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**HIGH RF POWER TEST OF A LOWER HYBRID MODULE  
MOCK-UP IN CARBON FIBER COMPOSITE**

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Sunao MAEBARA, Kimihiro KIYONO, Masami SEKI,  
Kazuaki SUGANUMA, Yoshitaka IKEDA, Tsuyoshi IMAI,  
Marc GONICHE\*, Jean BROSSAUD\*, Vincent CANO\*,  
Fabienne KAZARIAN-VIBERT\*, Philippe BIBET\* and Guy REY\*

**日本原子力研究所  
Japan Atomic Energy Research Institute**

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High RF Power Test of a Lower Hybrid Module  
Mock-up in Carbon Fiber Composite

Sunao MAEBARA, Kimihiro KIYONO<sup>+1</sup>, Masami SEKI<sup>+2</sup>, Kazuaki SUGANUMA<sup>+3</sup>  
Yoshitaka IKEDA<sup>+2</sup>, Tsuyoshi IMAI, Marc GONICHE\*, Jean BROSSAUD\*,  
Vincent CANO\*, Fabienne KAZARIAN-VIBERT\*, Philippe BIBET\* and Guy REY\*

Department of Fusion Engineering Research  
Naka Fusion Research Establishment  
Japan Atomic Energy Research Institute  
Naka-machi, Naka-gun, Ibaraki-ken

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A mock-up module of a Lower Hybrid Current Drive antenna module of a Carbon Fiber Composite (CFC) was fabricated for the development of heat resistive front facing the plasma. This module is made from CFC plates and rods which are copper coated to reduce the RF losses. The withstand-voltage, the RF properties and outgassing rates for long pulses and high RF power were tested at the Lower Hybrid test bed facility of Cadarache.

After the short pulse conditioning, long pulses with a power density ranging between 50 and 150 MW/m<sup>2</sup> were performed with no breakdowns. During these tests, the module temperature was increasing from 100-200 °C to 400-500 °C. It was also checked that high power density, up to 150 MW/m<sup>2</sup>, could be transmitted when the waveguides are filled with H<sub>2</sub> at a pressure of 5x10<sup>-2</sup> Pa. No significant change in the reflection coefficient is measured after the long pulse operation. During a long pulse, the power reflection increases during the pulse typically from 0.8 % to 1.3 %. It is concluded that the outgassing rate of Cu-plated CFC is about 6~7 times larger than of Dispersion Strengthened Copper (DSC) module at the module temperature of 300 °C. No significant increase of the global outgassing of the CFC module was measured after hydrogen prefilling. After the test, visual inspection

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+1 Office of ITER Project Promotion

+2 Department of Fusion Facilities, Naka Fusion Research Establishment

+3 Advanced photon Research Center, Kansai Research Establishment

\* Centre d'études de Cadarache Association Euratom-CEA

revealed that peeling of the copper coating occurred at one end of the module only on a very small area ( $0.2 \text{ cm}^2$ ). It is assessed that a CFC module is an attractive candidate for the hardening of the tip of the LHCD antenna.

**Keywords:** Carbon Fiber Composite (CFC), Plasma Facing Component  
Al<sub>2</sub>O<sub>3</sub>-Dispersion Strengthened Copper (DSC), Outgassing

炭素系繊維材で試作した低域混成波アンテナモジュールの大電力高周波試験

日本原子力研究所那珂研究所核融合工学部

前原 直・清野 公広<sup>+1</sup>・関 正美<sup>+2</sup>・菅沼 和明<sup>+3</sup>

池田 佳隆<sup>+2</sup>・今井 剛・Mark GONICHE\*・Jean BROSSAUD\*

Vincent CANO\*・Fabienne KAZARIAN-VIBERT\*・Philippe BIBET\*

Guy REY\*

(1997年10月24日受理)

プラズマ対向するアンテナ先端部モジュール開発のために、炭素系繊維材を用いたLHCDアンテナモジュールを試作した。このモジュールは、高周波損失を抑えるために、CFC材の表面に銅をメッキした仕切板及びロッドから作られている。試作モジュールの長パルス及び大電力の高周波伝送時における、耐電力特性、高周波特性及びガス放出特性について、カグラッシュ研究所のLH出力試験装置で試験した。

短パルスでのRFコンディショニングの後、高周波電力密度50–150MW/m<sup>2</sup>を放電なく伝送することができた。この時のモジュール温度は、100–200℃から400–500℃まで上昇した。またモジュール内に水素を $5 \times 10^{-2}$ Paまで充満した時にも、高周波電力密度150MW/m<sup>2</sup>まで伝送できることを確認した。長パルス運転後、モジュールの問題となる高周波特性（反射率）劣化は、測定されなかった。長パルス運転時の電力反射率は、0.8から1.3%に増加する程度である。CFCモジュールからのガス放出率は、モジュール温度300℃において、アルミナ分散強化銅（DSC）モジュールからのガス放出率の約6～7倍程度であった。高周波伝送前に水素を充満しても、重大な全体的なガス放出の増加は、観測されなかった。これらの実験の後、モジュール表面にメッキした銅の剝離は、0.2cm<sup>2</sup>の小規模であった。これらの結果から、CFCモジュールは、アンテナ先端部の材料として魅力的な候補であるとの結論に達した。

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那珂研究所：〒311-01 茨城県那珂郡那珂町向山801-1

+1 ITER業務推進室

+2 核融合装置試験部

+3 関西研究所光量子科学センター

\* フランス原子力庁、カグラッシュ研究所

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## 1. Introduction

In the frame of the collaboration between JAERI and CEA, two modules for a Lower Hybrid Antenna Modules have been previously tested. Test Module I was a 4-waveguide module made of Dispersion Strengthened Copper (DSC). A large data base of outgassing rate during high RF power injection was established [1]. Test Module II was composed of a Power Poloidal Divider (PPD) and a 3 x 6 waveguide Multijunction. RF properties at low and high power were studied in combination with an other poloidal divider (Mode converter or E/H plane multijunction). Outgassing properties of such a complex system was also documented [2].

For the next step machine, the mouth of the antenna facing the plasma is an important issue. In present day experiments the antenna waveguides are metallic. The metal is copper-coated stainless steel on JET and JT-60U, an alloy of copper and zirconium on Tore Supra, DSC on TdeV. Sputtering and erosion may lead to an undesirable contamination of the plasma with metallic impurities. The mouth of the Lower Hybrid antenna has to be hardened with a low Z material to reduce such impurities.

A possible solution is to fabricate the entire multijunction from a Carbon Fiber Composite (CFC). The inside surfaces of this multijunction has to be Cu-plated in order to reduce the RF losses. Because of the difference of thermal expansion coefficients between copper and CFC, the Cu-plating on CFC material is a key of fabrication technique. In order to establish this fabrication technique, heat resistivity test is first needed to check Cu adherence and bonding defect. Second, an RF test using long pulses at high power is needed for the RF properties, the withstand voltage and the outgassing rate.

For the first point, a 8GHz mock-up module using CFC was fabricated by JAERI, the heat resistivity test was done on JAERI Electron Beam Irradiation Stand(JEBIS)[3]. For irradiation heat fluxes up to 3.2 MW/m<sup>2</sup> for 2 min, no peeling of the Cu-plating and no bonding defects were observed. This heat loads of 3.2 MW/m<sup>2</sup> is about 13 times higher than the steady-state average heat load on the first wall of ITER.

The second point is the main part of this report. Test nodule III was a CFC module fabricated with the same techniques as used for the 8GHz mock-up module. In order to compare the RF properties, withstand-voltage and outgassing rates, a reference module in DSC was also fabricated by JAERI. These two components were tested at the Lower Hybrid test bed facility of Cadarache in the frame of the collaboration. The CFC and DSC modules are described in section 2. The experimental set-up for high power test is

presented in section 3. The low power measurements are given in section 4 whereas RF properties at high power are presented in section 5. The different outgassing measurements are discussed in section 6. Finally conclusions are drawn.

## 2. The Module III

Table 1 shows the main physical properties of CFC and Al<sub>2</sub>O<sub>3</sub>-Dispersion Strengthened Copper (DSC) material. The melting point of CFC material is about 3 times higher than that of DSC. However, electric resistivity of CFC material is about 70 times higher than DSC. The high resistivity is favorable from the point of the electro-magnetic force in case of the plasma disruption but it makes large ohmic loss as the wave guide. The Cu-plating is therefore indispensable on the inside surfaces of the CFC waveguide. Because of the difference of thermal expansion coefficients between copper and CFC, by almost a factor of 3, a compliant layer is needed and the Cu-plating is advantageously used as the interface layer for assembling the septum plates to rods and the cooling channel to the CFC module.

Item	Unit	CFC	Cu-Al <sub>2</sub> O <sub>3</sub>
Melting point	°C	3923	1356
Density	g cm <sup>-3</sup>	1.88	8.86
Resistivity ( at 400 °C )	x 10 <sup>-8</sup> Ω m	350	4.9
Specific heat ( at 400 °C )	J kg <sup>-1</sup> K <sup>-1</sup>	1450	413
Coefficient of thermal extension ( at 400 °C )	x 10 <sup>-6</sup> K <sup>-1</sup>	~ 6.3	17.1
Thermal conductivity ( at 400 °C )	W m <sup>-1</sup> K <sup>-1</sup>	320	300

Table 1 Main physical properties

The CFC is a two-dimensional structure (Toshiba Corporation). This CFC is knitted with an intersecting angle of 15° to increase bending strength of the septum plate. The carbon fiber diameter is 10 μm and the bundle number is 3000. The fabrication flow chart for the mock-up module using CFC material is shown in Fig.1. First, four septum plates and six rods were accurately cut from CFC material and the cleaning of pieces was done by ultrasonic waves. Second, each pieces were coated with titanium by physical vapor deposition. The thickness of Ti layer is 5 ~ 8 μm. Third, copper was electrochemically plated on top of the Ti layer, a copper plating of 20~30 μm was deposited by electrochemical deposition. After outgassing treatment of these pieces, septum plates and rods were assembled by diffusion bonding



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method at a temperature of 920 °C and an isostatic pressure of ~ 20 MPa. Finally, the copper cooling channels were silver brazed on the module at 820 °C. In these process, the CFC outgassing treatment is a crucial step. An incomplete treatment would lead to a loss of adherence of the copper when diffusion bonding or silver brazing is carried out.

Two test modules were fabricated by JAERI. These modules are only different by the material used for fabrication. The first one is made from DSC plates and rods and is used as a reference and the second is made from CFC copper-plated plates and rods. The DSC module is assembled by silver brazing. The CFC module is assembled and copper plated with the same techniques than for the 8GHz module mock-up.

The photo of 3.7 GHz module is shown in Fig.2. The length is 206 mm and the inner dimension of the secondary waveguides is 19 x 72 mm. At each end, a choke flange is used as shown in Fig.3. The choke length is 49.95 mm ( $\sim \lambda_g/2$ ), the nominal gap length is 3 mm.

### 3. The experimental set-up

The experimental set-up is sketched in Fig. 4. The total under vacuum length is 1.2 m. Each window is connected to a bellow for the vacuum tightness. Inside the bellow, a standard waveguide is in contact with the window. Between the waveguide and the test module, a taper waveguide (from 36 x 76 to 34 x 72 mm) is connected. This taper waveguide ( $L=200$  mm), made from brazed copper-coated plates, is vacuum tight. Because of the low conductance between the inside and the outside of the module, two pumping pipes were connected. Pumping of the inside was provided by a low conductance  $\phi 12$  mm port and pumping of the outside was provided by a  $\phi 114$  mm port. Taking into account the connection pipes between the ports and the turbo-molecular pump, the total effective pumping speed is estimated to be only 0.01 m<sup>3</sup>/sec. A ion gauge allows to measure the pressure outside of the module. Three thermocouples were connected to the module. These under vacuum thermocouples allow to control the temperature of the module at different locations of the same transverse section. One thermocouple was installed on each taper waveguide.

The DSC reference module was tested first. The module was baked at 300 °C for 50 h. After cooling, an air leak was detected and the system had to be vented in order to repair the leak. Then the module was baked at 300 °C for 16 h. For the CFC module, only a 300 °C/16h baking was achieved.

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#### 4. Low power measurements

Low power measurements were performed before and after the high power test. The  $S_{21}$  and  $S_{11}$  parameters were measured from the input RF window to the output window in the 3.6-3.8 GHz band. In the entire frequency range,  $S_{21}$  is below 0.1 dB for both modules, indicating a transmission factor higher than 0.98. The calculation of the RF losses of the 1.2 m long waveguides leads to a transmission factor of 0.994 at room temperature.  $S_{11}$  is larger than 20 dB in the frequency range. The different values at 3.70 and 3.73 GHz are listed in Table 2.

$S_{11}$ (dB)	DSC module		CFC module	
	3.70 GHz	3.73 GHz	3.70 GHz	3.73 GHz
Before experiment	-26.8	-28.4	-25.5	-21.7
After experiment	-28.8	-23.0	-24.9	-21.6

Table 2.  $S_{11}$  parameter of the 2 modules

For the CFC module, no significant change of  $S_{11}$  is measured.

#### 5. RF properties at high power

##### 5.1 RF conditioning

The same RF conditioning technique for the module I and II was used. First, the RF power was gradually increased using 10 msec every 100 ms. For the DSC module, the first breakdown occurred at 110 kW (48 MW/m<sup>2</sup>), and the 150 MW/m<sup>2</sup> level was obtained after 30,000 pulses. For the CFC module, the first breakdown was detected almost at the same power (100 kW) but the conditioning was faster. Only 10,000 pulses were needed to reach the 150 MW/m<sup>2</sup> level (Fig.5). This may indicate that conditioning of the connections waveguides or / and RF windows which were baked at a lower temperature than the module was the most demanding in terms of number of pulses.

After this short pulse conditioning, two second pulses at the 100 MW/m<sup>2</sup> level were easily achieved with no breakdowns and long pulse operation was started. After long pulse operation at the 100 MW/m<sup>2</sup> level, conditioning at higher power was achieved and a record power of 390 kW (170 MW/m<sup>2</sup>) was obtained for 6 sec (Fig.6). For each module, 12 long pulses with a power density ranging between 50 and 140 MW/m<sup>2</sup> were performed. During these long pulses, the module temperature was increasing between 100-200 °C and

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400-500 °C (Fig.7). A typical shot at 90 MW/m<sup>2</sup> is shown in Fig.8. All these tests were performed with an initial pressure of  $\sim 10^{-4}$  Pa. It was checked that high power density, at least up to 150 MW/m<sup>2</sup>, could be transmitted when the waveguides are filled with H<sub>2</sub> at a pressure of  $5 \times 10^{-2}$  Pa.

## 5.2 Power reflection coefficients

The power reflection coefficient (RC) was calculated from the RF measurements available at the output of the klystron,  $\sim 0.5$  m upstream the input RF window. For an input power of 120 kW, this reflection coefficient is quoted at the beginning of the high power test and after all the long pulse operation for the 2 modules. The results are given in Table 3.

RC. (%)	DSC module	CFC module
before experiment	0.55 % (T=50°C)	0.90 % (T=45°C)
after experiment	0.90% (T=175°C)	0.82 % (T=55°C)

Table 3 Reflection coefficients of the 2 modules

Low reflection is measured for the 2 modules ( $<1$  %). No significant modification of the RC is measured after the long pulse operation. During a long pulse, the reflection increases during the pulse typically from 0.8 % to 1.3 % (Fig.9). The same behavior was observed in previous long pulse RF tests. The main reason is the change of phase of the reflected wave due to thermal expansion which has to be taken into account for a directional coupler with a finite directivity. With the following parameters

$$L = 1.2 \text{ m}$$

$$a = 1.7 \cdot 10^{-5} \text{ K}^{-1}$$

$$\text{RC.} = 0.9 \%$$

$$D = -30 \text{ dB}$$

it is found that the measured reflected power increases by  $\sim 10$  % for  $\Delta T = 100$  °C. Experimentally, it is found  $\sim 20$  % for  $\Delta T = 100$  °C. The change of directivity of the coupler with the temperature may also influence the measurement of the reflected power.

## 5.3 RF losses

RF losses were estimated from the temperature increase rate measured on the thermocouples attached to the module. At  $f = 3.7$  GHz, the RF losses  $P$  (in W/m<sup>2</sup>) of a waveguide of height  $a$  and width  $b$  are given by :

$$P_a = \frac{0.3206 \sqrt{\rho_{el}} \left[ 1 - \left( 1 - \frac{\lambda_0^2}{2a^2} \right) \cos \frac{2\pi x}{a} \right]}{\sqrt{1 - \frac{\lambda_0^2}{4a^2}}} P \quad (1)$$

along the large side  $a$  of the waveguide.

$$P_b = \frac{0.3206 \sqrt{\rho_{el}} \left[ \frac{\lambda_0^2}{2a^2} \right]}{\sqrt{1 - \frac{\lambda_0^2}{4a^2}}} P \quad (2)$$

along the small side  $b$  of the waveguide, where  $\rho_{el}$  is the resistivity (in  $\Omega \cdot m$ ),  $P$  the RF power density (in  $W/m^2$ ) and  $\lambda$  the wavelength in vacuum (0.081m).  $P_a$  is maximum for  $x=a/2$  (mid-height) and is equal to

$$P_a^{\max} = \frac{0.3206 \sqrt{\rho_{el}} \left[ 2 - \frac{\lambda_0^2}{2a^2} \right]}{\sqrt{1 - \frac{\lambda_0^2}{4a^2}}} P \quad (3)$$

the average losses per unit of length  $\langle P \rangle$  (in  $W/m$ ) are given by

$$\langle P \rangle = \frac{0.3206 \sqrt{\rho_{el}} \left[ 2a + 2b \frac{\lambda_0^2}{2a^2} \right]}{\sqrt{1 - \frac{\lambda_0^2}{4a^2}}} P \quad (1 \text{ waveguide}) \quad (4)$$

For the module III, with  $a=0.072$  m and  $b=0.019$  m, the losses as a function of the temperature are given in Table 4 for  $P=100MW/m^2$ .

T (°C)	50	150	250	350
$\rho_{el}$ ( $10^{-8} \Omega \cdot m$ )	2.0	2.8	3.7	4.6
$P_a$ max ( $W/m^2$ )	7495	8870	10195	11365
$P_b$ ( $W/m^2$ )	3480	4120	4735	5280
$\langle P \rangle$ ( $W/m$ ) (for the 2 waveguides)	1845	2185	2510	2800

Table 4. Calculated RF losses for module III

The local heating rate is given by

$$\frac{dT}{dt} = \frac{P_a \text{ or } P_b}{\rho e C_p} \quad (5)$$

where  $e$  (in m) is the thickness of the walls

$\rho$  (in  $\text{kg m}^{-3}$ ) is the volume density of the material

$C_p$  (in  $\text{J kg}^{-1} \text{K}^{-1}$ )

The average heating rate is given by

$$\frac{dT}{dt} = \frac{P_a \text{ or } P_b}{\rho S C_p} \quad (6)$$

where  $S$  is the section surface of the module ( $S=8.80 \times 10^{-4} \text{ m}^2$ ). For the DSC module the heating rates corresponding to  $P_a^{\text{max}}$  ( $T_{\text{mod1}}$ ) or to  $P_b$  ( $T_{\text{mod2}}$  and  $T_{\text{mod3}}$ ) are given in Table 5 for  $P = 100 \text{ MW/m}^2$ .

T (°C)	50	150	250	350
$P_a^{\text{max}} (\text{W/m}^2)$	7495	8870	10195	11365
$\rho e C_p (\text{Jm}^{-2}\text{K}^{-1})$ ( $e=0.2 \text{ cm}$ , $\rho=8.86 \text{ gcm}^{-3}$ $c_p=0.39-0.41 \text{ J g}^{-1}\text{K}^{-1}$ )	6911	7088	7212	7301
$dT_{\text{mod1}} / dt (\text{K/s})$	1.08	1.25	1.41	1.56
$P_b (\text{W/m}^2)$	3480	4120	4735	5280
$\rho e C_p (\text{Jm}^{-2}\text{K}^{-1})$ ( $e=0.7 \text{ cm}$ , $\rho=8.86 \text{ gcm}^{-3}$ $c_p=0.39-0.41 \text{ J g}^{-1}\text{K}^{-1}$ )	24188	24808	25242	25553
$dT_{\text{mod2/3}} / dt (\text{K/s})$	0.14	0.17	0.19	0.21
$\langle P \rangle (\text{W/m})$ (for the 2 waveguides)	1845	2185	2510	2800
$\rho S C_p (\text{Jm}^{-2}\text{K}^{-1})$ ( $S=8.8 \text{ cm}^2$ , $\rho=8.86 \text{ gcm}^{-3}$ $c_p=0.39-0.41 \text{ J g}^{-1}\text{K}^{-1}$ )	3041	3119	3174	3213
$dT / dt (\text{K/s})$ ( $t \gg t$ )	0.61	0.70	0.79	0.87

Table 5. Heating rates for the DSC module at  $P=100 \text{ MW/m}^2$



For the CFC module, the heating rates at 100 MW / m<sup>2</sup> are given in Table 6.

$P_a \text{ max (W/m}^2\text{)}$	7495	8870	10195	11365
$\rho e C_p \text{ (J.m}^{-2}\text{K}^{-1}\text{)}$ ( $e=0.2 \text{ cm}$ , $\rho=1.79 \text{ gcm}^{-3}$ , $C_p=0.76\text{--}1.39 \text{ Jg}^{-1}\text{K}^{-1}$ )	2735	3755	4464	4976
$dT_{\text{mod1}} / dt \text{ (K/s)}$	2.74	2.36	2.28	2.28
$P_b \text{ (W/m}^2\text{)}$	3480	4120	4735	5280
$\rho e C_p \text{ (J.m}^{-2}\text{K}^{-1}\text{)}$ ( $e=0.7 \text{ cm}$ , $\rho=8.86 \text{ gcm}^{-3}$ , $C_p=0.76\text{--}1.39 \text{ Jg}^{-1}\text{K}^{-1}$ )	9572	13142	15624	17416
$dT_{\text{mod2/3}} / dt \text{ (K/s)}$	0.36	0.31	0.30	0.30
$\langle P \rangle \text{ (W/m)}$ (for the 2 waveguides)	1845	2185	2510	2800
$\rho S C_p \text{ (J.m}^{-2}\text{K}^{-1}\text{)}$ ( $S=8.8 \text{ cm}^2$ , $\rho=1.79 \text{ gcm}^{-3}$ , $C_p=0.76\text{--}1.39 \text{ Jg}^{-1}\text{K}^{-1}$ )	1203	1652	1964	2189
$dT / dt \text{ (K/s) (} t \gg \tau \text{)}$	1.53	1.32	1.28	1.28

Table 6 Heating rates for the CFC module at 100 MW/m<sup>2</sup>

In fact with heat diffusion, the thermocouple will indicate a heating rate given by Eq.6 after a time much larger than the time constant  $t = L^2 / D$  where  $L = a/2 = 0.036 \text{ m}$  and  $D (=k/\rho c_p)$  the diffusivity of the material.  $\tau$  for the DSC and the CFC are given in Table 7.

	DSC	CFC
$D \text{ (m}^2\text{/s)}$	$1.03 \times 10^{-4}$ at 20 °C $0.87 \times 10^{-4}$ at 300 °C	$2.23 \times 10^{-4}$ at 20 °C $0.76 \times 10^{-4}$ at 300 °C
$\tau \text{ (s)}$	12.6 s at 20 °C 14.9 s at 300 °C	5.8 s at 20 °C 17.0 s at 300 °C

Table 7. Diffusivity and time constant for DSC and CFC

For the DSC module, as expected, the thermocouple ( $T_{\text{mod1}}$ ) located at mid-height of the module has a much higher rate than the ones attached to the small side of the waveguide ( $T_{\text{mod2}}$  and  $T_{\text{mod3}}$ ). For different power densities, the experimental and the calculated rates are compared for  $T_{\text{mod1}}$  and  $T_{\text{mod2}}$ . For  $T_{\text{mod1}}$ , the experimental value, measured on  $\sim 10 \text{ s}$  is lower than the one given by the calculation, whereas the experimental value is higher than  $T_{\text{mod2}}$  (Fig.9). This can be consistent, if we consider that diffusion, from  $x = a/2$  to  $x = a$ , is not negligible on a time scale  $t=10 \text{ s}$  ( $t \sim \tau$ ).

After ~100 s, the heating rate is the same for the 3 thermocouples. The experimental and calculated heating rates are compared for three RF power densities. Experimental values are systematically lower than the calculated ones. This is not surprising if we consider diffusion, now along the RF propagation is effective on a time scale of 100 s (Fig.10).

For the CFC module, it was found anomalous time evaluations of temperature given by the 3 thermocouples attached to the module. Moreover, one of the TC blew off during the second long pulse. This anomalous behavior is due to direct heating of the TC by RF leaking from the input choke flange. This leak is likely to occur from time to time and an attempt to compare the experimental and calculated heating rates were nevertheless performed.

The initial heating rates, for 2 power densities, are indicated on Fig.11. The heating rate at 100 MW / m<sup>2</sup>, which is lower than predicted, is consistent with measurements made on the DSC module. After heat diffusion, at  $t \sim 100$  s ( $t \gg \tau$ ), the experimental and calculated heating rates are compared on Fig.12. For this case, when comparing to the results of the DSC module, lower heating rates should be measured.

## 6. Outgassing measurements

### 6.1 Outgassing measurement of CFC samples

Outgassing measurements of samples in CFC, CFC with Cu-plating and DSC were carried out at JAERI. The dimension of the test pieces is 100 x 20 x 5 mm<sup>3</sup>. Two pieces were installed to increase surface area. The total outgassing surface area of the test pieces is 0.01 m<sup>2</sup>. The test pieces and the experimental set-up are shown in Fig.13. Pressure is measured on both sides of a low conductance ( $C=1$  l/s) orifice. The outgassing flux  $Q$  is obtained from the pressure difference ( $Q=C \times \Delta P$ ) when the left-hand side gate valve is closed (Conductance method). Figure 14 indicates the outgassing flux at 300 °C as a function of the baking treatment time. It is deduced that the outgassing rate of Cu-plated CFC and CFC are almost the same after ~ 80 hours of baking ( $\sim 10^{-5}$  Pam<sup>-3</sup> s<sup>-1</sup>m<sup>-2</sup>). It is also inferred that the outgassing of CFC-based material is about one order of magnitude higher than that of DSC, after ~ 60 hours of 300 °C baking.

### 6.2 Outgassing measurements of module III

For the 15 long pulses performed on each module at high RF power, the outgassing flux was measured regularly during RF injection by the build-up

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### 6.2 Outgassing measurements of module III

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method. For these pulses, the outgassing flux is plotted versus the module temperature in Fig. 15.

For the two modules, the initial outgassing before RF injection is, average on the different shots, the same :  $2 \times 10^{-7} \text{ Pam}^{-3}\text{s}^{-1}$  at  $100^\circ\text{C}$  and  $6\text{--}7 \times 10^{-7} \text{ Pam}^{-3}\text{s}^{-1}$  at  $200^\circ\text{C}$ . During RF operation, the average outgassing flux is significantly higher for the CFC module by a factor  $\sim 2$  at  $200^\circ\text{C}$  and a factor  $\sim 4$  at  $350^\circ\text{C}$ . The outgassing rates were evaluated in both cases by taking into account the actual outgassing surface. In the case of DSC module, the 2 tapered waveguides and the 2 connection waveguides are almost at the same temperature as the module as shown in Fig.16. The total surface of these 5 components were considered ( $S=0.27 \text{ m}^2$ ). In the case of CFC module, because of the lower  $\rho C_p$  value for CFC, the module heating rate is about 2 times the heating rates of the other waveguides (Fig.17), and only the surface of CFC module is considered ( $S=0.12 \text{ m}^2$ ). With such assumptions, the outgassing rate versus the module temperature is plotted in Fig.18. When comparing the data base from the two modules, it is concluded that the outgassing rate of Cu-plated CFC is 8 times larger than of DSC at the module temperature of  $300\text{--}350^\circ\text{C}$ .

The effect of  $\text{H}_2$  prefilling on the outgassing rate was investigated. After prefilling at  $5 \times 10^{-4} \text{ Pa}$  for  $\sim 60\text{s}$ , the waveguides were pumped out for  $\sim 60 \text{ s}$ . The pressure did not fully recover and the initial outgassing flux after  $\text{H}_2$  prefilling was about 2-3.5 times higher than the outgassing flux before  $\text{H}_2$  prefilling. Due to this higher initial outgassing flux, the outgassing rate of CFC module during RF operation, was about 2-3 times larger than DSC module.

## 7. Module inspection

After experiments, two modules were inspected. Both modules show a good general aspect. No brazing defects were reported. This is consistent with the RF measurements which were made before and after the long pulse operation (see table 3 and 4). However on the CFC module peeling of the copper coating occurred on a very localized zone ( $S=0.8 \times 0.2\text{--}0.3 \text{ cm}^2$ ), located at one end of the module. In the vicinity of this zone, traces of arcing were also observed. Arcing is believed to be due to the smaller gap length of the choke flange. The gap was measured for the two flanges of two modules. For the DSC module this gap varies between 2.5 and 2.9 mm, but on the DSC the variation is larger, between 2.1 and 3.1 mm and the minimum is effectively measured where the arcs occurred.

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Black points were observed in the corners of the waveguides before the RF test. These black points were analyzed by X-ray diffraction technique. Only copper is seen and the change of color is assumed to be due to an oxidation of the surface.

## 8. Conclusions

A 2-waveguide, 206 mm long, module in Carbon Fiber Composite (CFC) was tested first at low RF power, then in vacuum at high RF power (up to 150 MW/m<sup>2</sup>). Good RF properties in the 3.67-3.73 GHz band was measured (VSWR >21 dB) with no resonance. The accuracy of the transmission factor measurement ( $\pm 0.1$  dB) allows only to assess that the RF losses are, at the most, 2 % (expected value 0.6 %).

At high power (50-150 MW/m<sup>2</sup>), RF losses were evaluated from thermocouples measurements during very long pulses (up to 2500 s). When compared to a reference module in a copper alloy (Dispersed Strengthened Copper), the RF losses seem slightly higher for the CFC module (~20%). However, because of an RF leak at one end of the module, the measurements on the CFC module has a poorer accuracy and this results has to be confirmed.

The outgassing rate was measured for the CFC module and the DSC reference module. Neglecting the outgassing of the vacuum vessel for the both cases and the outgassing of the connection waveguides when connected to the CFC module, it is found that, despite the same initial outgassing, the outgassing rate is about 8 times higher at 350 °C for the CFC module than for the DSC module. However, such assumption lead to an over evaluation of the outgassing rate of CFC. Taking into account the outgassing of these waveguides, the outgassing rate of CFC module is evaluated to 6~7 times the outgassing rate of DSC. Despite the same initial outgassing rate, the highest outgassing rate of CFC at high temperature might be due to a different composition of the desorbed gas for the 2 materials. It is deduced that the outgassing rate for CFC module is about  $4 \times 10^{-5}$  Pam<sup>3</sup>s<sup>-1</sup>m<sup>-2</sup> at 300 °C and  $3 \times 10^{-4}$  Pam<sup>3</sup>s<sup>-1</sup>m<sup>-2</sup> at 400 °C. For the DSC module the average value of  $5 \times 10^{-5}$  Pam<sup>3</sup>s<sup>-1</sup>m<sup>-2</sup> at 400 °C is comparable with the early measurements made on module I[1]. The effects of hydrogen prefilling was studied. When H<sub>2</sub> prefilling is performed, the outgassing during RF injection increases accordingly to the increase of the initial outgassing (factor 2-3.5). These results are not fully consistent with previous results[1] and it is concluded that for module IV, first a long baking time is required (60 hours instead of 15 hours at 300 °C), second a mass spectrometer is needed to monitor separately the outgassing of the hydrogen and the other species.

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Visual inspection as RF measurements before and after long pulse operation indicate no degradation of the thin copper plating. However arcing and a very localized copper peeling ( $S \sim 0.2 \text{ cm}^2$ ) occurred at one end. This may be avoided by the appropriate design of the choke gap.

We assessed that CFC module is a promising material for the multijunction module of an LHCD antenna. More reliable data are needed to evaluate accurately the RF losses and the outgassing has to further studied.

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- [2] S. Maebara et al., Development of a new lower hybrid antenna module using a poloidal power divider, JAERI-Research 96-036, Rep. EUR-CEA-FC-1590.
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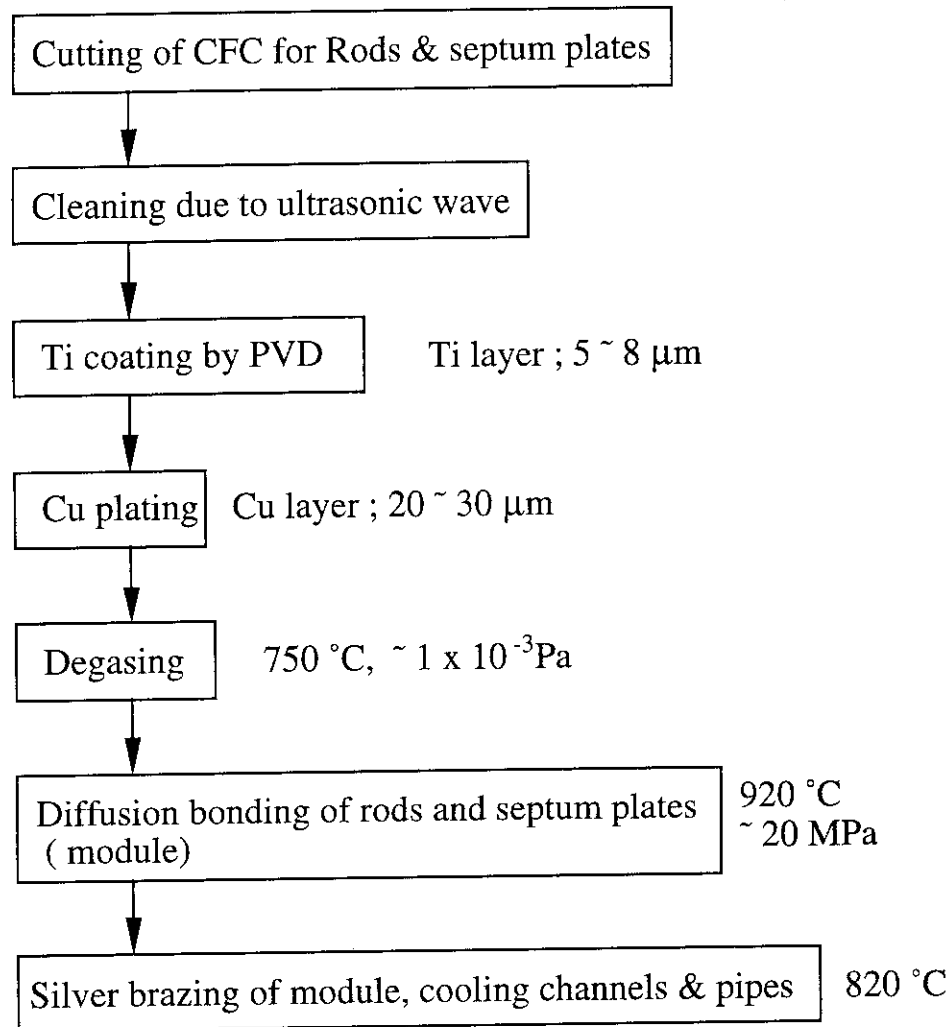


Fig. 1 Fabrication flow chart for the mock-up module using CFC material.

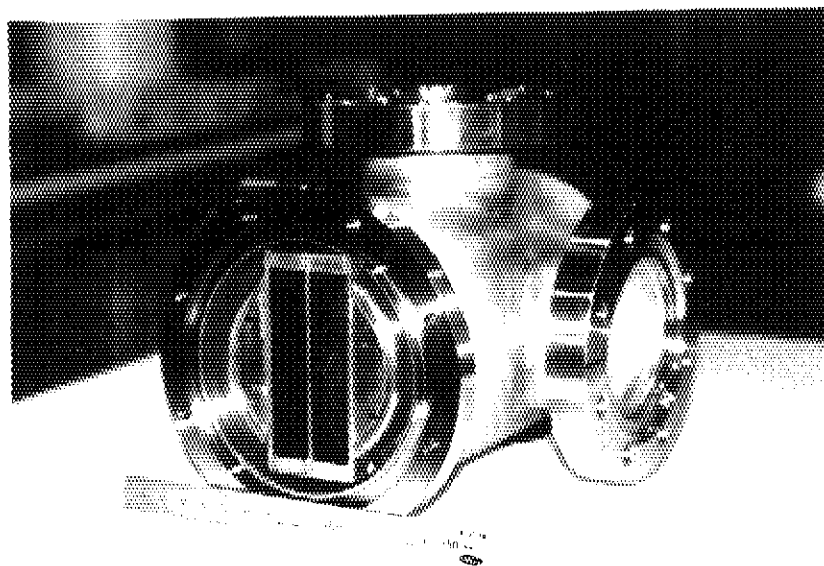


Fig.2 Photograph of 3.7 GHz CFC module.

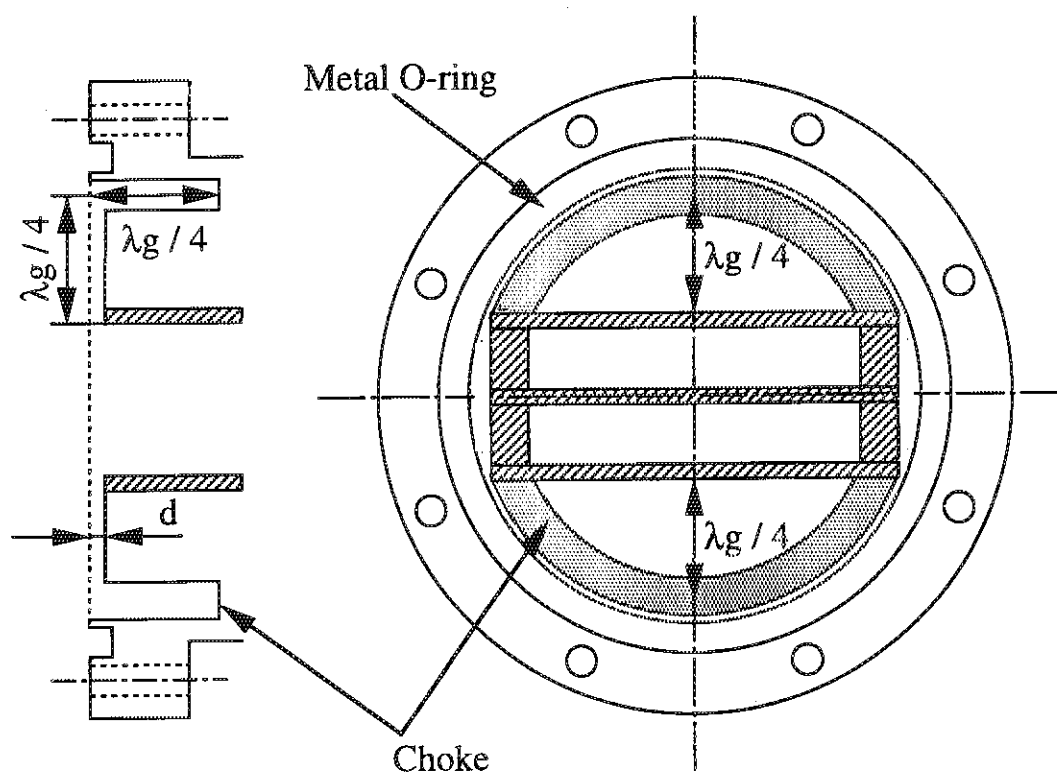


Fig.3 Cross section of the choke flange.

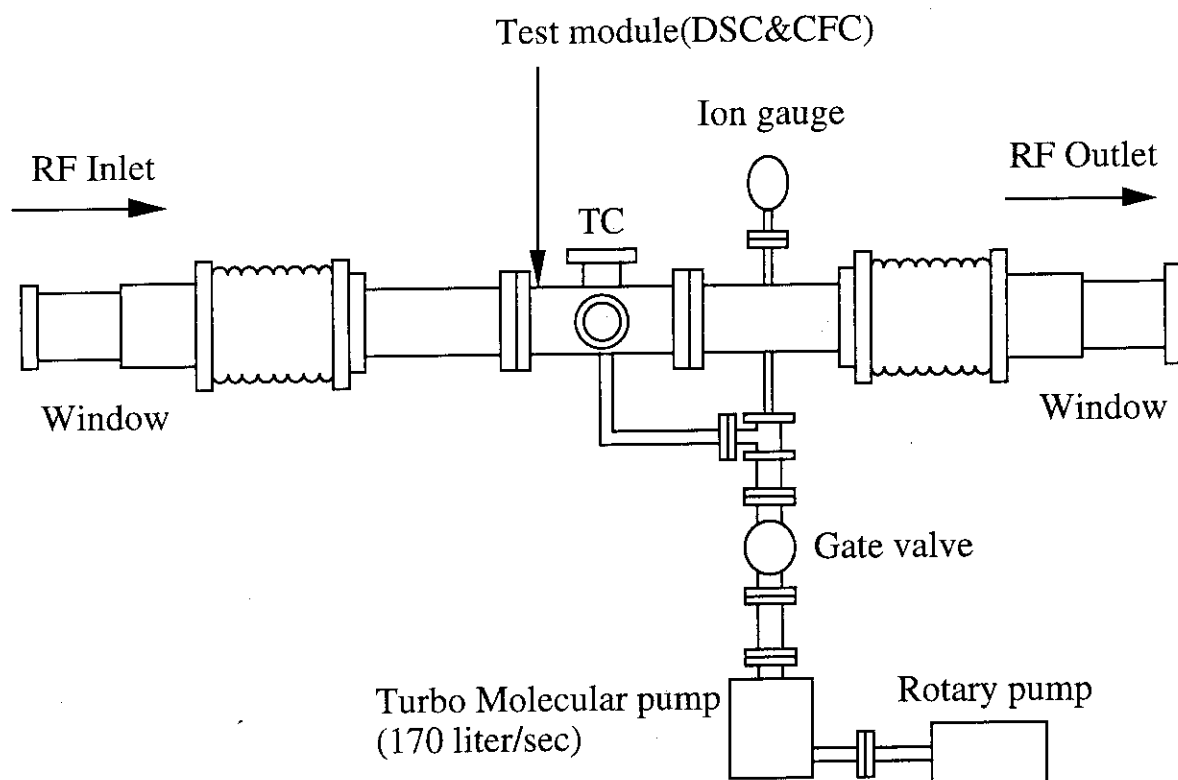


Fig. 4 Experimental Set-up.

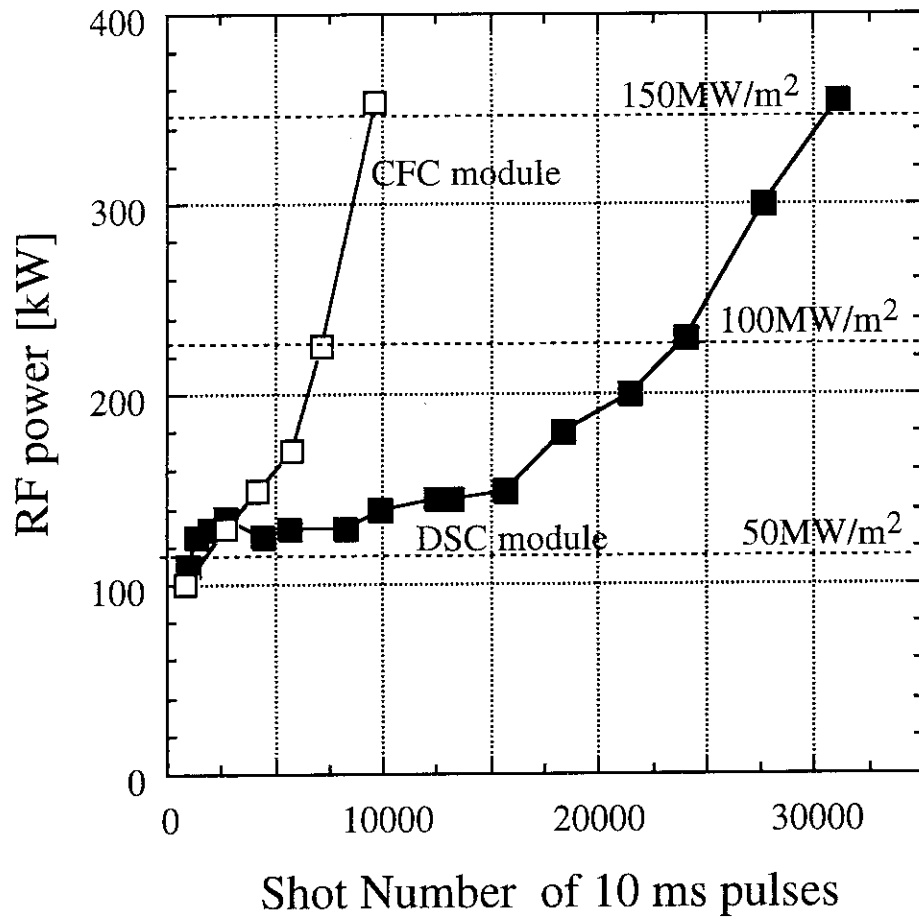
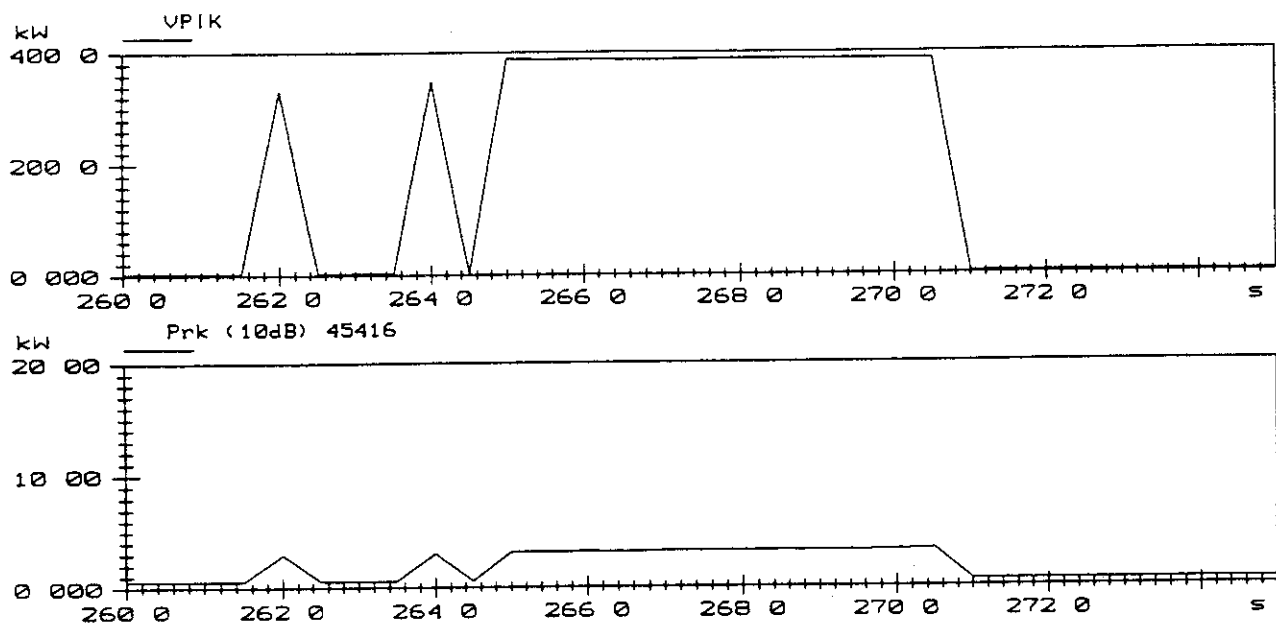


Fig. 5 RF conditioning of DSC and CFC modules

Fig. 6 High power test of the CFC module ( 170MW/m<sup>2</sup>-6s)

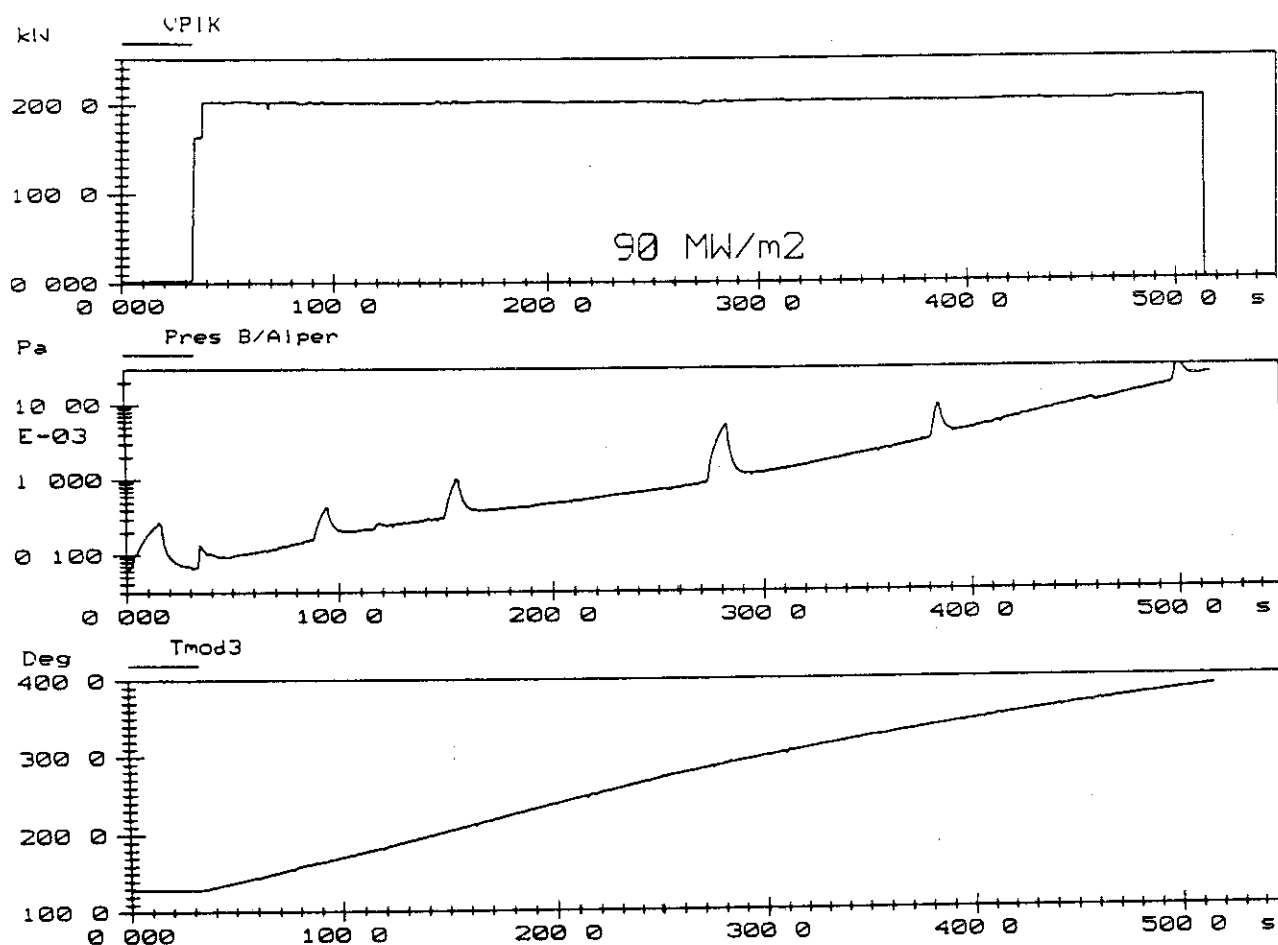


Fig.7 High power test of the CFC module (90MW/m<sup>2</sup>-480s)

- a) Incident power
- b) Pressure
- c) Temperature of the CFC module

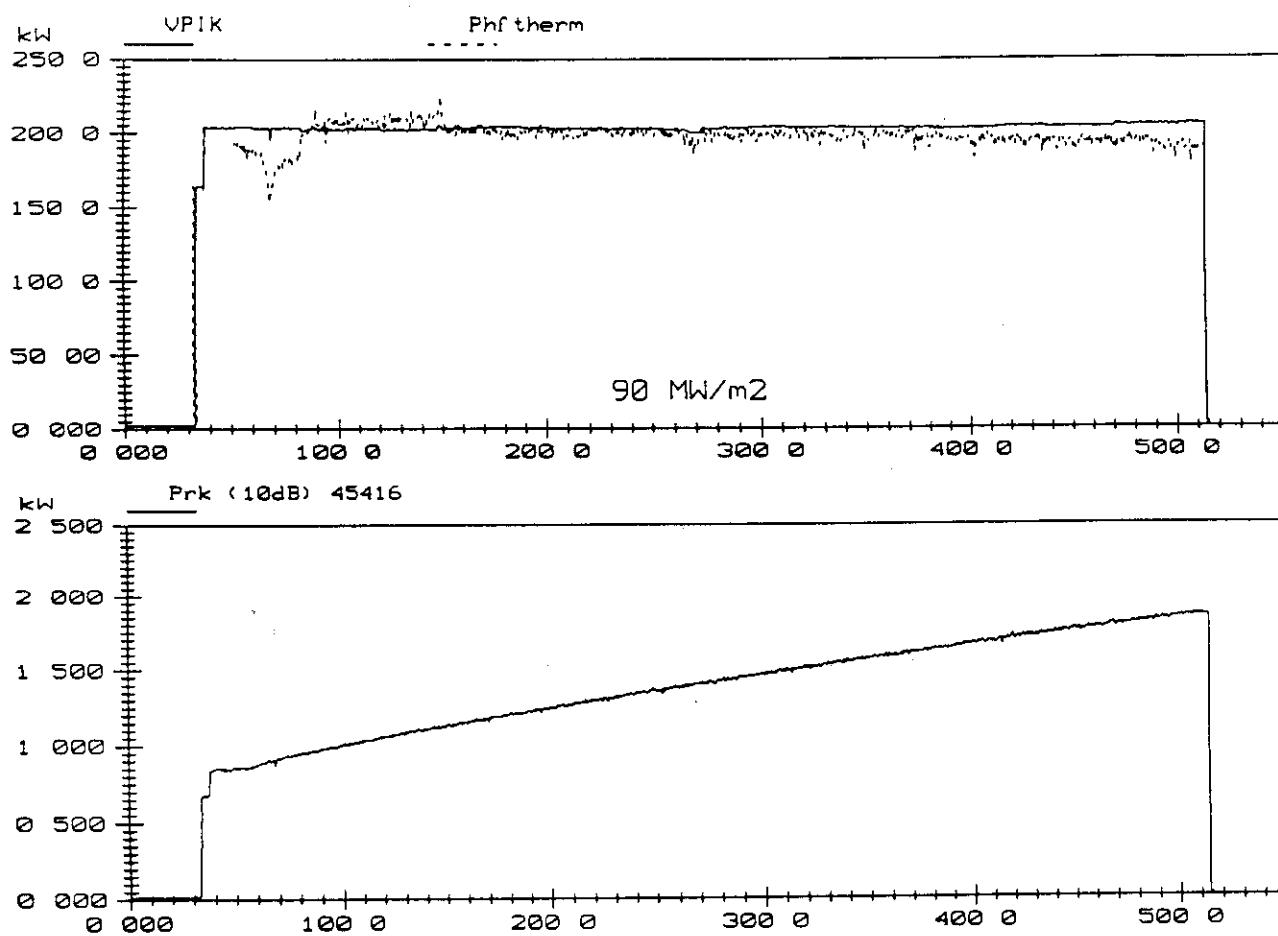


Fig.8 High power test of the CFC module (90MW/m<sup>2</sup>-480s)

- a) Incident power and calorimetric measurement of the power dumped in the end
- b) Reflected power



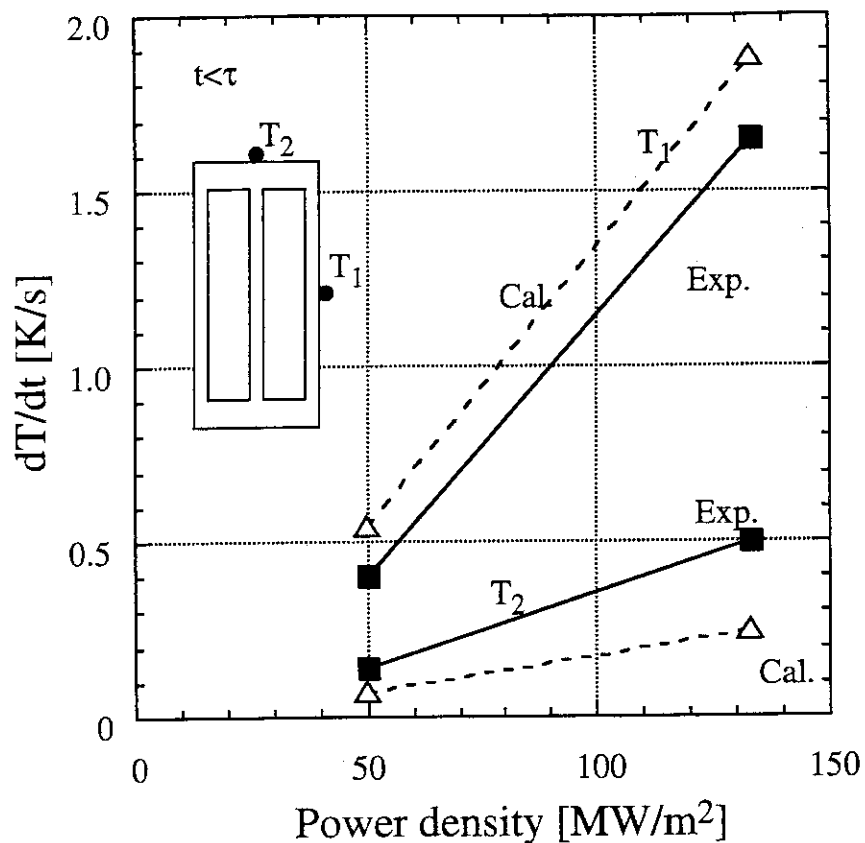


Fig.9 Calculated and experimental heating rates for the DSC module( $t < \tau$ ).

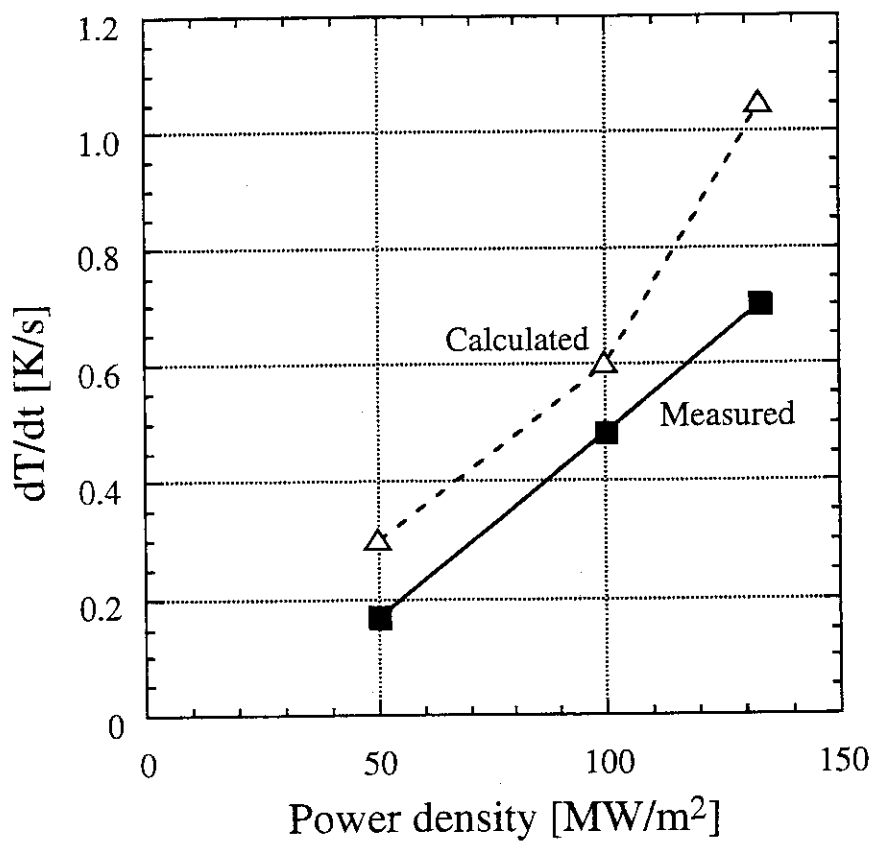


Fig.10 Calculated and experimental heating rates for the DSC module( $t \gg \tau$ ).

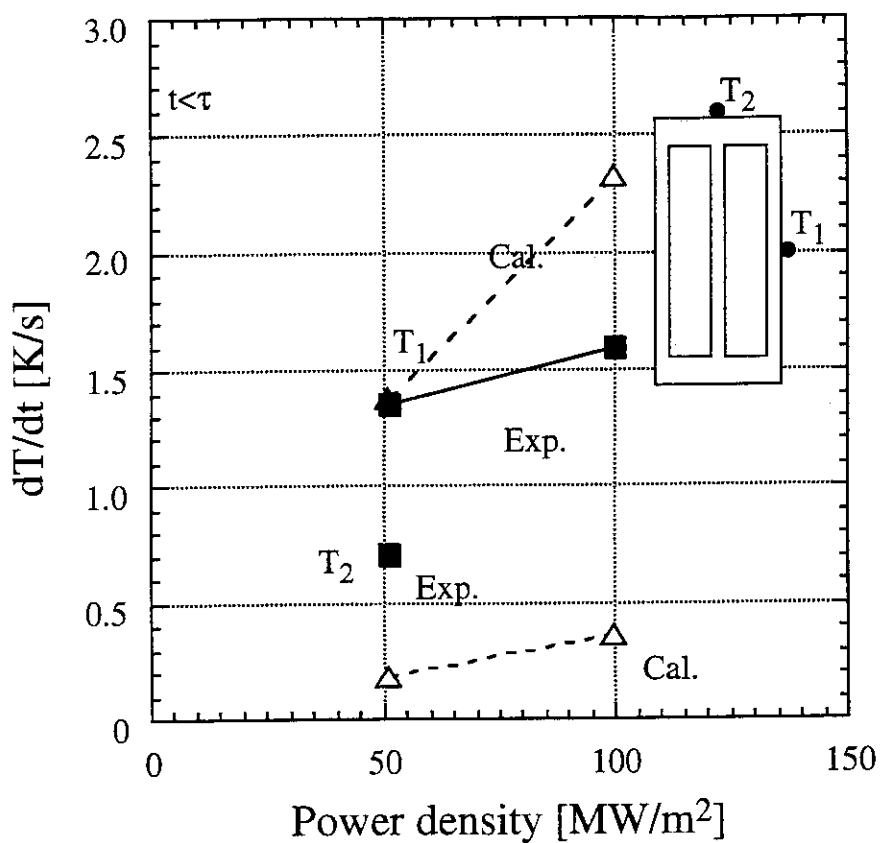


Fig.11 Calculated and experimental heating rates for the CFC module( $t < \tau$ ).

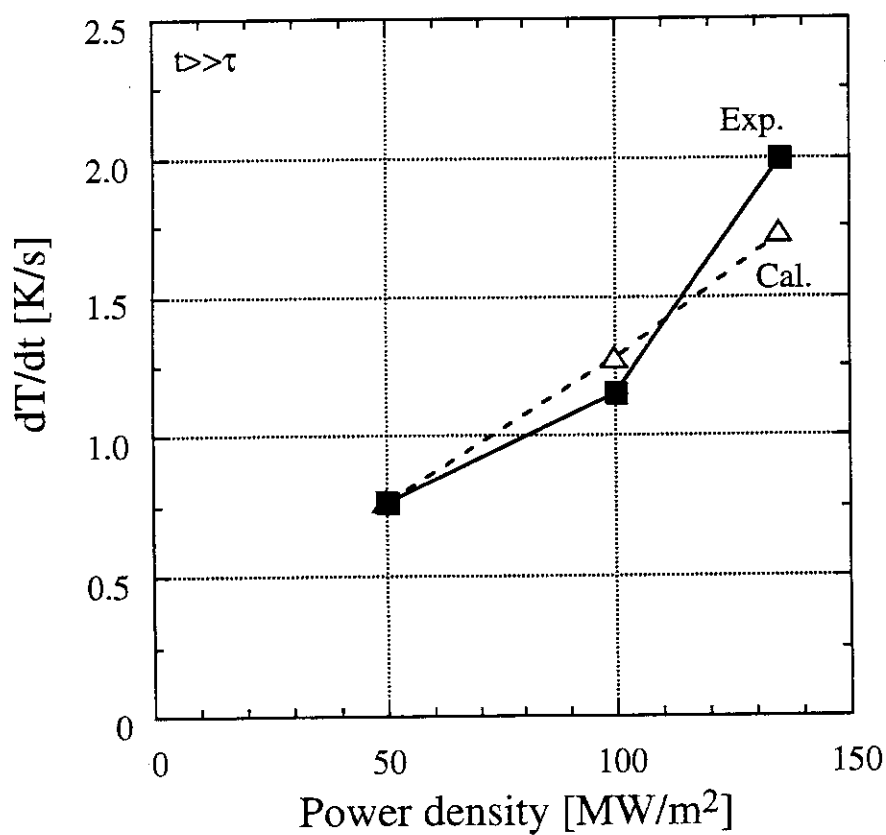
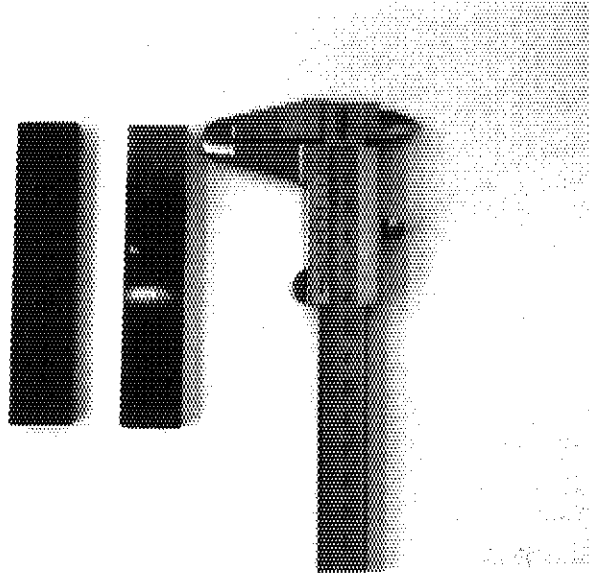


Fig.12 Calculated and experimental heating rates for the CFC module( $t \gg \tau$ ).



The test pieces of CFC and CFC with Cu plating

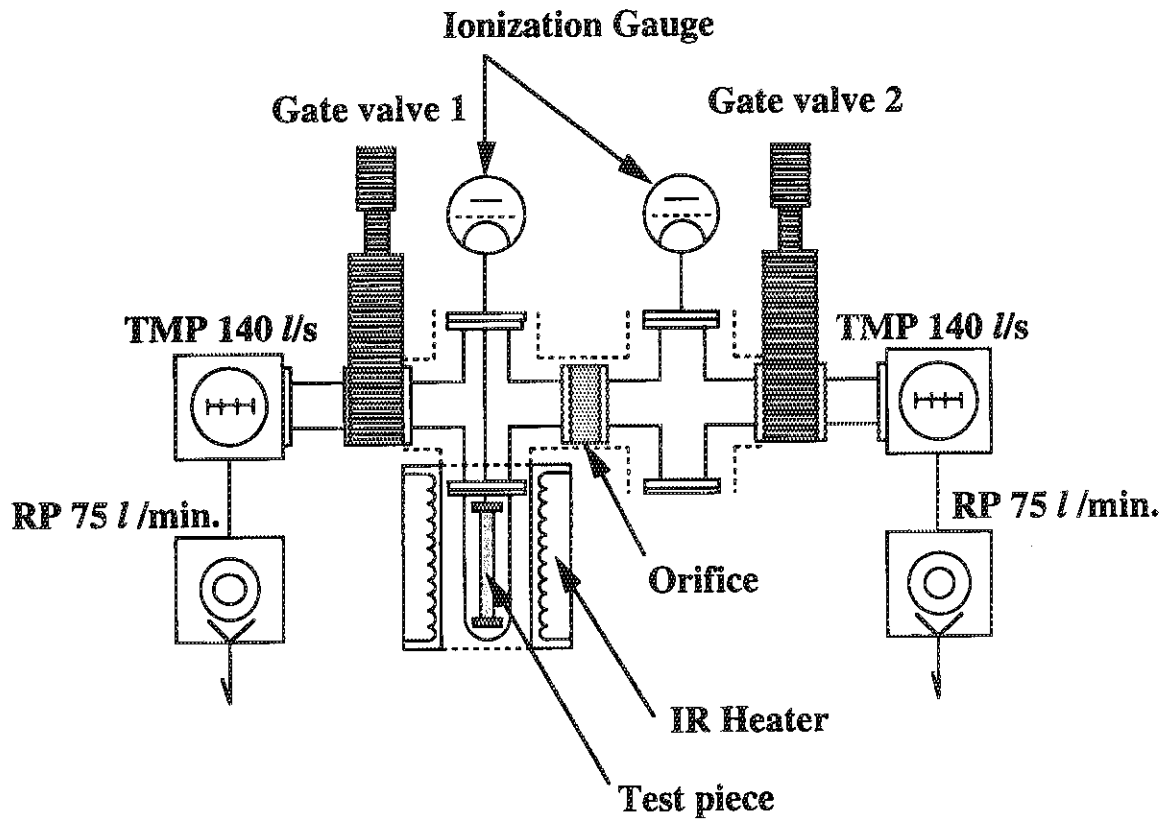


Fig.13 The experimental set-up for outgassing measurement of CFC, CFC with Cu plating and DSC test pieces.

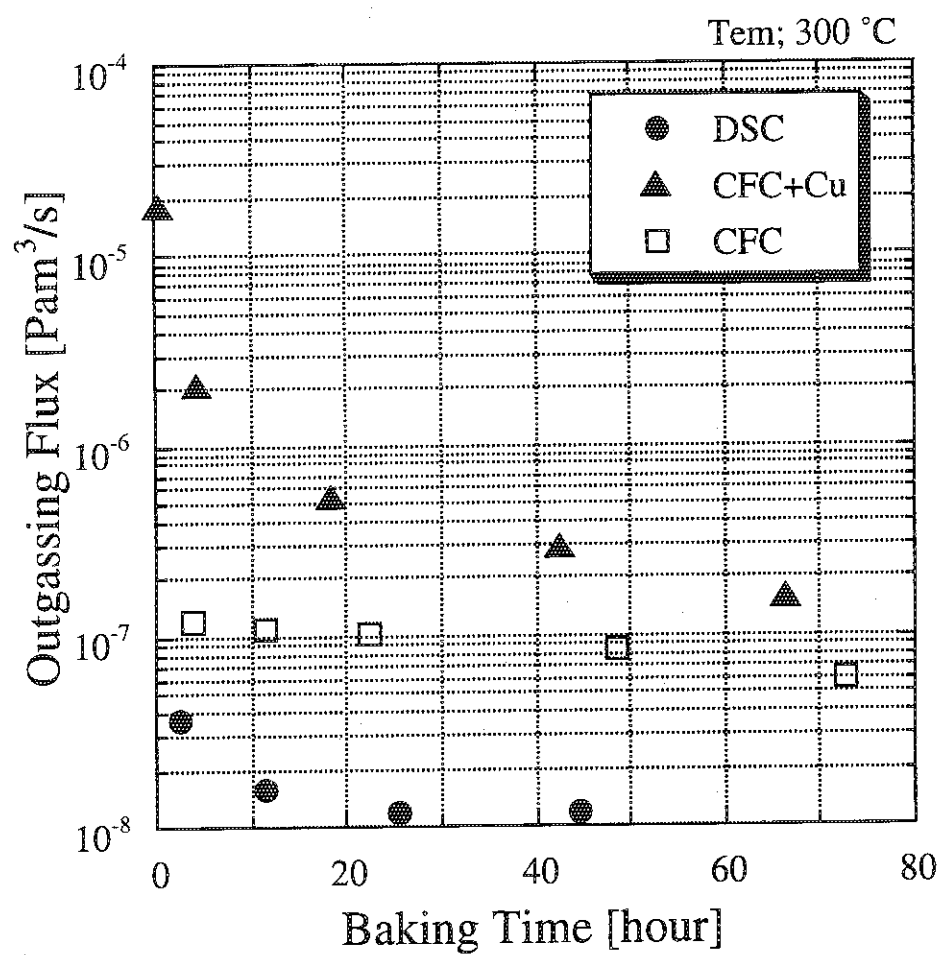


Fig.14 Outgassing flux as function of the baking treatment time.

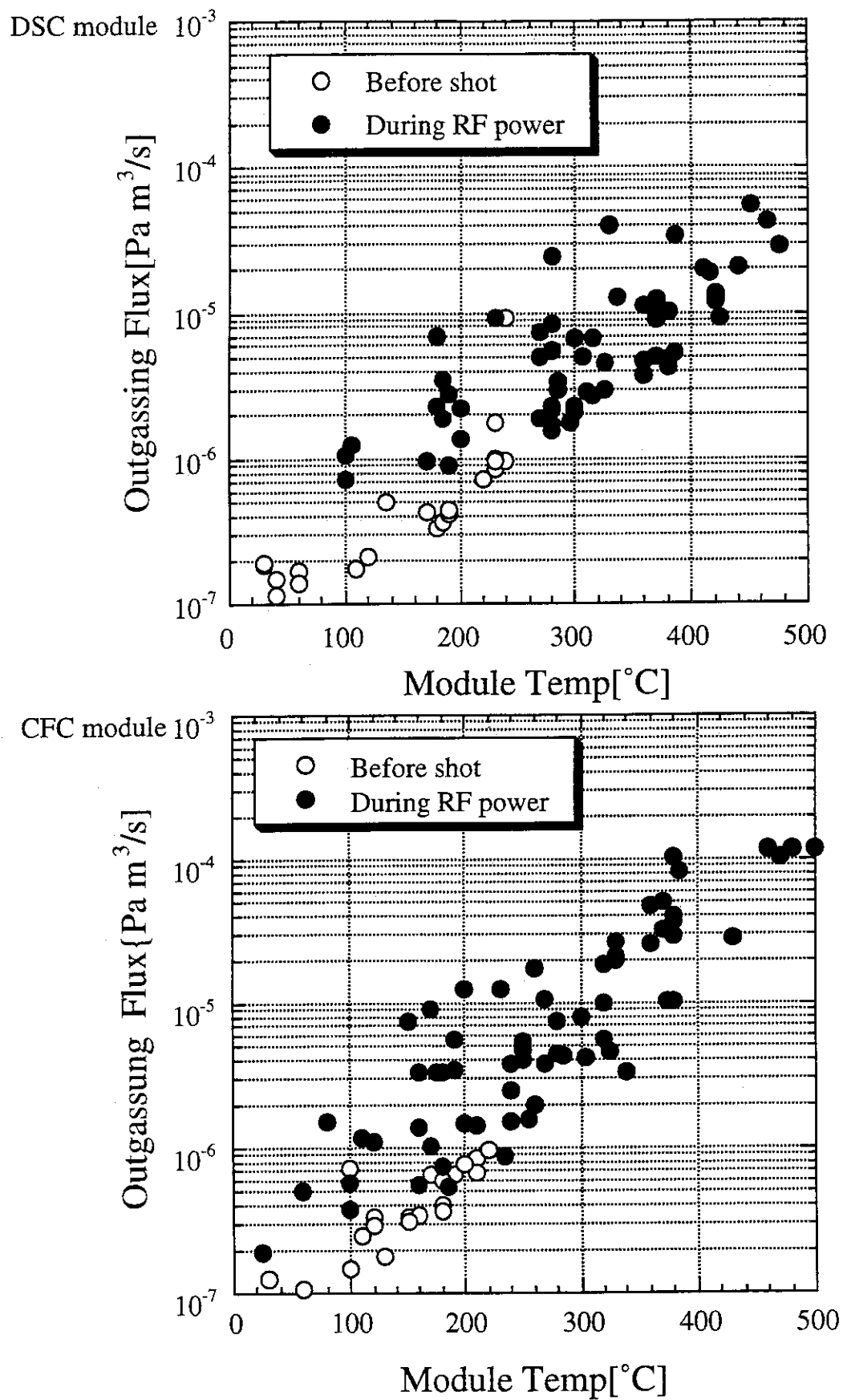


Fig.15 Outgassing flux is plotted versus the module temperature.

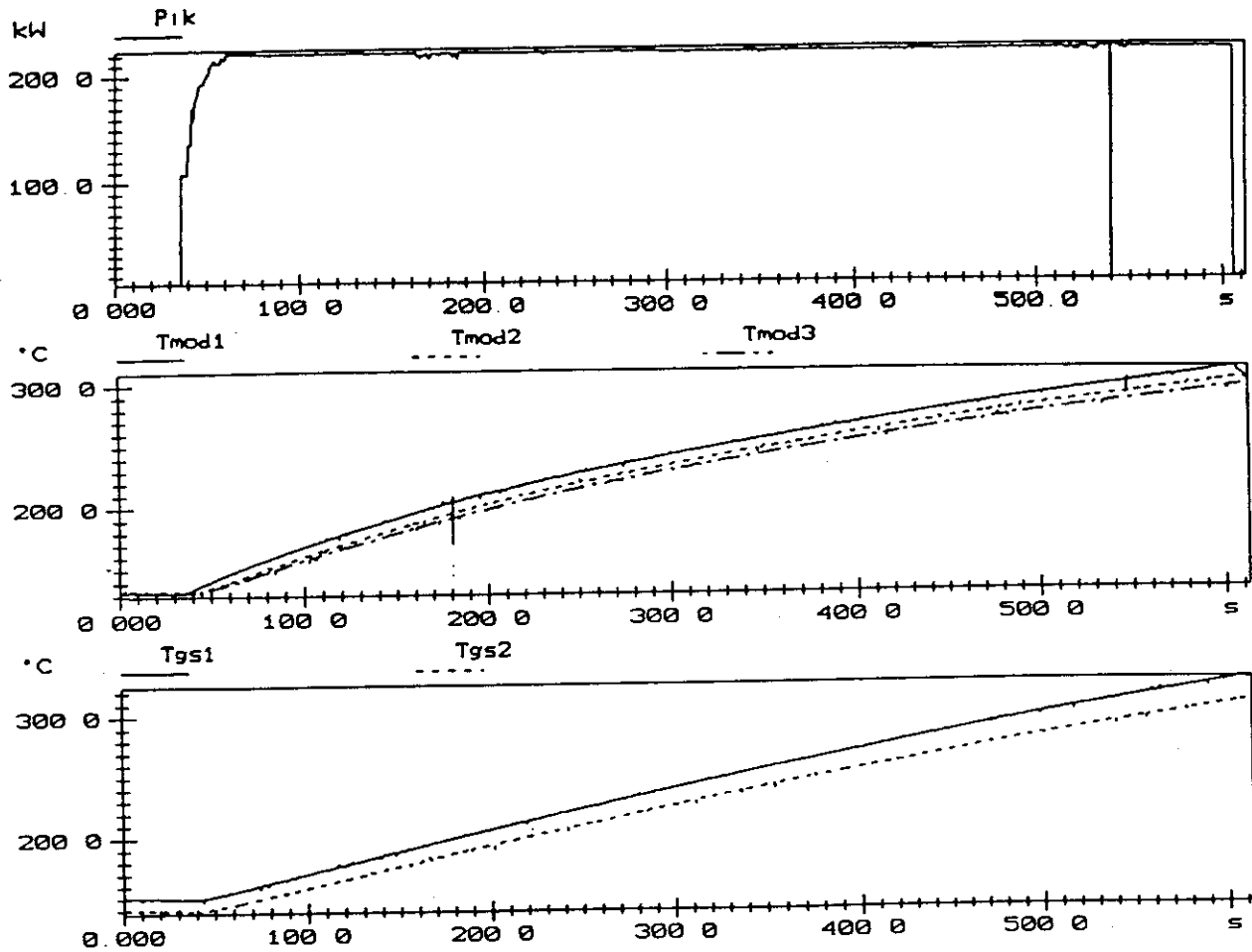


Fig.16 Time trajectory of incident RF power, DSC module temperature ( $T_{module1}$ ,  $T_{module2}$  and  $T_{module3}$ ) and the connection waveguides ( $T_{gs1}$  and  $T_{gs2}$ ).

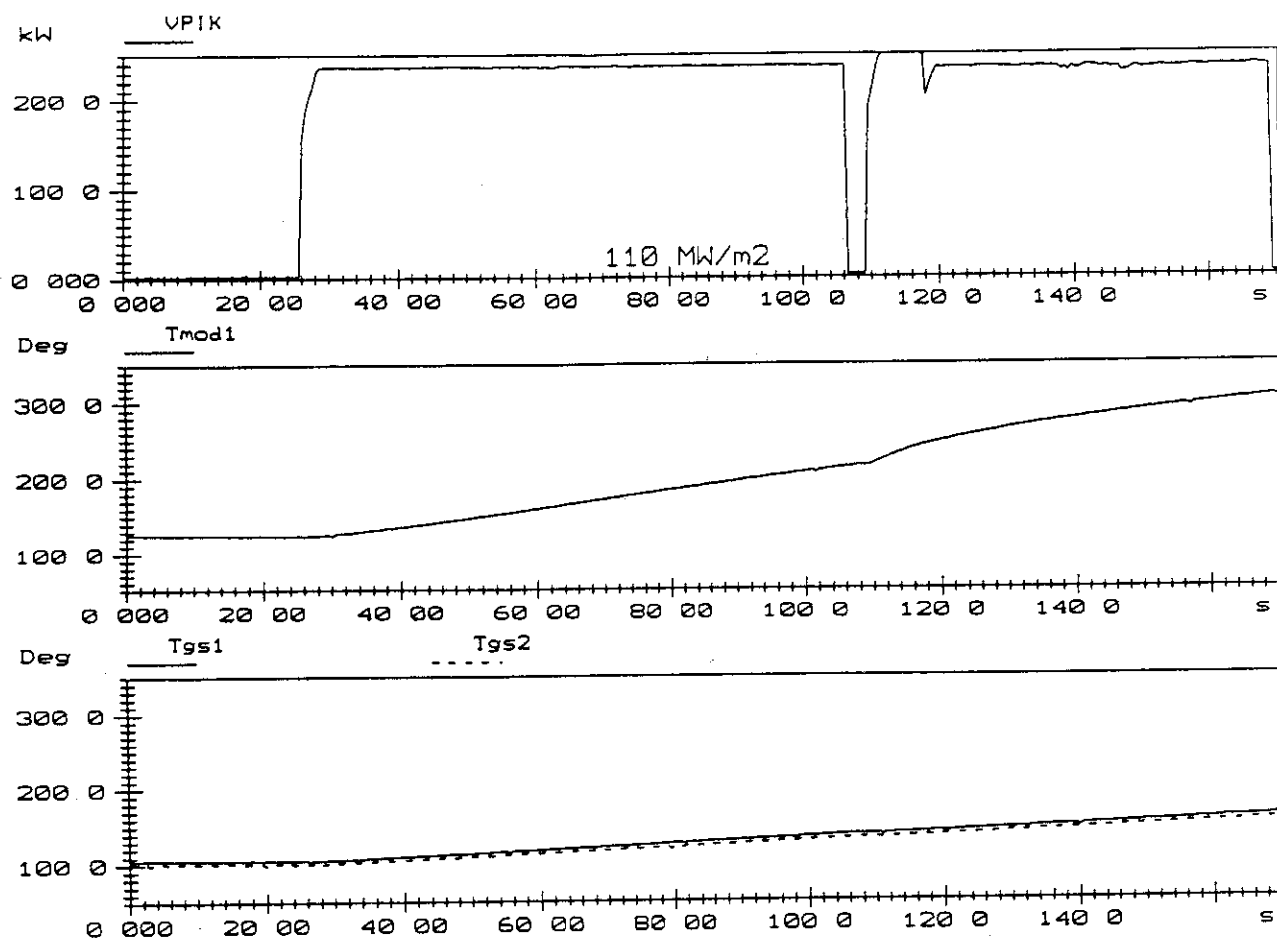


Fig.17 Time trajectory of incident RF power, CFC module temperature ( $T_{\text{module1}}$ ) and the connection waveguides ( $T_{\text{gs1}}$  and  $T_{\text{gs2}}$ ).

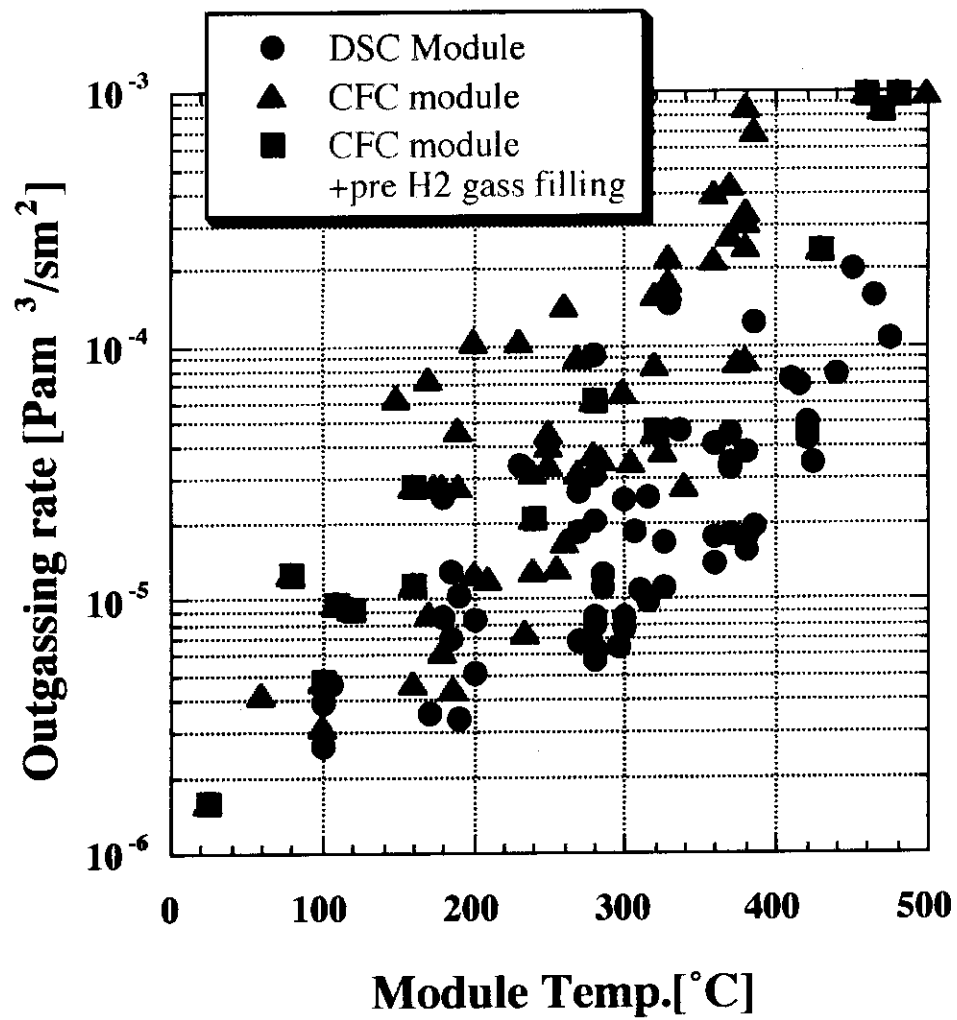


Fig.18 Outgassing rate versus the module temperature in three cases, DSC module, CFC module and CFC module with H<sub>2</sub> gas filling.