO ( 671, 681, 691 ) , J
C
671 ID(1) = 2
ID(2) = 1
ID(3) = 4
KS3 = 3
IF( MLENG(4).EQ.0 ) KS3 = 2
DO 672 I=1, LOOP
MLOOP = MLOOP+1
U = TODISP-XLENG(2,MLOOP,2)
IF( U.LE.0.0 ) GO TO 672
CALL MULTI ( U, XLENG(3,MLOOP,2), ID, KS3, SIGM, IERR )
IF( IERR.NE.0 ) GO TO 3000
UO = DISPX
IF( U.LT.DISPX ) UO = U
P(2) = P(2) + SIGM*XLENG(1,MLOOP,2)
E(2) = E(2) + SIGM*XLENG(1,MLOOP,2)*UO
672 CONTINUE
GO TO 700
681 ID(1) = 2
ID(2) = 1
ID(3) = 2
KS3 = 2
DO 682 I=1, LOOP
MLOOP = MLOOP+1
U = TODISP-XLENG(2,MLOOP,2)
IF( U.LE.0.0 ) GO TO 682
CALL MULTI ( U, XLENG(3,MLOOP,2), ID, KS3, SIGM, IERR )
IF( IERR.NE.0 ) GO TO 3000
UO = DISPX
IF( U.LT.DISPX ) UO = U
P(2) = P(2) + SIGM*XLENG(1,MLOOP,2)
E(2) = E(2) + SIGM*XLENG(1,MLOOP,2)*UO
682 CONTINUE
GO TO 700
691 CONTINUE
MLOOP = MLOOP+1
IF( U.LE.0.0 ) GO TO 692
EPS = U/XLENG(3,MLOOP,2)
CALL GETSIG ( EPS, 2, SIGM )
UO = DISPX
IF( U.LT.DISPX ) UO = U
P(2) = P(2) + SIGM*XLENG(1,MLOOP,2)
E(2) = E(2) + SIGM*XLENG(1,MLOOP,2)*UO
692 CONTINUE
C
700 CONTINUE
C
C
C
C
Z = Z2- Z3
P(3) = 0.0
E(3) = 0.0
MLOOP = 0
DO 800 J=1,2
LOOP = MESH2(J)
DO 782 I=1, LOOP

```

Appendix F (Continued)

```

MLOOP = MLOOP+1
U = TODISP-XLENG(2,MLOOP,3)
GO TO 782
IF( U.LE.0.0 )
EPS = U/XLENG(3,MLOOP,3)
CALL GETSIG ( EPS, 2, SIGM )
UO = DISPX
IF( U.LT.DISPX ) UO = U
P(3) = P(3) + SIGM*XLENG(1,MLOOP,3)
E(3) = E(3) + SIGM*XLENG(1,MLOOP,3)*UO
782 CONTINUE
800 CONTINUE
C
C... Z = Z3- Z4
P(4) = 0.0
E(4) = 0.0
MLOOP = 0
C
DO 900 J= 1, 2
LOOP = MESH2(J)
DO 882 I= 1, LOOP
MLOOP = MLOOP+1
U = TODISP-XLENG(2,MLOOP,4)
IF( U.LE.0.0 ) GO TO 882
EPS = U/XLENG(3,MLOOP,4)
CALL GETSIG ( EPS, 3, SIGM )
UO = DISPX
IF( U.LT.DISPX ) UO = U
P(4) = P(4) + SIGM*XLENG(1,MLOOP,4)
E(4) = E(4) + SIGM*XLENG(1,MLOOP,4)*UO
882 CONTINUE
900 CONTINUE
C
C... SUM OF EACH FORCES
SUM = 0.0
TUM = 0.0
DO 1300 J= 1, 4
SUM = SUM + 2.0*P(J)*DZ(J)*FACTK(J)
TUM = TUM + 2.0*E(J)*DZ(J)*FACTK(J)
1300 ENERGY = TUM
C
DEPTH(MXSTEP) = TODISP
FORCE(MXSTEP) = SUM
ACCEL(MXSTEP) = SUM/WEIGHT
ENERG(MXSTEP) = ENERGY
IF( MXSTEP.NE.1 ) ENERGY = ENERG(MXSTEP-1)+ENERGY
C
IF( ENERG(MXSTEP).LT.ENERGO ) GO TO 1000
IF( MINI.NE.0 ) GO TO 1800
MINI = 1
TODISP = TODISP - DISPX
MXSTEP = MXSTEP-1
DISPX = 0.1*DISP
GO TO 1000
C... INTERPOLATION OF LAST STEP

```

Appendix F (Continued)

```

1800 CONTINUE
CALL FITSTP
C... END OF HORIZONTAL
2000 CONTINUE
IF( MXSTEP.GT.500 ) MXSTEP = 500
RETURN
C*** ERROR
3000 CONTINUE
WRITE(MTOUT,3099) MXSTEP, TODISP, KS3, ID
3099 FORMAT(// 29H *** MULTI-MATERIAL ERROR. 5X,7H STEP =,I5,
+ 5X,7HDEPTH =,F10.4,5X,11HNO. OF MAT=,I3,5X,5I3)
CALL EXIT
RETURN
END
SUBROUTINE CONER
C**** /PURPOSE/
CALCULATION ENERGY OF CORNER SHAPE
C
C*** PROGRAMMED BY JAERI 1988.02.09
MODIFIED BY JAERI 1988.09.19
C*** MODIFIED BY JAERI 1989.01.05
C
IMPLICIT REAL*8 (A-H,O-Z)
COMMON /MODELS/
RCOOR(5), ZCOOR(9), WEIGHT, HEIGHT, ANGLE,
+ DISP, FACTK(5), ENERGO, GRAVIT, EPMAX,
+ MTINN, MTOUT, MODEL, IFLAG, KSOLV,
+ KOUT, NPART, KPART, NTITLE(18)
COMMON /MATDAT/
TABLE(100,2,5), MLENG(5)
COMMON /S/
S2, C2
COMMON /RESLT/
DEPTH(500), FORCE(500), ACCEL(500),
+ ENERG(500), ENERGY, TODISP,
+ EPSU(500,6), SIGU(500,6), MXSTEP,
+ MSUSE(6)
COMMON /SWITCH/
KEDF, KKTEST
C
DIMENSION ST(2,18), SLICE(6,400),
+ MESH(3,400)
C... FUNCTION
SQRT(Z) = DSQRT(Z)
C... INITIAL STATUS
MXSTEP = 0
TODISP = 0.0
MINI = 0

```



```

KFNO = 1
IF( XX.GT.ST(1,14) )
XL2 = 0.0
XL1 = S2X*XX
IF( MLENG(4).GT.0 )
IF( XX.GT.ST(1,7) )
GO TO 90
H1 = RCOOR(1)*SX
GO TO 90

55 KFNO = 5
IF( XX.LE.ST(1,18) )
IF( XX.LE.ST(1,17) )
IF( XX.GT.0.0 )
KTYPE = 1
XL1 = 0.0
H1 = -S2X*XX
GO TO 58
57 KTYPE = 2
H1 = 0.0
XL1 = S2X*XX
58 IF( XX.LE.ST(1,9) )
IF( XX.GT.ST(1,9) )
GO TO 90
H1 = RCOOR(1)*SX
GO TO 90

C... ( P5(S)<S<Q0(S) )
60 IF( XX.GT.ST(1,15) )
XL2 = S2X*(ZCOOR(2)*S-XX)
KFNO = 5
IF( MLENG(4).GT.0 )
IF( XX.GE.ST(1,2) .AND. XX.LE.ST(1,3) )
IF( XX.GT.0.0 )
KTYPE = 4
XL1 = 0.0
H1 = XL2
IF( XX.LE.ST(1,6) )
H1 = XL2
GO TO 90
61 H1 = RCOOR(1)*SX
IF( XX.LT.ST(1,2) )
GO TO 90
63 KTYPE = 3
KFNO = 1
IF( XX.GT.ST(1,14) )
XL2 = 0.0
XL1 = S2X*XX
H1 = RCOOR(1)*SX
GO TO 90

65 IF( XX.GT.0.0 )
KTYPE = 4
XL1 = 0.0
H1 = -S2X*XX
GO TO 90
67 KTYPE = 6
H1 = 0.0
XL1 = S2X*XX
GO TO 90

C... ( Q0(S)<S<P11(S) )
70 IF( XX.GT.ST(1,11) )
KTYPE = 5
KFNO = 5

```

```

XL2 = 0.0
XL1 = S2X*XX
IF( MLENG(4).GT.0 )
IF( XX.GT.ST(1,7) )
GO TO 75
GO TO 75
GO TO 75
GO TO 100

C
IF( XX.GE.ST(1,2) .AND. XX.LE.ST(1,3) )
IF( XX.LE.ST(1,3) )
GO TO 73
IF( XX.LE.ST(1,7) )
KFNO = 2
IF( XX.LE.ST(1,14) )
KTYPE = 3
H1 = RCOOR(1)*SX
GO TO 90
73 KTYPE = 4
H1 = RCOOR(1)*SX
XL2 = H1
XL1 = 0.0
GO TO 90

C
75 H1 = S2X*(XX-ZCOOR(2)*S)
IF( XX.GT.ST(1,8) )
H1 = RCOOR(3)*SX
GO TO 90

C... 80 KTYPE = 3
XL2 = 0.0
H1 = S2X*(XX-ZCOOR(4)*S)
XL1 = S2X*XX
IF( XX.LE.ST(1,12) )
KFNO = 4
IF( XX.LE.ST(1,16) )
KFNO = 3

C... 90 CONTINUE
MFACE = MFACE+1
MESH(2,MFACE) = KTYPE
MESH(3,MFACE) = KFNO
SLICE(1,MFACE) = H0
SLICE(2,MFACE) = XL1
SLICE(3,MFACE) = XL2
SLICE(4,MFACE) = H1
SLICE(5,MFACE) = DX1
SLICE(6,MFACE) = XX
100 XX = XX + DX1
C
C...
CRC
XX = S2 + 0.5*DX2
XX = S2 + 0.1*DX2
DO 200 J= 1, M2
H0 = CS*(XX-RCOOR(5)*C)
C...
IF( XX.GT.ST(1,5) )
IF( MLENG(4).GT.0 )
KTYPE = 3
KFNO = 1
IF( XX.GT.ST(1,14) )
XL2 = 0.0
XL1 = R5MAX
H1 = RCOOR(1)*SX
GO TO 190
C
155 KFNO = 5

```

Appendix F (Continued)

```

IF( XX.LE.ST(1,18) )      KFNO = 2
IF( XX.LE.ST(1,17) )      KFNO = 1
KTYPE = 2
H1 = 0.0
XL1 = R5MAX
IF( XX.LE.ST(1,9) )
IF( XX.GT.ST(1,9) )
GO TO 190
C.. ( P5(S)<S<Q0(S) )
160 IF( XX.GT.ST(1,15) )
IF( MLENG(4).GT.0 )
KTYPE = 3
H1 = RCOOR(1)*SX
XL1 = R5MAX
XL2 = 0.0
KFNO = 1
IF( XX.GT.ST(1,14) )
GO TO 190
C 165 KTYPE = 6
KFNO = 5
H1 = 0.0
XL1 = R5MAX
XL2 = S2X*(ZCOOR(2)*S-XX)
GO TO 190
C.. ( Q0(S)<S<P11(S) )
170 IF( XX.GT.ST(1,11) )
KTYPE = 5
KFNO = 5
XL2 = 0.0
XL1 = R5MAX
IF( MLENG(4).GT.0 )
IF( XX.GT.ST(1,7) )
C
KTYPE = 3
KFNO = 2
IF( XX.LE.ST(1,14) )
H1 = RCOOR(1)*SX
GO TO 190
C 175 H1 = S2X*(XX-ZCOOR(2)*S)
IF( XX.GT.ST(1,8) )
GO TO 190
C.. KTYPE = 3
XL2 = 0.0
H1 = S2X*(XX-ZCOOR(4)*S)
XL1 = R5MAX
IF( XX.LE.ST(1,12) )
IF( XX.LE.ST(1,16) )
KFNO = 4
KFNO = 3
C.. 190 CONTINUE
MFACE = MFACE+1
MESH(2,MFACE) = KTYPE
MESH(3,MFACE) = KFNO
SLICE(1,MFACE) = H0
SLICE(2,MFACE) = XL1
SLICE(3,MFACE) = XL2

```

Appendix F (Continued)

```

SLICE(4,MFACE) = H1
SLICE(5,MFACE) = DX2
SLICE(6,MFACE) = XX
200 XX = XX + DX2
C
C... GET (MLOOP)
C
MLOOP = 0
DO 600 J= 1, MFACE
IF( MLOOP.GT.18000 ) GO TO 4000
IF( KKTEST.NE.0 )
+ WRITE(6,6) J, MESH(2,J),
MESH(3,J), (SLICE(I,J),I=1,6)
6 FORMAT(/6H FACE=,I3,7H KTYPE=,I3,7H KFACT=,I3,7H SLICE=,6F13.4)
KTYPE = MESH(2,J)
MLOOP1 = MLOOP+1
GO TO ( 310, 320, 330, 340, 350, 360 ) , KTYPE
C
C.. (1)-TYPE
310 CONTINUE
H = SLICE(4,J)
XM = S*SQRT( B-H**2 )
IF( SLICE(3,J).NE.R5MAX ) GO TO 315
ML = XM/DXO
IF( ML.EQ.0 )
DPART = ML
DX = XM/DPART
XX = J - 0.5*DX
MESH(1,J) = ML
DO 311 I= 1, ML
XL = SX*SQRT( R5-XX**2 )
MLOOP = MLOOP+1
XLENG(1,MLOOP) = 0.0
XLENG(2,MLOOP) = XL-H
XLENG(3,MLOOP) = DX
311 XX = XX+DX
GO TO 500
C 315 XMO = S*SQRT( B-SLICE(3,J)**2 )
MLOOP = MLOOP+1
XLENG(1,MLOOP) = 0.0
XLENG(2,MLOOP) = SLICE(3,J)-H
XLENG(3,MLOOP) = XMO
C
ML = (XM-XMO)/DXO
IF( ML.EQ.0 )
DPART = ML
DX = (XM-XMO)/DPART
XX = XMO + 0.5*DX
MESH(1,J) = ML+1
DO 316 I= 1, ML
XL = SX*SQRT( R5-XX**2 )
MLOOP = MLOOP+1
XLENG(1,MLOOP) = 0.0
XLENG(2,MLOOP) = XL-H
XLENG(3,MLOOP) = DX
316 XX = XX+DX
GO TO 500

```

Appendix F (Continued)

```

C
C. (2)-TYPE
320 CONTINUE
ML = RCOOR(5)/DXO
IF( ML.EQ.0 )
DPART = ML
DX = RCOOR(5)/DPART
XX = 0.5*DX
MESH(1,J) = ML
ML = 1
GO TO 325
IF( SLICE(2,J).NE.R5MAX )
IF( SLICE(3,J).NE.R5MAX )
DO 321 I= 1, ML
XL = SX*SQRT( R5-XX**2 )
MLOOP = MLOOP+1
XLENG(1,MLOOP) = R5MAX-XL
XLENG(2,MLOOP) = 2.0*XL
XLENG(3,MLOOP) = DX
321 XX = XX+DX
GO TO 500
C
322 XM1 = S*SQRT( B-SLICE(3,J)**2 )
DO 323 I= 1, ML
XL = SX*SQRT( R5-XX**2 )
MLOOP = MLOOP+1
XLENG(1,MLOOP) = R5MAX-XL
XLENG(2,MLOOP) = XL+SLICE(3,J)
XLENG(3,MLOOP) = DX
IF( XX.GT.XM1 )
XLENG(2,MLOOP) = 2.0*XL
323 XX = XX+DX
GO TO 500
C
325 XMO = S*SQRT( B-SLICE(2,J)**2 )
IF( SLICE(3,J).NE.R5MAX )
DO 326 I= 1, ML
XL = SX*SQRT( R5-XX**2 )
MLOOP = MLOOP+1
XLENG(1,MLOOP) = 0.0
XLENG(2,MLOOP) = XL+SLICE(2,J)
XLENG(3,MLOOP) = DX
IF( XX.LE.XMO )
GO TO 326
XLENG(1,MLOOP) = SLICE(2,J)-XL
XLENG(2,MLOOP) = 2.0*XL
326 XX = XX+DX
GO TO 500
C
327 XM1 = S*SQRT( B-SLICE(3,J)**2 )
DO 329 I= 1, ML
XL = SX*SQRT( R5-XX**2 )
MLOOP = MLOOP+1
IF( XX.GT.XMO )
GO TO 328
XLENG(1,MLOOP) = 0.0
XLENG(2,MLOOP) = XL+SLICE(2,J)
XLENG(3,MLOOP) = DX
IF( XX.LE.XM1 )
XLENG(2,MLOOP) = SLICE(2,J)+SLICE(3,J)
GO TO 329
328 XLENG(1,MLOOP) = SLICE(2,J)-XL
XLENG(2,MLOOP) = 2.0*XL

```

Appendix F (Continued)

```

XLENG(3,MLOOP) = DX
IF( XX.LE.XM1 )
XLENG(2,MLOOP) = SLICE(3,J)+XL
329 XX = XX+DX
GO TO 500
C
C. (3)-TYPE
330 CONTINUE
H = SLICE(4,J)
XM = S*SQRT( B-H**2 )
IF( SLICE(2,J).NE.R5MAX )
ML = XM/DXO
IF( ML.EQ.0 )
DPART = ML
DX = XM/DPART
XX = 0.5*DX
MESH(1,J) = ML
DO 331 I= 1, ML
XL = SX*SQRT( R5-XX**2 )
MLOOP = MLOOP+1
XLENG(1,MLOOP) = R5MAX-XL
XLENG(2,MLOOP) = XL-H
XLENG(3,MLOOP) = DX
331 XX = XX+DX
GO TO 500
C
335 XMO = S*SQRT( B-SLICE(2,J)**2 )
MLOOP = MLOOP+1
XLENG(1,MLOOP) = 0.0
XLENG(2,MLOOP) = SLICE(2,J)-H
XLENG(3,MLOOP) = XMO
ML = (XM-XMO)/DXO
IF( ML.EQ.0 )
DPART = ML
DX = (XM-XMO)/DPART
XX = XMO + 0.5*DX
MESH(1,J) = ML+1
DO 336 I= 1, ML
XL = SX*SQRT( R5-XX**2 )
MLOOP = MLOOP+1
XLENG(1,MLOOP) = SLICE(2,J)-XL
XLENG(2,MLOOP) = XL-H
XLENG(3,MLOOP) = DX
336 XX = XX+DX
GO TO 500
C
C. (4)-TYPE
340 CONTINUE
XMM = S*SQRT( G-SLICE(3,J)**2 )
H = SLICE(4,J)
XM = S*SQRT( B-H**2 )
ML = (XM-XMM)/DXO
IF( ML.EQ.0 )
DPART = ML
DX = (XM-XMM)/DPART
XX = XMM + 0.5*DX
MESH(1,J) = ML
DO 341 I= 1, ML

```

```

XL = SX*SQRT( R5-XX**2 )
MLOOP = MLOOP+1
XLENG(1,MLOOP) = 0.0
XLENG(2,MLOOP) = XL-H
XLENG(3,MLOOP) = DX
341 XX = XX+DX
GO TO 500
C
C.(5)-TYPE
350 CONTINUE
H = SLICE(4,J)
XMO = S*SQRT( B-SLICE(2,J)**2 )
XMM = S*SQRT( G-H**2 )
XXX = XMO
IF( XMM.LT.XMO ) XXX = XMM
MLOOP = MLOOP+1
XLENG(1,MLOOP) = 0.0
XLENG(2,MLOOP) = SLICE(2,J)-H
XLENG(3,MLOOP) = XXX
ML1 = 0
IF( XMO.GE.XMM ) GO TO 355
ML1 = (XMM-XMO)/DXO
IF( ML1.EQ.0 ) ML1 = 1
DPART = ML1
DX = (XMM-XMO)/DPART
XX = XMO + 0.5*DX
DO 351 I=1, ML1
XL = SX*SQRT( R5-XX**2 )
MLOOP = MLOOP+1
XLENG(1,MLOOP) = SLICE(2,J)-XL
XLENG(2,MLOOP) = XL-SLICE(4,J)
XLENG(3,MLOOP) = DX
351 XX = XX+DX
XXX = XMM
C
355 ML2 = 0
IF( XXX.GE.RCOOR(3) ) GO TO 357
ML2 = (RCOOR(3)-XXX)/DXO
IF( ML2.EQ.0 ) ML2 = 1
DPART = ML2
DX = (RCOOR(3)-XXX)/DPART
XX = XX + 0.5*DX
DO 352 I=1, ML2
XL = SX*SQRT( R5-XX**2 )
XMM = SX*SQRT( R3-XX**2 )
IF( XL.GT.SLICE(2,J) ) XL = SLICE(2,J)
MLOOP = MLOOP+1
XLENG(1,MLOOP) = SLICE(2,J)-XL
XLENG(2,MLOOP) = XL-XLM
XLENG(3,MLOOP) = DX
IF( XX.LE.XMO ) XLENG(2,MLOOP) = SLICE(2,J)-XLM
352 XX = XX+DX
C
357 ML3 = (RCOOR(5)-RCOOR(3))/DXO
IF( ML3.EQ.0 ) ML3 = 1
DPART = ML3
DX = (RCOOR(5)-RCOOR(3))/DPART

```

```

XX = RCOOR(3) + 0.5*DX
DO 353 I=1, ML3
XL = SX*SQRT( R5-XX**2 )
IF( XL.GT.SLICE(2,J) ) XL = SLICE(2,J)
MLOOP = MLOOP+1
XLENG(1,MLOOP) = SLICE(2,J)-XL
XLENG(2,MLOOP) = 2.0*XL
XLENG(3,MLOOP) = DX
353 XX = XX+DX
C
MESH(1,J) = ML1+ML2+ML3+1
GO TO 500
C
C.(6)-TYPE
360 CONTINUE
XMO = S*SQRT( B-SLICE(2,J)**2 )
XMM = S*SQRT( G-SLICE(3,J)**2 )
XXX = XMO
IF( XMM.LT.XMO ) XXX = XMM
MLOOP = MLOOP+1
XLENG(1,MLOOP) = 0.0
XLENG(2,MLOOP) = SLICE(2,J)+SLICE(3,J)
XLENG(3,MLOOP) = XXX
ML = (RCOOR(5)-XXX)/DXO
IF( ML.EQ.0 ) ML = 1
DPART = ML
DX = (RCOOR(5)-XXX)/DPART
XX = XX + 0.5*DX
DO 363 I=1, ML
XL = SX*SQRT( R5-XX**2 )
IF( XX.GT.XMO ) GO TO 361
MLOOP = MLOOP+1
XLENG(1,MLOOP) = 0.0
XLENG(2,MLOOP) = XL+SLICE(2,J)
XLENG(3,MLOOP) = DX
IF( XX.LE.XMM ) XLENG(2,MLOOP) = SLICE(2,J)+SLICE(3,J)
GO TO 363
361 MLOOP = MLOOP+1
XLENG(1,MLOOP) = SLICE(2,J)-XL
XLENG(2,MLOOP) = 2.0*XL
XLENG(3,MLOOP) = DX
IF( XX.LE.XMM ) XLENG(2,MLOOP) = SLICE(3,J)+XL
363 XX = XX+DX
MESH(1,J) = ML+1
C
500 CONTINUE
MLOOP2 = MLOOP
XMIN = 1.0D10
XMAX = 0.0
DO 550 I=MLOOP1,MLOOP2
IF( XLENG(2,I).LE.0.0 )
7 FORMAT(6H XLNG=I5,3F12.5)
IF( XLENG(2,I).GE.XMIN ) XMIN = XLENG(2,I)
IF( XLENG(2,I).LE.XMAX ) XMAX = XLENG(2,I)
550 IF( XLENG(2,I).LE.XMAX ) GO TO 550
WRITE(6,7) I, (XLENG(K,I),K=1,3)
GO TO 540
540 IF( XLENG(2,I).LE.XMAX ) GO TO 550

```

Appendix F (Continued)

```

XMAX = XLENG(2,I)
HMAX = XLENG(1,I)
550 CONTINUE
IF( KKTST.NE.0 )
+
8 FORMAT(6H LOOP=,2I5,6H LMIN=,F12.5,6H H1 =,F12.5,
+
6H LMAX=,F12.5,6H H1 =,F12.5)
600 CONTINUE
C
WRITE(6,9) MFACE, MLOOP
9 FORMAT(14H *TOTAL MESH *,5X,18HNO. OF (S-T) FACE=,I6,10X,
+
18HNO. OF (X-Y) FACE=,I6)
C
C... GET CURRENT FORCE
1000 CONTINUE
MXSTEP = MXSTEP+1
IF( MXSTEP.GT.500 ) GO TO 2000
TODISP = TODISP + DISPX
C
C...
SUM = 0.0
TUM = 0.0
MLOOP = 0
C
DO 1500 J= 1, MFACE
UO = TODISP - SLICE(1,J) GO TO 1450
IF( UO.LE.0.0 )
C
ML = MESH(1,J)
KFNO = MESH(3,J)
PSUM = 0.0
ESUM = 0.0
C
M6USE = 0
IF( J.LT.M6M1 ) GO TO 1100
IF( J.GT.M6M2 ) GO TO 1100
M6USE = 1
1100 CONTINUE
DO 1400 I= 1, ML
MLOOP = MLOOP+1
U = UO-XLENG(1,MLOOP)
IF( U.LE.0.0 ) GO TO 1400
C
EPS = U/XLENG(2,MLOOP)
IF( MXSTEP.GT.5 ) GO TO 1190
CRC IF( EPS.LE.0.30 ) GO TO 1190
CRC WRITE(6,19) MXSTEP, MLOOP, (XLENG(K,MLOOP),K=1,3), U, TODISP, UO
CRC19 FORMAT(6H OVER=,2I5,8F12.5)
C1190 CONTINUE
IF( EPS.LT.EPMAX ) GO TO 1200
EPS = EPMAX
U = EPMAX*XLENG(2,MLOOP)
C
1200 CONTINUE
CALL GETSIG ( EPS, 5, SIGM )
UU = DISPX

```

Appendix F (Continued)

```

IF( U.LT.DISPX ) UU = U
PSUM = PSUM + SIGM*XLENG(3,MLOOP)
ESUM = ESUM + SIGM*XLENG(3,MLOOP)*UU
C
IF( M6USE.EQ.0 ) GO TO 1400
IF( EPSU(MXSTEP,6).GT.EPS ) GO TO 1400
MSUSE(6) = MXSTEP
EPSU(MXSTEP,6) = EPS
SIGU(MXSTEP,6) = SIGM
1400 CONTINUE
SUM = SUM + 2.0*PSUM*SLICE(5,J)*FACTK(KFNO)
TUM = TUM + 2.0*ESUM*SLICE(5,J)*FACTK(KFNO)
GO TO 1500
C... (UPDATE MLOOP)
1450 ML = MESH(1,J)
MLOOP = MLOOP+ML
1500 CONTINUE
C
C... STEP RESULTANT
ENERGY = TUM
C
DEPTH(MXSTEP) = TODISP
FORCE(MXSTEP) = SUM
ACCEL(MXSTEP) = SUM/WEIGHT
ENERG(MXSTEP) = ENERGY
IF( MXSTEP.NE.1 ) ENERG(MXSTEP) = ENERG(MXSTEP-1)+ENERGY
C
IF( ENERG(MXSTEP).LT.ENERGO ) GO TO 1000
IF( MINI.NE.0 ) GO TO 1800
MINI = 1
TODISP = TODISP - DISPX
MXSTEP = MXSTEP-1
DISPX = 0.1*DISP
GO TO 1000
C
C... INTERPOLATION OF LAST STEP
1800 CONTINUE
CALL FITSTP
C
C... END OF CORNER
2000 CONTINUE
IF( MXSTEP.GT.500 ) MXSTEP = 500
RETURN
C
C... ERROR
4000 CONTINUE
WRITE(MTOUT,4099)
4099 FORMAT(/ / 29H *** (XLENG) CAPACITY OVER. )
RETURN
END

```

Appendix F (Continued)

Appendix F (Continued)

```

SUBROUTINE MULTI ( DISP, DL, ID, LENGTH, S I G M, IERR )
C**** /PURPOSE/
C GET STRESS FROM MULTI-MATERIAL TRUSS
C**** /OUTPUT/
C SIGM = TRUSS STRESS
C IERR = ERROR FLAG
C**** /INPUT/
C DISP = TOTAL DISPLACEMENT
C DL = ELEMENT LENGTH
C ID = MATERIAL IDENT
C LENGTH = NO. OF MATERIALS
C**** /PROGRAMMED BY JAERI 1988.01.29
C**** /MODIFIED BY JAERI 1988.09.19
C
C IMPLICIT REAL*8 (A-H,O-Z)
COMMON /SIGCONV/ CONV, MXITER
DIMENSION DL(1), ID(1)
DIMENSION EPS(3), B(3), SO(3), EPS1(3),
+ SK(9), DU(3), F(3),
+ KNAME(2)
DATA DLT/ 1.0E-10 /
C
C... FUNCTION
C ABS(Z) = DABS(Z)
C... INITIAL STATUS
C
EPS(1) = 0.0
EPS(2) = 0.0
EPS(3) = 0.0
IERR = 0
ITER = 0
C... INITIAL STATUS
C 100 CONTINUE
C ITER = ITER+1
C DO 200 J= 1, LENGTH
C CALL GETMAT ( EPS(J), ID(J), B(J), SO(J) )
C 200 CONTINUE
C... FIND SAME MATERIAL
C DO 300 J= 2, LENGTH
C I = J-1
C SAME(I) =ABS(B(J))-B(I))
C KNAME(I) = 0
C IF( SAME(I).LE.DLT ) KNAME(I) =1
C 300 CONTINUE
C... CALCULATION OF TRUSS DISP
C IF( LENGTH.NE.2 ) GO TO 400

```

```

IF( KNAME(1).NE.0 ) GO TO 3000
SK( 1) = 1.0
SK( 3) = 1.0
SK( 2) =-B(1)/DL(1)
SK( 4) = B(2)/DL(2)
F( 1) = DISP
F( 2) = SO(1)-SO(2)
CALL EGSOLV ( SK, F, 2, 0, IERR )
GO TO 500
C.... (3)-TRUSS
C 400 CONTINUE
C IF( KNAME(1).NE.0 ) GO TO 3000
C IF( KNAME(2).NE.0 ) GO TO 3000
SK( 1) = 1.0
SK( 4) = 1.0
SK( 7) = 1.0
SK( 2) =-B(1)/DL(1)
SK( 5) = B(2)/DL(2)
SK( 8) = 0.0
SK( 3) =-B(1)/DL(1)
SK( 6) = 0.0
SK( 9) = B(3)/DL(3)
F( 1) = DISP
F( 2) = SO(1)-SO(2)
F( 3) = SO(1)-SO(3)
CALL EGSOLV ( SK, F, 3, 0, IERR )
C
C... CONVERGENCE
C 500 CONTINUE
C IF( IERR.NE.0 ) GO TO 3000
C NCV = 0
C DO 600 J= 1, LENGTH
C EPS1(J) = DU(J)/DL(J)
C DCONV = EPS(J)-EPS1(J)
C IF( EPS1(J).NE.0.0 ) DCONV = DCONV/EPS1(J)
C IF( ABS(DCONV).LT.CONV ) NCV = NCV+1
C 600 EPS(J) = EPS1(J)
C IF( NCV.EQ.LENGTH ) GO TO 700
C... IF( ITER.LE.MXITER ) GO TO 100
C GO TO 1000
C... END OF MULTI
C 700 CONTINUE
C CRC WRITE(6,1) (ID(I),DL(I),DU(I),EPS1(I),I=1,LENGTH)
C CRC 1 FORMAT(6H MULT=,I4,6H DL =,F12.4,6H DU =,F12.4,6H EPS=,F12.4)
C
C DO 800 J= 1, LENGTH
C CALL GETSIG ( EPS(J), ID(J), S I G M )
C 800 CONTINUE
C RETURN
C... ERROR MULTI
C 1000 IERR = 1

```

Appendix F (Continued)

Appendix F (Continued)

```

SIGM = 0.0
RETURN
C
C
C... SAME MATERIAL OPTION
C
3000 CONTINUE
BLENGX = 0.0
DO 3100 J= 1, LENGTH
3100 BLENGX = BLENGX+DL(J)
EPS(I) = DISP/BLENGX
C
DO 3200 J= 1, LENGTH
CALL GETSIG ( EPS(I), ID(J), S I G M )
3200 CONTINUE
IERR = 0
RETURN
END

SUBROUTINE GETMAT ( EPS, ID, B K , S I G O )
C
C**** /PURPOSE/
GET STIFFNESS AND INITIAL SIGM
C
C**** /OUTPUT/
BK = STIFFNESS
SIGO = INITIAL SIGM
C
C**** /INPUT/
EPS = STRAIN
ID = MATERIAL IDENT
C
C**** /PROGRAMMED BY JAERI
1988.01.29
C**** /MODIFIED BY JAERI
1988.09.19
C
IMPLICIT REAL*8 (A-H,O-Z)
COMMON /MATDAT/ TABLE(100,2,5), MLENG(5)
COMMON /RESLT/
+ DEPTH(500), FORCE(500),
+ ENRG(500), ENERGY,
+ EPSU(500,6), SIGU(500,6),
+ MSUSE(6)
DATA DLT/ 1.0E-20 /
C
FUNCTION
ABS(Z) = DABS(Z)
C
C... INITIAL VALUE
C
LENGTH = MLENG(ID)
IF ( LENGTH.GT.1 )
BK = 1.0
SIGO = 0.0
RETURN
C

```

Appendix F (Continued)

```

C      RETURN
C      ERROR
C***  1600 CONTINUE
      SIGO = 0.0
      RETURN
      END

SUBROUTINE EGSOLV ( BK, F, NN, D, IERR )
      1988.01.29

C**** /PURPOSE/
      EQUATION SOLVER (BK)*D) = (F)

C*** /OUTPUT/
      D = DISP VECTOR
      IERR = ERROR FLAG

C*** /INPUT/
      BK = MATRIX
      F = LOAD VECTOR

C*** /PROGRAMMED BY JAERI
      1988.01.29

C      IMPLICIT REAL*8 (A-H,O-Z)
      DIMENSION BK(MN,NN), F(NN), D(NN)
      DATA EPS/ 1.0D-20 /

C      DO 500 J= 1, NN
      PIVOT = BK(J,J)
      IF( DABS(PIVOT).LE.EPS ) GO TO 1000

      DO 300 I= 1, NN
      IF( I.EQ.J ) GO TO 300
      IF( DABS(BK(I,J)).LT.EPS ) GO TO 300
      SHIFT = BK(I,J)/PIVOT

      DO 100 K= 1, NN
      BK(I,K) = BK(I,K) - SHIFT*BK(J,K)

      F(I) = F(I) - SHIFT*F(J)
300 CONTINUE
500 CONTINUE

C....
      DO 600 J= 1, NN
      D(J) = F(J)/BK(J,J)
      IERR = 0
      RETURN

C*** ERROR MATRIX
      1000 CONTINUE
    
```

Appendix F (Continued)

```

IERR = 1
      RETURN
      END

SUBROUTINE FITSTP
      1988.09.19

C**** /PURPOSE/
      RESET CALCULATED ENERGY OF LAST STEP

C*** PROGRAMMED BY JAERI
      1988.09.19

C      IMPLICIT REAL*8 (A-H,O-Z)
      COMMON /MODELS/
      + RCOORD(5), ZCOORD(9), WEIGHT, HEIGHT, ANGLE,
      + DISP, FACTK(5), ENERGO, GRAVIT, MTINN,
      + MTOUT, MODEL, IFLAG, KSOLV, KOUT,
      + COMMON /RESLT/
      + DEPTH(500), FORCE(500), ACCEL(500),
      + ENERGL(500), ENERGY, TODISP,
      + EPSU(500,6), SIGU(500,6), MXSTEP,
      + MSUSE(6)
      DATA EPS/ 0.05 /

C      FUNCTION
      ABS(Z) = DABS(Z)

C....
      GO TO 200

C      100 MXSTEP = MXSTEP-1

C      200 CONTINUE
      NXSTEP = MXSTEP-1
      DL1 = ENERGL(MXSTEP)-ENERG(NXSTEP)
      DL2 = ENERGO -ENERG(NXSTEP)
      IF( ABS(DL2).LE.EPS ) GO TO 100
      DLT = DL2/DL1

      DEPTH(MXSTEP) = DLT*(DEPTH(MXSTEP)-DEPTH(NXSTEP)) + DEPTH(NXSTEP)
      FORCE(MXSTEP) = DLT*(FORCE(MXSTEP)-FORCE(NXSTEP)) + FORCE(NXSTEP)
      ACCEL(MXSTEP) = DLT*(ACCEL(MXSTEP)-ACCEL(NXSTEP)) + ACCEL(NXSTEP)
      ENERGL(MXSTEP) = ENERGO

      RETURN
      END

SUBROUTINE MPLOT ( LABEL )

C**** /PURPOSE/
      GEOMETRY PLOT

C*** /INPUT/
    
```


Appendix F (Continued)

```

C
CALL PLMARK ( X(J), Y(J), RR )
700 CONTINUE
C
IF( NN.EQ.NODE ) GO TO 1000
MM = NODE+1
CALL PLINES ( X(MM), Y(MM), NODE )
DO 800 J= MM, NN
CALL PLMARK ( X(J), Y(J), RR )
800 CONTINUE
C
C... AXES PLOT
C
1000 CONTINUE
DO 1200 J= 1, 2
XV(1) = X(NN+1)
YV(1) = Y(NN+1)
XV(2) = X(NN+2)
YV(2) = Y(NN+2)
XV(5) = X(NN+2)
YV(5) = Y(NN+2)
NN = NN+2
C
X(1) = XV(2)-XV(1)
Y(1) = YV(2)-YV(1)
DLL = SQRT( X(1)**2+Y(1)**2 )
F1 = (DLL-XL)/DLL
F2 = (DLL+XL)/DLL
X(2) = XV(1) + F1*X(1)
Y(2) = YV(1) + F1*Y(1)
XV(6) = XV(1) + F2*X(1)
YV(6) = YV(1) + F2*Y(1)
X(1) = X(1)/DLL
Y(1) = Y(1)/DLL
C
XV(3) = X(2) - Y(1)*YL
XV(4) = X(2) + Y(1)*YL
YV(3) = Y(2) + X(1)*YL
YV(4) = Y(2) - X(1)*YL
CALL PLINES ( XV, YV, 5 )
1200 CONTINUE
C
C... HEADER PLOT
C
XX = 10.0
YY = 0.25*YO
CALL PSYM ( XX, YY, HH, NTITLE, 0.000, 70 )
C
XX = 150.0
YY = 0.50*YO
CALL PSYM ( XX, YY, HH, LABEL, 0.000, 12 )
C
C...
CALL PLTEOR
RETURN
END

```

Appendix F (Continued)

```

C
SUBROUTINE DPLOTT ( LABEL, NTITLE, XVAL, YVAL, LENGTH, KTYPE,
+ IFLAG, ID )
C
C**** /PURPOSE/
XYPLOT PROGRAM
C
C**** /INPUT/
LABEL = ANALYSIS TYPE LABEL
NTITLE = MODEL TITLE MESSAGE
XVAL = X VARIABLES
YVAL = Y VARIABLES
LENGTH = NO. OF VARIABLES
KTYPE = 1 ENERGY PLOT
= 2 ACCELERATION PLOT
= 3 STRAIN-STRESS PLOT
= 4 STRAIN-STRESS PLOT (WITH CALCULATED)
= 5 FORCE PLOT
IFLAG = 0, MM
ID = MATERIAL IDENT
C
C**** PROGRAMMED BY JAERI 1988.02.01
C**** MODIFIED BY JAERI 1988.09.19
C
IMPLICIT REAL*8 (A-H,O-Z)
DIMENSION LABEL(3), NTITLE(18)
DIMENSION XVAL(1), YVAL(1)
DIMENSION X(5), Y(5), NP(2,5), KSCL(2), KHEAD(3,6),
+ MLAB(6), KKCM(2), MH2H(2), MHEAD(2,6), MCALC(2),
+ KKCM2(2)
DATA NP/ 1, 2, 1, 3, 4, 5, 4, 5, 1, 6 /
DATA KHEAD/ 4HDEF0, 4HRMAT, 4HION,
+ 4HENER, 4HGY, 4H,
+ 4HACCE, 4HLERA, 4HTION,
+ 4HSTRA, 4HIN, 4H,
+ 4HSTRE, 4HSS, 4H,
+ 4HFORC, 4HE, 4H /
DATA MHEAD/ 4H, 4H,
+ 4H G, 4H,
+ 4H KG/, 4H,
+ 4H(10, 4H) /
DATA KSCL/ 4HSIGA, 4HSIGB, 4HSIGC, 4HSIGD, 4HSIG, 4HSIGX/
DATA MLAB/ 4HMM, 4HMM, 4HMM /
DATA KKCM/ 4HMM, 4HMM, 4HMM /
DATA KKCM2/ 4HMM2, 4HMM2 /
DATA MH2H/ 4HMM/M, 4HMM/C /
DATA MCALC/ 4H(CAL, 4HC.) /
DATA NPARTX, NPARTY/ 10, 8 /
DATA XL, YL/ 200.0, 150.0 /
DATA XO, YO, DH/ 40.0, 30.0, 20.0 /
DATA HH, H2, H3/ 2.5, 2.0, 1.5 /
C
C... DATA RANGE
CALL GOSCAL ( LENGTH, XVAL, YVAL, XNN, YNN, XMX, YMX, FACT )
CALL XYSICAL ( XNN, XMX, NPARTX, DX, NEXPX )
CALL XYSICAL ( YNN, YMX, NPARTY, DY, NEXPY )
C

```

Appendix F (Continued)

Appendix F (Continued)

```

C.... GRID MESH
C
CALL PLPEN( 1 )
X(1) = X0
X(2) = X0+XL
X(3) = X0+XL
X(4) = X0
X(5) = X0
Y(1) = Y0
Y(2) = Y0
Y(3) = Y0+YL
Y(4) = Y0+YL
Y(5) = Y0
CALL PLINES ( X, Y, 5 )
C. (X) SCALE
Y(1) = Y0-2.0
Y(2) = Y0+2.0
Y(3) = Y0-5.0
Y(4) = Y0+YL-2.0
Y(5) = Y0+YL
X(1) = X0
X(2) = X0
DNO = 0.0
LPARTX = NPARTX+1
DO 300 J= 1, LPARTX
CALL PLINES ( X, Y, 2 )
CALL PLINES ( X, Y(4), 2 )
X(3) = X(1)-H2
CALL PNUMB ( X(3), Y(3), H2, DNO, 0.000, 1 )
X(1) = X(1)+DH
X(2) = X(2)+DH
DNO = DNO+DX
300 CONTINUE
IF( NEXPX.EQ.0 ) GO TO 350
X(3) = X0+XL+4.0*H2
CALL PSYM ( X(3), Y(3), H2, KSCL, 0.000, 6 )
X(3) = X0+XL+7.0*H2
Y(3) = Y(3)+H3
CALL PNUMB ( X(3), Y(3), H3, NEXPX, 0.000 )
C. (Y) SCALE
350 X(1) = X0-2.0
X(2) = X0+2.0
X(3) = X0-5.0*H2
X(4) = X0+XL-2.0
X(5) = X0+XL
Y(1) = Y0
Y(2) = Y0
DNO = 0.0
LPARTY = NPARTY+1
DO 400 J= 1, LPARTY
CALL PLINES ( X, Y, 2 )
CALL PLINES ( X(4), Y, 2 )
Y(3) = Y(1)
CALL PNUMB ( X(3), Y(3), H2, DNO, 0.000, 1 )
Y(1) = Y(1)+DH
Y(2) = Y(2)+DH
DNO = DNO+DY
400 CONTINUE

```

```

IF( NEXPY.EQ.0 ) GO TO 500
X(3) = X0
Y(3) = Y0+YL+H2
CALL PSYM ( X(3), Y(3), H2, KSCL, 0.000, 6 )
X(3) = X0+3.0*H2
Y(3) = Y(3)+H3
CALL PNUMB ( X(3), Y(3), H3, NEXPY, 0.000 )
C.... HEADER PLOT
C
500 CONTINUE
XX = X0+20.0*HH
YY = Y0+YL+3.0*HH
CALL PSYM ( XX, YY, HH, MTITLE, 0.000, 70 )
C. (X-TITLE)
IPX = NP(1,KTYPE)
XX = X0+20.0*HH
YY = 0.6*Y0
CALL PSYM ( XX, YY, HH, KHEAD(1,IPX), 0.000, 12 )
C
XX = XX+15.0*HH
IF( IPX.EQ.1 ) MHEAD(1,IPX) = KKMCM(IFLAG+1)
IF( IPX.EQ.4 ) MHEAD(1,IPX) = MH2H(IFLAG+1)
CALL PSYM ( XX, YY, HH, MHEAD(1,IPX), 0.000, 8 )
C. (Y-TITLE)
IPY = NP(2,KTYPE)
XX = 0.4*X0
YY = Y0+20.0*HH
CALL PSYM ( XX, YY, HH, KHEAD(1,IPY), 90.000, 12 )
C
YY = YY+15.0*HH
IF( KTYPE.EQ.1 ) GO TO 510
IF( KTYPE.EQ.2 ) MHEAD(2,IPY) = KKMCM(IFLAG+1)
IF( IPY.EQ.5 ) MHEAD(2,IPY) = KKMCM2(IFLAG+1)
510 CONTINUE
CALL PSYM ( XX, YY, HH, MHEAD(1,IPY), 90.000, 8 )
C. (ANALYSIS LABEL)
XX = X0+160.0
YY = 0.6*Y0
CALL PSYM ( XX, YY, HH, LABEL, 0.000, 12 )
IF( KTYPE.LE.2 ) GO TO 600
IF( KTYPE.GE.5 ) GO TO 600
C. (MATERIAL ID)
XX = XX+13.0*HH
CALL PSYM ( XX, YY, HH, MLAB(ID), 0.000, 4 )
IF( KTYPE.EQ.3 ) GO TO 600
XX = XX+5.0*HH
CALL PSYM ( XX, YY, HH, MCALC, 0.000, 7 )
C
C.... XY PLOT
C
600 CONTINUE
CALL PLPEN( 2 )
PX = DX* 10.0**NEXPX
PY = DY* 10.0**NEXPY
XFACT = DH/PX
YFACT = DH/PY

```

Appendix F (Continued)

Appendix F (Continued)

```

C
C(1) = X0
Y(1) = Y0
R = 1.0
DO 700 J= 1, LENGTH
X(2) = XFACT*XVAL(J) + X0
Y(2) = YFACT*YVAL(J) + Y0
CALL PLINES ( X, Y, 2 )
CALL PLMARK ( X(2), Y(2), R )
X(1) = X(2)
Y(1) = Y(2)
700 CONTINUE
C
CALL PLTEOR
RETURN
END

SUBROUTINE UPLOT ( LABEL, DEPTH, ACCEL )
C
C**** /PURPOSE/
GEOMETRY PLOT AND MAXIMUM DISPLACEMENT VALUE PLOT
C
C*** /INPUT/
LABEL = ANALYSIS TYPE LABEL
DEPTH = MAXIMUM DISPLACEMENT VALUE
ACCEL = MAXIMUM ACCELERATION VALUE
C
C*** PROGRAMMED BY JAERI 1988.09.20
C
IMPLICIT REAL*8 (A-H,O-Z)
COMMON /MODELS/
+ RCOORD(5), ZCOORD(9), WEIGHT, HEIGHT, ANGLE,
+ DISP, FACTK(5), ENERGO, GRAVIT, EPMAX,
+ MTINN, MTOU, MODEL, IFLAG, KSOLV,
+ KOUT, NPART, KPART, NTITLE(18)
COMMON /MODEL1/
+ S, C, S2, C2
C
DIMENSION LABEL(3)
DIMENSION X(40), Y(40), XV(6), YV(6), LDIR(2),
+ NP(2,17), KKMCM(2),
+ KDEPTH(2), KACCEL(2)
DATA NP/ 1, 0, 2, 0, 3, 0, 4, 0, 5, 0, 5, 1,
+ 5, 2, 5, 3, 5, 4, 4, 4, 3, 4, 3, 3,
+ 3, 2, 2, 2, 1, 2, 1, 1, 1, 0/
DATA LDIR/ 4HR, 4HZ, 4H, 4H, /
DATA KDEPTH/ 4HDEPT, 4HH= /
DATA KACCEL/ 4HACCE, 4HL= /
DATA KKMCM/ 4H(KM), 4H(CM) /
DATA KKACCG/ 4H(G) /
DATA XL,YL/ 3.8637, 1.0353 /
DATA XO,YO/ 40.0, 20.0 /
DATA HH/ 2.5 /
DATA H2/ 2.0 /

```

```

C
C... FUNCTION
SORT(Z) = DSORT(Z)
C
C... CLEAR (X,Y)
C
DO 100 J= 1, 35
X(J) = 0.0
Y(J) = 0.0
100 CONTINUE
C
C... GET GEOMETRY
C
DO 200 J= 1, NODE
IX = NP(1,J)
IY = NP(2,J)
X(J) = RCOORD(IX)
Y(J) = ZCOORD(IY)
200 CONTINUE
MN = NODE
ANGXX = 0.0
C
IF( MODEL.NE.2 ) GO TO 400
IF( ZCOORD(5).LT.ZCOORD(4) ) GO TO 310
MN = 2*NODE
DO 300 J= 1, NODE
IX = NP(1,J)
IY = NP(2,J)
X(J+NODE) = RCOORD(IX)
Y(J+NODE) = ZCOORD(9-IY)
300 CONTINUE
310 ZWIDE = 3.0*ZCOORD(4)
DO 320 J= 1, NODE
X(J+NODE) = X(J)
Y(J+NODE) = ZWIDE-Y(J)
320 CONTINUE
C
330 MM = MN+4
X(MN+1) = 0.0
X(MN+2) = 0.5*RCOOR(1)
X(MN+3) = 0.0
X(MN+4) = 0.0
Y(MN+1) = 0.0
Y(MN+2) = 0.0
Y(MN+3) = 0.0
Y(MN+4) = 0.5*RCOOR(1)
DO 350 J= 1, MM
R = X(J)
Z = Y(J)
X(J) = Z
Y(J) = -R
350 CONTINUE
GO TO 500
C
C... CORNER MODEL
C
400 MM = MN+4

```

Appendix F (Continued)

```

X(NN+1) = 0.0
X(NN+2) = 0.5*RCOOR(1)
X(NN+3) = 0.0
X(NN+4) = 0.0
Y(NN+1) = 0.0
Y(NN+2) = 0.0
Y(NN+3) = 0.0
Y(NN+4) = 0.5*RCOOR(1)
IF( MODEL.NE.3 ) GO TO 500
ANGXX = -ANGLE
DO 450 J = 1, MM
R = X(J)
Z = Y(J)
X(J) = R*C + Z*S
450 Y(J) = -R*S + Z*C
C
C.... GEOMETRY SCALING
C
500 CONTINUE
CALL GOSCAL ( MM, X, Y, XMN, YMN, XMN, YMN, XMX, YMY, FACTOR )
C
C.... PLOT COORDINATES
C
DO 600 J = 1, MM
X(J) = FACTOR*(X(J)-XMN) + XO
Y(J) = FACTOR*(Y(J)-YMN) + YO
600 CONTINUE
C
C.... GEOMETRY PLOT
C
CALL PLPEN (1)
CALL PLINES ( X, Y, NODE )
RR = 1.0
DO 700 J = 1, NODE
CALL PLMARK ( X(J), Y(J), RR )
700 CONTINUE
C
IF( NN.EQ.NODE ) GO TO 1000
MM = NODE+1
CALL PLINES ( X(MM), Y(MM), NODE )
DO 800 J = MM, NN
CALL PLMARK ( X(J), Y(J), RR )
800 CONTINUE
C
C.... AXES PLOT
C
1000 CONTINUE
DO 1200 J = 1, 2
XV(1) = X(NN+1)
YV(1) = Y(NN+1)
XV(2) = X(NN+2)
YV(2) = Y(NN+2)
XV(3) = X(NN+2)
YV(3) = Y(NN+2)
YV(5) = Y(NN+2)
NN = NN+2
C
X(1) = XV(2)-XV(1)
Y(1) = YV(2)-YV(1)

```

Appendix F (Continued)

```

DLL = SQRT( X(1)**2+Y(1)**2 )
IF( DLL .LE. 1.0D-2 ) DLL = 1.0D-2
F1 = (DLL-XL)/DLL
F2 = (DLL+XL)/DLL
X(2) = XV(1) + F1*X(1)
Y(2) = YV(1) + F1*Y(1)
XV(6) = XV(1) + F2*X(1)
YV(6) = YV(1) + F2*Y(1)
X(1) = X(1)/DLL
Y(1) = Y(1)/DLL
C
XV(3) = X(2) - Y(1)*YL
XV(4) = X(2) + Y(1)*YL
YV(3) = Y(2) + X(1)*YL
YV(4) = Y(2) - X(1)*YL
CALL PLINES ( XV, YV, 5 )
CALL PSYM ( XV(6), YV(6), HH, LDIR(J), ANGXX, 1 )
1200 CONTINUE
C
C.... DISPLACEMENT VALUE PLOT
C
DLL = 0.1*DEPTH*FACTOR
X(1) = 230.0
X(2) = X(1)+6.0
X(3) = X(1)
X(4) = X(2)
X(6) = X(1)+3.0
X(7) = X(6)
X(5) = X(6)-0.25*YL*DLL
X(8) = X(6)+0.25*YL*DLL
Y(1) = YO
Y(2) = Y(1)
Y(3) = Y(1)+DEPTH*FACTOR
Y(4) = Y(3)
Y(6) = Y(1)
Y(7) = Y(3)
Y(5) = Y(6)+0.25*XL*DLL
Y(8) = Y(7)-0.25*XL*DLL
C
CALL PLPEN (2)
CALL PLINES ( X(1), Y(1), 2 )
CALL PLINES ( X(3), Y(3), 2 )
CALL PLINES ( X(5), Y(5), 4 )
XX = X(1)-10.0
YY = YO-1.5*H2
CALL PSYM ( XX, YY, H2, KDEPTH, 0.000, 7 )
XX = XX+7.0*H2
CALL PNUMB ( XX, YY, H2, DEPTH, 0.000, 2 )
FXU = 6.0
IF( DEPTH.GE.100.0 ) FXU = 7.0
IF( DEPTH.GE.1000.0 ) FXU = 8.0
XX = XX+FXU*H2
CALL PSYM ( XX, YY, H2, KCMCM(IFLAG+1), 0.000, 4 )
C
XX = 120.0
YY = YO+H2+FACTOR*(YMX-YMN)
CALL PSYM ( XX, YY, H2, KACCEL, 0.000, 7 )
XX = XX+7.0*H2

```


Appendix F (Continued)

```

SUBROUTINE PLTBGN
IMPLICIT REAL*8 (A-H,O-Z)
CALL PLOTS (0.000,0.000,10)
RETURN
END
    
```

```

SUBROUTINE PLPEN (NO)
CALL NEWPEN (NO)
RETURN
END
    
```

```

SUBROUTINE PLINES ( X,Y,NN )
IMPLICIT REAL*8 (A-H,O-Z)
DIMENSION X(1), Y(1)
CALL PLOT (X(1),Y(1),3)
IF( NN.EQ.1 ) RETURN
DO 10 I=2,NN
CALL PLOT (X(I),Y(I),2)
10 CONTINUE
RETURN
END
    
```

```

SUBROUTINE PLMARK ( X0,Y0,R )
IMPLICIT REAL*8 (A-H,O-Z)
DIMENSION CR(2,9), X(9), Y(9)
DATA CR/ 0.92388, 0.38268, 0.38268, 0.92388,
+ -0.38268, 0.92388, -0.38268, -0.92388,
+ 0.38268, -0.92388, 0.92388, -0.38268,
+ 0.92388, 0.38268/
DO 10 J=1,9
X(J) =R*CR(1,J) + X0
10 Y(J) =R*CR(2,J) + Y0
CALL PLINES ( X, Y, 9 )
RETURN
END
    
```

C

C

```

SUBROUTINE PSYM ( X,Y,H,TIT,ANG,NL )
IMPLICIT REAL*8 (A-H,O-Z)
DIMENSION TIT(1)
CALL SYMBOL (X,Y,H,TIT,ANG,NL)
RETURN
END
    
```

```

SUBROUTINE PNUMB ( X,Y,H,AN,ANG,N )
IMPLICIT REAL*8 (A-H,O-Z)
CALL NUMBER (X,Y,H,AN,ANG,N)
RETURN
END
    
```

Appendix F (Continued)

```

SUBROUTINE PNUMB ( X,Y,H,AN,ANG )
IMPLICIT REAL*8 (A-H,O-Z)
AN =N
CALL NUMBER (X,Y,H,AN,ANG,-1)
RETURN
END
    
```

```

SUBROUTINE PLTEOR
IMPLICIT REAL*8 (A-H,O-Z)
CALL PLOT (300.000,0.000,-3)
CALL PLOT ( 0.000, 0.000, 666 )
CALL PLOT ( 0.000, 0.000, 777 )
CALL PLOT ( 0.000, 0.000, 888 )
RETURN
END
    
```

C

```

SUBROUTINE PLTEND
IMPLICIT REAL*8 (A-H,O-Z)
CALL PLOT (0.000,0.000,999)
RETURN
END
    
```