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## **Applicability of Two-Dimensional Shielding Code**

**August, 1974**

**POWER REACTOR AND NUCLEAR FUEL DEVELOPMENT CORPORATION**

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## Applicability of Two-Dimensional Shielding Code\*

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### Abstract

Two-dimensional shielding code RASC-2D has been improved to treat line of sight calculation in annular duct with axial symmetrical geometry.

As the application of the improved RASC-2D code, the analysis of the main shield of the Prototype Fast Reactor "MONJU" has been performed and the effects of neutron streaming along voids and the shielding effects of top shield and the adjacent concrete shield are shown and re-evaluated comparing with previous results. (J213 73 01 report).

It has been made clear that Albedo effects at void surface are small at high energy region and large at low energy region.

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This is the translation of the report, No. J213 73-02, issued in November, 1973.

\* Work performed by Kawasaki Heavy Industries, Ltd. under contract with Power Reactor and Nuclear Fuel Development Corporation.

\*\* Kawasaki Heavy Industries, Ltd.

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## I. Purpose and Outline

In previous study, two-dimensional diffusion calculation code DIAC was improved to make it possible to analize large two-dimensional shielding system including voids using line of sight and diffusion method, and two-dimensional analysis of "MONJU" main shield including voids was done, and it was shown that voids in shield region gave a considerable effects to neutron flux distribution. In that analysis, however, neutrons entering into voids from void surface were assumed to have cosine angular distribution, so when the void intersected with radial center axis as N<sub>2</sub> gas layer of "MONJU", it was considered to overestimate the neutrons streaming from void surface on radial center axis to upper part of void.

In this study, the code has been improved to estimate exactly the neutron streaming from void surface on radial center axis to upper part of void under condition of axial symmetrical geometry. In this study following works have been performed.

### I. Improvement of DIAC Code

- i) Addition of axial symmetrical line of sight calculation in void.
- ii) Removalent of minus neutron current entering into void from void surface at high energy region.

II. Two-dimensional analysis of "MONJU" main shield including void.

From these works, it has been made clear that the analysis including void is necessary and streaming neutrons in void affect to neutron flux distribution as far as energy is high. These results are reported.

## III. Improvement of RASC-2D Code

### (III-1) Addition of Axial Symmetrical Line of sight Calculation

#### 1) Calculation method of axial symmetric void

According to the assumption of axial symmetric geometry to the void intersecting with radial center axis, space meshes around void are assumed to be axial symmetric.

Neutron current  $J_i^-$  entering into an element  $i$  on the void surface can be obtained from neutron current  $J_0^+$  flowing out from other surface element  $0$ , as follows. (Previous report J213 73 01)

$$J_i^- = \frac{1}{F_i} \sum_{0=1}^n F_0 P_{0 \rightarrow i} J_0^+ \quad (1)$$

At axial symmetrical void, neutron current  $J_i^-$  entering to the upper element than radial center axis, which is necessary to diffusion calculation, can be devided to two contributions from upper and lower part. That is

$$J_i^- = \frac{1}{F_i} \sum_{0=1}^n F_0 P_{0 \rightarrow i} J_0^+ + \frac{1}{F_i} \sum_{0'=1}^n F_{0'} P_{0' \rightarrow i} J_{0'}^+. \quad (2)$$

$\sum_{0=1}^n$  : Sum of upper element

$\sum_{0=1}^n$  : Sum of lower element

$P_{0 \rightarrow i}$  Probability entering from upper element 0 to  $i$

$P_{0' \rightarrow i}$  Probability entering from lower element 0' to  $i$

Here, if lower element  $0'$  is assumed to exist at symmetric position with upper element  $0$ , neutron current  $J_0^+$  and surface cross-section  $F_0$ , equal to  $J_0^+$  and  $F_0$ . So equation (2) can be represented as follows;

$$J_i^- = \frac{1}{F_i} \sum_{0=1}^n F_0 (P_{0 \rightarrow i} + P_{0' \rightarrow i}) J_0^+. \quad (3)$$

And if

$$P_{0 \rightarrow i} = P_{0 \rightarrow i} + P_{0' \rightarrow i}, \quad (4)$$

equation (3) and equation (1) are same.

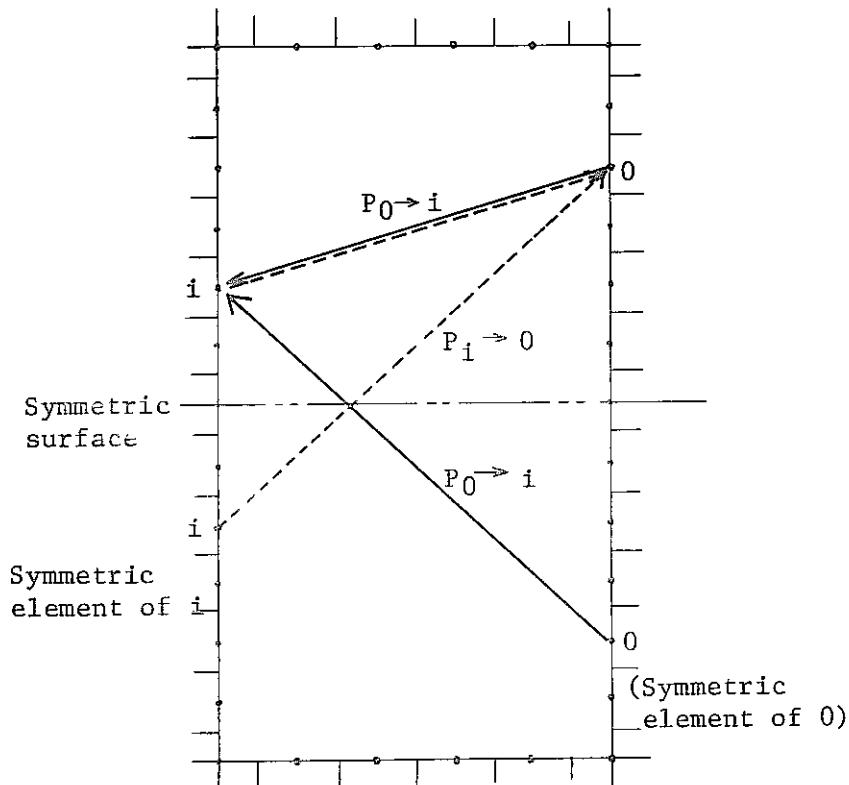


Fig. (II-1) Geometry of Probability Calculation

Consequently, neutron current entering from lower element (accordingly from all element) can be obtained by using as  $P_{0 \rightarrow i}$  outward probability obtained from summation of true probability  $P_{0 \rightarrow i}$  and  $P_{0' \rightarrow i}$  when neutron current  $J_i^-$  entering from upper element is calculated by equation (1).

As true probability  $P_{0 \rightarrow i}$  and  $P_{0' \rightarrow i}$  are

$$P_{0 \rightarrow i} = \frac{F_i}{F_0} P_{i \rightarrow 0}, \quad P_{0' \rightarrow i} = \frac{F_i}{F_0} P_{i' \rightarrow 0'},$$

outward probability  $P_{0 \rightarrow i}$  from element 0 to i is represented as follows by equation (4).

$$P_{0 \rightarrow i} = \frac{F_i}{F_0} (P_{i \rightarrow 0} + P_{i' \rightarrow 0'})$$

Here  $P_{i' \rightarrow 0'}$  equals to  $P_{i \rightarrow 0}$  because of symmetry of void, then  $P_{0 \rightarrow i}$  is same as true probability, that is

$$P_{0 \rightarrow i} = \frac{F_i}{F_0} (P_{i \rightarrow 0} + P_{i' \rightarrow 0}),$$

and

$$P_{0 \rightarrow i} = \frac{F_i}{F_0} P_{i \rightarrow 0}.$$

Finally, outward probability  $P_{0 \rightarrow i}$  and  $P_{i \rightarrow 0}$  are sufficient by calculating either of the two.

## 2) Modification of Code

Modification has been done mainly in subroutine PROFAC, and other elemental modification has been

unnecessary.

Fig. (II-2) shows subroutine PRØFAC's flow diagram after modification, and right-hand parts have been newly added.

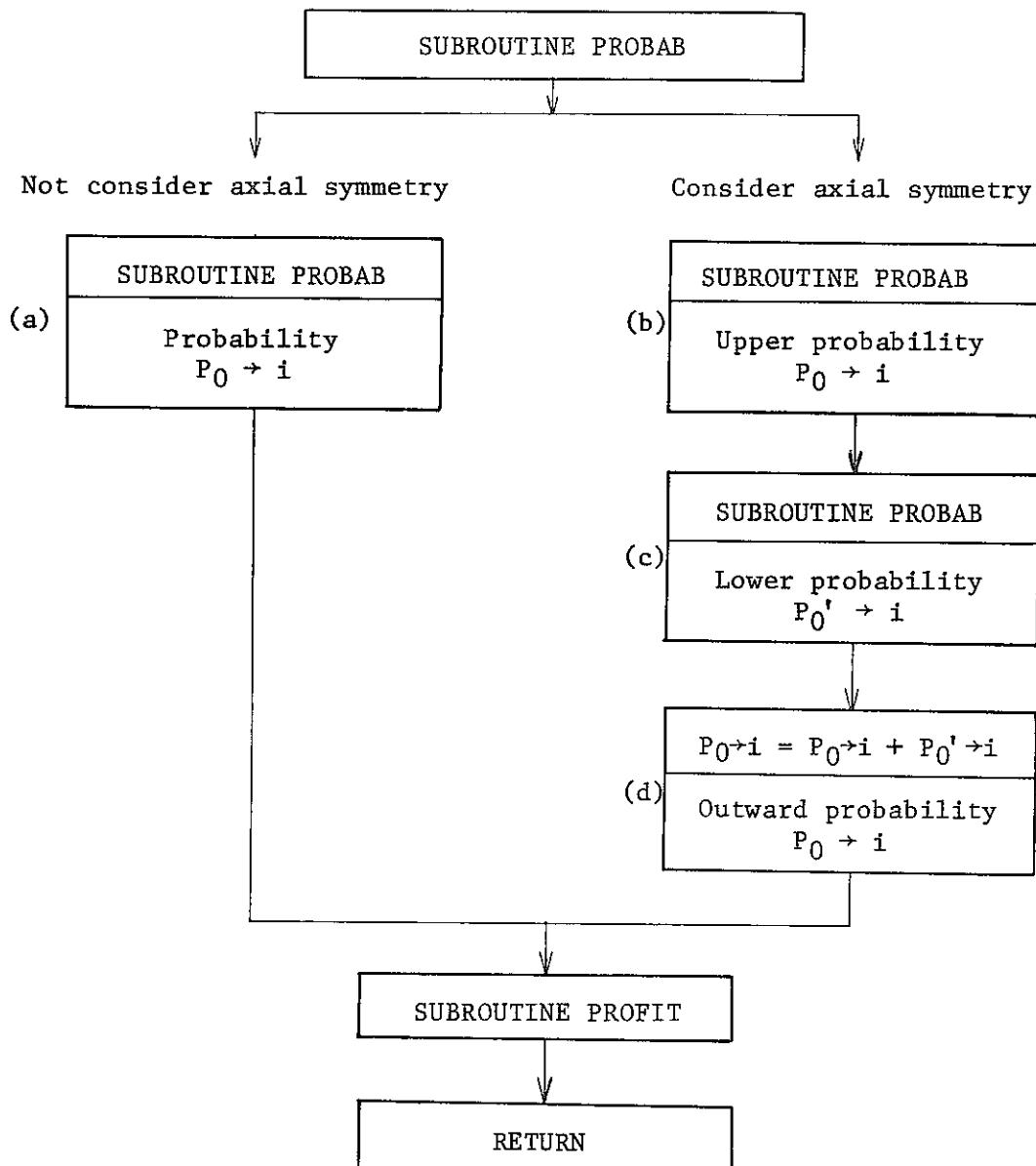


Fig. (II-2) Flow Diagram of Subroutine PROFAC

### 3) Test calculation of probability

Test calculation has been done to examine whether the probability between lower element and upper one of axial symmetrical void is calculated exactly or not. Fig. (II-3) shows the Test Calculation configuration same as previous one.

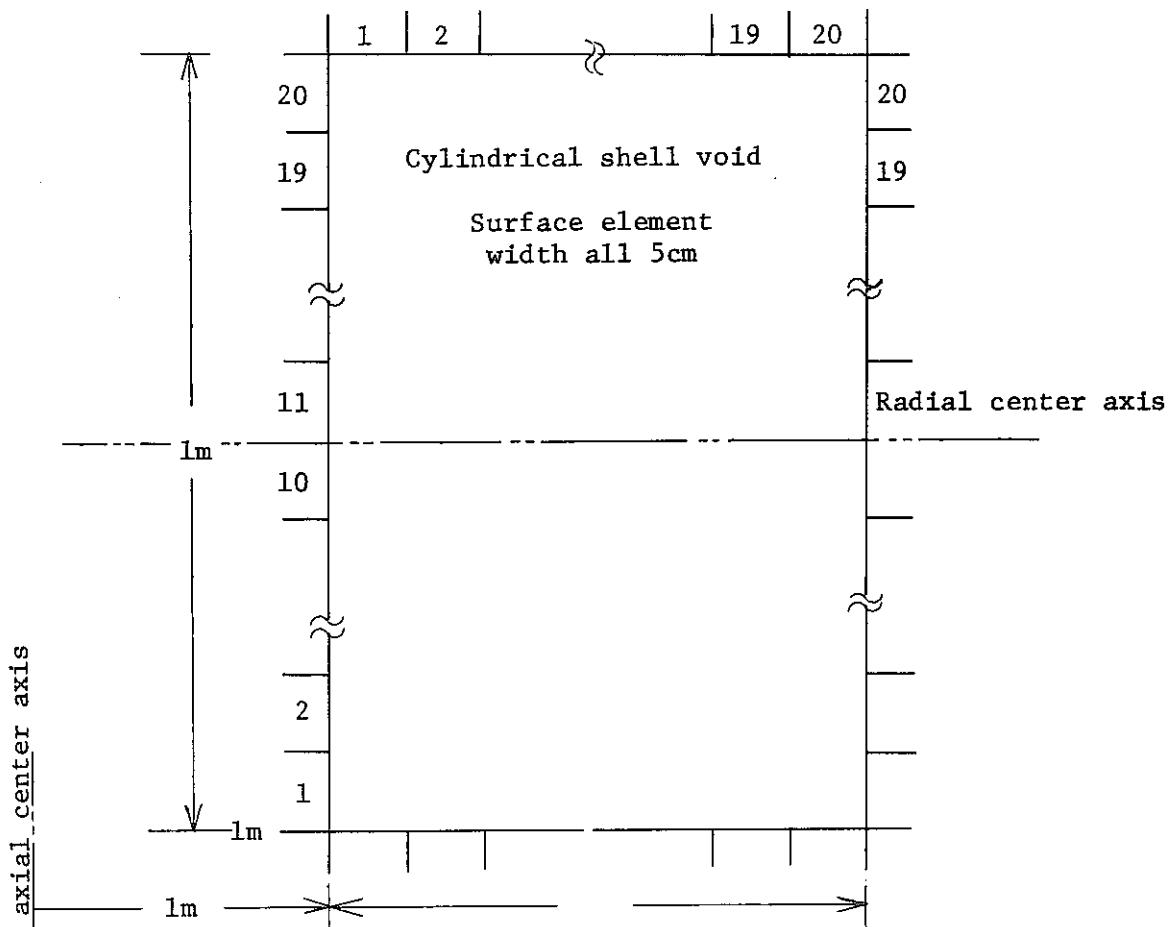


Fig. (II-3) Geometry of Probability Test Calculation

- (i) The probability calculation result has agreed with previous calculation using ordinary method ((a) of Fig. (II-2)) when it has been assumed that this void exists above radial center axis.
- (ii) Probability has been calculated on the assumption that the upper half of the void exists above radial center axis. The results of upper probability given by (b) in Fig. (II-2) and the results of lower one given by (c) in Fig. (II-2) have agreed each corresponding results of (i).
- (iii) And in (ii), total sum of outward probabilities given by (b) in Fig. (II-2) and outward probabilities between upper element lower one has been equal to 1.00.

Fig. (II-4) shows examples of calculated probabilities. The symbol of (a), (b), (c) and (d) represents each of the results of (a), (b), (c) and (d) in Fig. (II-2).

From (i), (ii) and (iii), it is understood that modification based on the chapter (II-1) is done exactly.

### (II-2) Removalent of Minus Neutron Current

In high energy region, minus neutron current entering into void from void surface appears sometimes on the calculation procedure. Albedo  $\beta$  calculated according to the approximate formula  $\beta = \frac{1-2D/L}{1+2D/L}$  using the constant of "MONJU" main shield is shown in Table (II-1). Albedo of first energy group is minus. That the neutron current entering into void from void surface is minus is consistent with minus albedo. Actually  $J^+$  (or  $\beta$ ) is determined by a diffusion constant and the gradient of neutron flux, so in some cases neutron current entering into void from some space meshes on the void surface becomes minus number. Physically minus neutron current can not exist, and in this energy region diffusion approximation does not formed because forward scattering is much greater than back scattering. In this code, however, minus neutron current  $J^+$  is considered to be the diffusion approximation expression that the neutrons entering into void are almost negligible, and when the minus current appears it is put into zero.

Before this modification of the code, neutron current is put into zero when the current entering into some mesh becomes minus after suming up all mesh of void surface. This method ignores the contribution of all other mesh when the minus contribution of some mesh is large, so sometimes distorted neutron flux distribution is calculated under opposite boundary conditions that the gradient of neutron flux in some mesh is minus and in adjacent mesh is plus (see Fig. (II-5)).

From above reason, DIAC code has been modified to put zero the minus current entering into void immediately after the current is calculated. From this, the distortion of neutron flux is removed as is shown in Fig. (II-6).

Table(II-1) Neutron Albedo

$$\beta = \frac{1-2KD}{1+2KD} \quad K = \frac{1}{L} \sqrt{\sum a_D}$$

Energy Group No.	Albedo $\beta$			
	SUS	Fe	Na	C
1	-0.02	-0.03	-0.02	-0.16
2	0.10	0.10	0.13	0.20
3	0.39	0.38	0.31	0.29
4	0.60	0.68	0.49	0.47
5	0.68	0.61	0.58	0.47
6	0.66	0.68	0.54	0.49
7	0.68	0.66	0.61	0.51
8	0.61	0.57	0.69	0.52
9	0.37	0.23	0.47	0.94

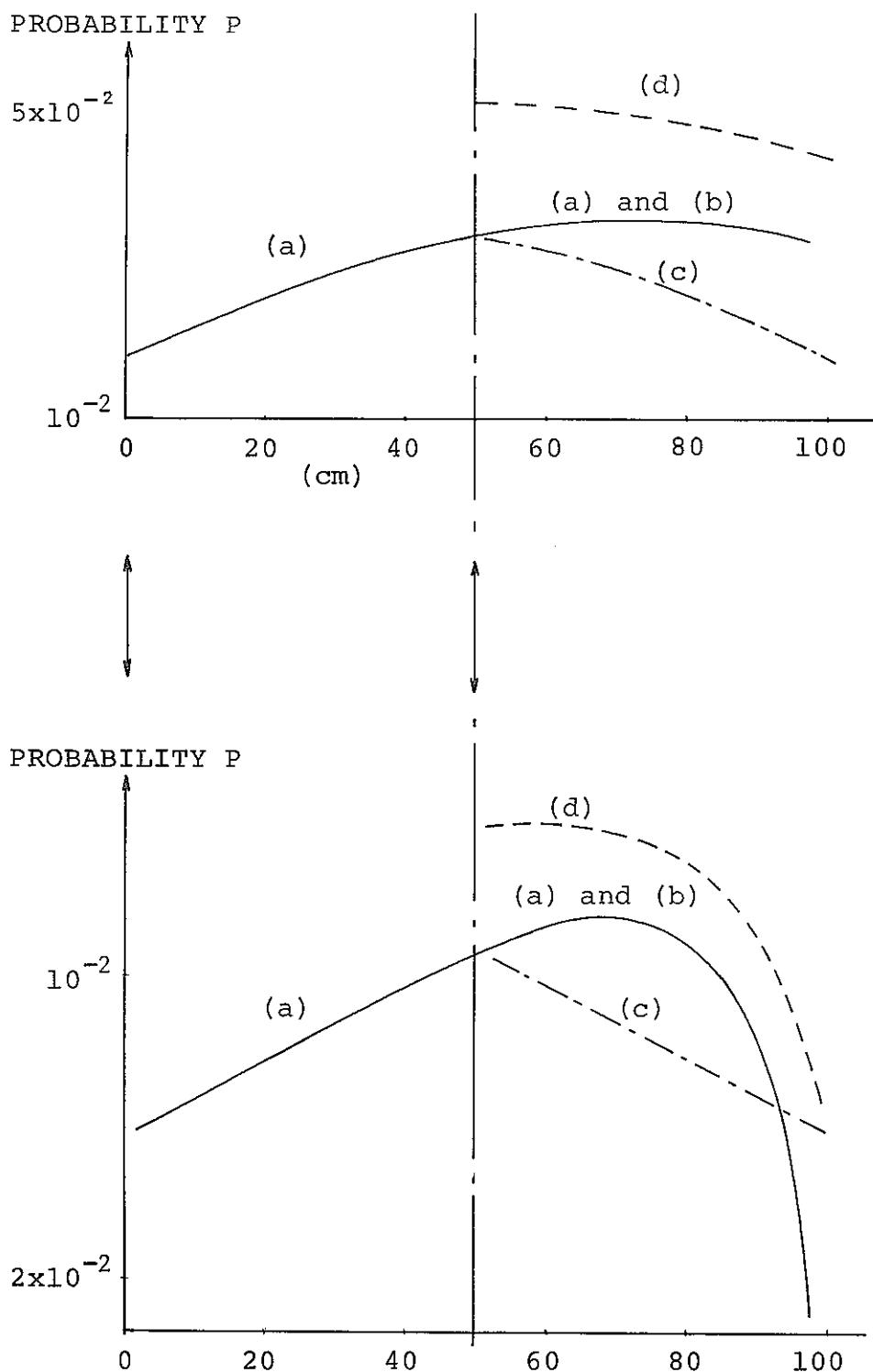


Fig.II-4 Test Calculation of Probability

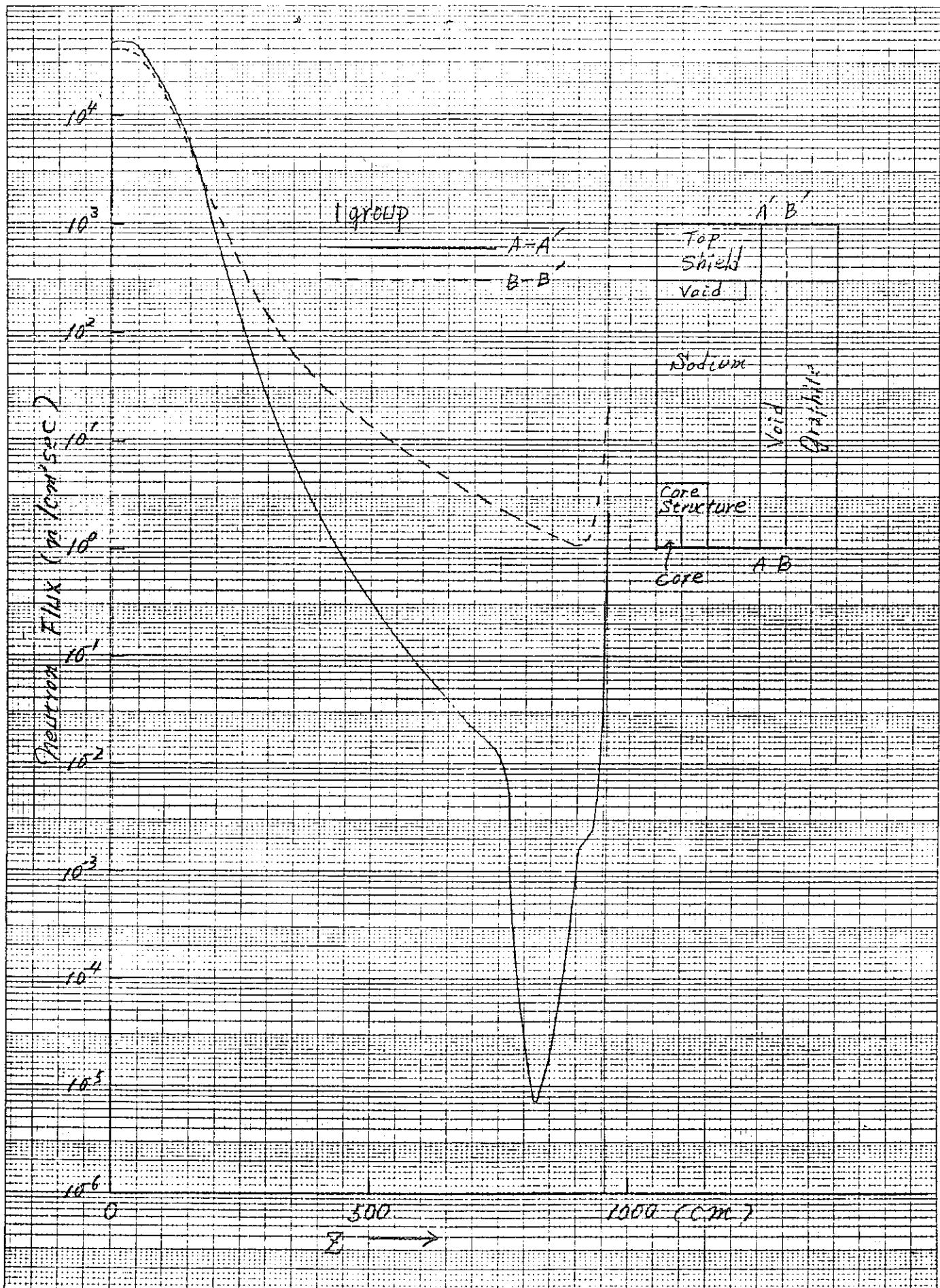


Fig.(II-5) Neutron Flux around Void  
before Modification of Code

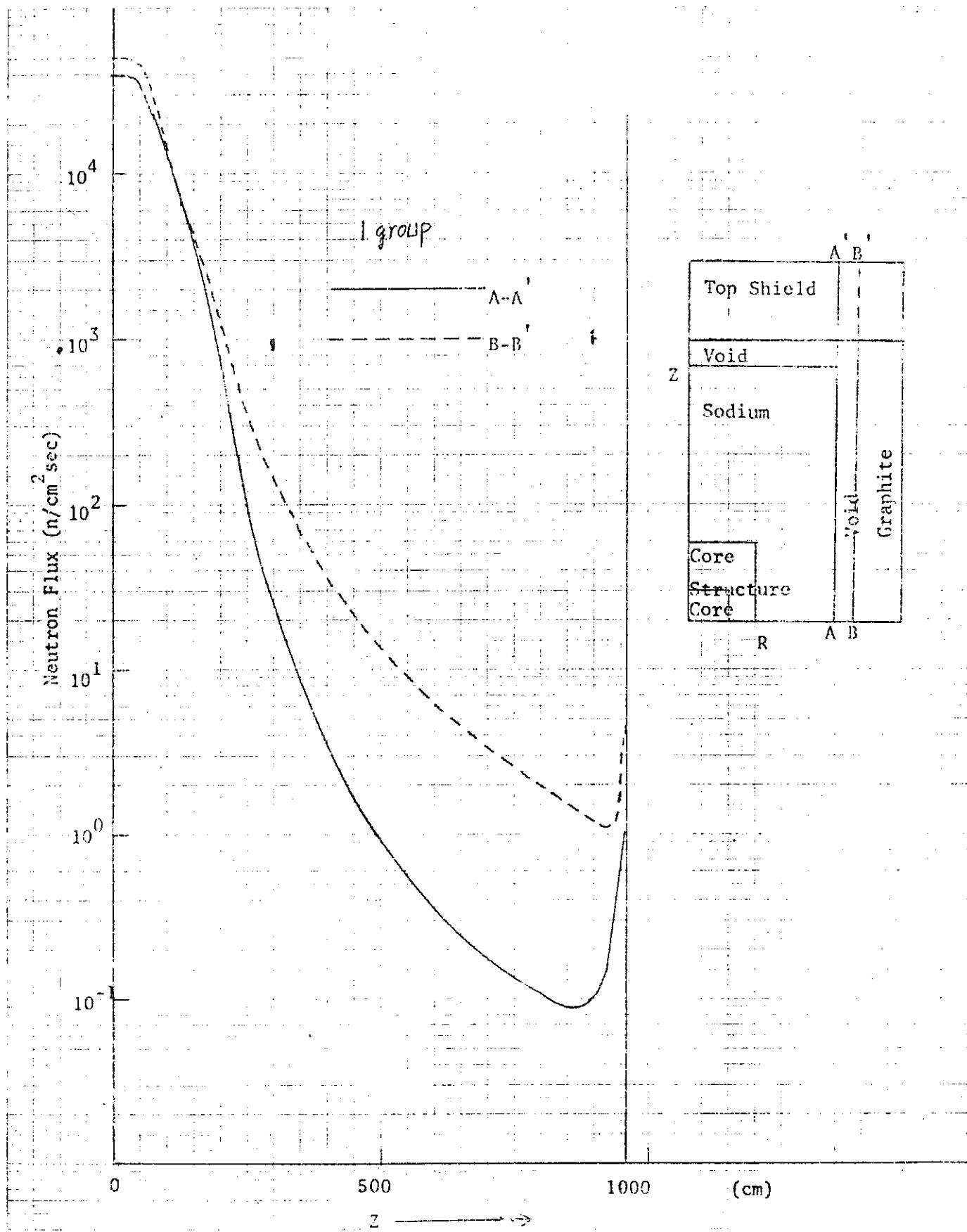


Fig. (II-6) Neutron Flux around Void after Modification of Code

### III. Evaluation of Void Effect of Prototype Fast Reactor "MONJU" Main Shield

#### (III-1) Outline

By using the RASC-2D (DIAC) code as described in the previous chapter, two-dimensional neutron flux distribution in the main shield of the fast prototype reactor "MONJU" has been calculated under condition that in N<sub>2</sub> gas void between reactor vessel and graphite shield, neutrons streaming from lower part below radial central axis are calculated according axial symmetrical geometry.

The following two points has been the main concern in this study.

- 1) Comparison between calculation of not including void (J213 72-03-01 before previous calculation) and this calculation. Comparing before previous analysis, this analysis contains following newly added four layers.
  - i) Top shield plug.
  - ii) Concrete shield adjacent to above plug.
  - iii) Ar gas void layer on the sodium liquid surface inside reactor vessel.
  - iv) N<sub>2</sub> gas void layer between the reactor vessel and graphite shield.

The problems have been that what extent the neutron flux disturbance occurs by the newly added layers, and especially that whether the neutrons streaming in the N<sub>2</sub> gas void raise the flux level at the top shield plug or not.

2) Comparison between this calculation and the calculation under condition that the neutrons enter into N<sub>2</sub> gas void from the void surface intersecting with radial center axis. (J213 73-01, previous calculation). The problem has been that to what extent overestimate the neutron streaming to upper part, and that whether the neutron flux distribution in shields will be affected or not.

### (III-2) Calculation Method

#### 1) Calculation Configuration

Fig. (III-1) shows the calculation configuration of prototype fast reactor "MONJU" main shield. Analysis has been done covering radial direction up to the graphite shield outside of the reactor vessel and axial direction up to the top-side of the upper shield plug. As the voids, N<sub>2</sub> gas layer between reactor vessel and graphite shield, and Ar gas layer on the sodium liquid surface have been taken into account. As the analysable forms of voids in RASC-2D code are limited to only the cylindrical shape and cylindrical shell shape, a simplification of actual void configuration has been done.

#### 2) Energy Group Structure

It has been analysed entirely under the identical energy structure as before previous and previous calculations. It has been with seven groups of removal

neutrons and nine groups of diffusion neutrons of which energy structures are shown in Table (III-1) and (III-2) respectively.

### 3) Group Constant

The densities of atomic elements composing each layer are given in Table (III-3).

#### i) Removal Cross Section

For the removal cross section in each layer below the dip-plate, those which have been prepared in the before previous calculation as the previous calculation have been employed. These have been prepared so as to adjust the removal-diffusion calculation result to the result of one dimensional transport calculation code ANISN. The removal cross sections of the top shield layers have not been adjusted other than these have been collapsed from the basic constants of 25-group to 7-group. Table (III-4) shows the removal cross sections employed for the analysis.

#### ii) Diffusion Group Constant

For the diffusion group constant at each layer below the dip-plate, those which have been prepared in the before previous calculation as previous calculation have been used the same as in the case of removal cross section. That is, for the core and blanket regions, the 9-group collapsed constants produced as an output from one dimensional core calculation code EXPANDA-2 have been obtained by converting into a

continuous slowing-down type from a matrix type. For other shield regions, 31-group one-dimensional calculations have been done in both r-direction and z-direction, and with these 31 group spectrums as a weight, 31-group constants have been collapsed into 9 groups, and the slowing down model has been converted into continuous type from the matrix one. The constants of the upper shield have been newly prepared by the above mentioned procedures. Table (III-5) shows the diffusion parameters.

#### 4) Calculation Input

##### i) Removal source calculation input

For the removal source, power distributions of the core calculation result in the core and blanket region have been inputted as an external source, and for the shield regions, the removal sources have been calculated by the removal source calculation code REAC. Fig. (III-2) shows the power distributions of the core and blanket regions, while Table (III-6) shows the total power. Fig. (III-3) shows the removal source calculation points. The gauss integral division points have been taken as  $Z_4 \times R_4 \times Q_4$  in the four source regions (core 1, core 2 and two axial blanket regions), and as  $Z_4 \times R_4 \times Q_6$  in the two radial blanket regions.

ii) Diffusion calculation input

The material distribution for diffusion calculation is shown in Fig. (III-4), and at the same time space meshes are also shown. The space meshes have been determined from the results of one dimensional calculation along both radial and axial center axis. Fig. (III-5) and (III-6) show the comparison between the finely cut mesh and roughly cut mesh respectively. There is seen an error of about ~30%. Finally, a space mesh having the size of 130 maximum along axial direction and 78 maximum to radial direction has been selected considering the capacity of the code as shown in Fig. (III-4).

iii) Input for line of sight calculation in void

a) Void Condition

Line of sight calculation in N<sub>2</sub> gas void layer outside reactor vessel has been done using the option of axial symmetrical condition, and it differs from previous calculation.

b) Gauss integral division point number

16 points have been taken the same as previous calculation.

c) Order of multinomial fitting of the neutron transition probability in voids

The orders of multinomial fitting of probabilities obtained at each diffusion mesh have been selected as follows, the same as previous calculation:

		<u>N<sub>2</sub> Gas Void</u>	<u>Air Gas Void</u>
(1)	Inner Circle ↔ Outer Circle	14	
(2)	"      ↔ Upper(Lower) End	12	
(3)	Outer Circle ↔ "	12	9
(4)	"      ↔ Outer Circle	9	12
(5)	Upper(Lower) ↔ Lower(Upper) End	9	12

The fitting error has been largest in the case of N<sub>2</sub> gas void "Inner circle → Outer circle", and has been as much as 30% at a certain location. This might be because of the extremely long shape to axial direction. But as a whole, it is considered the error will not be too large because the neutron current is calculated as the total sum of the contribution from other area, and because noted neutron streaming will come from the lower part of void which is high neutron flux level. In other cases, the errors have been below ~ 10%. The degree of the errors does not differ between this calculation and previous one.

d) Frequency of line of sight calculation during line iteration

The calculation performed this time has been such a large geometry that it has been assumed that the neutron flux attenuation might be more than 25 figures at the maximum, and in addition, the neutron flux variation would be great due to void effect, which would make convergency extremely difficult and would require

so many repeated calculation iteration (line iteration). There has been a possibility that any recalculation made by line of sight method for incident neutrons at the void surface at each line iteration would invite a larger neutron flux variation, and would not make the calculation convergence. For this reason, this time the following convergency calculation method has been adopted:

<u>Line Iteration No.</u>	<u>Line of Sight Calculation in Void</u>	<u>Energy Group</u>
1	Every time	
5		1st group
	Every 5 times	
50		8th group
	None	
100		
1	None	
30		
	Every time	
35		9th group (thermal)
	Every 10 times	
100		
	None	
150		

In 1 group to 8 group, line of sight calculation has been inserted to every five times of line iteration from the experience of previous calculation that neutron flux around void converges in about five line

iterations. The reason of stopping line of sight calculation in 50 line iterations has been that the neutron flux below void region converges.

In 9 group (thermal), the method of inserting line of sight calculation has differed from above 8 groups.

Line of sight calculation of 9 group has been inserted after neutron flux has converged, because as the neutron flux distributions of those above 8th group are relatively resembled to the removal neutron source distribution or the slowing-down neutron source distribution, those neutron fluxes in the neighbourhood of void surface have converged at the early stage of the line iteration while those of the ninth group seemed not like that. The final convergence accuracy has been 1% and each group needed 120 - 140 times of iterations. Particularly it has been difficult to have the neutron flux convergence inside top shield, and it has been the cause of so many iterations. The reason for this has been assumed to be the large neutron flux attenuation inside upper shield and the relatively rough space meshes.

##### 5) Computing time

For this calculation, it took about one hour by CDC-6600.

### (III-3) Calculation Result

Fig. (III-7) to Fig. (III-33) show the results.

#### 1) Contour distribution of neutron flux

Nine groups neutron flux contour distribution mappings are shown in Fig. (III-7) ~ Fig. (III-15). Three groups neutron flux contour distribution mappings combined into fast, intermediate and thermal neutron groups are shown in Fig. (III-16) ~ Fig. (III-18).

In above 8 groups (fast and intermediate neutrons), neutrons streaming through N<sub>2</sub> gas void outside reactor vessel from bottom to top raise the neutron flux level at both side surface of void and lower part of top shield. From the part of high neutron flux level, neutrons enter into the adjacent Ar gas void above sodium surface, then on the axial center axis, neutron flux level in Ar gas void becomes higher than that in sodium liquid. Neutron flux contour distribution having peak point at the entrance surface of the streaming neutron is newly formed. The decay rate of neutron flux in sodium is largest in energy group 4, so the streaming neutrons give the greatest effect to the neutron flux in sodium.

In 9 group (thermal neutron), streaming neutrons are very little because as the N<sub>2</sub> gas void surface is composed by stainless steel many thermal neutrons are absorbed, so thermal neutron flux distribution is

formed from the slowing down neutrons caused by the streaming neutron distribution above 8 groups. Peak of thermal neutron flux formed at the position near the carbon shield.

Fig. (III-19) shows the neutron fluxes at both  $N_2$  gas void surface on radial center axis and the top of  $N_2$  gas void, and at the same time decay rate of neutron flux along void is shown. It seems that the decay rate becomes smaller as energy becomes lower. Streaming neutrons entering into the top of the void come mostly from the part of high neutron flux at lower side surface of void, while streaming effects are decreased as energy becomes lower because the contribution of slowing down neutron increases. This is considered the spectrum softening seen in Fig. (III-19).

## 2) Decay distribution of neutron flux

In Fig. (III-20) ~ Fig. (III-29), neutron flux decay distributions on the surface parallel to  $r$ ,  $z$  axis are shown comparing between calculation including void and that not including void. On the  $r$ -axis, almost no difference exists between both calculations, and also on the  $z$ -axis there seems only little difference at the neighbourhood of Ar gas void. Decay distribution on the surface parallel to  $r$ -axis, the difference becomes greater from the neighbourhood of  $N_2$  gas void to whole part as it remoted from  $r$ -axis

and energy become lower. This is because the effects of streaming are greater as energy becomes higher. This is same as on the surface parallel to z axis.

3) Comparison between this calculation and before previous one

Fig. (III-30) shows the comparison at the top of N<sub>2</sub> gas void between the calculation including void and that not including void. There seems that the streaming neutrons raise the neutron flux level at the top of N<sub>2</sub> gas void by maximum 5 figures at high energy region and 3 figures at low energy region. However, the increased neutron flux is fully shielded by the top shield and adjacent concrete shield, and it seems that there exists any shield design problem, within the assumption of simplified void geometry.

4) Comparison between this calculation and previous one

Fig. (III-31) shows the comparison between the calculation using axial symmetric void option and the calculation under condition that neutron streams to upper part from the surface of N<sub>2</sub> gas void intersected radial center axis with cosine distribution. Neutron flux is compared at the top of N<sub>2</sub> gas void. It seems that over whole energy groups previous calculation is greater by about 1 figure than this one. Previous calculation evidently overestimates the neutron flux

in long narrow N<sub>2</sub> gas void because the neutron transition probability becomes larger between upper part and lower one.

### 5) Neutron flux around N<sub>2</sub> gas void

As is shown in Fig. (III-32) ~ Fig. (III-33), the neutron flux level at inner surface of N<sub>2</sub> gas void becomes lower than that at outer surface as going to upper part. Table (III-7) represents ratio of neutron flux at inner surface to that of outer surface of 60th mesh (703.7 cm). The ratio becomes 10 to 2 according to high to low energy. The reason is that neutron flux distribution at upper part of void is determined by the neutron streaming from lower part of void which is high flux level. As is shown in Table (III-8), the neutron transition probability from the lower part of the inner void surface to the upper part of void is greater by maximum 9 times than that from lower part of outer void surface, and this comes from the difference between both solid angles. Further, as is shown Table (III-9), neutrons entering into void from inner surface are more by about 10 times at high energy than those from the outer surface. Consequently, at high energy region the difference between neutron currents entering into outer surface and into inner surface is more than ten times and the difference of neutron fluxes is caused. As going to lower energy region, the difference of neutron flux between inner

surface and outer one becomes small because the difference between these neutron currents entering into the void becomes little and because the neutron flux in shields does not decay largely and because the exchange of neutrons between inner surface and outer one increases.

**Table(III-1) Energy Structure of Removal Calculation**

No.	Original Energy Structure (Mev)	No.	Collapsed Energy Structure (mev)
1	18.00-13.74	1	18.00- 8.25
2	13.74-10.50		
3	10.50- 9.30		
4	9.30- 8.25		
5	8.25- 7.32	2	8.25- 6.50
6	7.32- 6.50		
7	6.50- 5.75	3	6.50- 5.10
8	5.75- 5.10		
9	5.10- 4.52	4	5.10- 4.00
10	4.52- 4.00		
11	4.00- 3.55	5	4.00- 2.50
12	3.55- 3.16		
13	3.16- 2.81		
14	2.81- 2.50		
15	2.50- 2.16	6	2.50- 1.40
16	2.16- 1.87		
17	1.87- 1.61		
18	1.61- 1.40		
19	1.40- 1.21	7	1.40- 0.05
20	1.21- 1.05		
21	1.05- 0.92		
22	0.92- 0.80		
23	0.80- 0.57		
24	0.57- 0.40		
25	0.40- 0.05		

Table (III-2) Energy Structure of Diffusion Calculation

Gr.	Energy Range (Mev)	Lethagy	Gr.	Energy Range (Mev)	Lethagy
1	1.05 (+1) - 8.25 (+1)	0.241	1	1.05 (+1) - 6.50 (+0)	0.475
2	8.25 (+1) - 6.50 (+0)	0.234	2	6.50 (+0) - 2.50 (+0)	0.956
3	6.50 (+0) - 5.10 (+0)	0.243	3	2.50 (+0) - 8.00 (-1)	1.130
4	5.10 (+0) - 4.00 (+0)	0.243	4	8.00 (-1) - 4.65 (-2)	2.845
5	4.00 (+0) - 3.16 (+0)	0.235	5	4.65 (-2) - 2.15 (-3)	3.074
6	3.16 (+0) - 2.50 (+0)	0.235	6	2.15 (-3) - 1.00 (-4)	3.073
7	2.50 (+0) - 1.87 (+0)	0.281	7	1.00 (-4) - 4.65 (-6)	3.070
8	1.37 (+0) - 1.40 (+0)	0.290	8	4.65 (-6) - 2.15 (-7)	3.070
9	1.40 (+0) - 1.05 (+0)	0.287			
10	1.05 (+0) - 8.00 (-1)	0.272			
11	8.00 (-1) - 4.00 (-1)	0.693			
12	4.00 (-1) - 2.00 (-1)	0.693			
13	2.00 (-1) - 1.00 (-1)	0.694			
14	1.00 (-1) - 4.65 (-2)	0.765			
15	4.65 (-2) - 2.15 (-2)	0.772			
16	2.15 (-2) - 1.00 (-2)	0.765			
17	1.00 (-2) - 4.65 (-3)	0.771			
18	4.65 (-3) - 2.15 (-3)	0.766			
19	2.15 (-3) - 1.00 (-3)	0.763			
20	1.00 (-3) - 4.65 (-4)	0.770			
21	4.65 (-4) - 2.15 (-4)	0.770			
22	2.15 (-4) - 1.00 (-4)	0.770			
23	1.00 (-4) - 4.65 (-5)	0.770			
24	4.65 (-5) - 2.15 (-5)	0.770			
25	2.15 (-5) - 1.00 (-5)	0.760			
26	1.00 (-5) - 4.65 (-6)	0.770			
27	4.65 (-6) - 2.15 (-6)	0.770			
28	2.15 (-6) - 1.00 (-6)	0.760			
29	1.00 (-6) - 4.65 (-7)	0.770			
30	4.65 (-7) - 2.15 (-7)	0.770			
31	2.15 (-7) - 0.00		9	2.15 (-7) - 0.00	

Table (III-3) (1) Composition of Core, Blanket Materials

Materials Element	Core		Radial Blanket	Axial Blanket
	I	II		
U-235	9.8615 (-6)	1.0816 (-5)	2.0785 (-5)	1.3920 (-5)
U-238	4.9209 (-3)	5.3972 (-3)	1.0372 (-2)	6.9460 (-3)
Pu239	8.6042 (-4)	1.3883 (-3)		
Pu240	22.945 (-4)	3.7021 (-4)		
Pu241	5.7362 (-5)	0.2552 (-5)		
Na	1.0806 (-2)	0.3071 (-3)	7.6948 (-3)	8.7460 (-3)
Fe	1.2160 (-2)	1.2841 (-2)	1.0839 (-2)	1.4770 (-2)
Cr	3.1293 (-3)	3.3048 (-3)	2.7895 (-3)	4.1810 (-3)
Ni	2.2352 (-3)	2.3606 (-3)	1.9925 (-3)	2.2060 (-3)
Mo	3.5763 (-4)	3.7769 (-4)	3.1880 (-4)	
O	1.2156 (-2)	1.4518 (-2)	2.0785 (-2)	1.3890 (-2)

Table (III-3) (2): Composition of Shielding Materials

( $\text{cm}^3/\text{cc.}$ )

Material Element \	Nickel Shield	Graphit Shield	Core Barrel	Gas Plenum	Fuel Element Hold Down Part	Handlings Head	Guard Tube and Thermo couple
Ni	6.2340	0.0925	0.9346	0.2370	3.3320	0.1869	0.0935
Fe	0.5402	0.5428	5.1800	1.5100	2.1100	1.0360	0.5180
Cr	0.1457	0.1465	1.4800	0.3970	0.5950	0.2060	0.1480
Mo	0.0191	0.0192	0.1947			0.0387	0.0195
Na	0.1838	0.1806		0.3140	0.3400	0.0934	0.7834
C		1.1090					
Total	7.0928	2.0606	7.7893	2.4780	4.2670	2.2540	1.5624

Material Element \	Sodium	Dop Plate	Graphite	Thermal Shield	Core Vessel and Thermal Shield	Carbon Steel
Ni		0.5608		0.5607	0.0346	
Fe		3.1030		3.1030	5.1800	2.7790
Cr		0.8881		0.8880	1.4800	
Mo		0.1168		0.1168	0.1947	
Na	0.3705	0.3482		0.3482		
C			1.6000			0.0785
Total	0.8705	5.0219	1.6000	5.0217	7.7893	2.8485

Table (III-3) (3) Composition of Shielding Materials

	121 Fe	122,123 Graphite, Ar Gas Fe	124 $N_2$ Gas, Fe	125 Ar Gas, Fe
Ni	0.4226	0.1797		
Fe	2.3423	0.9962	4.3917	7.0528
Cr	0.7573	0.3221		
Mo				
C		1.2308	0.0044	0.0071

	126 Ordinary Concrete
Fe	0.0620
C	0.0065
O	1.2040
Na	0.0268
H	0.0117
Si	0.8610
S	0.0039
K	0.0194
Ca	0.2000
Mn	0.0060

Table (III-4) (1) Removal Cross-section for "MONJU"

101 Core. 1		
Gr.	(A)	(B)
1	0.08478	0.08478
2	0.08862	0.08862
3	0.09514	0.09514
4	0.10230	0.10230
5	0.11300	
6	0.11640	
7	0.14940	

102 Core2		
Gr.	(A)	(B)
1	0.08478	0.08478
2	0.08862	0.08862
3	0.09514	0.09514
4	0.10230	0.10230
5	0.11300	
6	0.11640	
7	0.14940	

103 Radial Blanket		
Gr.	(A)	(B)
1	0.09490	0.09490
2	0.09941	0.09941
3	0.10680	0.10680
4	0.11480	0.11480
5	0.12870	
6	0.12970	
7	0.18050	

104 Axial Blanket		
Gr.	(A)	(B)
1	0.08471	0.08471
2	0.08853	0.08853
3	0.09504	0.09504
4	0.10220	0.10220
5	0.11300	
6	0.11630	
7	0.14930	

105 Nickel Shield		
Gr.	(A)	(B)
1	0.15680	0.13260
2	0.17080	0.14230
3	0.18030	0.15030
4	0.19340	0.16120
5	0.17060	
6	0.18510	
7	0.20200	

106 Carbon Shield		
Gr.	(A)	(B)
1	0.05798	0.06772
2	0.08031	0.07295
3	0.08807	0.07809
4	0.11380	0.09484
5	0.13640	
6	0.12270	
7	0.15490	

107 Gas Plenum		
Gr.	(A)	(B)
1	0.06270	0.05225
2	0.06515	0.05429
3	0.06986	0.05821
4	0.07518	0.06265
5	0.06664	
6	0.07247	
7	0.07255	

108 Fuel Hold-down Mechanism		
	(A)	(B)
1	0.08133	0.06778
2	0.08466	0.07055
3	0.09087	0.07573
4	0.08755	0.08154
5	0.08665	
6	0.09346	
7	0.09141	

(A) After Adjust  
(B) Before Adjust

Table (III-4) (2) Removal Cross-section for "MONJU"

109 Sodium 1		
Gr.	(A)	(B)
1	0.03380	0.02816
2	0.03499	0.02916
3	0.03811	0.03176
4	0.04088	0.03460
5	0.03556	
6	0.04458	
7	0.05974	

110 Core Barrel		
Gr.	(A)	(B)
1	0.18200	0.15160
2	0.14630	0.15710
3	0.13340	0.16680
4	0.14360	0.17950
5	0.19200	
6	0.20150	
7	0.18180	

111 Handlings Head		
Gr.	(A)	(B)
1	0.05833	0.05193
2	0.05931	0.05384
3	0.06083	0.05771
4	0.06070	0.06210
5	0.06592	
6	0.07457	
7	0.08230	

112 Sodium 2		
Gr.	(A)	(B)
1	0.03380	0.02816
2	0.03499	0.02916
3	0.03811	0.03176
4	0.04088	0.03406
5	0.03556	
6	0.04458	
7	0.05974	

113 Sodium 3		
Gr.	(A)	(B)
1	0.03237	0.02816
2	0.03301	0.02916
3	0.03495	0.03176
4	0.03658	0.03406
5	0.03556	
6	0.04458	
7	0.05973	

114 Sodium 4		
Gr.	(A)	(B)
1	0.03237	0.02816
2	0.03301	0.02916
3	0.03495	0.03176
4	0.03658	0.03406
5	0.03556	
6	0.04458	
7	0.05973	

115 Dip Plate		
Gr.	(A)	(B)
1	0.11640	0.10220
2	0.11630	0.10590
3	0.13530	0.11280
4	0.09707	0.12130
5	0.13000	
6	0.13870	
7	0.13300	

116 Thermal Shield		
Gr.	(A)	(B)
1	0.12270	0.10220
2	0.12710	0.10590
3	0.13530	0.11280
4	0.14560	0.12130
5	0.13000	
6	0.13870	
7	0.13300	

(A) After Adjust

(B) Before Adjust

Table(III-4) (3) Removal Cross-section for "MONJU"

117 Core Vessel		
Gr.	(A)	(B)
1	0.12130	0.15170
2	0.12570	0.15710
3	0.13350	0.16680
4	0.14360	0.17960
5	0.19300	
6	0.20150	
7	0.18180	

118 Graphite		
Gr.	(A)	(B)
1	0.06970	0.06762
2	0.06878	0.07408
3	0.07996	0.07930
4	0.12120	
5	0.15620	
6	0.13480	
7	0.17960	

119 Carbon Steel		
Gr.	(A)	(B)
1	0.15250	0.14710
2	0.12550	0.15690
3	0.20880	0.17400
4	0.15130	0.18910
5	0.19810	
6	0.20560	
7	0.17680	

120 Guid Tube and Thermo Couple		
Gr.	(A)	(B)
1	0.04861	0.04051
2	0.05031	0.04195
3	0.05353	0.04526
4	0.05821	0.04861
5	0.05129	
6	0.06026	
7	0.07194	

(A) After Adjust  
 (B) Before Adjust

Table (III-4) (4) Removal Cross-section for "MONJU" Shield

121 Fe	
Gr.	
1	0.0686
2	0.0710
3	0.0754
4	0.0812
5	0.0873
6	0.0911
7	0.0941

122 Graphite, Ar Gas Fe (1)	
Gr.	
1	0.0812
2	0.0872
3	0.0931
4	0.1120
5	0.1560
6	0.1420
7	0.2030

123 Graphite,Ar Gas Fe (2)	
Gr.	
1	0.0812
2	0.0872
3	0.0931
4	0.1120
5	0.1560
6	0.1420
7	0.2030

124 N <sub>2</sub> Gas, Fe	
Gr.	
1	0.0832
2	0.0887
3	0.0983
4	0.1070
5	0.1120
6	0.1160
7	0.1200

125 Ar Gas,Fe	
Gr.	
1	0.1340
2	0.1420
3	0.1580
4	0.1720
5	0.1870
6	0.1870
7	0.1920

126 Ordinary Concrete	
Gr.	
1	0.0807
2	0.0813
3	0.0875
4	0.0963
5	0.1260
6	0.1370
7	0.2650

Table (III-5) (1) Diffusion Constant of Core and Blanket

101 Core I

Gr.	D.C.	$\Sigma_R$	$\Sigma_{SL}$
1	4.34151400E+00	5.28422570E-02	1.44427370E-02
2	3.29080530E+00	5.11417871E-02	2.99233731E-02
3	2.56049230E+00	4.11801304E-02	2.82841187E-02
4	1.71327300E+00	1.16580028E-02	3.79478048E-03
5	1.17117590E+00	1.71133778E-02	3.72681100E-03
6	9.87421930E-01	3.74824992E-02	8.00886200E-04
7	8.79196080E-01	1.43969511E-01	8.96853070E-05
8	9.15573640E-01	9.00087097E-02	3.51126900E-05
9	1.91600543E-01	9.39315201E-01	0.

102 Core II

Gr.	D.C.	$\Sigma_R$	$\Sigma_{SL}$
1	4.03343250E+00	6.32500624E-02	1.48988560E-02
2	3.05560030E+00	5.96577370E-02	3.16285304E-02
3	2.37379110E+00	4.93574833E-02	3.08423923E-02
4	1.58169040E+00	1.59941386E-02	3.94916381E-03
5	1.08127180E+00	2.34918294E-02	3.60319550E-03
6	9.06313410E-01	5.40532393E-02	8.32760260E-04
7	7.82691510E-01	2.02161530E-01	1.91617740E-04
8	7.81381780E-01	2.32994905E-01	7.46753490E-05
9	1.91600543E-01	9.39315201E-01	0.

103 Radial Blanket

Gr.	D.C.	$\Sigma_R$	$\Sigma_{SL}$
1	3.66649810E+00	6.81142330E-02	1.54966880E-02
2	2.76686130E+00	5.91926863E-02	3.44163682E-02
3	2.06548020E+00	4.65613751E-02	3.75936515E-02
4	1.33074740E+00	8.37901158E-03	5.85415010E-03
5	9.35450390E-01	1.10742136E-02	4.58728740E-03
6	8.77312220E-01	1.56977931E-02	2.43057420E-03
7	8.55837570E-01	2.60347922E-02	8.57823070E-04
8	8.88159250E-01	1.76261634E-02	3.77768760E-04
9	7.80448860E-01	5.53614268E-02	0.

104 Axial Blanket

	D.C.	$\Sigma_R$	$\Sigma_{SL}$
1	3.97363500E+00	5.14545630E-02	1.54472690E-02
2	3.00161140E+00	5.05679339E-02	3.38299332E-02
3	2.34868700E+00	3.68787647E-02	3.10459921E-02
4	1.54968280E+00	7.03434887E-03	5.27241384E-03
5	1.06791370E+00	9.16044686E-03	4.67134590E-03
6	9.17734710E-01	1.26225632E-02	2.59848470E-03
7	8.72060800E-01	1.97228522E-02	8.75877690E-04
8	8.90243940E-01	1.64938592E-02	4.33910680E-04
9	7.90251150E-01	5.74612536E-02	0.

Table(III-5) (2) Diffusion Constant of Shield Region

105 Nickel Shield

Gr.	D.C.	$\Sigma_R$	$\Sigma_{SL}$
1	2.38747058E+00	1.16745337E-01	2.55200000E-02
2	1.99234000E+00	8.98452500E-02	6.10447510E-02
3	1.72897346E+00	1.64599132E-02	1.67061775E-02
4	1.26999581E+00	9.55769814E-03	8.88603782E-03
5	4.07557190E-01	1.95443190E-02	1.80200180E-02
6	3.05623952E-01	1.63637142E-02	1.30700007E-02
7	2.82202824E-01	2.12299374E-02	1.02700001E-02
8	2.74492109E-01	4.45399017E-02	4.35342818E-03
9	2.28333924E-01	2.16104981E-01	0.

106 Carbon Shield

Gr.	D.C.	$\Sigma_R$	$\Sigma_{SL}$
1	4.97508233E+00	8.38697667E-02	4.71200000E-02
2	3.07061243E+00	4.73717190E-02	4.37827622E-02
3	2.28918270E+00	4.15597084E-02	4.16865009E-02
4	1.29274152E+00	2.41400582E-02	2.40716201E-02
5	9.71623360E-01	1.86352036E-02	1.82800006E-02
6	0.52025099E-01	1.66178431E-02	1.60400000E-02
7	9.37031637E-01	1.56133866E-02	1.45300000E-02
8	9.29751785E-01	1.59706589E-02	1.17220461E-02
9	8.81803682E-01	1.85260932E-02	0.

107 Gas Plenum

Gr.	D.C.	$\Sigma_R$	$\Sigma_{SL}$
1	6.20847211E+00	4.37507863E-02	1.33500000E-02
2	5.12053374E+00	3.16896153E-02	2.53032446E-02
3	4.70413520E+00	1.26434920E-02	1.29876798E-02
4	3.29913121E+00	4.49294783E-03	4.34710533E-03
5	2.24072822E+00	6.25570993E-03	5.50200194E-03
6	1.32567185E+00	5.21979666E-03	3.73300007E-03
7	1.22763955E+00	5.72295956E-03	3.00700000E-03
8	1.20394801E+00	1.05038786E-02	1.37651274E-03
9	1.01521201E+00	4.59207150E-02	0.

108 Fuel Hold-down Mechanism

Gr.	D.C.	$\Sigma_R$	$\Sigma_{SL}$
1	4.78371801E+00	5.67437432E-02	1.66100000E-02
2	3.91973355E+00	4.20132600E-02	3.40800288E-02
3	3.73992607E+00	1.70823001E-02	1.70281610E-02
4	2.58667041E+00	5.91835961E-03	5.71533857E-03
5	1.77748564E+00	7.35281527E-03	6.33200042E-03
6	9.87016629E-01	7.25535041E-03	5.29600001E-03
7	9.07871096E-01	7.94335029E-03	4.24700000E-03
8	8.91215589E-01	1.44041293E-02	1.06665538E-03
9	7.53706911E-01	6.59595974E-02	0.

Table III-5) (3) Diffusion Constant of Shield Region

109              Sodium 1

Gr.	D.C.	$\Sigma_R$	$\Sigma_{SL}$
1	1.15252049E+01	2.41373468E-02	1.18600000E-02
2	9.65392945E+00	1.54209104E-02	1.36195407E-02
3	6.28270084E+00	1.29068509E-02	1.30967404E-02
4	4.01357230E+00	5.53060085E-03	5.51401108E-03
5	2.82168396E+00	7.30570694E-03	7.22900001E-03
6	4.44869886E+00	2.16295238E-03	2.00500000E-03
7	4.91194610E+00	2.02456329E-03	1.61900000E-03
8	4.59936778E+00	3.28852880E-03	1.39930500E-03
9	3.89004144E+00	8.30051767E-03	0.

110              Core Barrel

Gr.	D.C.	$\Sigma_R$	$\Sigma_{SL}$
1	2.14509375E+00	1.25331839E-01	3.19100000E-02
2	1.79097699E+00	9.43071508E-02	7.53789173E-02
3	1.81602760E+00	2.50911485E-02	2.63204249E-02
4	1.31663510E+00	9.91425275E-03	9.40621387E-03
5	8.91632821E-01	4.5694462E-03	1.80200594E-03
6	4.18987456E-01	1.7067318E-02	1.18400001E-02
7	3.78995126E-01	1.7622149E-02	8.17600001E-03
8	3.73241357E-01	3.3765613E-02	3.70324655E-03
9	3.12810154E-01	1.63477156E-01	0.

111              Handlings Head

Gr.	D.C.	$\Sigma_R$	$\Sigma_{SL}$
1	6.25638900E+00	4.34597249E-02	1.53500000E-02
2	5.14598976E+00	3.15565159E-02	2.61744495E-02
3	4.21539273E+00	1.85655366E-02	1.85534391E-02
4	2.89538178E+00	6.37214397E-03	6.25479661E-03
5	1.97318086E+00	7.25993443E-03	6.67600012E-03
6	1.50556721E+00	5.23132050E-03	4.05900000E-03
7	1.46131019E+00	5.48535677E-03	3.22200000E-03
8	1.42281471E+00	9.36407076E-03	1.50443199E-03
9	1.19473237E+00	4.00751984E-02	0.

112              Sodium 2

Gr.	D.C.	$\Sigma_R$	$\Sigma_{SL}$
1	1.15441121E+01	2.39344755E-02	1.16400000E-02
2	9.61498549E+00	1.55153032E-02	1.36952260E-02
3	6.39273032E+00	1.17435036E-02	1.19787631E-02
4	4.03874602E+00	5.22701397E-03	5.21000847E-03
5	2.90866657E+00	5.47713496E-03	5.41300001E-03
6	4.73168910E+00	2.70441093E-03	2.55600000E-03
7	4.91350824E+00	2.03713691E-03	1.63400000E-03
8	4.61573935E+00	2.76733346E-03	1.01349553E-03
9	3.89004144E+00	8.30051767E-03	0.

Table (III-5) (4) Diffusion Constant of Shield Region

119                    Sodium 3

Gr.	D.C.	$\Sigma_R$	$\Sigma_{SL}$
1	1.15473800E+01	2.38407688E-02	1.16000000E-02
2	9.50711257E+00	1.65017711E-02	1.43534125E-02
3	6.41049797E+00	1.14668375E-02	1.16761848E-02
4	3.91433686E+00	7.43997730E-03	7.42100028E-03
5	2.84226529E+00	6.16659366E-03	6.09700000E-03
6	4.98574789E+00	4.73108706E-03	4.58100000E-03
7	4.88235694E+00	3.25394664E-03	2.77900000E-03
8	4.58444559E+00	3.69470716E-03	1.67376269E-03
9	3.89004144E+00	8.30051757E-03	0.

114                    Sodium 4

Gr.	D.C.	$\Sigma_R$	$\Sigma_{SL}$
1	1.15593923E+01	2.36801124E-02	1.14500000E-02
2	9.53490127E+00	1.66747476E-02	1.51242955E-02
3	6.43780705E+00	1.11113593E-02	1.13260372E-02
4	3.90130125E+00	7.64200108E-03	7.62300155E-03
5	2.80911691E+00	6.52747426E-03	6.45500000E-03
6	5.04771401E+00	6.29127457E-03	6.13500000E-03
7	4.85632007E+00	4.53335091E-03	3.99600000E-03
8	4.56178951E+00	4.44369691E-03	2.23291792E-03
9	3.89004144E+00	8.30051757E-03	0.

115                    Dip Plate

Gr.	D.C.	$\Sigma_R$	$\Sigma_{SL}$
1	3.19248051E+00	8.38123083E-02	2.28000000E-02
2	2.62317037E+00	6.35396732E-02	5.85084136E-02
3	2.53168118E+00	2.06357153E-02	2.11148029E-02
4	1.71806681E+00	1.12454171E-02	1.09127687E-02
5	1.08488620E+00	1.30136456E-02	1.12700004E-02
6	6.05977838E-01	2.89275505E-02	2.55400000E-02
7	6.00433243E-01	2.43580248E-02	1.67600000E-02
8	5.86417635E-01	3.23505551E-02	6.52424906E-03
9	4.94817017E-01	1.04407749E-01	0.

116                    Thermal Shield and Vessel

Gr.	D.C.	$\Sigma_R$	$\Sigma_{SL}$
1	3.18708002E+00	8.42733632E-02	2.32900000E-02
2	2.62087097E+00	6.30319297E-02	5.1103841E-02
3	2.52383784E+00	1.99147573E-02	2.06281511E-02
4	1.75994794E+00	9.26280943E-03	8.94143046E-03
5	1.11556680E+00	1.22381278E-02	1.05200017E-02
6	6.25547423E-01	1.679069320E-02	1.47700000E-02
7	6.03707294E-01	1.44113144E-02	8.09400000E-03
8	5.88881359E-01	2.53657683E-02	3.65990977E-03
9	4.94833005E-01	1.04404559E-01	0.

Table (III-5) (5)

Diffusion Constant of Shield Region

## 117 Reactor Vessel

Gr.	D.C.	$\Sigma_R$	$\Sigma_{SL}$
1	2.14914958E+00	1.24589214E-01	3.10700000E-02
2	1.76346877E+00	9.49156726E-02	7.70376094E-02
3	1.82884111E+00	2.60063498E-02	2.62773287E-02
4	1.28086736E+00	1.19525266E-02	1.14275932E-02
5	8.00231794E-01	1.14910929E-02	8.61300146E-03
6	4.01307956E-01	2.68418757E-02	2.15700000E-02
7	3.78984736E-01	2.35220532E-02	1.30800000E-02
8	3.72496761E-01	3.88782623E-02	4.71863068E-03
9	3.12761575E-01	1.68526130E-01	0.

## 118 Graphite

Gr.	D.C.	$\Sigma_R$	$\Sigma_{SL}$
1	5.07131099E+00	9.71886482E-02	6.12100000E-02
2	3.07837207E+00	3.77704635E-02	3.49194028E-02
3	2.07970527E+00	4.00325854E-02	4.01760649E-02
4	1.07791991E+00	3.23138954E-02	3.23000000E-02
5	9.17633811E-01	3.50511767E-02	3.50500000E-02
6	9.08217738E-01	3.11205685E-02	3.11200000E-02
7	9.07551687E-01	2.87365252E-02	2.87100000E-02
8	9.07442894E-01	2.81019784E-02	2.80550512E-02
9	9.06903419E-01	2.94331045E-04	0.

## 119 Carbon Steel

Gr.	D.C.	$\Sigma_R$	$\Sigma_{SL}$
1	2.182891649E+00	1.29125073E-01	3.48000000E-02
2	1.77626315E+00	9.32235874E-02	8.90901204E-02
3	1.80222272E+00	2.93862924E-02	3.83565424E-02
4	1.32074609E+00	7.54363909E-03	7.21359936E-03
5	1.12541840E+00	1.32337247E-02	1.12201415E-02
6	3.82109352E-01	2.45613852E-02	2.13500019E-02
7	3.46576930E-01	2.97350483E-02	2.06600000E-02
8	3.42217006E-01	5.51223859E-02	1.60562708E-02
9	3.02098843E-01	3.26603748E-01	0.

## 120 Guid Tube and Thermo Couple

Gr.	D.C.	$\Sigma_R$	$\Sigma_{SL}$
1	8.02774062E+00	3.40445703E-02	1.36700000E-02
2	6.60642142E+00	2.42794530E-02	2.05109630E-02
3	4.90093922E+00	1.75199550E-02	1.75541653E-02
4	3.29580574E+00	6.54964325E-03	6.48118114E-03
5	2.29725103E+00	6.48162283E-03	6.15600002E-03
6	2.24342799E+00	4.34292801E-03	3.69400000E-03
7	2.23565541E+00	4.06529601E-03	2.70000000E-03
8	2.15742167E+00	6.42984678E-03	1.40314005E-03
9	1.81478996E+00	2.43177048E-02	0.

Table (III-5) (6) Diffusion Constant of Shield Region

121 Fe			
Gr.	D.C.	$\Sigma_R$	$\Sigma_{SL}$
1	.4775E+.1	.5562E-01	.1328E-01
2	.3964E+01	.4254E-01	.5104E-01
3	.4057E+01	.1180E-01	.1242E-01
4	.2867E+01	.5217E-02	.4985E-02
5	.1724E+01	.5417E-02	.4103E-02
6	.3467E+00	.1955E-01	.1705E-01
7	.8383E+00	.1714E-01	.1154E-01
8	.3213E+00	.2149E-01	.3553E-02
9	.6913E+00	.7832E-01	0.

122 Graphite , Ar Gas , Fe (1)			
Gr.	D.C.	$\Sigma_R$	$\Sigma_{SL}$
1	.4151E+01	.1029E+00	.5735E-01
2	.2913E+01	.4437E-01	.4098E-01
3	.1897E+01	.7887E-01	.7916E-01
4	.1152E+01	.2652E-01	.2645E-01
5	.8503E+00	.2605E-01	.2541E-01
6	.7614E+00	.2531E-01	.2427E-01
7	.7380E+00	.4576E-01	.4332E-01
8	.7300E+00	.4153E-01	.3148E-01
9	.6835E+00	.3346E-01	0.

123 Graphite , Ar Gas , Fe (2)			
Gr.	D.C.	$\Sigma_R$	$\Sigma_{SL}$
1	.4147E+01	.1014E+00	.5583E-01
2	.2893E+01	.4455E-01	.4239E-01
3	.2120E+01	.3013E-01	.3295E-01
4	.1217E+01	.1821E-01	.1812E-01
5	.3701E+00	.2042E-01	.1982E-01
6	.7655E+00	.2131E-01	.2025E-01
7	.7387E+00	.2263E-01	.2061E-01
8	.7317E+00	.2642E-01	.1849E-01
9	.6835E+00	.3346E-01	0.

Table (III-5) (7) Diffusion Constant of Shield Region

124 N <sub>2</sub> Gas , Fe			
Gr.	D.C.	$\Sigma_R$	$\Sigma_{SL}$
1	.3855E+01	.7313E-01	.1984E-01
2	.3136E+01	.5279E-01	.7132E-01
3	.3243E+01	.1755E-01	.2210E-01
4	.2342E+01	.4079E-02	.3863E-02
5	.2117E+01	.6034E-02	.4910E-02
6	.7124E+00	.9985E-02	.7868E-02
7	.6129E+00	.1182E-01	.7350E-02
8	.6166E+00	.2176E-01	.4354E-02
9	.5345E+00	.8488E-01	0.

125 Ar Gas, Fe			
Gr.	D.C.	$\Sigma_R$	$\Sigma_{SL}$
1	.2416E+01	.1166E+00	.3093E-01
2	.1931E+01	.9563E-01	.1502E+00
3	.2131E+01	.3200E-01	.3304E-01
4	.1427E+01	.7803E-02	.7382E-02
5	.1261E+01	.9324E-02	.7553E-02
6	.4543E+00	.1383E-01	.1020E-01
7	.3816E+00	.1633E-01	.9539E-02
8	.3780E+00	.2947E-01	.4838E-02
9	.3328E+00	.1363E+00	0.

126 Ordinary Concrete			
Gr.	D.C.	$\Sigma_R$	$\Sigma_{SL}$
1	.4623E+01	.3671E-01	.2277E-01
2	.3636E+01	.2225E-01	.1866E-01
3	.2678E+01	.2545E-01	.2712E-01
4	.1379E+01	.2951E-01	.2911E-01
5	.1287E+01	.4878E-01	.4844E-01
6	.1201E+01	.5173E-01	.5134E-01
7	.1223E+01	.5266E-01	.5294E-01
8	.1219E+01	.5364E-01	.5375E-01
9	.4619E+00	.7132E-02	0.

Table (III-6) Power of "MONJU"

Name of Region	Power(Mw)
Core I	391.40
Core II	296.40
Radial Blanket (Upper half)	14.79
Axial Blanket (Upper half)	5.67

Table (III-7) Neutron Flux Ratio between Inside and Outside of Void

Z=703.76cm(57mesh)

Energy Group	Outside Flux
	Inside Flux
1	17.74
2	6.31
3	1.91
4	1.69
5	1.70
6	1.58
7	1.57
8	1.56
9	1.66

Table(III-8) Neutron Transition Probability Inside N<sub>2</sub> Gas Void

$$\text{Inner Side} \leftrightarrow \text{Outer Side} \quad P_i = \frac{F_i P_{i \rightarrow o}}{F_i F_o}$$

$$\text{Outer Side} \leftrightarrow \text{Outer Side} \quad P_o = \frac{F_o P_{o \rightarrow o}}{F_o F_o}$$

$$J_{(i \rightarrow o)} = P_i J_i F_i$$

$$J_{(o \rightarrow o)} = P_o J_o F_o$$

P <sub>i</sub> or P <sub>o</sub>						
Mesh No.	1		10		20	
	Inner Side ↑ Outer Side	Outer Side ↑ Outer Side	Inner Side ↑ Outer Side	Outer Side ↑ Outer Side	Inner Side ↑ Outer Side	Inner Side ↑ Outer Side
1	1.67x10 <sup>-5</sup>	3.38x10 <sup>-7</sup>				
10			8.29x10 <sup>-6</sup>	2.51x10 <sup>-7</sup>		
20	1.13x10 <sup>-7</sup>	1.02x10 <sup>-7</sup>	2.50x10 <sup>-7</sup>	1.13x10 <sup>-7</sup>	7.51x10 <sup>-6</sup>	1.86x10 <sup>-7</sup>
40	2.79x10 <sup>-10</sup>	9.24x10 <sup>-9</sup>	2.28x10 <sup>-8</sup>	1.02x10 <sup>-9</sup>	5.47x10 <sup>-9</sup>	1.52x10 <sup>-8</sup>
60	1.53x10 <sup>-10</sup>	1.02x10 <sup>-9</sup>	1.53x10 <sup>-10</sup>	8.38x10 <sup>-10</sup>	2.07x10 <sup>-10</sup>	1.25x10 <sup>-9</sup>
80	6.22x10 <sup>-11</sup>	4.16x10 <sup>-10</sup>	6.22x10 <sup>-11</sup>	4.60x10 <sup>-10</sup>	7.60x10 <sup>-11</sup>	5.08x10 <sup>-10</sup>

Table(III-9) Neutron Current Around N<sub>2</sub> Gas Void (1 Group)

Axial Direction (cm)	Mesh Num.	Number	Upper	Middle	J <sup>+</sup>
			Lower	Neutron Flux J <sup>-</sup>	J <sup>+</sup>
962.00	82	-1.31x10 <sup>-1</sup> 5.22x10 <sup>0</sup> 2.74x10 <sup>0</sup>	-3.0x10 <sup>-1</sup> 1.04x10 <sup>1</sup> 5.55x10 <sup>0</sup>	-2.48x10 <sup>-1</sup> 1.40x10 <sup>1</sup> 9.48x10 <sup>0</sup>	0.0 1.01x10 <sup>0</sup> -5.05x10 <sup>-2</sup>
917.00	73	0.0 1.53x10 <sup>-3</sup> 1.53x10 <sup>0</sup>			5.08x10 <sup>-1</sup> 1.51x10 <sup>-1</sup> -2.95x10 <sup>-2</sup>
600.00	51	3.05x10 <sup>-2</sup> 6.64x10 <sup>-2</sup> 2.70x10 <sup>-3</sup>			3.16x10 <sup>0</sup> 3.96x10 <sup>0</sup> -1.80x10 <sup>-1</sup>
392.29	40	4.75x10 <sup>1</sup> 2.32x10 <sup>0</sup> 6.85x10 <sup>-1</sup>			1.92x10 <sup>1</sup> 3.64x10 <sup>1</sup> -1.00x10 <sup>0</sup>
204.75	28	7.56x10 <sup>1</sup> 7.70x10 <sup>2</sup> 3.09x10 <sup>2</sup>			6.55x10 <sup>2</sup> 1.36x10 <sup>3</sup> 2.50x10 <sup>1</sup>
15.00	4	1.21x10 <sup>3</sup> 6.20x10 <sup>4</sup> 2.78x10 <sup>0</sup>			2.48x10 <sup>4</sup> 5.44x10 <sup>3</sup> 2.40x10 <sup>0</sup>
0.00	1	9.64x10 <sup>4</sup> 6.49x10 <sup>4</sup> 2.25x10 <sup>0</sup>			2.52x10 <sup>4</sup> 4.05x10 <sup>3</sup> 2.04x10 <sup>0</sup>

325.50      332.90      342.30

Radial Direction(cm) →

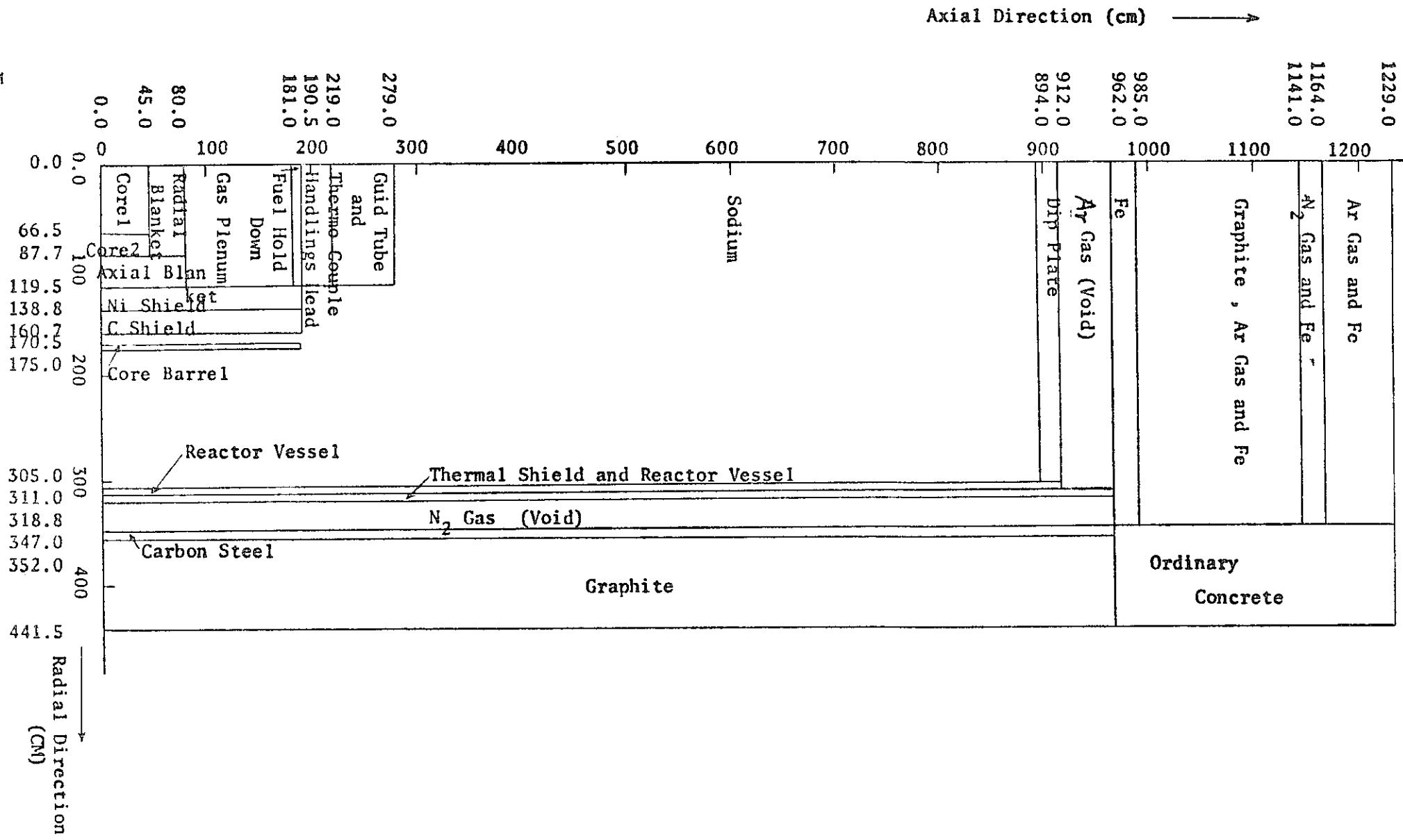
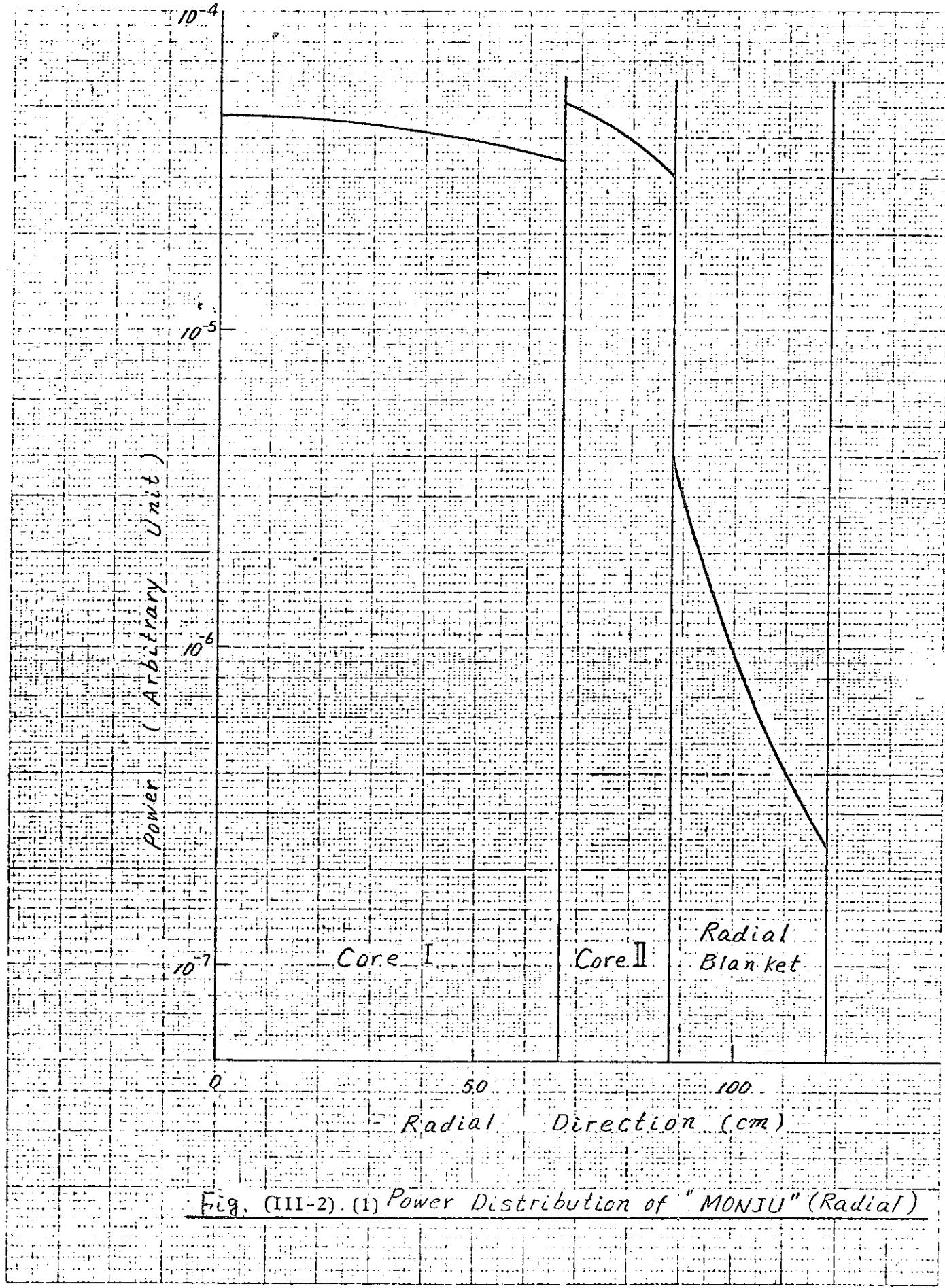


Fig. (III-1) "MONJU" Main Shield Configuration



Power Arbitrary Unit

$10^{-2}$

$10^{-3}$

$10^{-4}$

Core

Axial  
Blanket

0

50

100

Axial Direction (cm)

Fig. (III-2) (2) Power Distribution of "MONJU" (Axial)

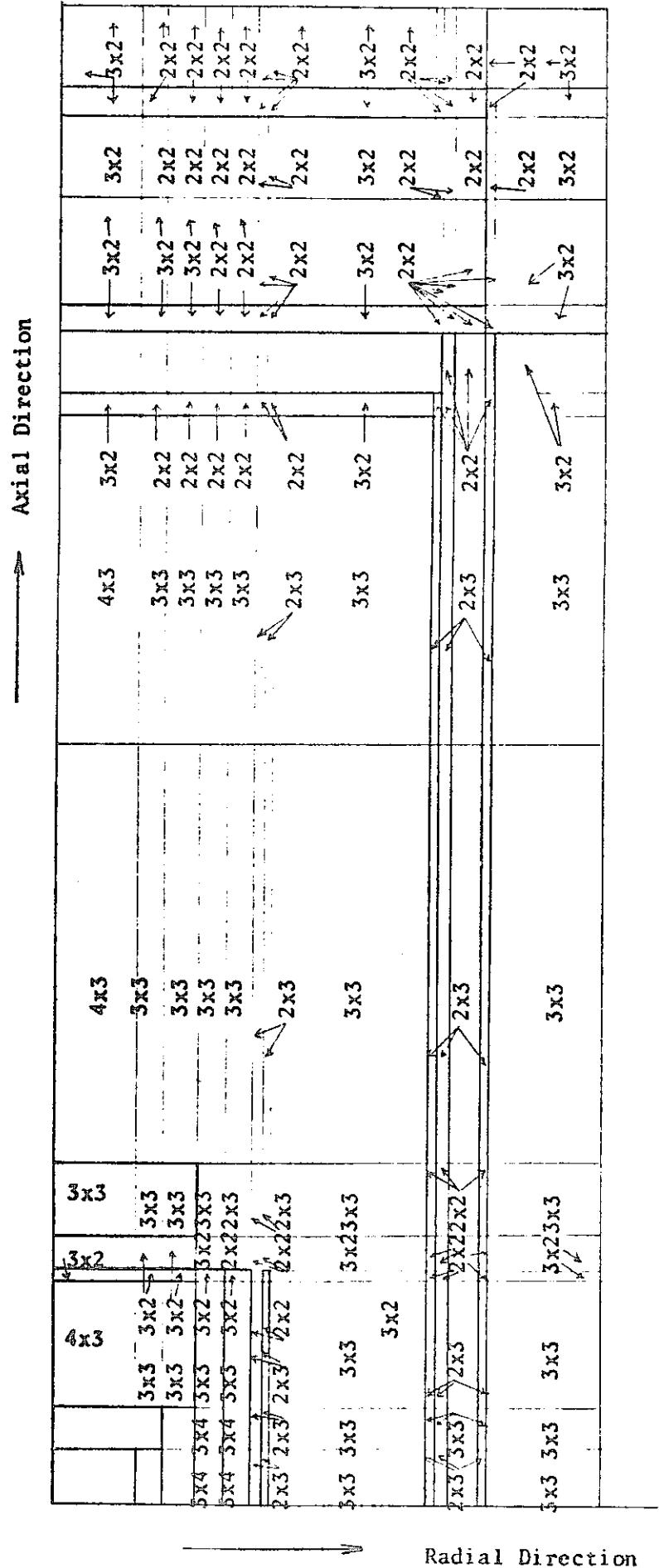
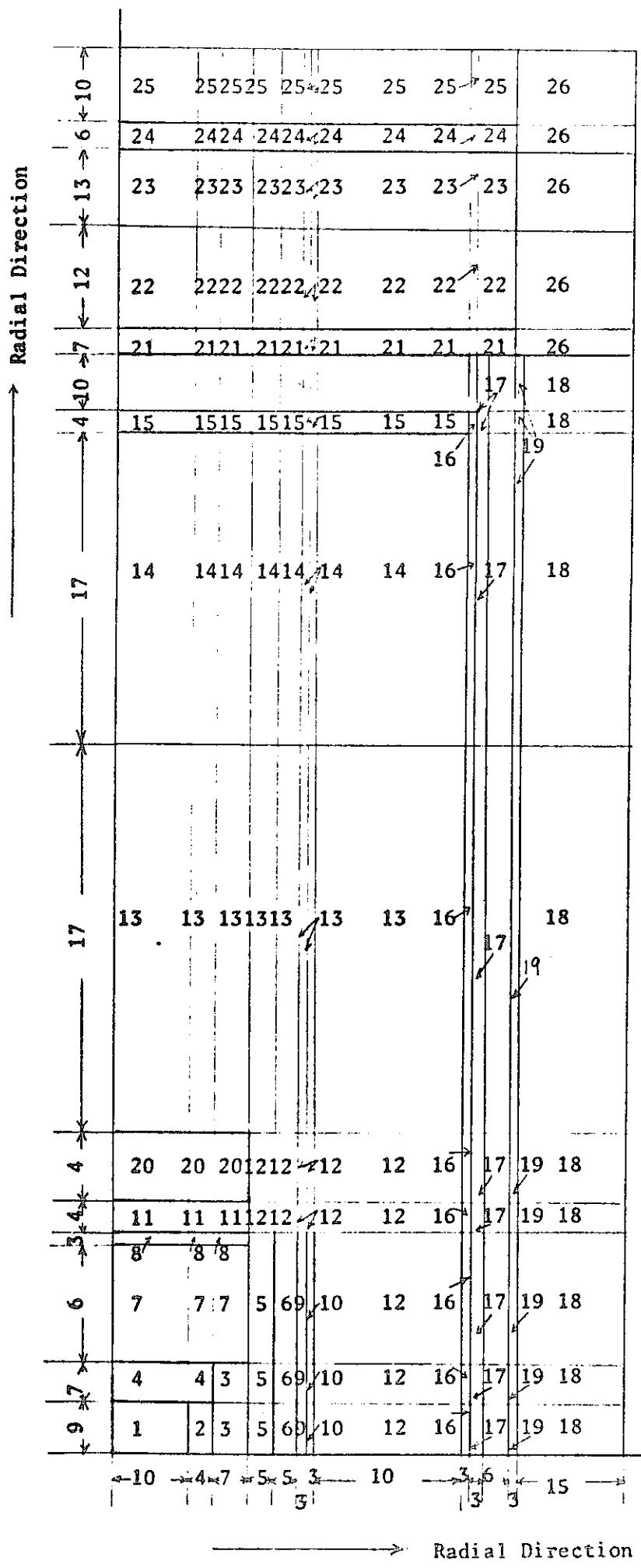


Fig. (III-3) Calculation Point of Removal Sources



### Fig. (III-4) Input of Diffusion Calculation

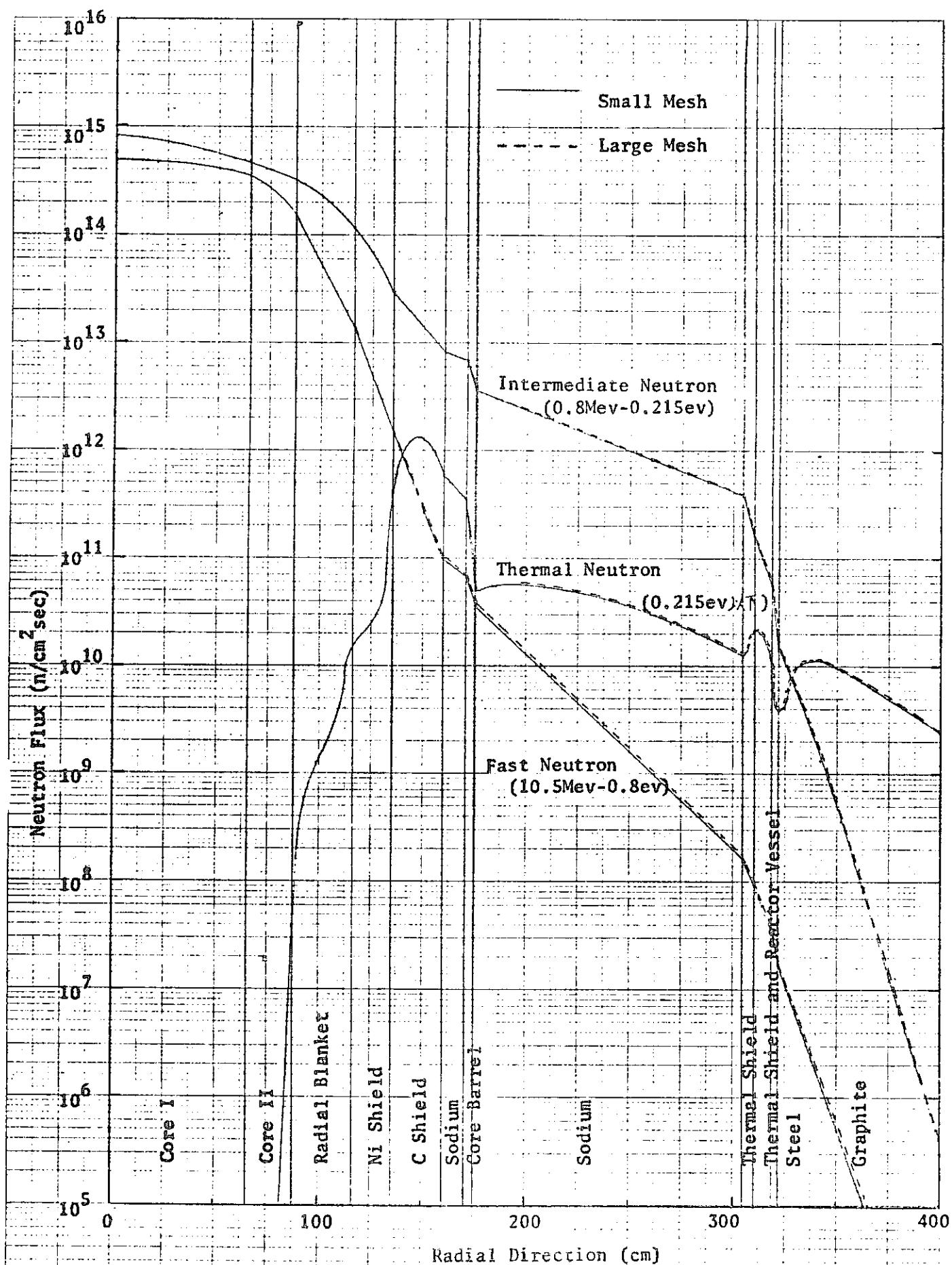


Fig. (III-5) Mesh Test Result ( $r$ -Direction)

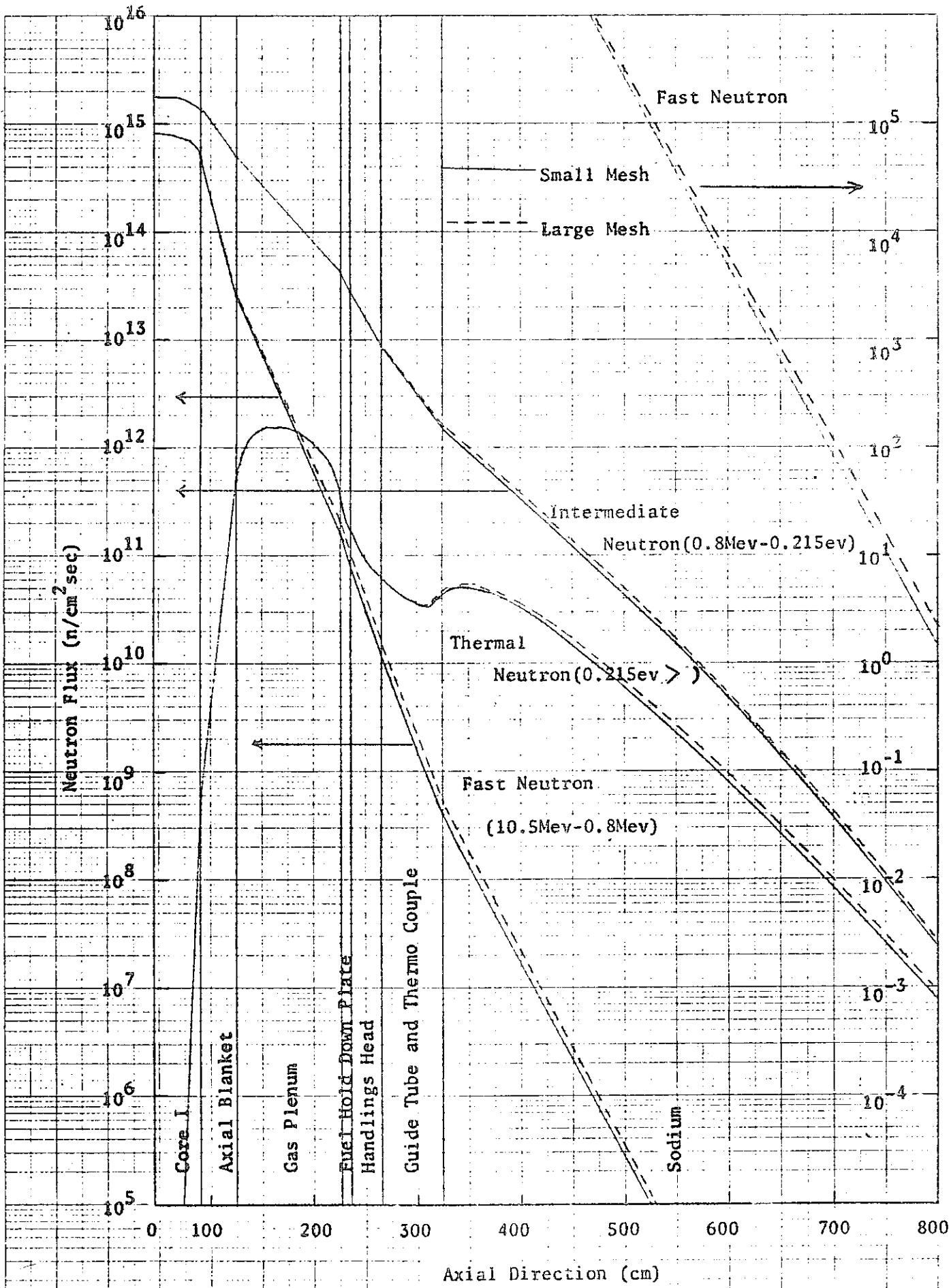


Fig. (III-6) Mesh Test Result (Z-Direction)(1)

A4 12段

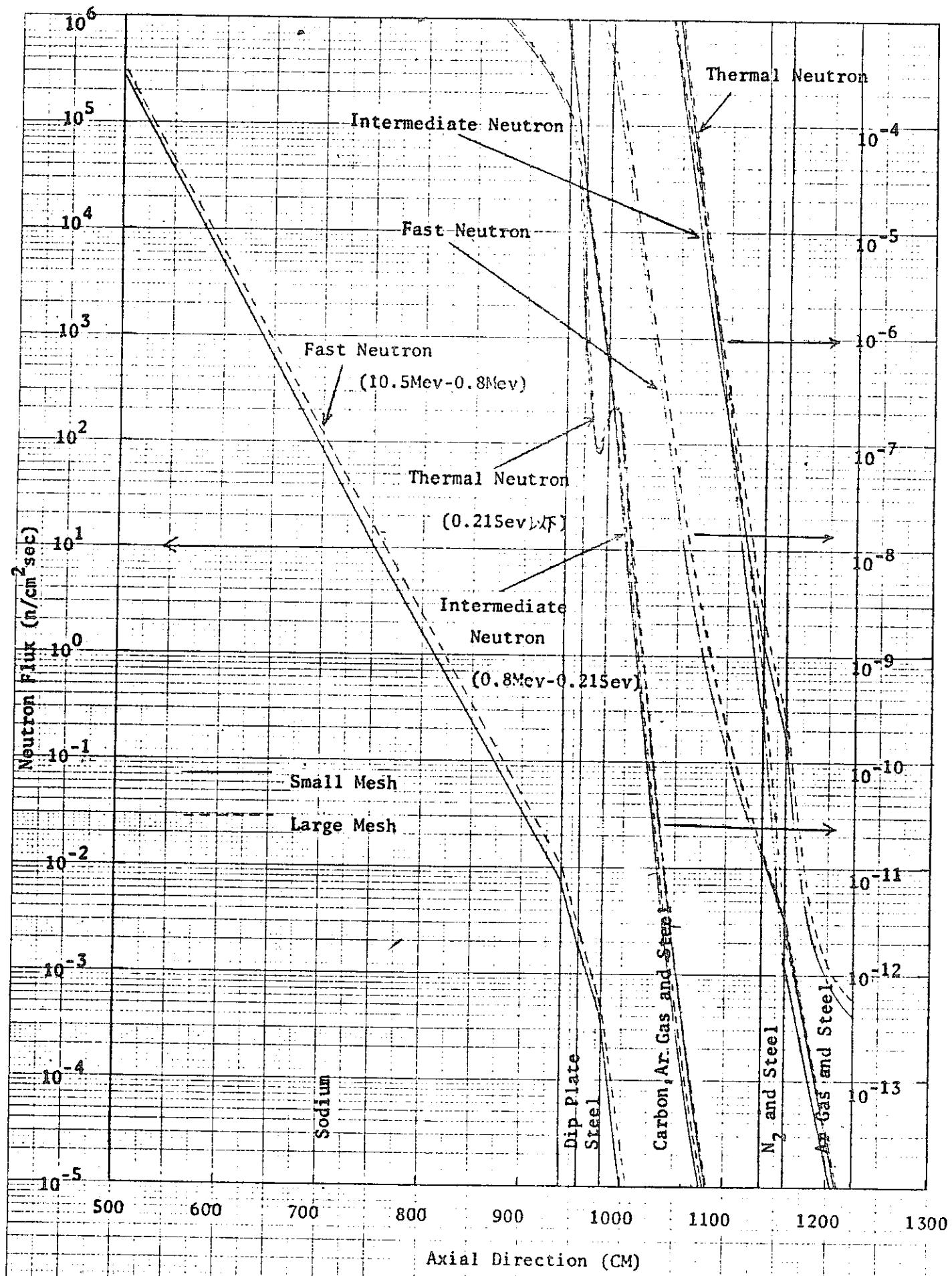


Fig. (III-6). Mesh Test Result (Z-Direction) (2) EISI RA

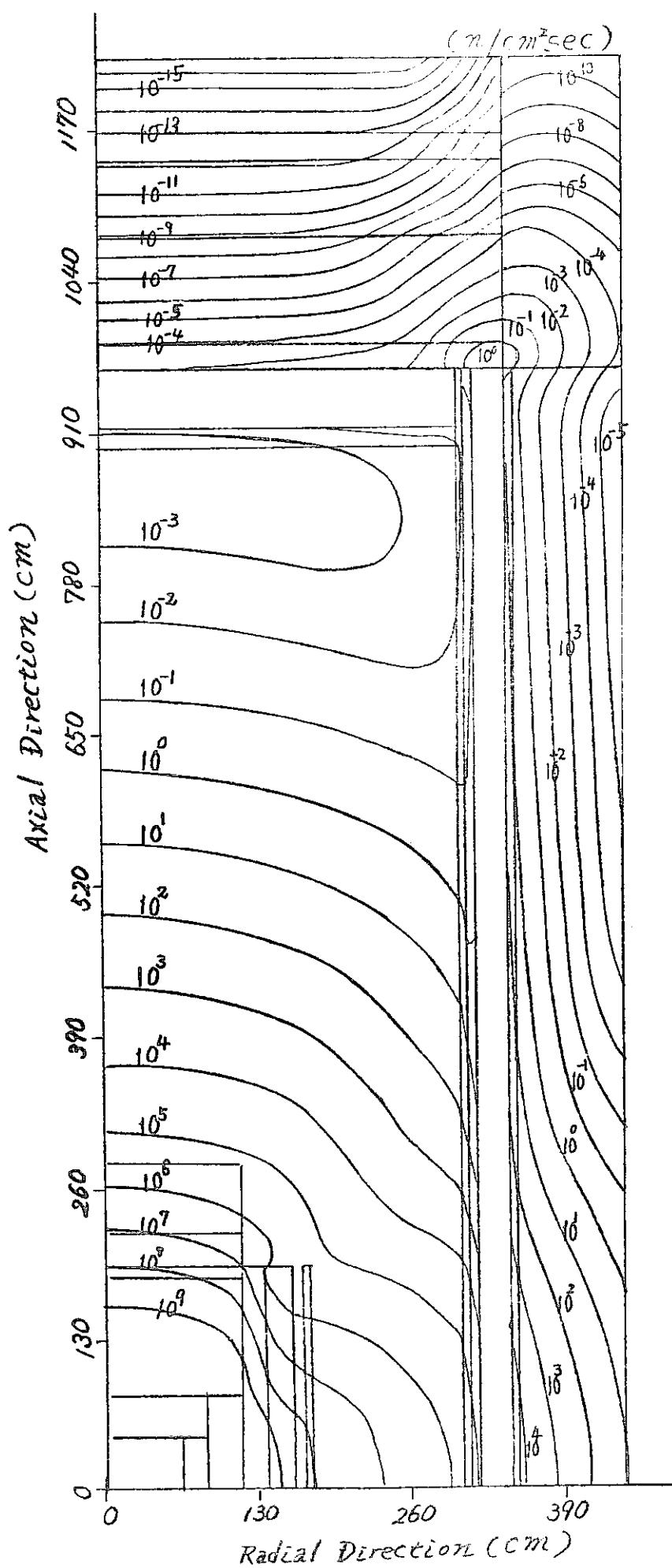
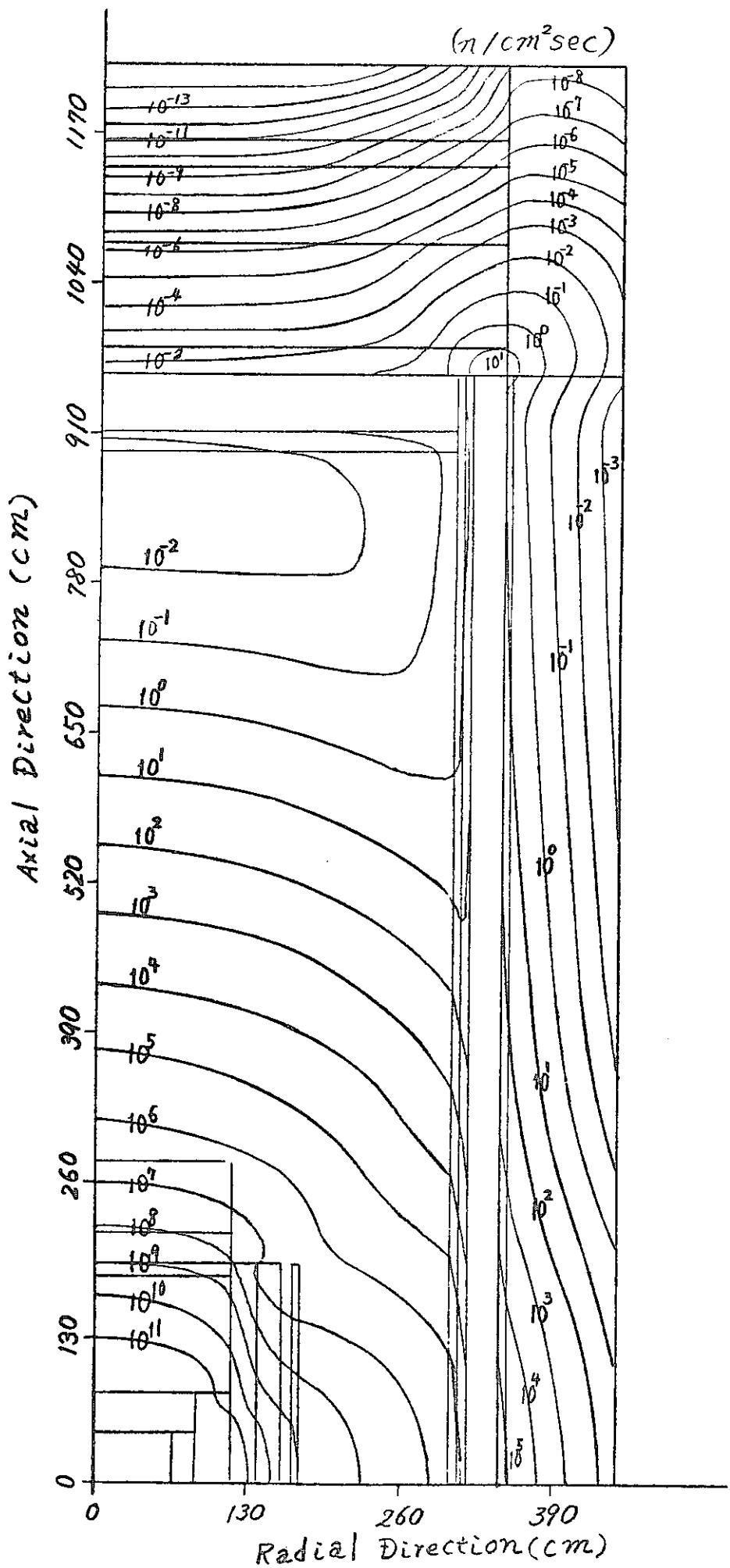


Fig. (III-7) Neutron Contour Mapping | Group (10.5 MeV - 6.5 MeV)



Fig(III-8) Neutron Contour Mapping 2 Group (6.5 Mev-25 Mev)

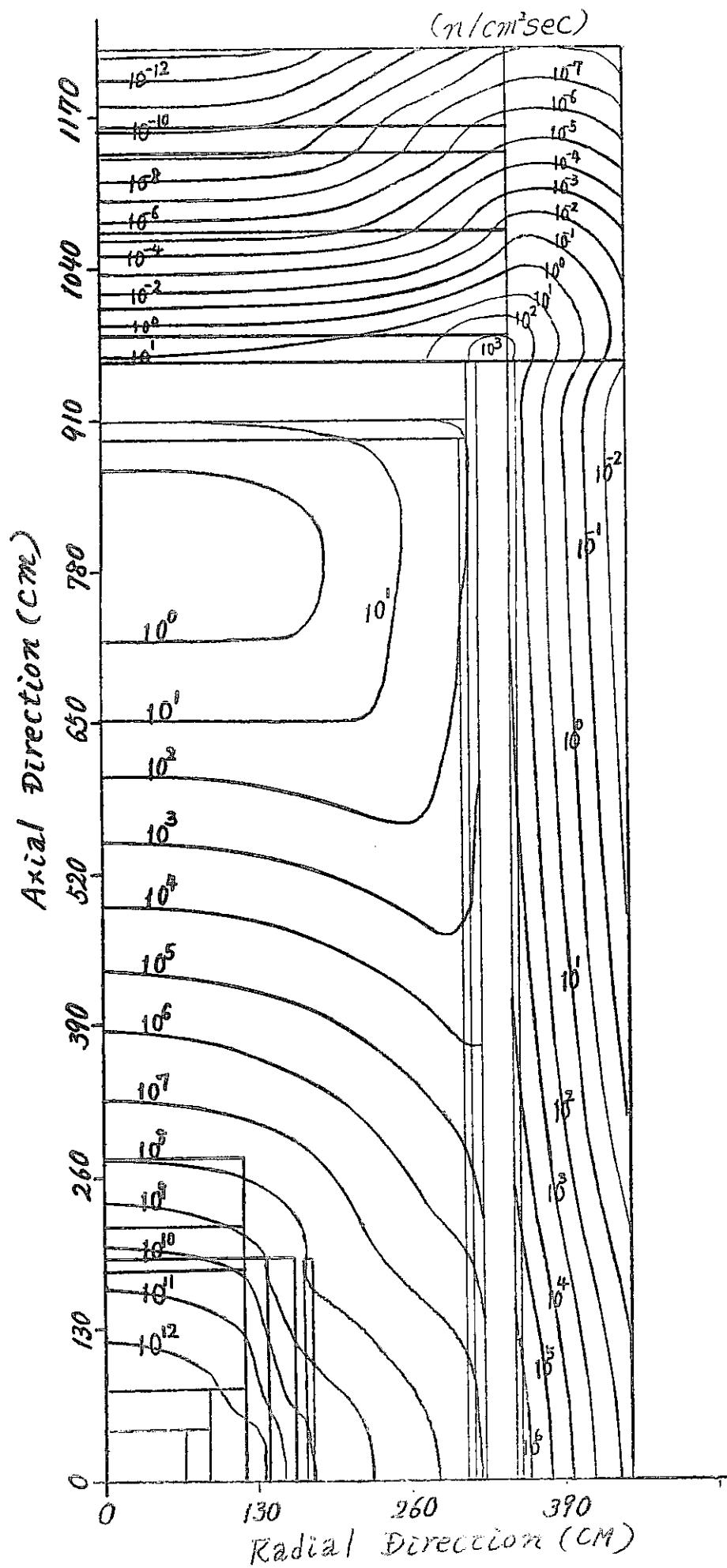


Fig.(III-9) Neutron Contour Mapping 3 Group (2.5 Mev-0.8 Mev)

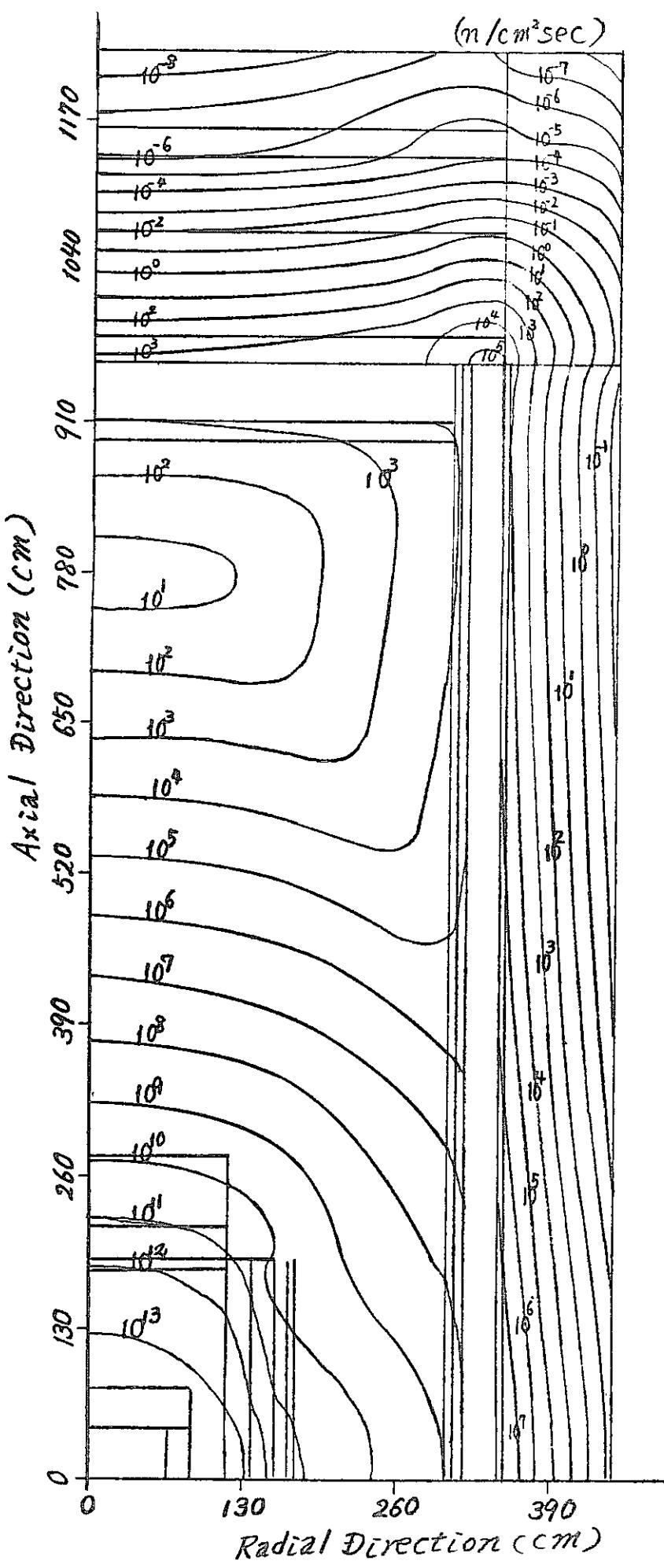


Fig. (III-10) Neutron Contour Mapping 4 Group (0.8 Mev - 6.5 Kev)

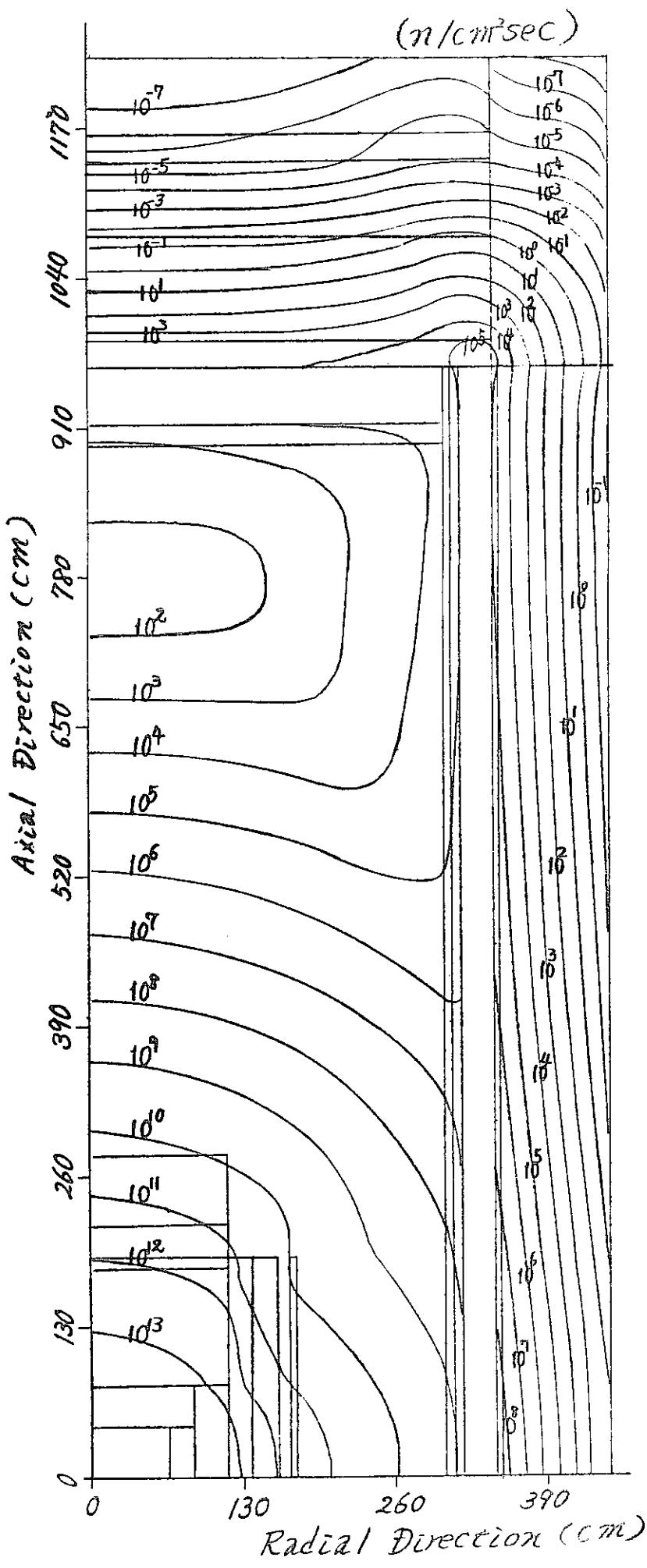


Fig.(III-11) Neutron Contour Mapping 5 Group (46.5 KeV - 2.15 KeV)

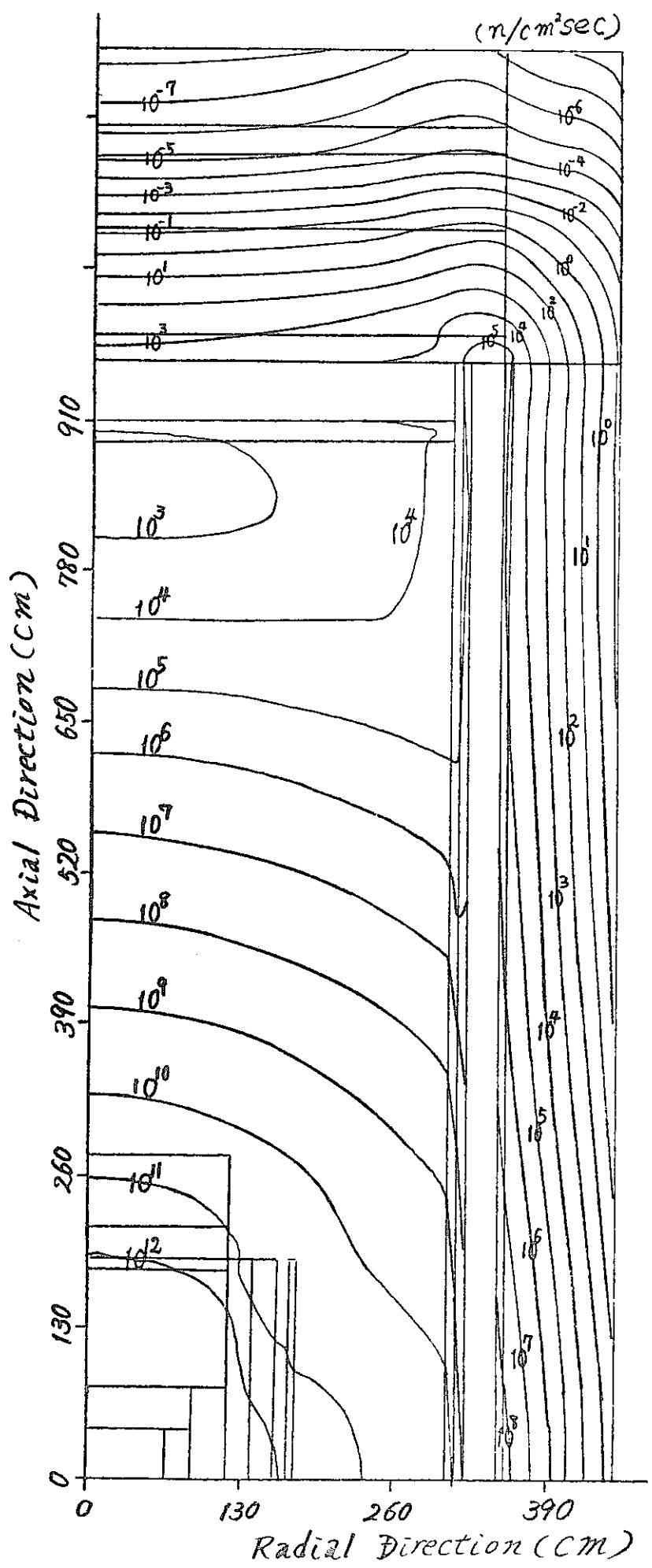


Fig.(III-12) Neutron Contour Mapping 6 Group (2.15Kev-0.1Kev)

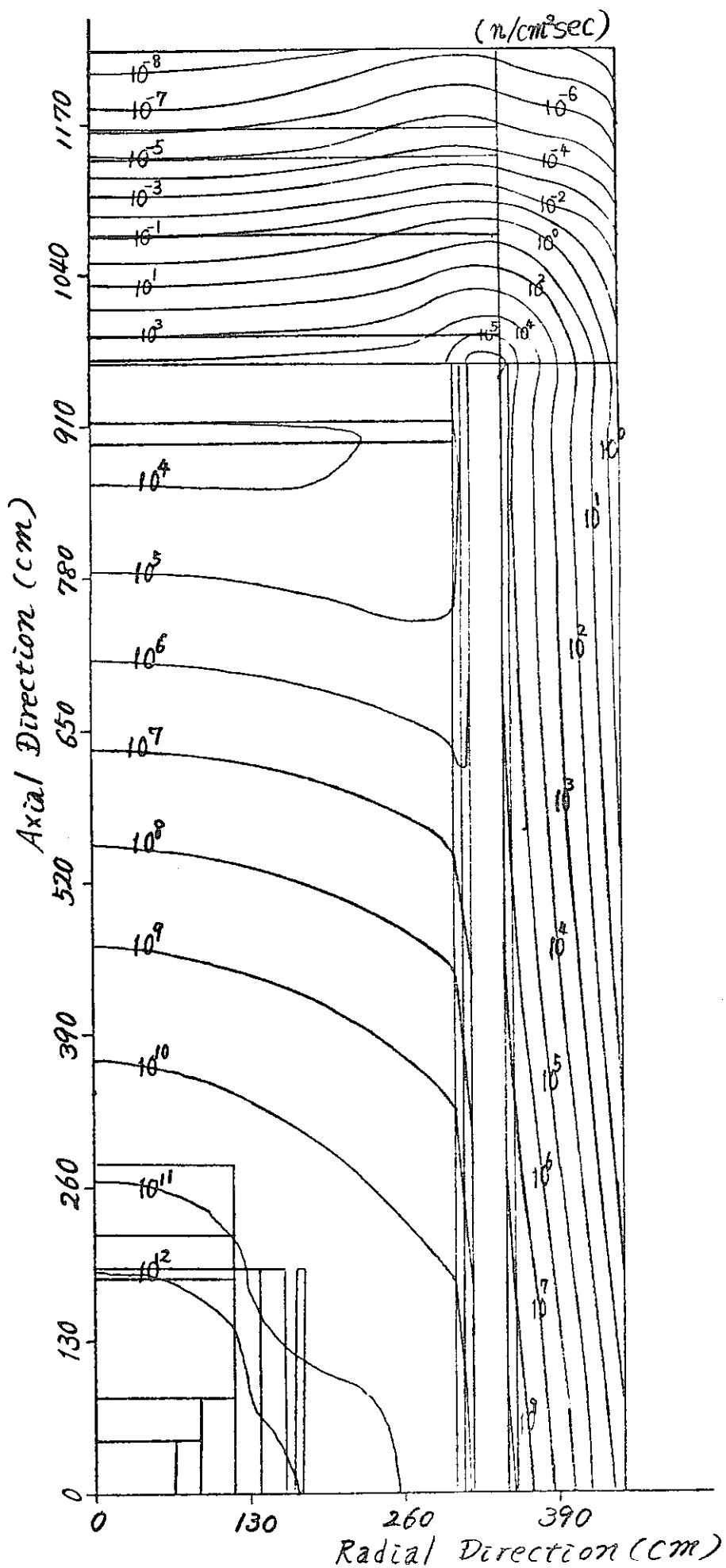


Fig.(III-13) Neutron Contour Mapping 7 Group (0.1 Kev - 4.65 ev)

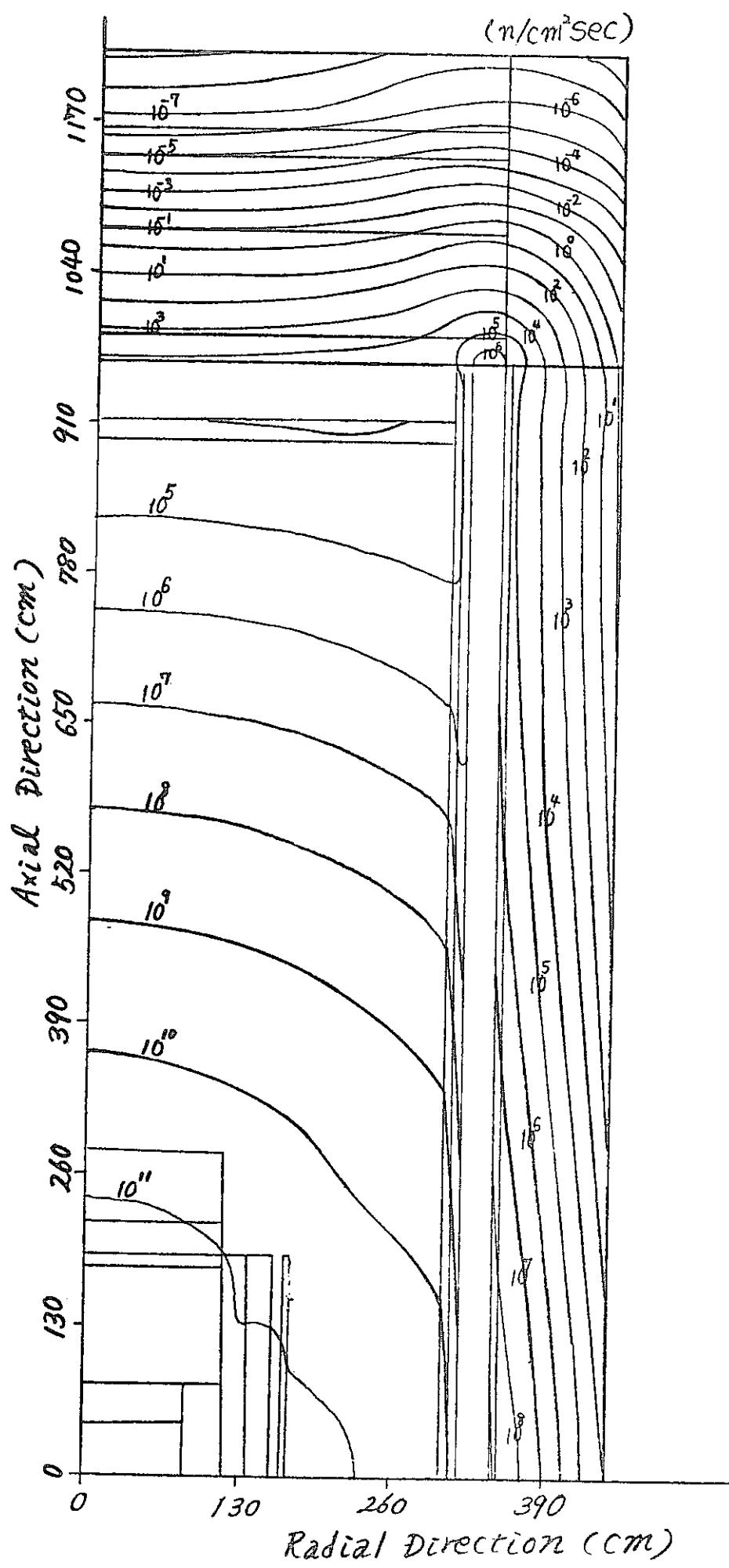
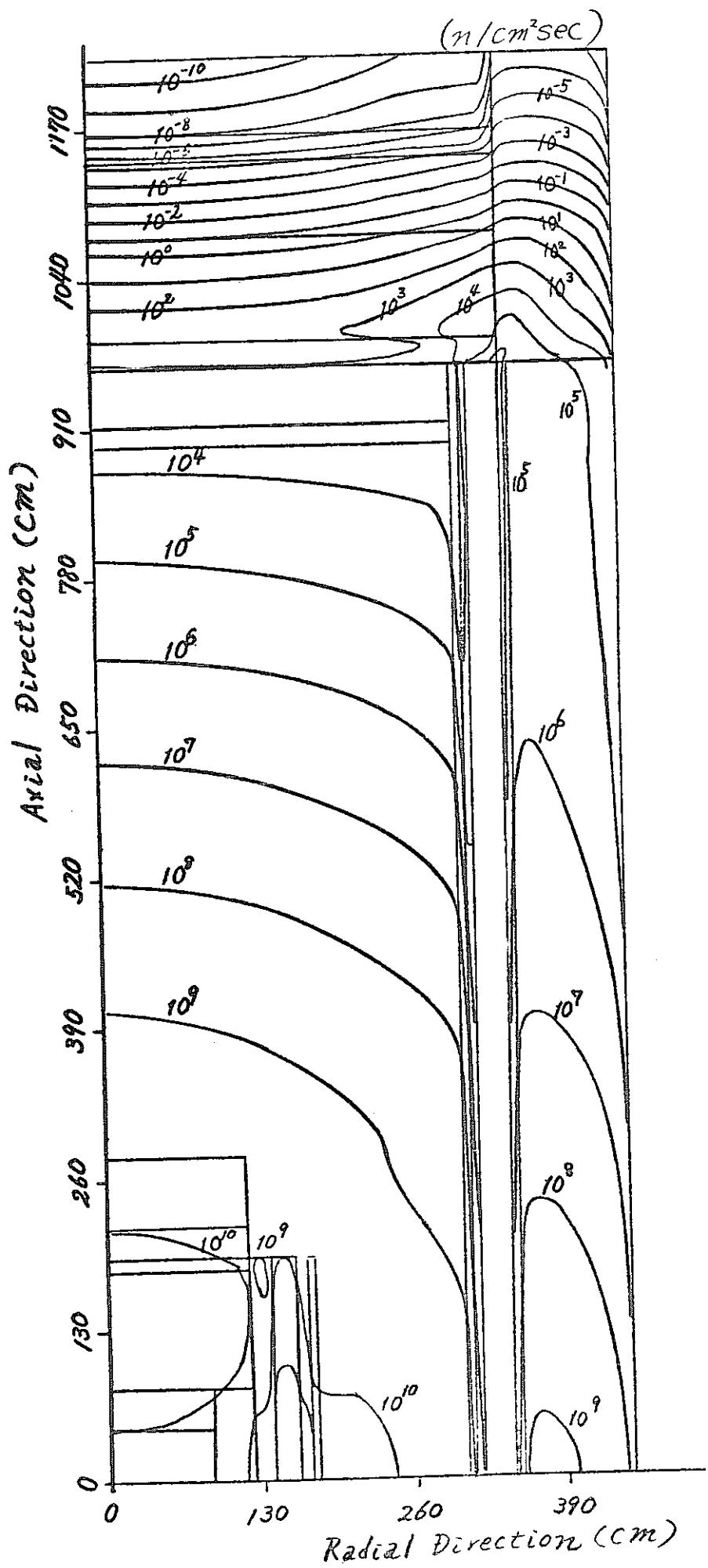


Fig.(III-14) Neutron Contour Mapping 8 Group (4.65ev-0.215ev)



Fig(III-15) Neutron Contour Mapping 9 GROUP (0.215 eV >)

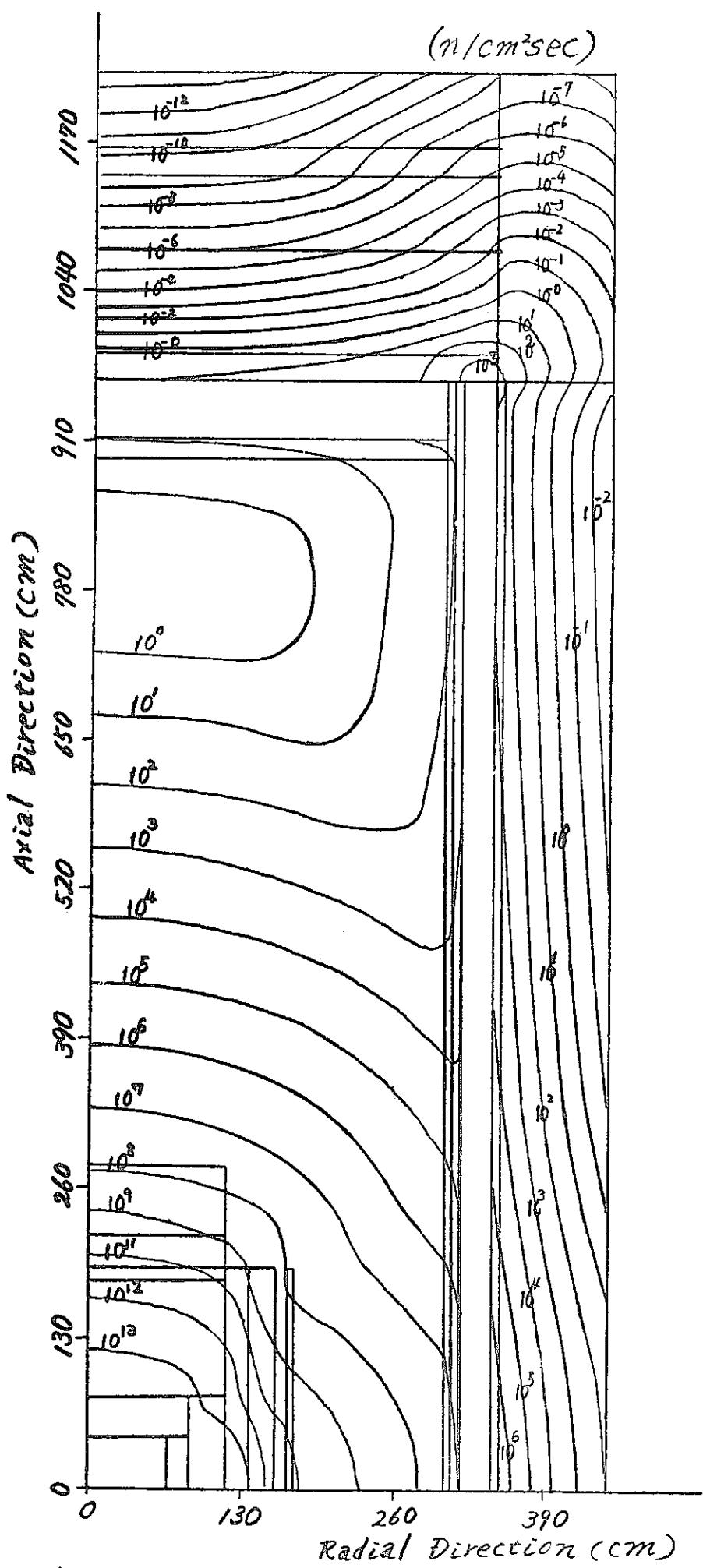
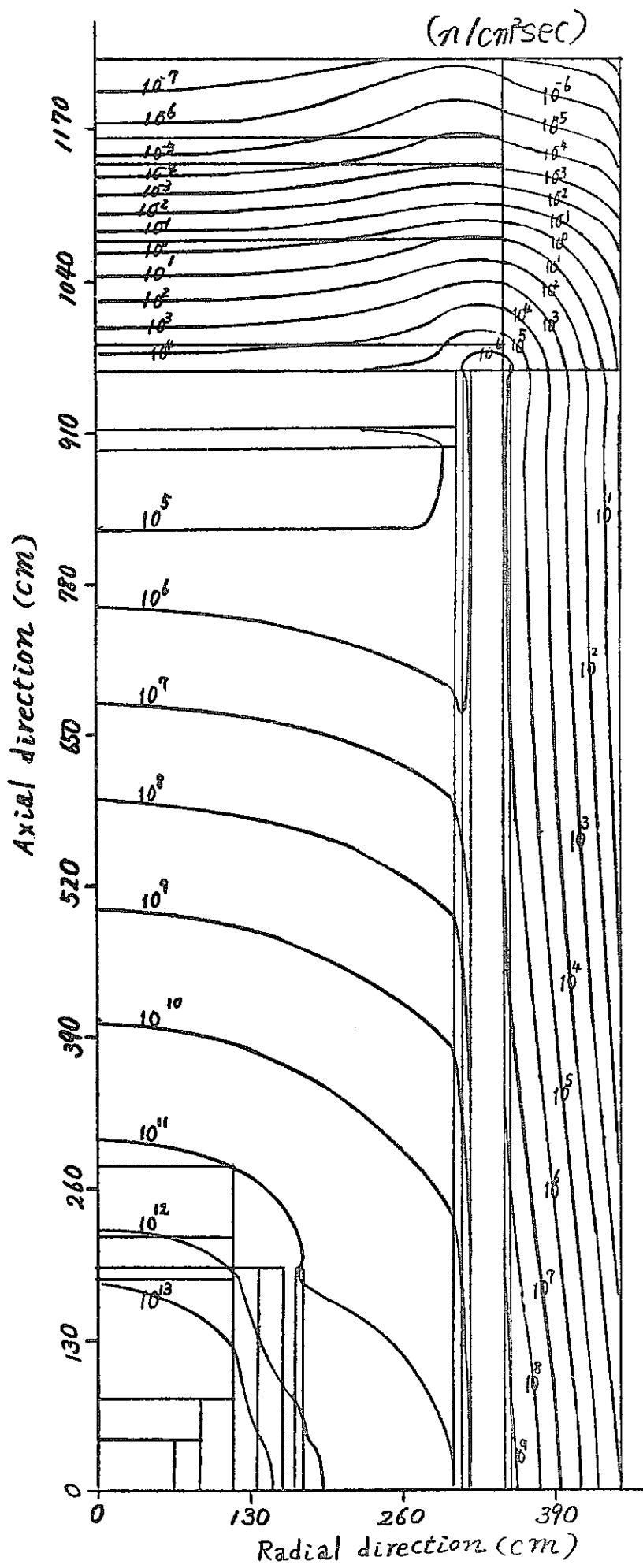


Fig.(III-16) Neutron Contour Mapping Fast Neutron  
(10.5 Mev - 0.8 Mev)



Fig(III-17) Neutron Contour Mapping Intermediate Neutron

(0.8 Mev - 0.215 eV)

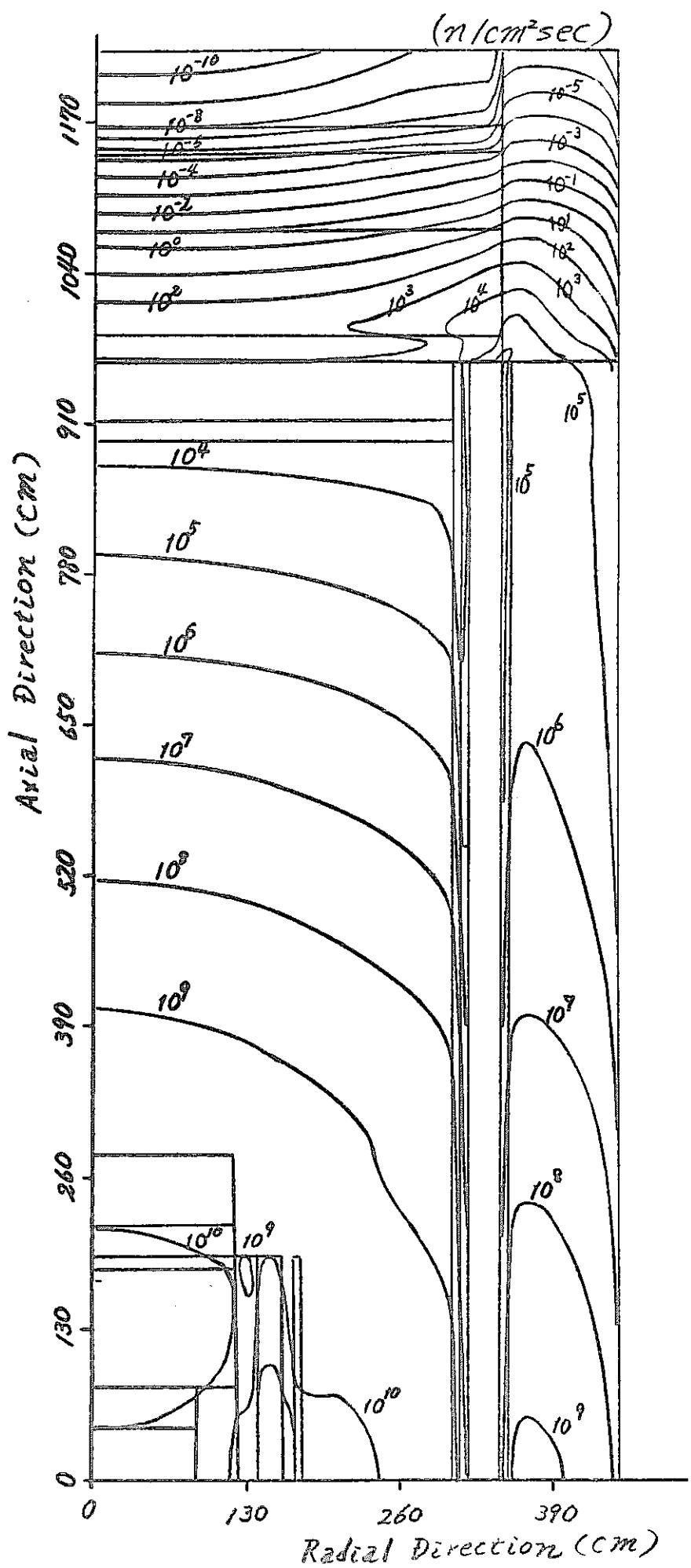


Fig.(III-18) Neutron Contour Mapping Thermal / Neutron (0.215ev)

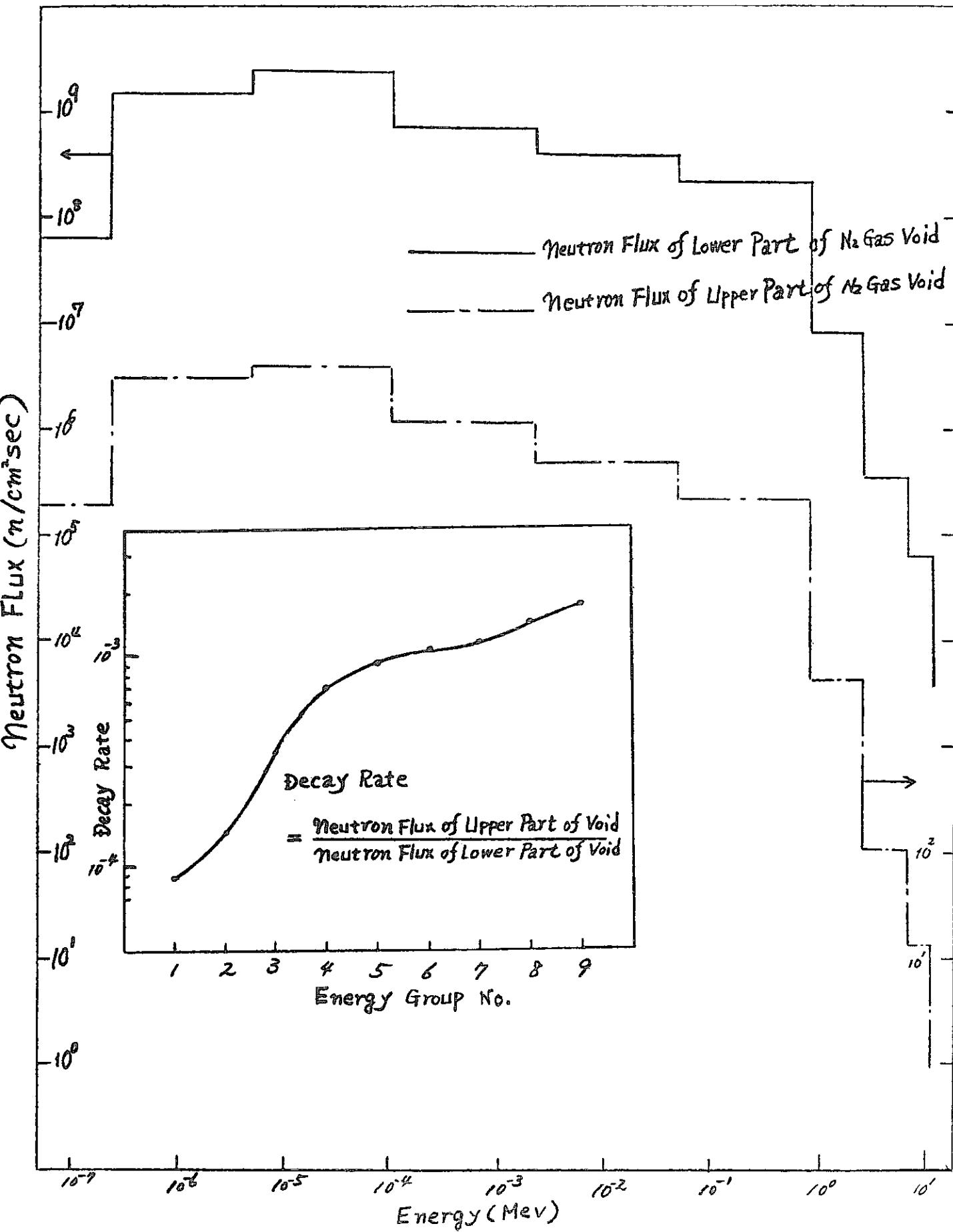


Fig.(III-19) Comparison of Neutron Flux between at Lower Part of N<sub>2</sub> Gas Void and at Upper Part

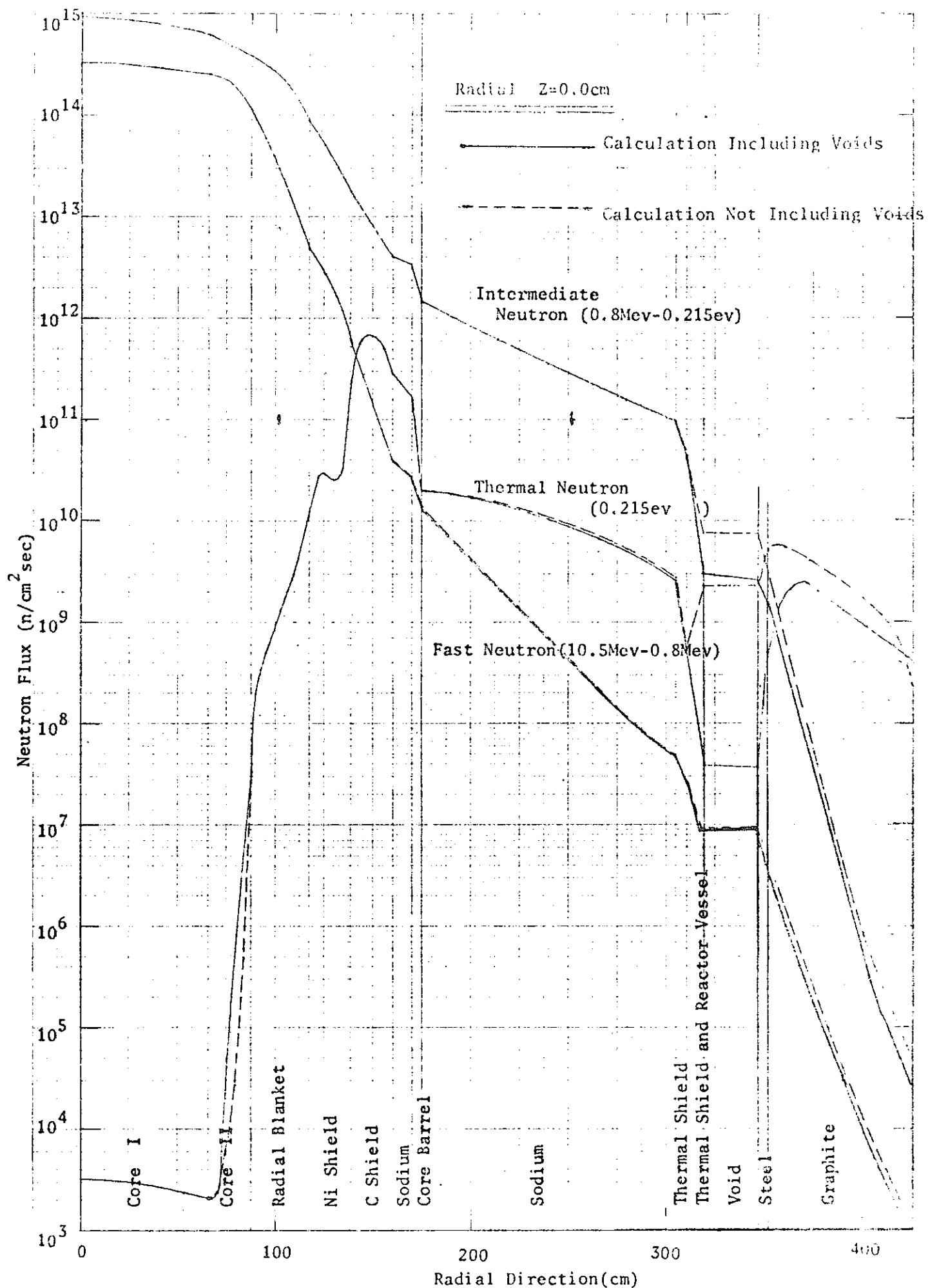


Fig.(III-20) Neutron Attenuation Distribution  $Z=0.0\text{cm}$

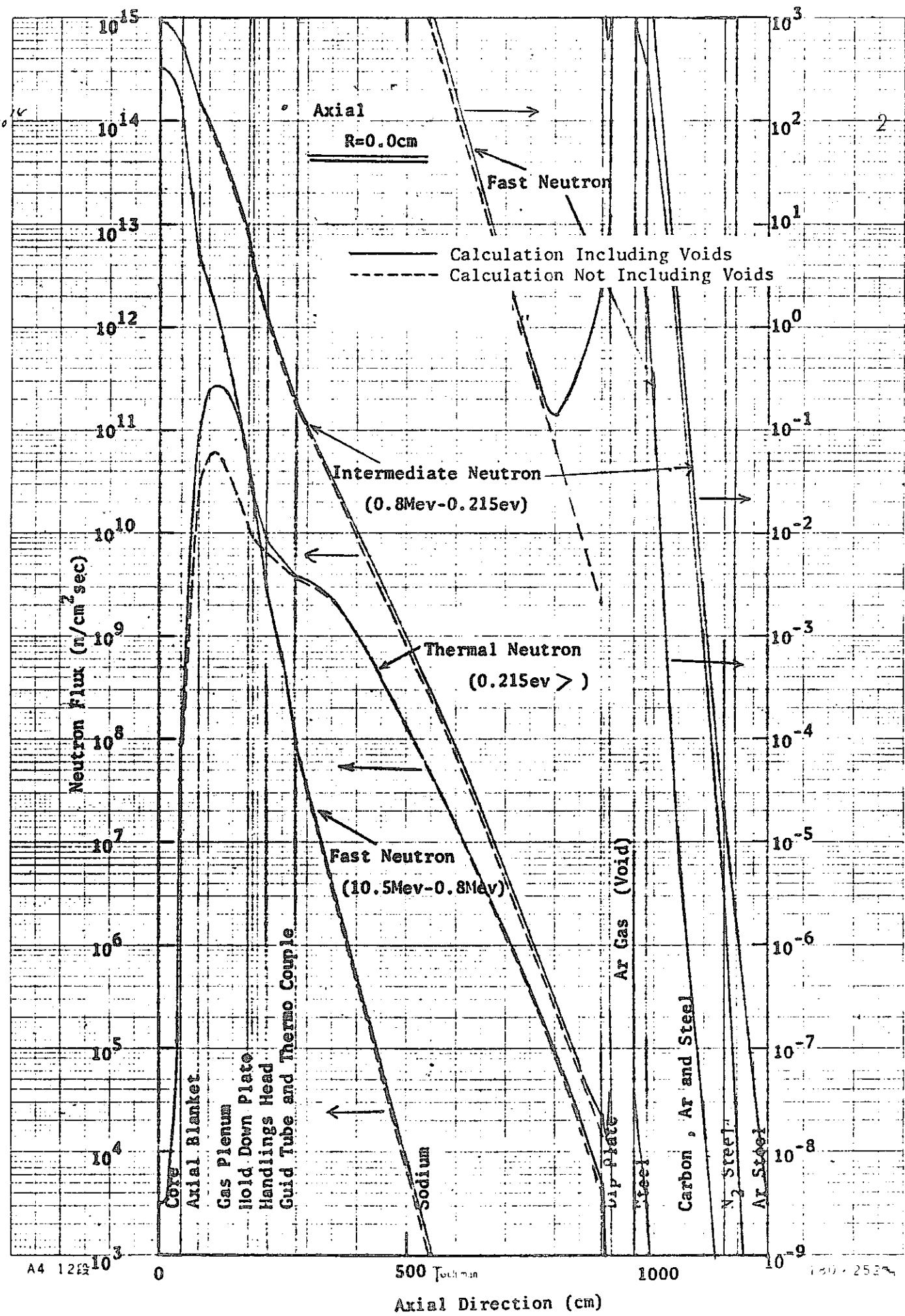


Fig. (II-21) Neutron Attenuation Distribution  $\gamma=0.0$

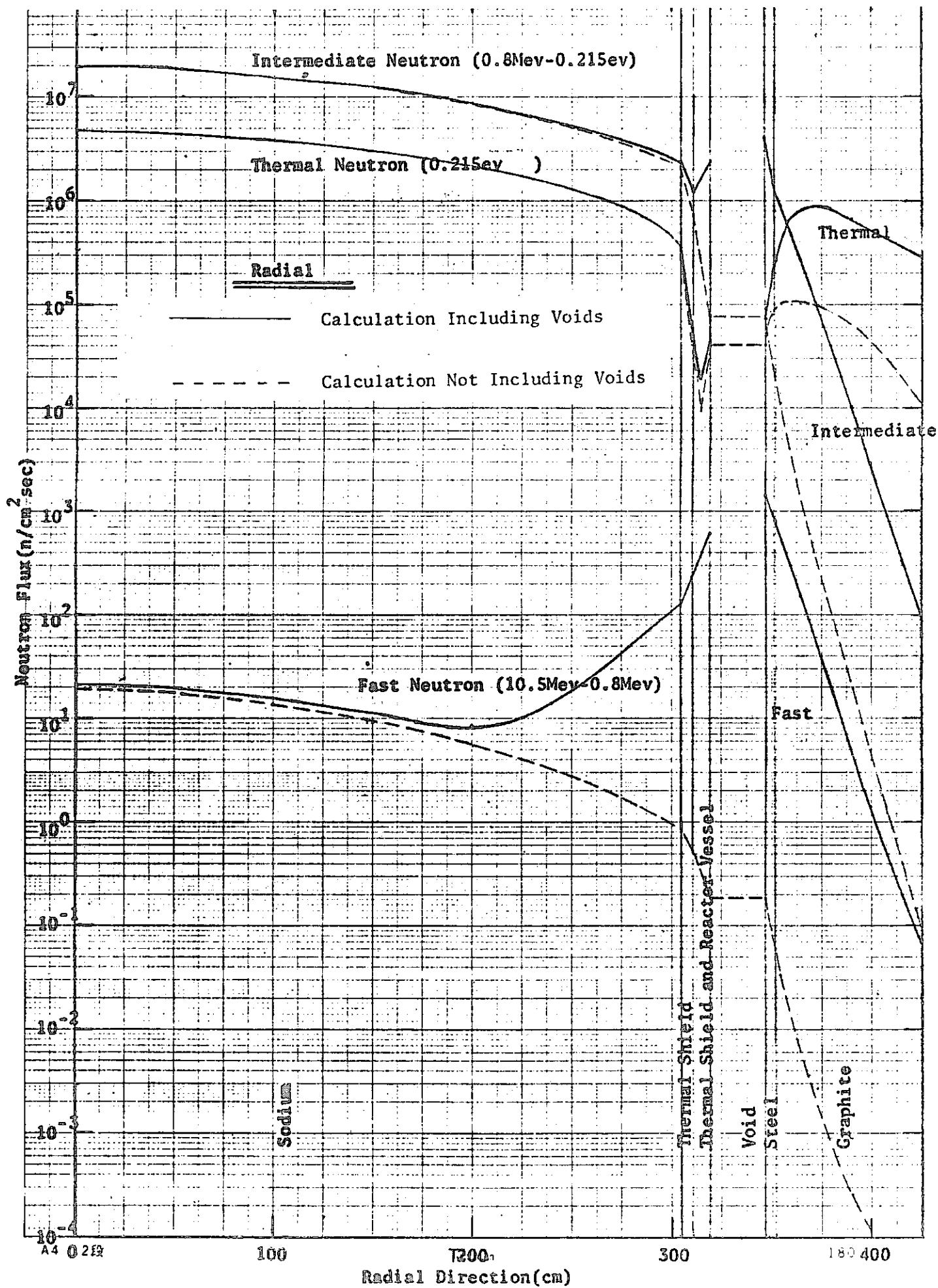


Fig. (III-22) Neutron Attenuation Distribution  $Z = \sim 650_{C2L}$

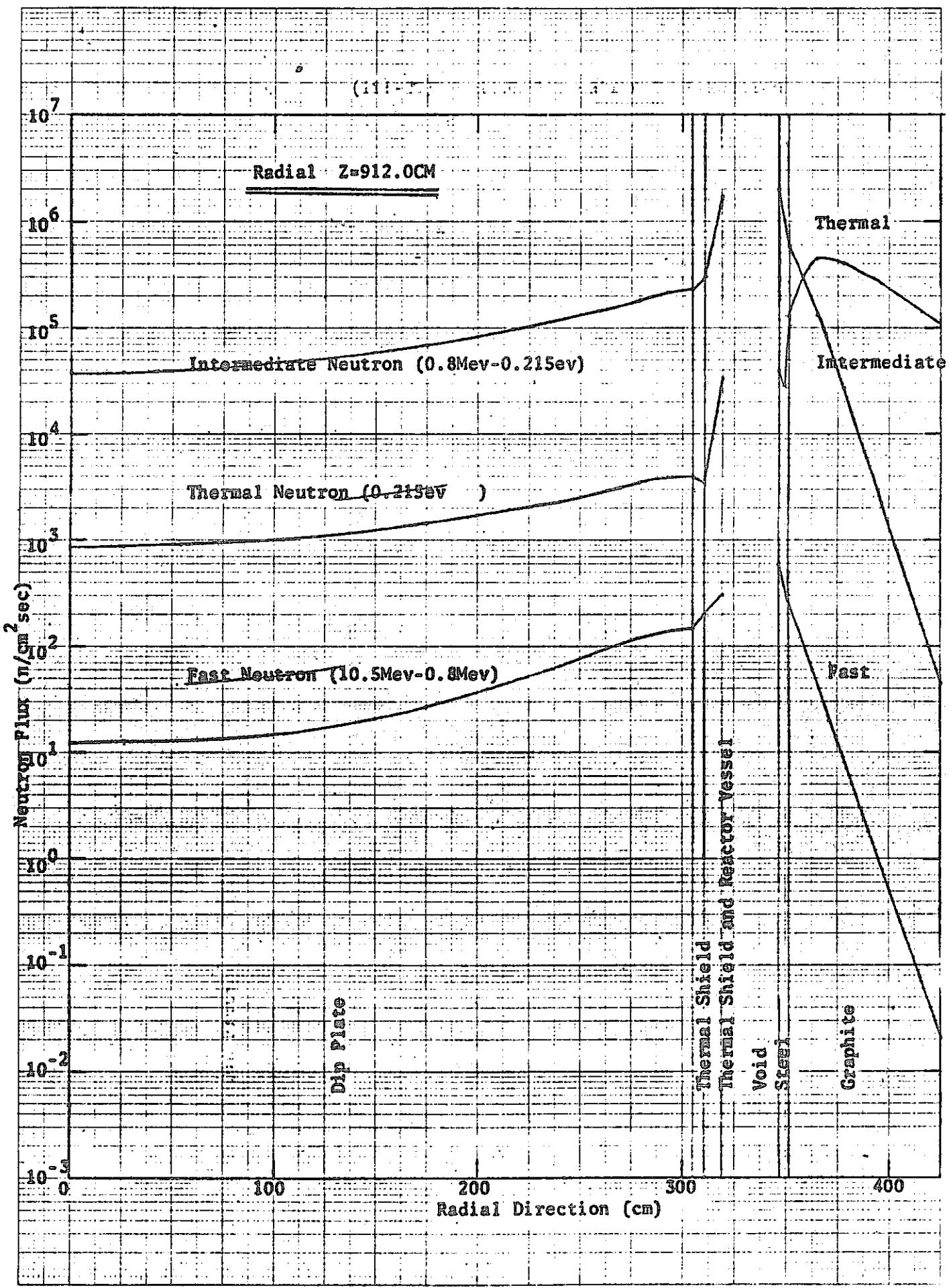
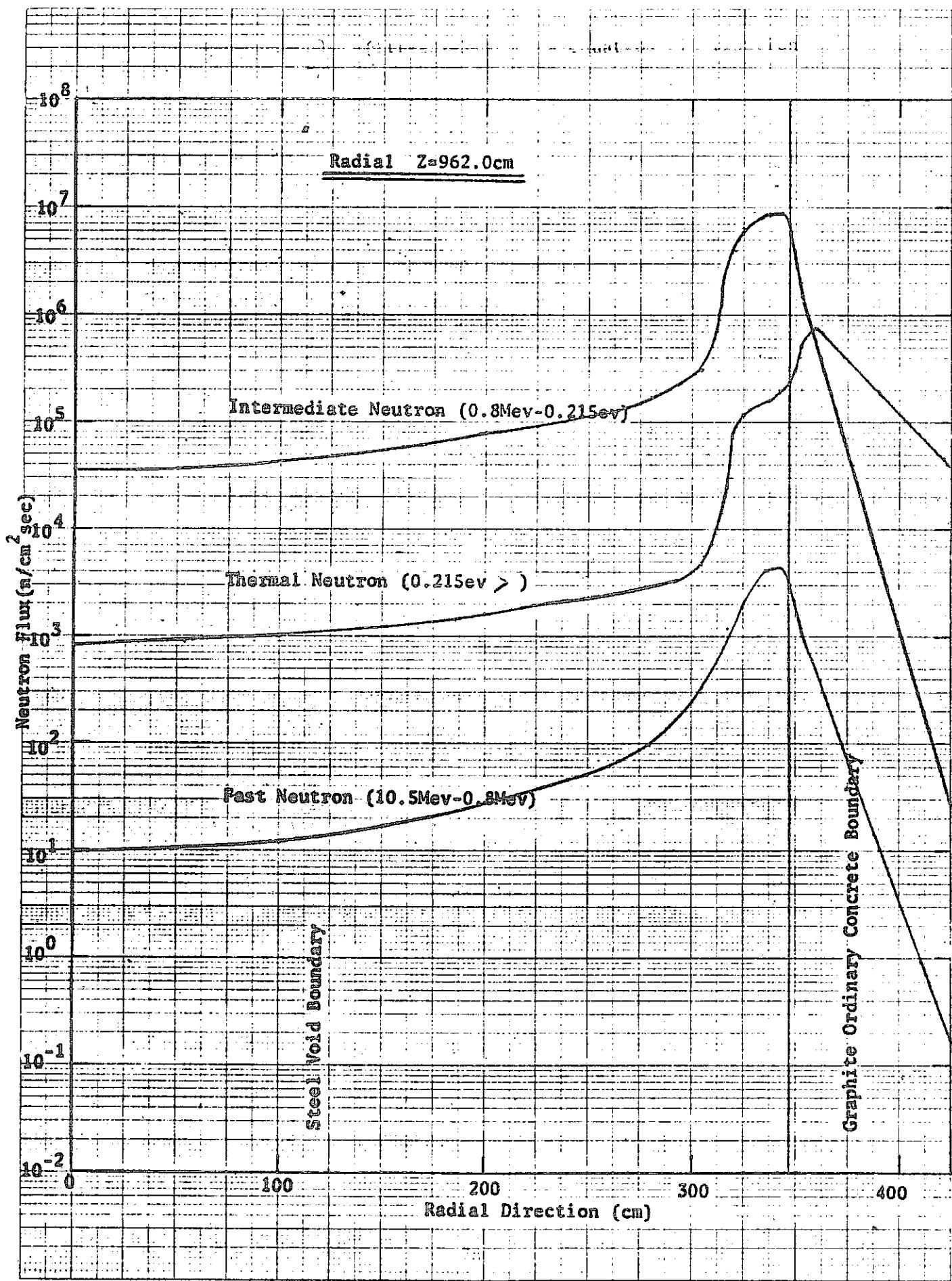


Fig.(II-23) Neutron Attenuation Distribution Z=912.0CM

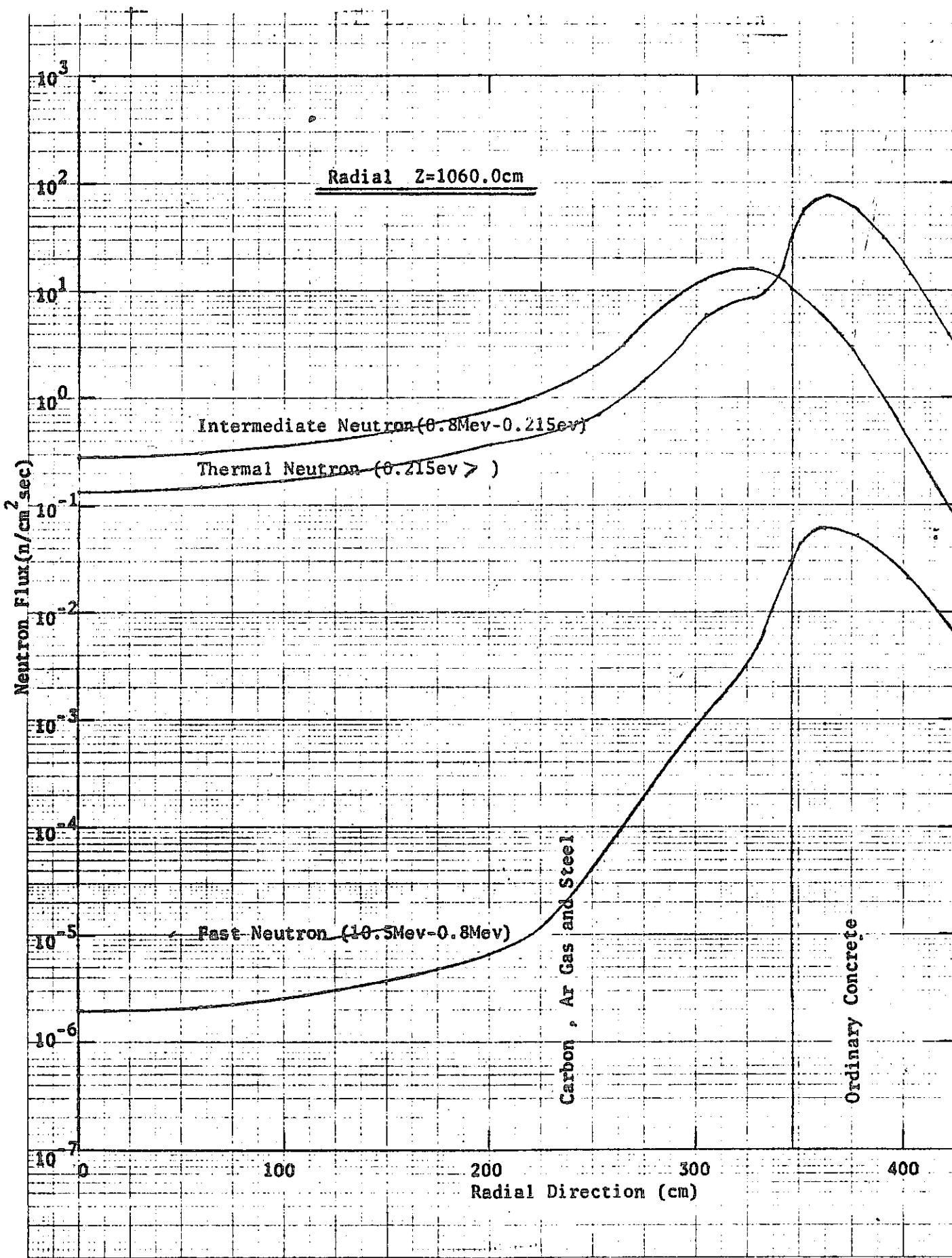


A4 12版

Tschirhart

180 × 252mm

Fig. (III-24) Neutron Attenuation Distribution  $Z=962.0\text{cm}$



A4 12段

Tocamem

180~252~

Fig. (III-25) Neutron Attenuation Distribution  $Z=1060.0\text{cm}$

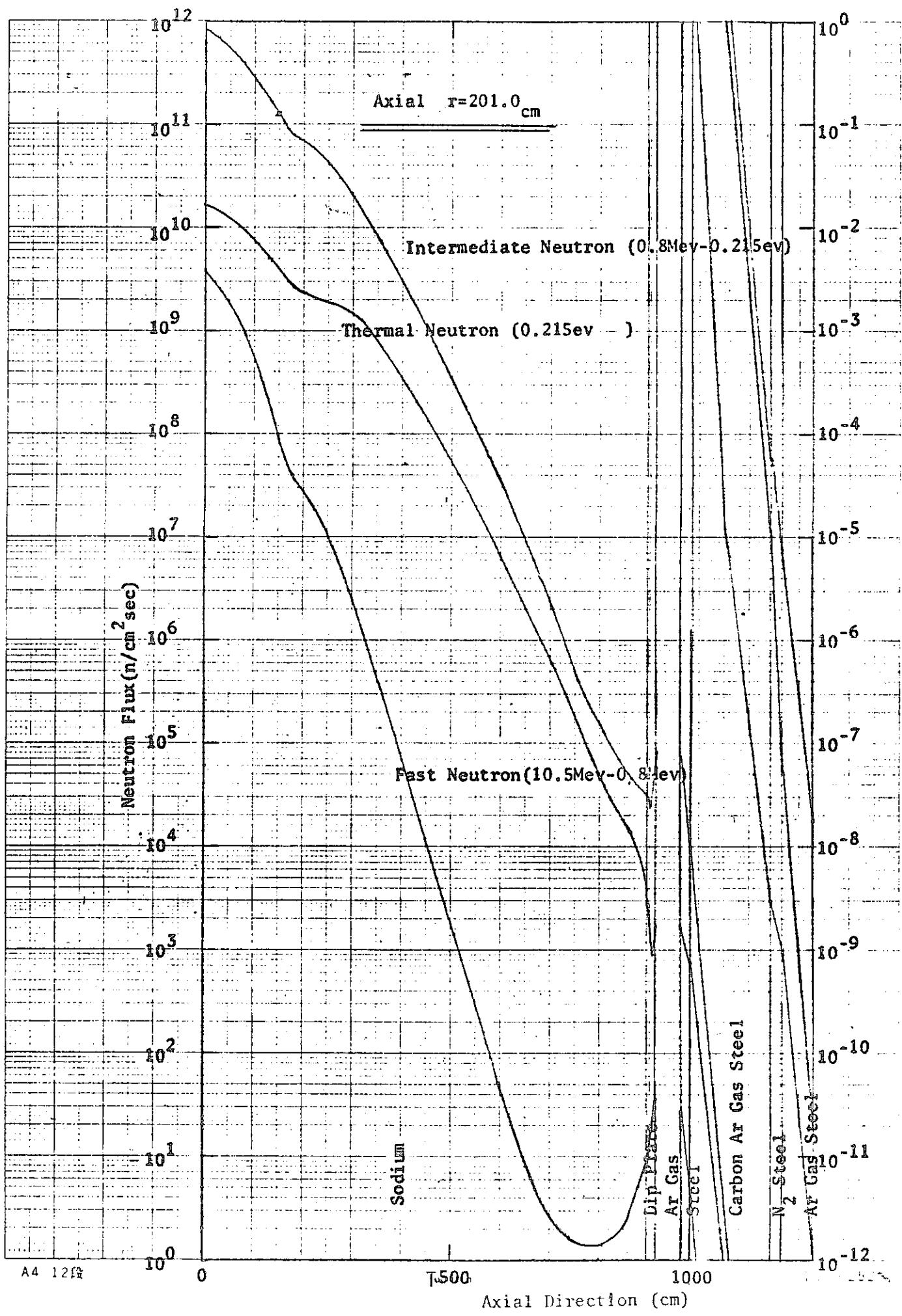


Fig.(II-26) Neutron Attenuation Distribution  $r=201.0\text{ cm}$

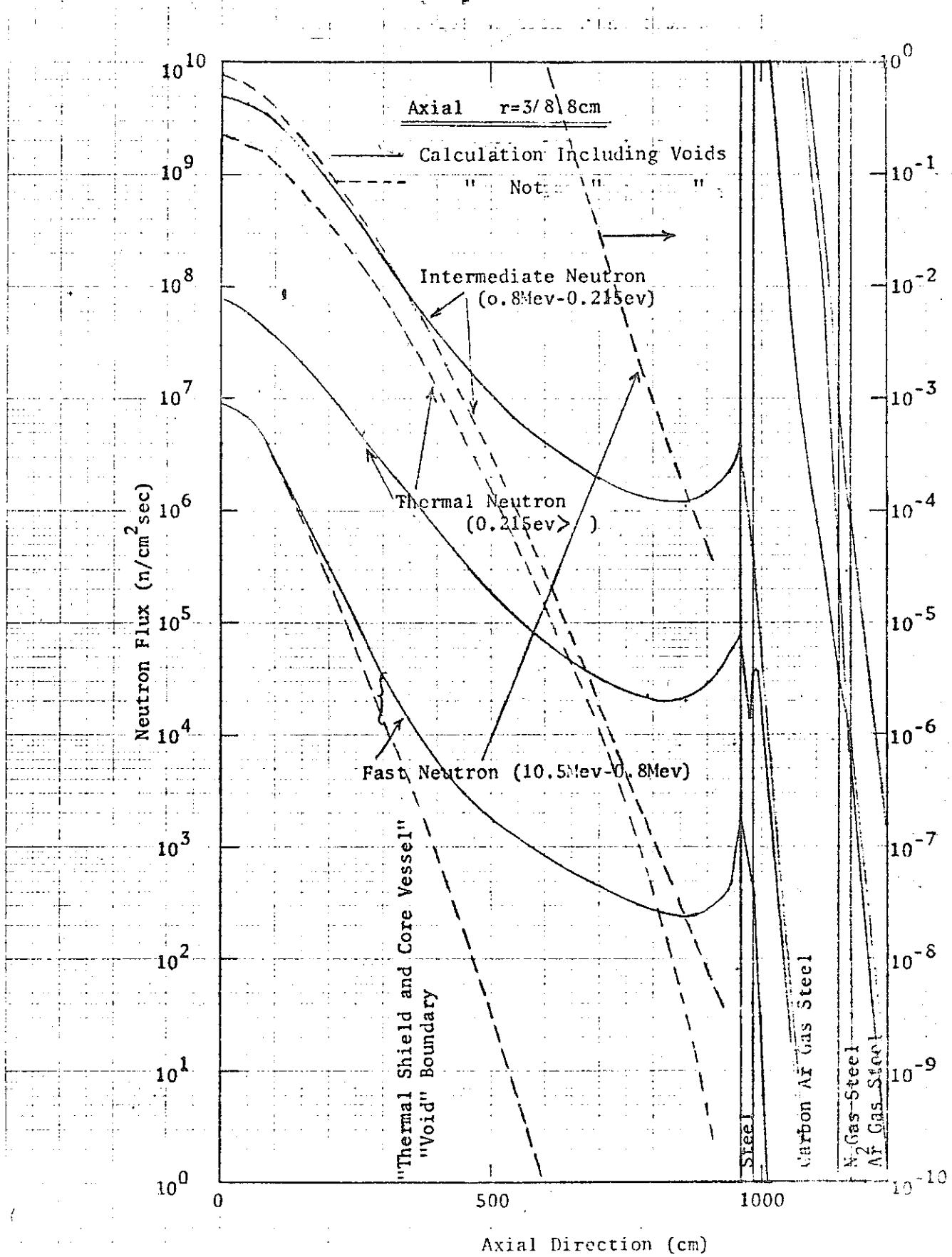
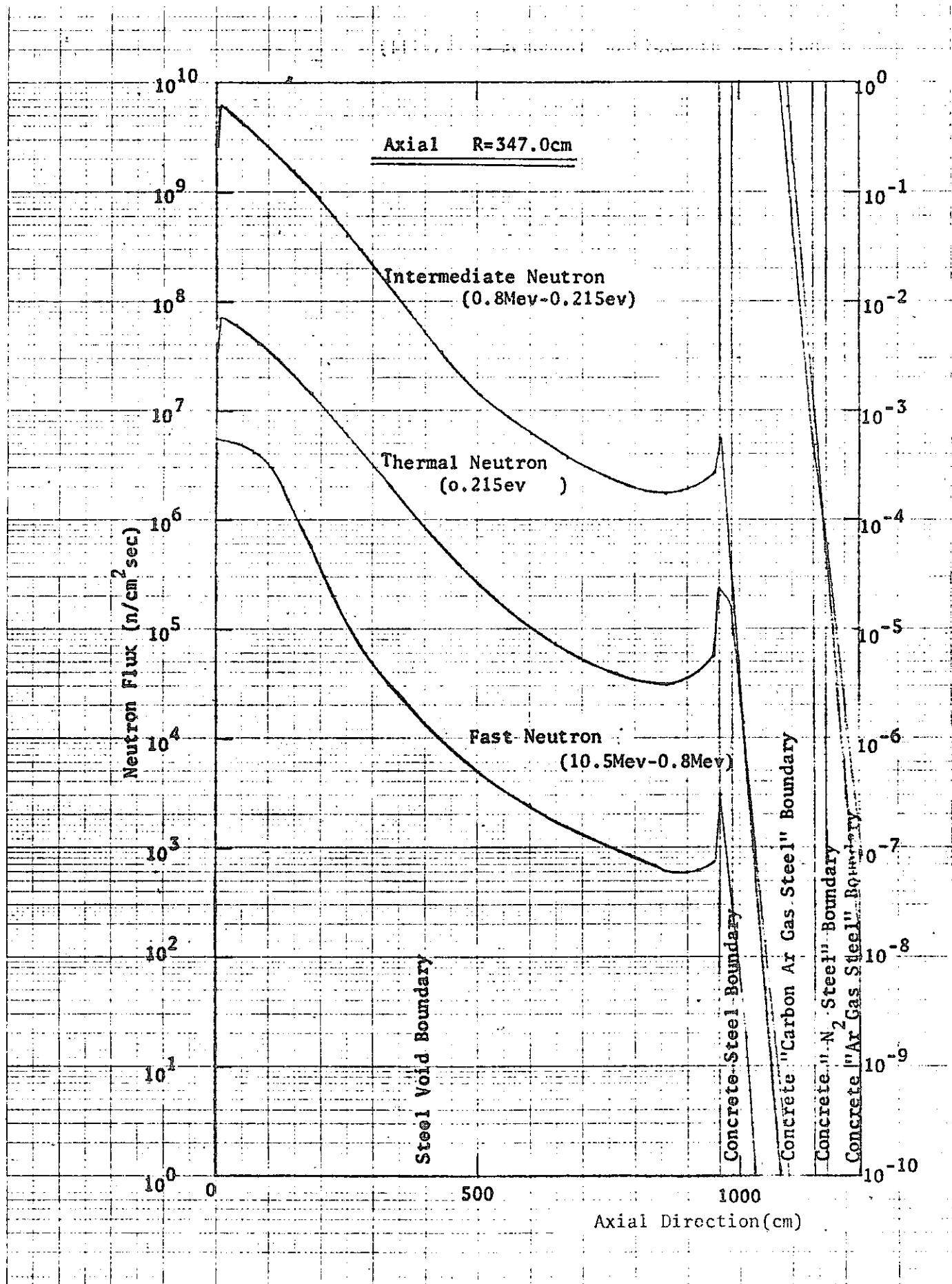


Fig.(III-27) Neutron Attenuation Distribution  $r=318.8\text{cm}$

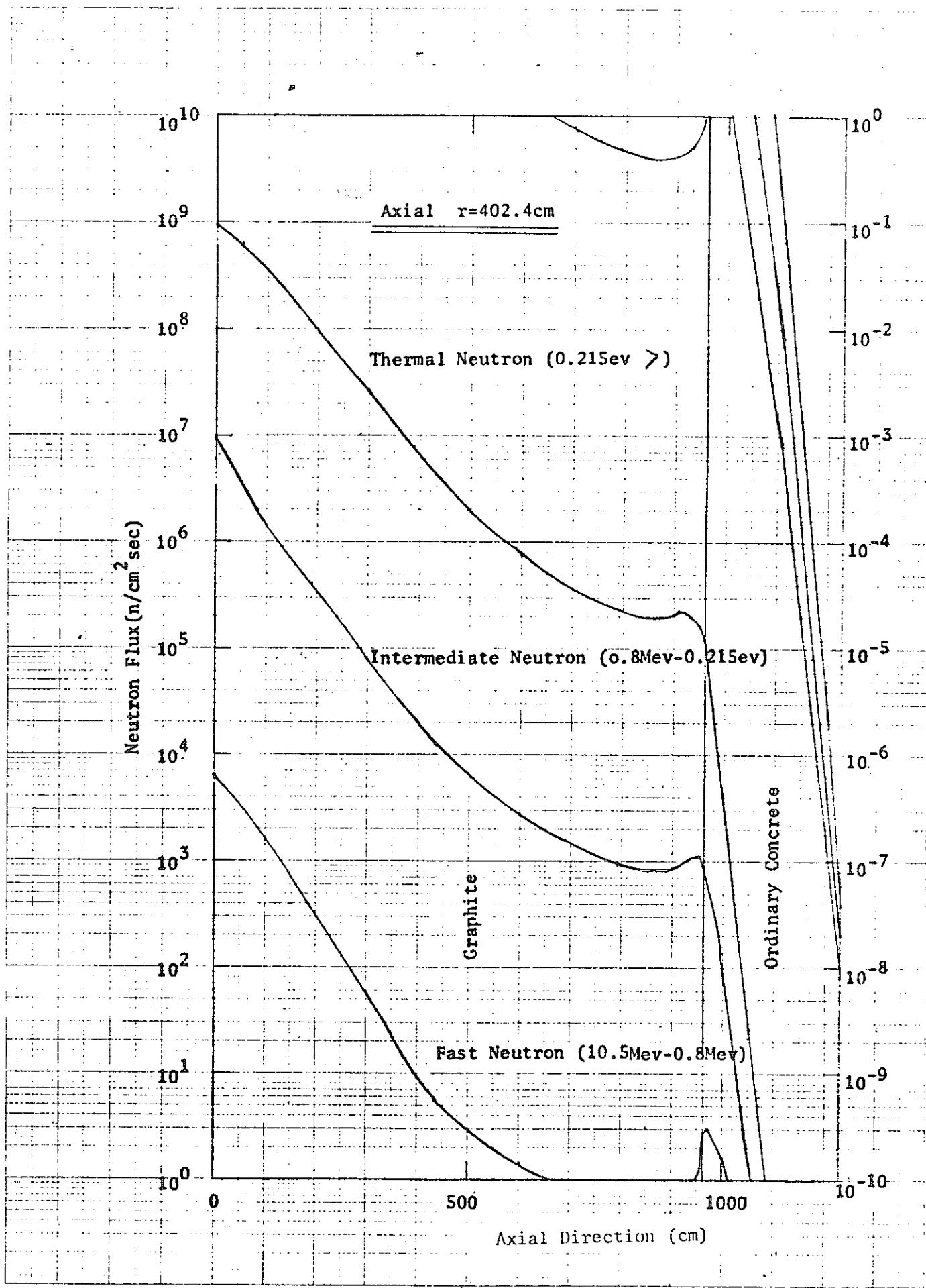


A4 1222

Tschman

180° 253°

Fig. (III-28) Neutron Attenuation Distribution R=347.0cm

Fig.(III-29) Neutron Attenuation Distribution  $r=402.4\text{cm}$

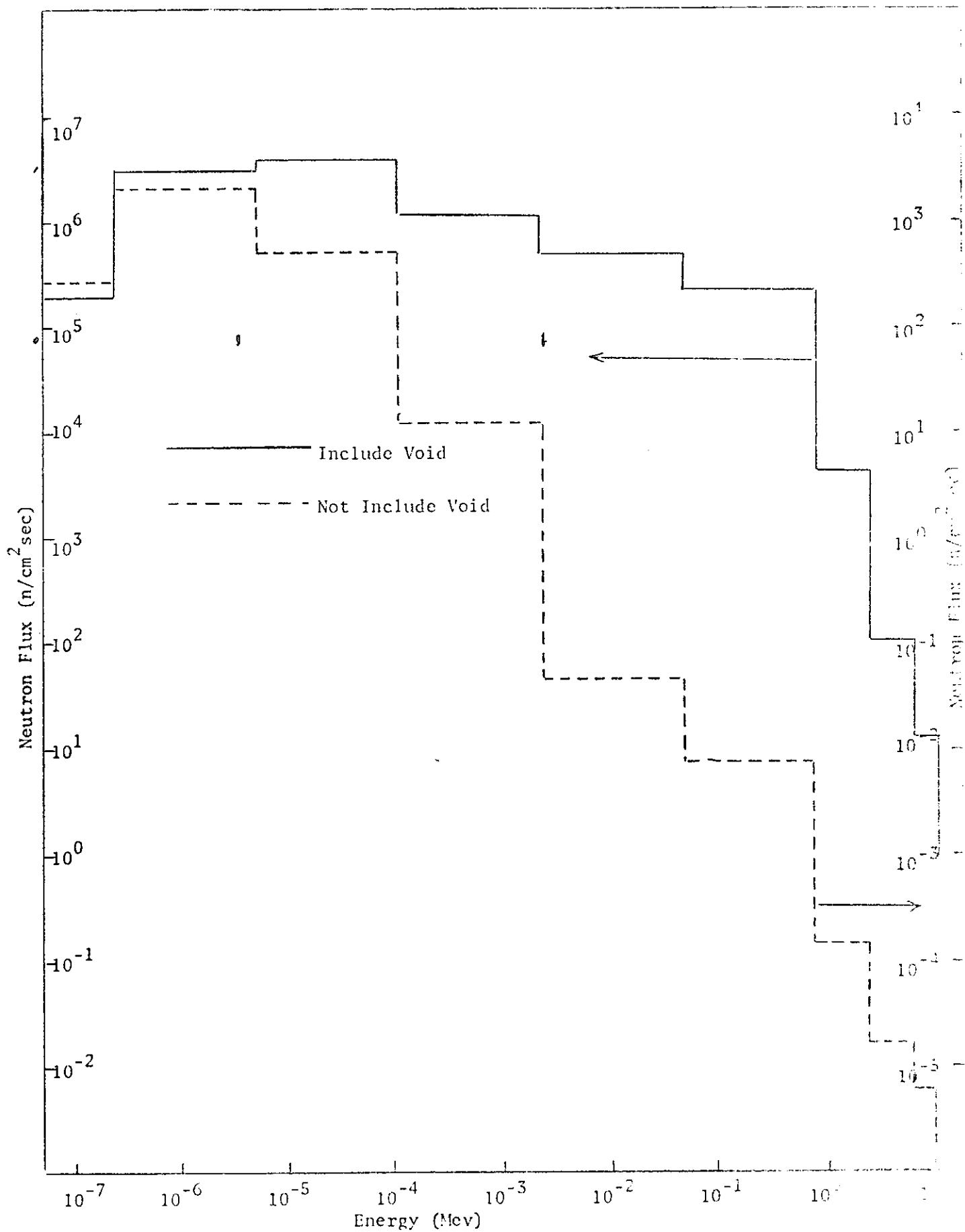


Fig.(III-30) Comparison of Neutron Flux between Calculation Including Void and Not Including Void

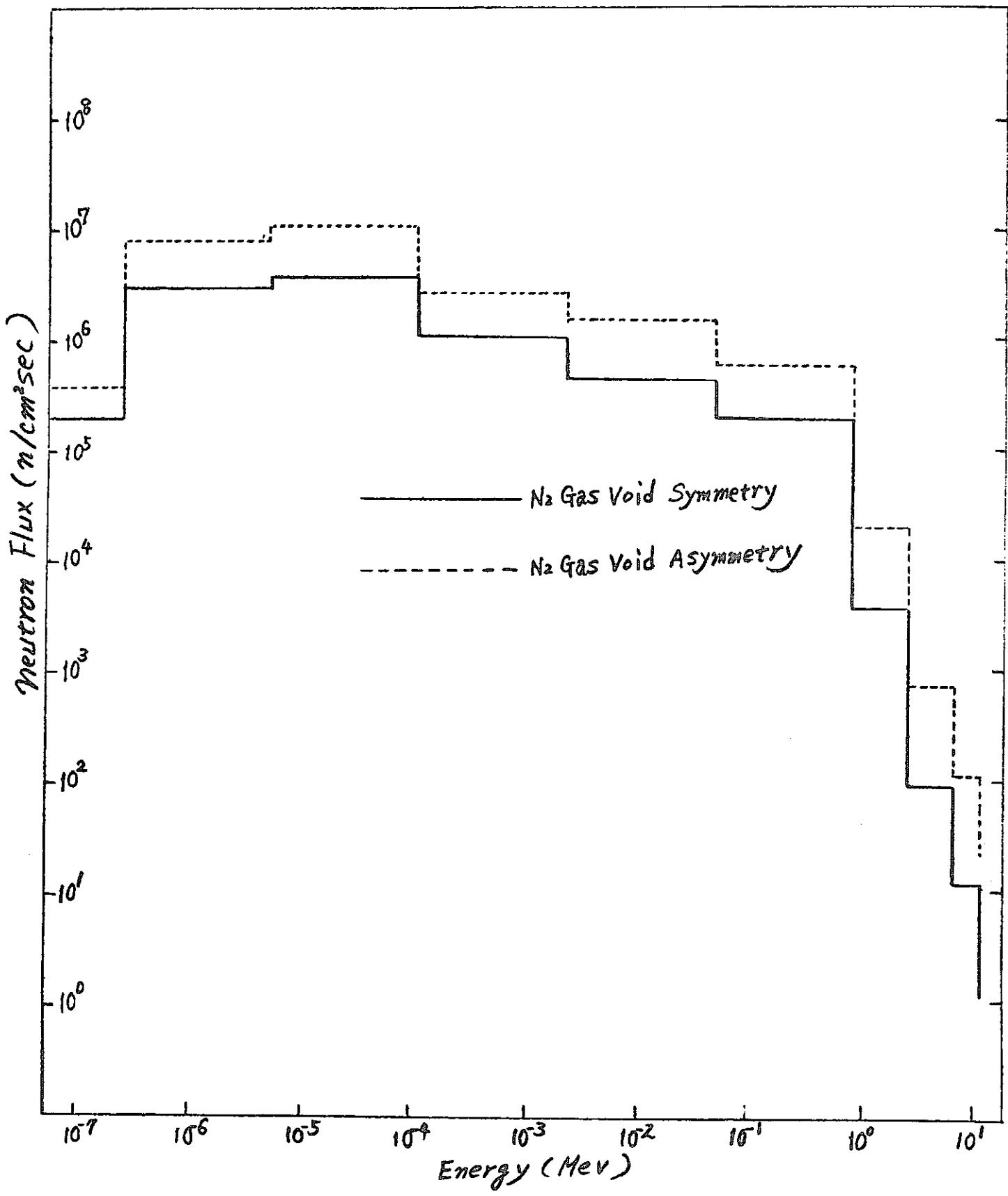
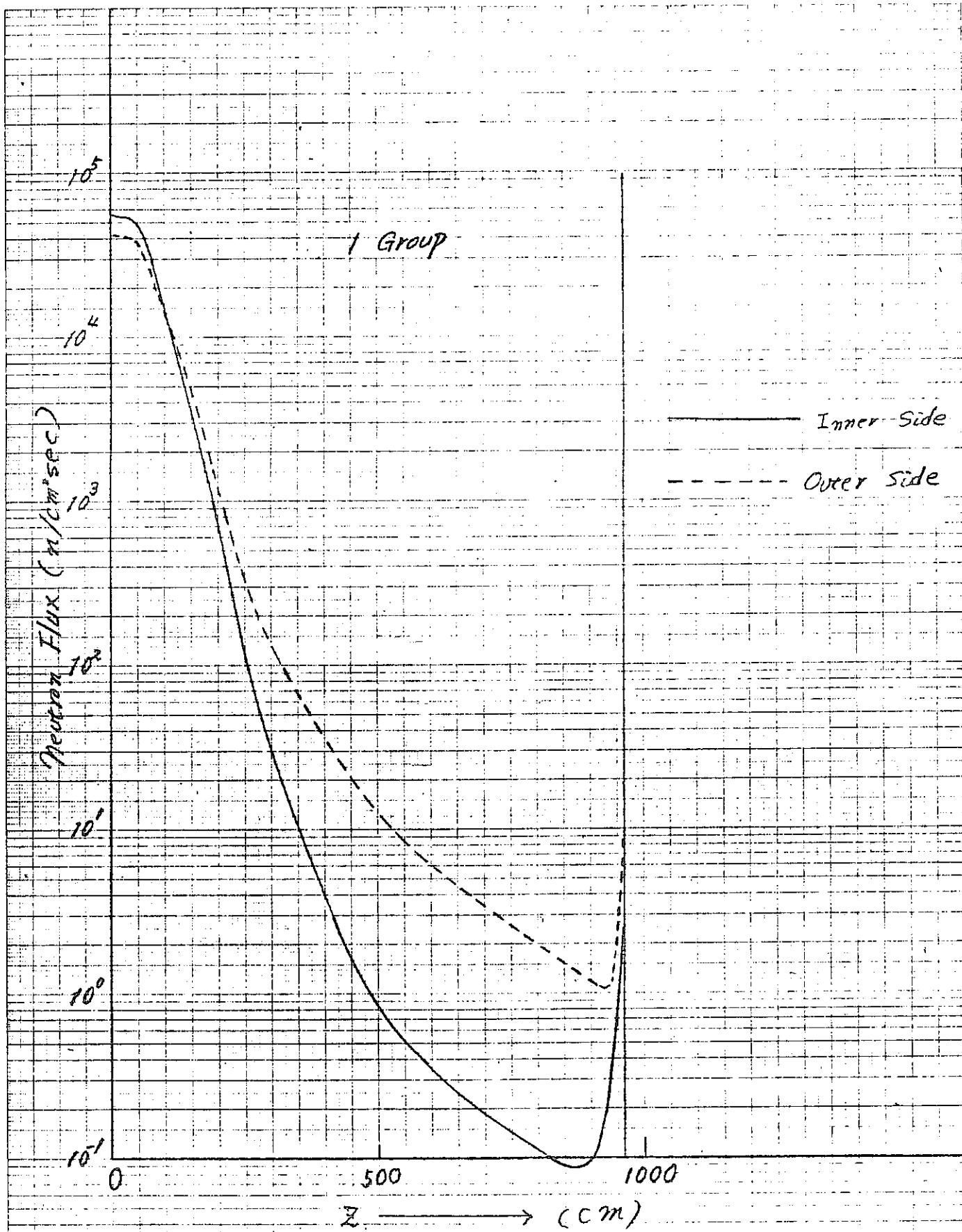


Fig.(III-31) Difference of Void Boundary Condition



Fig(III-32) Neutron Flux Distribution  
around Void (1 Group)

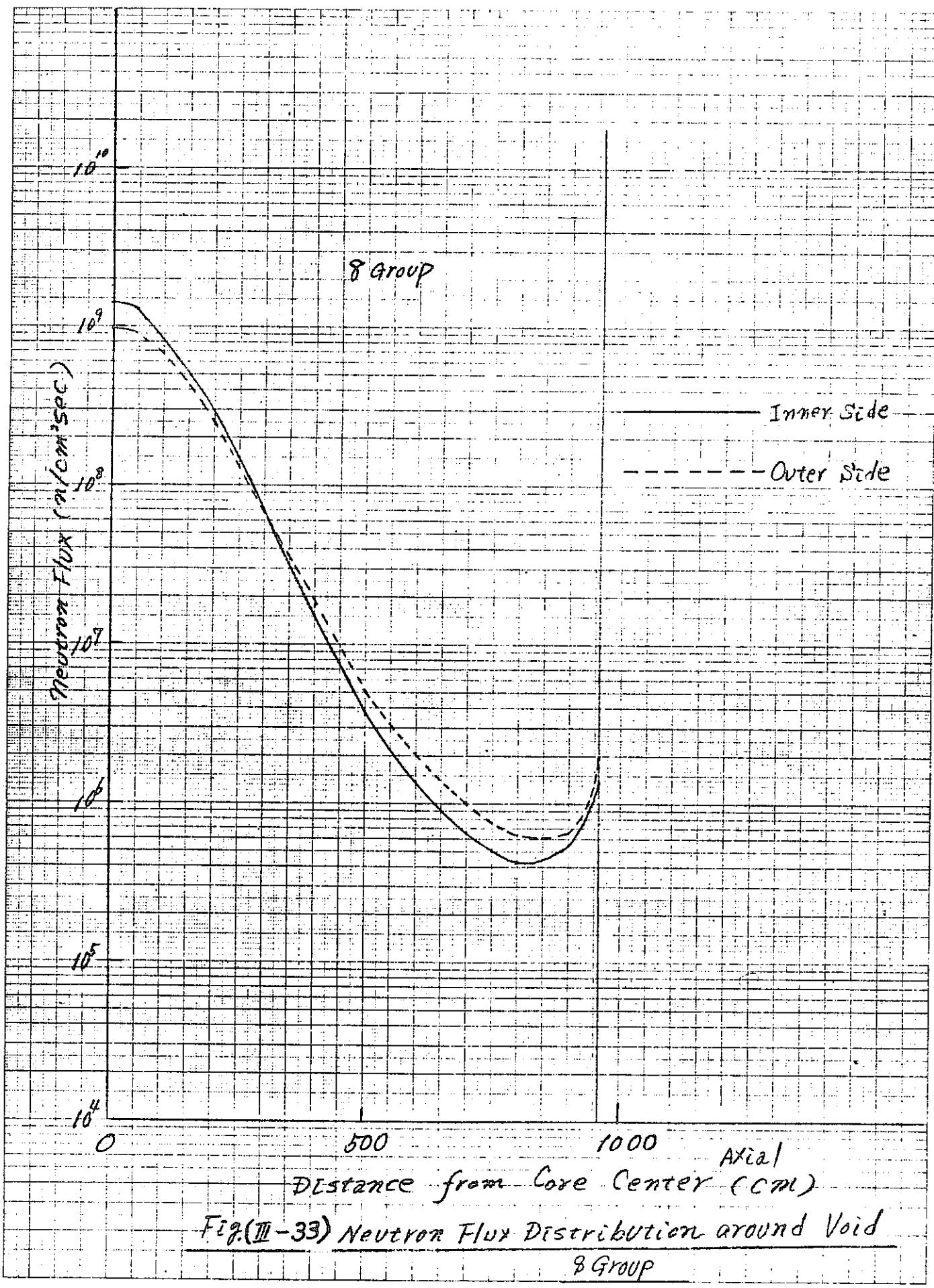


FIG.(III-33) Neutron Flux Distribution around Void  
 8 Group

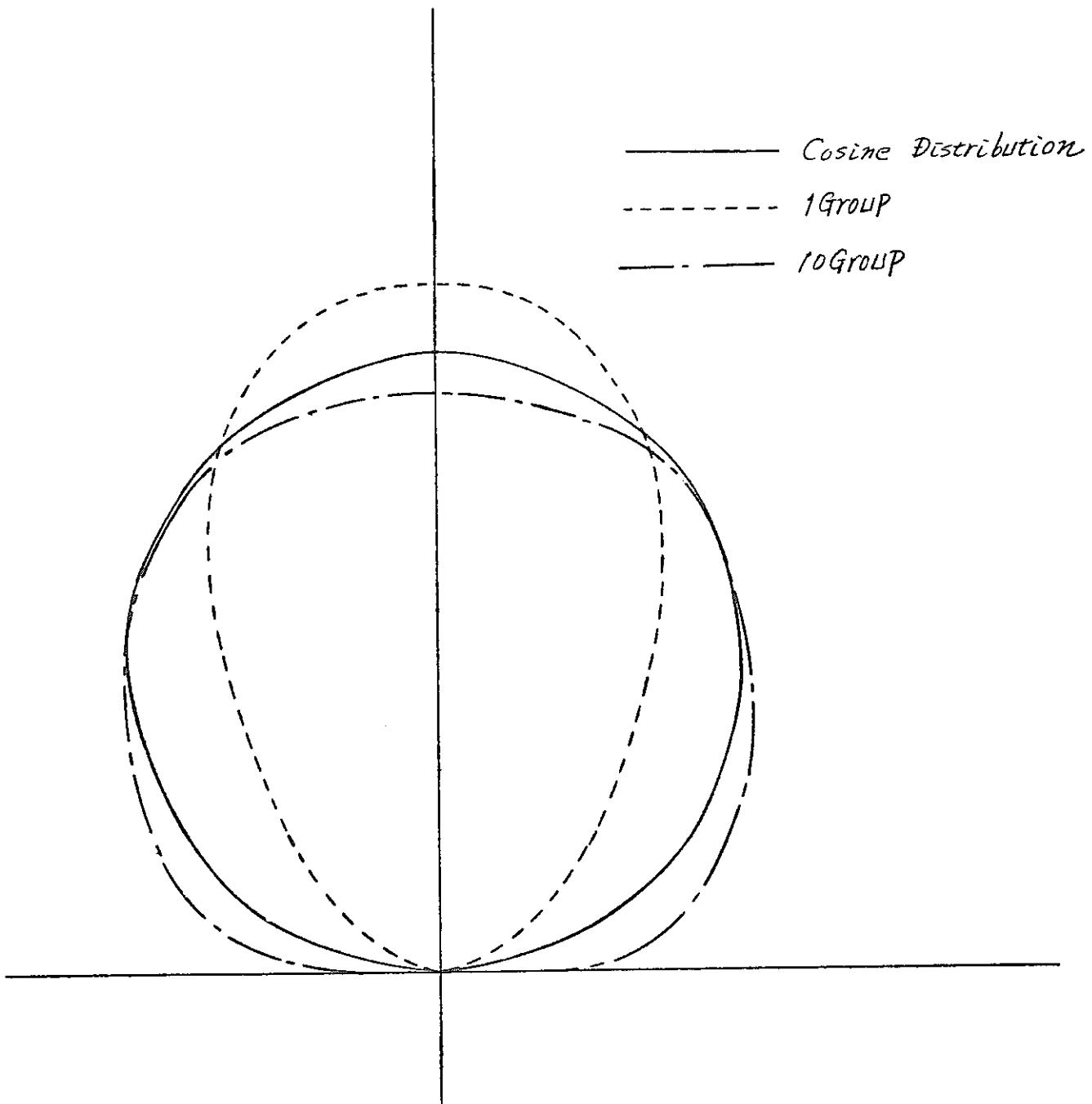


Fig.(III-34) Angular Flux Distribution at  
Reactor Vessel of "Monju" by ANISN

#### IV. Summary

The following works have been performed as the research and development under contract for 1972:

1. In order to make possible the exact analysis of the geometries including voids, two-dimensional diffusion calculation code DIAC has been modified taking into account of the treatment of the axial symmetrical void geometry.
2. Unreasonable treatment of minus neutron current entering into void has been removed.
3. As an application example of the modified code, an analysis of the main shield of Fast Prototype Reactor "MONJU" including voids has been performed and the evaluation of void effect has been made. From this result, following points have been cleared.
  - a) Comparing the result not including voids (before previous analysis, there has been noted an increase of 3 ~ 7 figures neutron flux level at the upper part of N<sub>2</sub> gas void (lowest part of top shield)).
  - b) Comparing the result on the assumption that the neutrons enter into void from N<sub>2</sub> gas void lower surface on the radial central axis (previous analysis), there has been noted an decrease of ~ 1 figure neutron flux level at the upper part of N<sub>2</sub> gas void.
  - c) That the neutron flux level at outer surface of N<sub>2</sub> gas void has been high than that at inner surface has been caused from the cosine angular distribution assumption of

neutron entering into void and from that the neutron current entering into void at inner void surface was greater than that at outer surface.

4. From the above works of 1 to 3, it has been found out that the presence of voids in the shield gives a conspicuous effect to the neutron flux distribution which could not be ignored from the standpoint of shield design. Future problems may be as follows:

- a) Evaluation of diffusion approximation at high energy groups.
- b) Evaluation of cosine angular distribution assumption of neutron entering into void.
- c) Treatment of complex geometrical voids
- d) Improvement of the fitting method of the neutron transition probability within voids calculated by line of sight method.

## V. Appreciation

This research work on "Applicability of Shielding Design Code" has been performed as follow up and further modification of those works of "Development of Two-Dimensional Shielding Design Code (I)--Evaluation of Two-Dimensional Analysis"--in 1970, "Development of Two-Dimensional Shielding Design Code (II)--Group Constant Collapsing Method and Establishment of Core Boundary Connection Method--" in 1971, "Evaluation and Application Study of Fast Reactor's Two-Dimensional Shielding Code" in 1972, and "Improvement and Evaluation of Two-Dimensional Shielding Design Code" in 1973.

Taking this opportunity, we hereby wish to express our most sincere appreciation of the assistance and useful advice provided to us for this research work by Dr. Kobayashi, General Manager and other members of the Nuclear Research and Development Department of Kawasaki Heavy Industries, and Mr. Yoshiharu Higashihara, Director General of the Technical Development Headquarters of PNC.

## VI. Reference

- 1) Y. Tanaka, I. Suzuki: Development of Two-Dimensional Shielding Design Code (I)--Code Manual--, J213 70 1-1 (1970).
- 2) Y. Tanaka, I. Suzuki: Development of Two-Dimensional Shielding Design Code(II)--Evaluation of Two-Dimensional Analysis--, J213 70 1-2 (1970).
- 3) Y. Tanaka, I. Suzuki and N. Ohtani: Development of Two-Dimensional Shielding Design Code (II)--Group Constant Collapsing Method and Establishment of Core Boundary Connection Method--, J213 71-02 (1971).
- 4) Y. Tanaka, I. Suzuki and N. Ohtani: Evaluation and Application Study of Fast Reactor's Two-Dimensional Shielding Design Code, J213 72-03-01 (1972).
- 5) Y. Tanaka, I. Suzuki and N. Ohtani: Evaluation and Application Study of Fast Reactor's Twp-Dimensional Shielding Design Code--Code Manual--, J213 72-03-02 (1972).
- 6) Y. Tanaka, I. Suzuki and M. Takemura: Improvement and Evaluation of Two-Dimensional Shielding Design Code, J213 73-01 (1973).
- 7) Von H. Vossebrecker: Two-dimensional shielding calculations using a combination of diffusion theory and line of sight method, Atomkernenergie (ATKE) Bd. 19 (1972).

## VII. Appendix Code Manual of DIAC Code

### 1) Outline

As described in previous chapter, following two modification has been done in DIAC.

- i) Line of sight calculation in axial symmetrical void geometry.
- ii) Removalent of minus neutron current at void surface.

Its elemental style has not been changed, so DIAC can be used in the same way as old two-dimensional calculation system of RASC-2D.

### 2) Calculation Flow

Fig. (VII-1) shows the Flow Diagram of DIAC code.

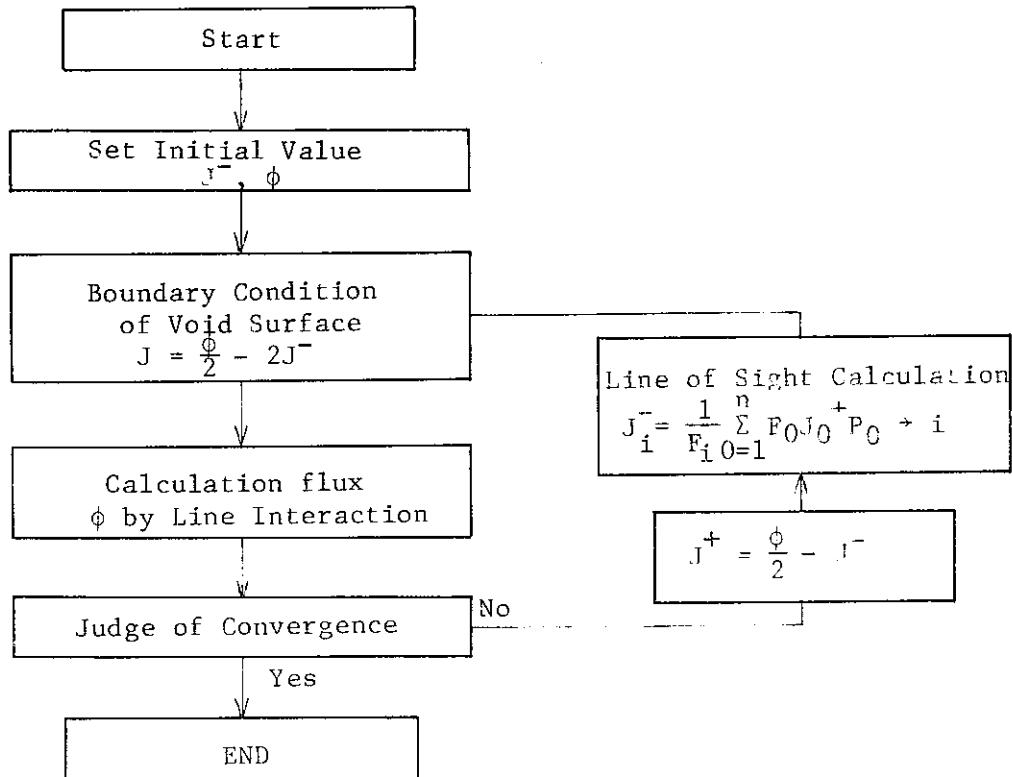


Fig. (VII-1) Flow Diagram of DIAC Code

#### 4) Format of Input and Output

Table (VII-2) shows the Input Format of DIAC Code after Modification. Output format is same as previous one.

#### 5) Test Calculation

Fig. (VII-2) shows the test calculation configuration of DIAC code. Table (VII-2) shows input data, and Table (VII-4) shows some of the output.

Probability Calculation	Gauss Division Points Approximate Polynomial Order	$\leq 16$ $\leq 14$
Line of Sight Calculation	Void Mesh Number Void Number Void Geometry	$\leq 85$ $\leq 2$ Cylinder, Cylindrical Shell
Diffusion Calculation	Mesh Number (P x Z) Region Number R Z Material Number Energy Group Number	$\leq 10140$ $\leq 16$ $\leq 13$ $\leq 26$ $\leq 10$

Table (VII-1) Limit Condition of DIAC Code

**CONTROL DATA 3600 DATA INPUT FORM**

Table (VII-2) DIAC Input Format (1)

Table (VII-2) DIAC Input Format (1)									
TITLE(I), I=1,10	Title Card								FORMAT(10A8)
MATREGS(1)	MATREGS(1).....Put 8 characters as GEOMETRY.								FORMAT(A8)
MATREGS(1)	MATREGS(1).....Put 8 characters as bX-Ybbbb when xy geometry, or bR-Zbbbb when RZ geometry.								FORMAT(A8)
IGROUP	IG	IGROUP.....Put 8 characters as GROUPSbb.							FORMAT(A8, 8X, I4)
	IG	.....Put maximum energy group number.							
MATREGS(I), I=1,2	MR	MATREGS(I).....Put 16 characters as MATERIALbbbbbbbb.							FORMAT(2A8,I4)
	MR	.....Put maximum material number.							
IACCUR	A1	A2	A3						FORMAT(A8,2X,3E10.3)
		IACCUR.....Put 8 characters as ACCURACY							
	A1,A2,A3	.....Put Convergence accuracy condition, (A1 > A2 > A3). final accuracy = A3.							
IOMEGA	W	IOMEGA.....Put 8 characters as OMEGAbbb.							FORMAT(A8,2X,3E10.3)
	W	.....Put Initial value of accerelation parameter							
ICOUNT	IT	NIT	ICOUNT.....Put 8 characters as COUNTbbb.						FORMAT(A8,2X,2I5)
	IT	.....Put muximum iteration number.							
	NIT	.....Put iteration times with use same accerelation parameter.							

**DATA DESCRIPTION**

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DATE \_\_\_\_\_ PAGE \_\_\_\_\_ OF \_\_\_\_\_

# **CONTROL DATA 3600 DATA INPUT FORM**

Table (VII-2) DIAC Input Format (2)

Table (VII-2) DIAC Input Format (2)	
1    2    3    4    5    6    7    8    9    10    11    12    13    14    15    16    17    18    19    20    21    22    23    24    25    26    27    28    29    30    31    32    33    34    35    36    37    38    39    40    41    42    43    44    45    46    47    48    49    50    51    52    53    54    55    56    57    58    59    60    61    62    63    64    65    66    67    68    69    70    71    72    73    74    75    76    77    78    79    80	0
ISOURCE(I), I=1, 3	FORMAT (3A8)
ISOURCE(I) .... Put 24 characters as when removal source is inputed.	
REMOVALSOURCESbbbbbbbbb or when removal source is not inputed.	
NObREMOVALbSOURCERSbbbb	
ISOURCE(I), I=1, 3	KGR
ISOURCE(I) .... Put 24 characters as GROUPSbOfREMOVALbbbbbbb.	
KGR	.... Put maximum energy group number of removal source (KGR<30)
MESH	1 = R(2)
J	= R(J+1)
JME	= R(JME+1)
MESH	Put 8 characters as MESHbbbbMR on first card. On the second card and thereafter, blank card is desirable.
J	* on the last card from the first column.
R(J)	Put mesh numbers in the boundaries of different mesh width, (begins from J=1).
JME	Put distance from the original point at J mesh number (radial direction or horizontal direction) (cm). R(2) proportionates with the original point.
	It is easier understandable to put = in 13 column.

**DATA DESCRIPTION**

PREPARED BY

DATE \_\_\_\_\_ PAGE \_\_\_\_\_ OF \_\_\_\_\_

**CONTROL DATA 3600 DATA INPUT FORM**

Table (VII-2) DIAC Input Format (3)

#### **DATA DESCRIPTION**

634552863 8

**CONTROL DATA 3600 DATA UNIT**

Table (VII-2) DIAC Input Format (4)

**DATA L'ESCAPTION**

**RECORDED BY**

DATE \_\_\_\_\_ PAGE \_\_\_\_\_ OF \_\_\_\_\_

# **CONTROL DATA 3600 DATA INPUT FORM**

Table (VII-2) DIAC INPUT FORMAT (5)

#### DATA DESCRIPTION

PREPARED BY

 Century Research Center Corporation

**C. DATA 3600 DATA INPUT FORM**

Table (VII-2) DIAC Input Format (6)

#### **DATA DESCRIPTION**

PREPARED BY

DATE \_\_\_\_\_ PAGE \_\_\_\_\_ OF \_\_\_\_\_

**CONTROL DATA 3600 DATA INPUT FORM**

Table (VII-2) DIAC Input Format (7)

Table (VII-2) DIAC Input Format (7)	
1    2    3    4    5    E    7    8    9    10	11    12    13    14    15    16    17    18    19
24    25    26    27    28    29    30	31    32    33    34    35    36    37    38    39    40    41    42    43    44    45    46    47    48    49    50
51    52    53    54    55    56    57    58    59    60	61    62    63    64    65    66    67    68    69    70
71    72    73    74    75    76    77    78    79    80	
MBK (1)    MBK (2)	MBK (IBK)
MBK (I)	Put designated number of material in consideration of leak. FORMAT(10I5)
BK1(1)    BK2(1)    BK3(1)	The leak effect is expressed by 2-degree formula as to perpendicular direction co-ordinate Z. FORMAT(3E10.3)
BK1(IBK)    BK2(IBK)    BK3(IBK)	
Leak = $BK1(MT) + BK2(MT) * Z + BK3(MT) * Z^2 * D(MT)$ MT = designated material number. D(MT) = diffusion constants.	
VOID(I), I=1, 4	FORMAT(4A8)
VOID(I)	Put 32 characters as VOIDbCALCULATIONbCONSTANTbbbbbbb.
GAUSS(I), I=1, 3	FORMAT(3A8)
GAUSS(I)	Put 24 characters as GAUSSbDIVISIONbNUMBERbb
NION(L)    NICN(1)    NOCN(1)    NCIN(1)    NCON(1)	FORMAT(5I5)
NION(NVOID)    NICN(NVOID)    NOCN(NVOID)    NCIN(NVOID)    NCON(NVOID)	
NVOID	Void number Inner Side $\leftrightarrow$ Outer Side Inner Side $\leftrightarrow$ Edge Outer Side $\leftrightarrow$ Edge Outer Side $\leftrightarrow$ Outer Side Edge $\leftrightarrow$ Edge
NION(N)	Put gauss division
NICN(N)	number of Nth Void
NOCN(N)	for probability calculation
NCIN(N)	
NCON(N)	

## DATA DESCRIPTION

PREPARED BY

# **CONTROL DATA 3600 DATA INPUT FORM**

Table (VII-2) DIAC Input Format (8)

Table (VII-2) DIAC Input Format (8)		
1 2 3 4 5 6 7 8 9 10	11 12 13 14 15 16 17 18 19 20	21 22 23 24 25 26 27 28 29 30
31 32 33 34 35 36 37 38 39 40	41 42 43 44 45 46 47 48 49 50	51 52 53 54 55 56 57 58 59 60
61 62 63 64 65 66 67 68 69 70	71 72 73 74 75 76 77 78 79 80	
ORDER(I), I=1,4	ORDER(I) .....	Put 32 characters as APPROXIMATE POLYNOMIAL BORDER
MION(1) MICN(1) MOCN(1) MOON(1)	MCCN(1)	
↓ ↓ ↓ ↓		
MION(NVOID) MICN(NVOID) MOCN(NVOID) MOON(NVOID)	MCCN(NVOID)	
MION(N) MICN(N)	..... Put Approximate polynomial order of probability fitting for Nth Vbid.	
MOCN(N)	(order + 1)	
MOON(N)		
MCCN(N)		
BOUND(I), I=1,3	BOUND(I)	Put 24 characters as VOIDbVASEbBOUNDARYbbbbbb.
MSYM-D(I), I=1,2	MSYM-D(I) .....	Put 16 characters as when consider symmetry of void bbSYMMETRYbbbbbb of neutron flux at bottom of void when consider bbASYMETRYbbbbbb when consider other cases bbSYMMETRY or Other any 16 characters

#### DATA DESCRIPTION

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PAGE        OF

**CONTROL DATA 3600 DATA INPUT FORM**

Table (VII-2) DIAC Input Format (8)

**DATA DESCRIPTION**

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DA FORM 1 PAGE 9 OF

Table(VII-3) Example of DIAC Input

\*\*\* DIAC 848 VOID SYSTEM TEST CALCULATION BY KUL 4/3/11/84

GEOMETRY

P-Z

GROUPS	1
MATERIAL	2

ACCURACY	0.1	1.01	1.01
----------	-----	------	------

OMEGA	1.5
-------	-----

CHRT	100
------	-----

NO. REMOVAL SOURCES

MESH MR	1 = 0.1
---------	---------

*	21 = 100.0
---	------------

*	VZ 1 = 0.0
---	------------

*	Z1 = 150.0
---	------------

ZONE	2
------	---

1 4 1 9 2 6 5 8 4 1
---------------------

9 ( 21, 1, 26, 11)
--------------------

1 ( 1, 13, 12, 17)
--------------------

SOURCES IN GROUP 1 BY MATERIAL 2

*	1 1.0+15 -2 -1.0 +5
---	---------------------

BOUNDS BR LOW TWOZERO *	21
-------------------------	----

BR HIGH ONEZERO 2 *	21
---------------------	----

BZ LOW TWOZERO *	21
------------------	----

BZ HIGH ONEZERO 3 *	21
---------------------	----

OUTPUT	2 0 2
--------	-------

DE-GROUP	1
----------	---

1 1
-----

VOID CALCULATION CONSTANT

GAUSS DIVISION NUMBER

4 4 4 4 4
-----------

4 4 4 4 4
-----------

APPROXIMATE POLYNOMIAL ORDER

5 5 5 5 10
------------

12 12 12 12 4
---------------

VOID BASE BOUNDARY

ASYMMETRY

SYMMETRY

MACROSCOPIC DATA IN CARD

101		
-----	--	--

3.82	0.0742	0.0152
------	--------	--------

102		
-----	--	--

11.57	0.0744	0.0155
-------	--------	--------

\*\*\* DIAC \*\*\* VOID SYSTEM TEST CALCULATION BY KHI. 48/11/24

GEOOMETRY

R-Z

GROUPS

1

MATERIAL

2

ACCURACY .10E+00 .10E-01 .10E-02

OMEGA .15E+01

CCOUNT 100 5

NO REMOVAL SOURCES

MESH MR 1 = 0.00  
\* 21 = 100.00

MZ 1 = 0.00  
\* 31 = 150.00

ZONE 2

1 ( 1, 1, 5, 4)  
0 ( 21, 1, 26, 11)  
\* 0 ( 1, 13, 13, 17)

RMESH

0.000	5.000	10.000	15.000	20.000	25.000	30.000	35.000	40.000	45.000
50.000	55.000	60.000	65.000	70.000	75.000	80.000	85.000	90.000	95.000
100.000									

ZMESH

0.000	5.000	10.000	15.000	20.000	25.000	30.000	35.000	40.000	45.000
50.000	55.000	60.000	65.000	70.000	75.000	80.000	85.000	90.000	95.000
100.000	105.000	110.000	115.000	120.000	125.000	130.000	135.000	140.000	145.000
150.000									

Table (VII-4) Example of DIAC Output (1)

Table (VII-4) Example of DIAC Output (2)

ZONES		(1,1) MESH AT BOTTOM LEFT	
2	2	2	2
2	2	2	2
2	2	2	2
2	2	2	2
2	2	2	2
2	2	2	2
2	2	2	2
2	2	2	2
0	0	0	0
0	0	0	0
0	0	0	0
0	0	0	0
0	0	0	0
0	0	0	0
2	2	2	2
2	2	2	2
2	2	2	2
2	2	2	2
2	2	2	2
2	2	2	2
2	2	2	2
2	2	2	2
2	2	2	2
2	2	2	2
2	2	2	2
2	2	2	2
2	2	2	2
2	2	2	2
2	2	2	2
2	2	2	2
2	2	2	2
2	2	2	2
1	1	1	2
1	1	1	2
1	1	1	2
1	1	1	2

Table (VII-4) Example of DIAC Output (3)

SOURCES IN GROUP 18Y MATERIALS 2  
\* 1 .100E+16 2 .100E+10

BOUNDS BR LOW TWOZERO \* \* 31 -0. -0.  
BR HIGH ONEZERO 3 \* 31 .50E+00 -0.  
BZ LOW TWOZERO \* \* 21 -0. -0.  
BZ HIGH ONEZERO 3 \* 21 .50E+00 -0.

OUTPUT 2 0 2

DE-GROUP 1

1 1

Table (VII-4) Example of DIAC Output (4)

VOID CALCULATION CONSTANT

GAUSS DIVISION NUMBER

4	4	4	4	4
4	4	4	4	4

APPROXIMATE POLYNOMIAL ORDER

5	5	5	5	10
12	12	12	12	4

VOID BASE BOUNDARY

ASYMMETRY

SYMMETRY

Table (VII-4) Example of DIAC Output (5)

MACROSCOPIC DATA IN CARD

MATERIAL	DIFF.COEFF.	REM.C.S.	SL-DWN.C.S.
101	.3820E+01	.7430E-01	.1530E-01
102	.1157E+02	.3440E-01	.1550E-01

Table (VII-4) Table of DIAC Output (6)

	X(I)	X**K*LOG(Y)	ERROR	RESID**2
1	.125E+01	-.121E+02	-.800E-03	
2	.500E+01	-.121E+02	.477E-02	
3	.100E+02	-.120E+02	-.178E-01	
4	.150E+02	-.119E+02	.417E-01	
5	.200E+02	-.117E+02	-.585E-01	
6	.250E+02	-.115E+02	.386E-01	
7	.300E+02	-.113E+02	.192E-01	
8	.350E+02	-.111E+02	-.716E-01	
9	.400E+02	-.109E+02	.797E-01	
10	.450E+02	-.108E+02	-.514E-01	
11	.500E+02	-.108E+02	.208E-01	
12	.550E+02	-.112E+02	-.537E-02	
13	.588E+02	-.124E+02	.839E-03	
				.2193912E-05

P-FITTING

M= 12 L= 1 N= 13

	X(I)	X**K*LOG(Y)	ERROR	RESID**2
1	.125E+01	-.114E+02	-.682E-03	
2	.500E+01	-.114E+02	.407E-02	
3	.100E+02	-.113E+02	-.152E-01	
4	.150E+02	-.113E+02	.357E-01	
5	.200E+02	-.112E+02	-.501E-01	
6	.250E+02	-.110E+02	.331E-01	
7	.300E+02	-.109E+02	.162E-01	
8	.350E+02	-.107E+02	-.611E-01	
9	.400E+02	-.106E+02	.681E-01	
10	.450E+02	-.105E+02	-.440E-01	
11	.500E+02	-.106E+02	.178E-01	
12	.550E+02	-.110E+02	-.457E-02	
13	.588E+02	-.121E+02	.767E-03	
				.1603700E-05

P-FITTING

M= 12 L= 2 N= 13

	X(I)	X**K*LOG(Y)	ERROR	RESID**2
1	.125E+01	-.114E+02	-.675E-03	
2	.500E+01	-.114E+02	.402E-02	
3	.100E+02	-.113E+02	-.150E-01	
4	.150E+02	-.113E+02	.352E-01	
5	.200E+02	-.112E+02	-.494E-01	
6	.250E+02	-.110E+02	.326E-01	
7	.300E+02	-.109E+02	.162E-01	
8	.350E+02	-.107E+02	-.696E-01	
9	.400E+02	-.105E+02	.674E-01	
10	.450E+02	-.104E+02	-.435E-01	
11	.500E+02	-.103E+02	.176E-01	
12	.550E+02	-.106E+02	-.453E-02	
13	.588E+02	-.117E+02	.745E-03	
				.1569138E-05

Table(VII-4) Example of DIAC Output (7)

M= 10 L= 11 N= 11

X(I)	X**K*LOG(Y)	ERROR	RESID**2
1 .125E+01	-.107E+02	-.385E-05	
2 .500E+01	-.107E+02	.104E-04	
3 .100E+02	-.106E+02	.125E-04	
4 .150E+02	-.105E+02	-.136E-03	
5 .200E+02	-.104E+02	.376E-03	
6 .250E+02	-.102E+02	.585E-03	
7 .300E+02	-.100E+02	.593E-03	
8 .350E+02	-.982E+01	-.383E-03	
9 .400E+02	-.968E+01	.165E-03	
10 .450E+02	-.960E+01	-.448E-04	
11 .488E+02	-.961E+01	.958E-05	
			.1018867E-09

P-FITTING

M= 12 L= 1 N= 13

X(I)	X**K*LOG(Y)	ERROR	RESID**2
1 .125E+01	-.915E+01	-.426E-04	
2 .500E+01	-.924E+01	.230E-03	
3 .100E+02	-.949E+01	-.105E-02	
4 .150E+02	-.984E+01	.349E-02	
5 .200E+02	-.102E+02	-.705E-02	
6 .250E+02	-.106E+02	.773E-02	
7 .300E+02	-.110E+02	-.203E-02	
8 .350E+02	-.114E+02	.651E-02	
9 .400E+02	-.118E+02	.105E-01	
10 .450E+02	-.121E+02	-.801E-02	
11 .500E+02	-.124E+02	.362E-02	
12 .550E+02	-.127E+02	-.102E-02	
13 .588E+02	-.129E+02	.162E-03	
			.3570359E-07

P-FITTING

M= 12 L= 2 N= 13

X(I)	X**K*LOG(Y)	ERROR	RESID**2
1 .125E+01	-.924E+01	.258E-04	
2 .500E+01	-.931E+01	-.145E-03	
3 .100E+02	-.950E+01	.496E-03	
4 .150E+02	-.979E+01	-.108E-02	
5 .200E+02	-.102E+02	.146E-02	
6 .250E+02	-.106E+02	.105E-02	
7 .300E+02	-.110E+02	-.367E-04	
8 .350E+02	-.113E+02	.980E-03	
9 .400E+02	-.117E+02	-.115E-02	
10 .450E+02	-.120E+02	.730E-03	
11 .500E+02	-.124E+02	-.301E-03	
12 .550E+02	-.127E+02	.561E-04	
13 .588E+02	-.129E+02	.354E-04	
			.7582590E-09

Table(VII-4) Example of DIAC Output(8)

	X(I)	X**K*LOG(Y)	ERROR	RESID**2
1	.125E+01	-.769E+01	.232E-05	
2	.500E+01	-.769E+01	.270E-04	
3	.100E+02	-.790E+01	-.273E-03	
4	.150E+02	-.821E+01	.103E-02	
5	.200E+02	-.857E+01	-.223E-02	
6	.250E+02	-.897E+01	.311E-02	
7	.300E+02	-.936E+01	-.291E-02	
8	.350E+02	-.975E+01	.184E-02	
9	.400E+02	-.101E+02	-.766E-03	
10	.450E+02	-.105E+02	.210E-03	
11	.488E+02	-.107E+02	-.341E-04	
				.2827319E-08

P-FITTING

M= 10 L= 2 N= 11

	X(I)	X**K*LOG(Y)	ERROR	RESID**2
1	.125E+01	-.769E+01	-.168E-05	
2	.500E+01	-.776E+01	.395E-04	
3	.100E+02	-.794E+01	-.273E-03	
4	.150E+02	-.822E+01	.937E-03	
5	.200E+02	-.857E+01	-.194E-02	
6	.250E+02	-.895E+01	.263E-02	
7	.300E+02	-.933E+01	-.242E-02	
8	.350E+02	-.972E+01	.151E-02	
9	.400E+02	-.101E+02	-.625E-03	
10	.450E+02	-.104E+02	.170E-03	
11	.488E+02	-.107E+02	-.268E-04	
				.2020598E-08

P-FITTING

M= 10 L= 3 N= 11

	X(I)	X**K*LOG(Y)	ERROR	RESID**2
1	.125E+01	-.790E+01	-.957E-05	
2	.500E+01	-.794E+01	.718E-04	
3	.100E+02	-.807E+01	-.334E-03	
4	.150E+02	-.828E+01	.965E-03	
5	.200E+02	-.856E+01	-.181E-02	
6	.250E+02	-.890E+01	.232E-02	
7	.300E+02	-.927E+01	-.204E-02	
8	.350E+02	-.964E+01	.123E-02	
9	.400E+02	-.100E+02	-.499E-03	
10	.450E+02	-.104E+02	.133E-03	
11	.488E+02	-.106E+02	-.207E-04	
				.1566137E-08

P-FITTING

M= 10 L= 4 N= 11

Table (VII-4) Example of DIAC-Output (9)

P-FITTING

M= 5 L= 1 N= 6

	X(I)	X**K*LOG(Y)	ERROR	RESID**2
1	.125E+01	-.101E+02	-.465E+00	
2	.500E+01	-.102E+02	.170E+01	
3	.100E+02	-.103E+02	-.291E+01	
4	.150E+02	-.107E+02	.300E+01	
5	.200E+02	-.113E+02	-.167E+01	
6	.238E+02	-.127E+02	.468E+00	
				.2352807E-02

P-FITTING

M= 5 L= 2 N= 6

	X(I)	X**K*LOG(Y)	ERROR	RESID**2
1	.125E+01	-.101E+02	-.465E+00	
2	.500E+01	-.102E+02	.170E+01	
3	.100E+02	-.103E+02	-.291E+01	
4	.150E+02	-.106E+02	.299E+01	
5	.200E+02	-.113E+02	-.167E+01	
6	.238E+02	-.126E+02	.467E+00	
				.2350398E-02

P-FITTING

M= 5 L= 3 N= 6

	X(I)	X**K*LOG(Y)	ERROR	RESID**2
1	.125E+01	-.101E+02	-.464E+00	
2	.500E+01	-.101E+02	.169E+01	
3	.100E+02	-.103E+02	-.290E+01	
4	.150E+02	-.106E+02	.299E+01	
5	.200E+02	-.112E+02	-.167E+01	
6	.238E+02	-.126E+02	.467E+00	
				.2343189E-02

P-FITTING

M= 5 L= 4 N= 6

	X(I)	X**K*LOG(Y)	ERROR	RESID**2
1	.125E+01	-.101E+02	-.463E+00	
2	.500E+01	-.101E+02	.169E+01	
3	.100E+02	-.102E+02	-.289E+01	
4	.150E+02	-.105E+02	.298E+01	
5	.200E+02	-.111E+02	-.166E+01	

Table(VII-4) Example of DIAC Output (0)

	X(I)	X**K*LOG(Y)	ERROR	RESID**2
1	.125E+01	-.114E+02	.577E-04	
2	.500E+01	-.115E+02	-.322E-03	
3	.100E+02	-.116E+02	.137E-02	
4	.150E+02	-.117E+02	-.413E-02	
5	.200E+02	-.118E+02	.782E-02	
6	.250E+02	-.119E+02	-.831E-02	
7	.300E+02	-.121E+02	.242E-02	
8	.350E+02	-.122E+02	.606E-02	
9	.400E+02	-.123E+02	-.991E-02	
10	.450E+02	-.124E+02	.758E-02	
11	.500E+02	-.125E+02	-.340E-02	
12	.550E+02	-.127E+02	.969E-03	
13	.588E+02	-.128E+02	-.135E-03	
				.3600592E-07

P-FITTING

M= 12      L= 2      N= 13

	X(I)	X**K*LOG(Y)	ERROR	RESID**2
1	.125E+01	-.115E+02	-.174E-03	
2	.500E+01	-.115E+02	.102E-02	
3	.100E+02	-.116E+02	-.397E-02	
4	.150E+02	-.117E+02	.103E-01	
5	.200E+02	-.118E+02	-.170E-01	
6	.250E+02	-.119E+02	.164E-01	
7	.300E+02	-.120E+02	-.493E-02	
8	.350E+02	-.122E+02	-.938E-02	
9	.400E+02	-.123E+02	.152E-01	
10	.450E+02	-.124E+02	-.113E-01	
11	.500E+02	-.125E+02	.496E-02	
12	.550E+02	-.127E+02	-.133E-02	
13	.588E+02	-.128E+02	.254E-03	
				.1179838E-06

P-FITTING

M= 12      L= 3      N= 13

	X(I)	X**K*LOG(Y)	ERROR	RESID**2
1	.125E+01	-.116E+02	-.405E-03	
2	.500E+01	-.116E+02	.214E-02	
3	.100E+02	-.116E+02	-.934E-02	
4	.150E+02	-.117E+02	.300E-01	
5	.200E+02	-.118E+02	-.593E-01	
6	.250E+02	-.119E+02	.633E-01	
7	.300E+02	-.120E+02	-.128E-01	
8	.350E+02	-.121E+02	-.605E-01	
9	.400E+02	-.123E+02	.931E-01	
10	.450E+02	-.124E+02	-.703E-01	
11	.500E+02	-.125E+02	.317E-01	
12	.550E+02	-.126E+02	-.878E-02	
13	.588E+02	-.127E+02	.155E-02	

Table(VII-4) Example of DIAC Output (II)

	X(I)	X**K*LOG(Y)	ERROR	RESID**2
1	.125E+01	-.128E+02	-.425E-04	
2	.500E+01	-.128E+02	.255E-03	
3	.100E+02	-.127E+02	-.952E-03	
4	.150E+02	-.127E+02	.222E-02	
5	.200E+02	-.126E+02	-.307E-02	
6	.250E+02	-.126E+02	.192E-02	
7	.300E+02	-.125E+02	.124E-02	
8	.350E+02	-.124E+02	-.404E-02	
9	.400E+02	-.123E+02	.439E-02	
10	.450E+02	-.123E+02	-.280E-02	
11	.500E+02	-.122E+02	.113E-02	
12	.550E+02	-.121E+02	-.289E-03	
13	.588E+02	-.120E+02	.377E-04	
				.6532871E-08

P-FITTING

M= 4 L= 1 N= 5

	X(I)	X**K*LOG(Y)	ERROR	RESID**2
1	.125E+01	-.129E+02	.390E+00	
2	.500E+01	-.125E+02	-.119E+01	
3	.100E+02	-.123E+02	.162E+01	
4	.150E+02	-.122E+02	-.119E+01	
5	.187E+02	-.122E+02	.390E+00	
				.5732252E-03

P-FITTING

M= 4 L= 2 N= 5

	X(I)	X**K*LOG(Y)	ERROR	RESID**2
1	.125E+01	-.125E+02	.232E+00	
2	.500E+01	-.122E+02	-.706E+00	
3	.100E+02	-.121E+02	.959E+00	
4	.150E+02	-.120E+02	-.706E+00	
5	.187E+02	-.120E+02	.232E+00	
				.2021633E-03

P-FITTING

M= 4 L= 3 N= 5

	X(I)	X**K*LOG(Y)	ERRCR	RESID**2
1	.125E+01	-.123E+02	.169E+00	
2	.500E+01	-.121E+02	-.517E+00	
3	.100E+02	-.120E+02	.701E+00	
4	.150E+02	-.120E+02	-.517E+00	
5	.187E+02	-.119E+02	.169E+00	
				.1081745E-03

P-FITTING

Table (VII-4) Example of DIAC Output (12)

ITERATION PARAMETERS FOR GROUP 1

RESIDUAL	LAMBDA	OMEGA	DIFF	PRODIF	THE PROPORTIONAL DIFFERENCE CRITERION IS SATISFIED AS FAR AS MESH LINE.....			
.1824390E+17	.3816310E+00	1.500000000	.1637367E+16	.9958554E+00		0	0	0
.9093266E+16	.4984277E+00	1.500000000	.3649363E+15	.7872491E+00		0	0	0
.5785027E+16	.6361440E+00	1.500000000	.2919330E+15	.7203373E+00		0	0	0
.3196195E+16	.5524944E+00	1.500000000	.1672903E+15	.6336648E+00		0	0	0
.25039L25E+16	.7852782E+00	1.500000000	.1815062E+15	.5331325E+00		0	0	0
.1556278E+16	.6200553E+00	1.546781178	.4883282E+14	.4322495E+00		0	0	0
.1018128E+16	.6542970E+00	1.546781178	.1669177E+14	.3398014E+00		0	0	0
.6415736E+15	.6301532E+00	1.546781178	.1620144E+14	.2488395E+00		0	0	0
.3699120E+15	.5609432E+00	1.546781178	.5687634E+13	.1679761E+00		0	0	0
.2168054E+15	.6023477E+00	1.546781178	.3712971E+13	.1034504E+00		0	0	0
.1231536E+15	.5680376E+00	1.548991140	.2067030E+13	.5858732E-01		31	0	0
.7023844E+14	.5703320E+00	1.548991140	.1503074E+13	.3158435E-01		31	0	0
.3437849E+14	.4994541E+00	1.548991140	.6087220E+12	.1648307E-01		31	0	0
.1946840E+14	.5062966E+00	1.548991140	.3150507E+12	.7958058E-02		31	31	0
.1131348E+14	.5811199E+00	1.548991140	.1657544E+12	.3340814E-02		31	31	0
.6266281E+13	.5534465E+00	1.549756167	.8803896E+11	.1330692E-02		31	31	0
.3728265E+13	.5949630E+00	1.549756167	.4840635E+11	.6806722E-03		31	31	31

Table (III-4) Example of DIAC Output (3)

FLUXES IN GROUP 1 FOR COLUMNS 1 TO 10												
	Z / R	0.00	5.00	10.00	15.00	20.00	25.00	30.00	35.00	40.00	45.00	
1	0.00	.889E+16	.841E+16	.678E+16	.338E+16	.213E+16	.137E+16	.907E+15	.610E+15	.417E+15	.290E+15	
2	5.00	.874E+16	.827E+16	.668E+16	.331E+16	.238E+16	.134E+16	.888E+15	.598E+15	.410E+15	.285E+15	
3	10.00	.819E+16	.777E+16	.630E+16	.311E+16	.194E+16	.125E+16	.832E+15	.563E+15	.388E+15	.272E+15	
4	15.00	.683E+16	.647E+16	.533E+16	.269E+16	.168E+16	.110E+16	.741E+15	.508E+15	.354E+15	.251E+15	
5	20.00	.349E+16	.334E+16	.284E+16	.195E+16	.132E+16	.907E+15	.628E+15	.440E+15	.311E+15	.224E+15	
6	25.00	.215E+16	.205E+16	.177E+16	.135E+16	.987E+15	.711E+15	.510E+15	.367E+15	.265E+15	.194E+15	
7	30.00	.135E+16	.129E+16	.114E+16	.926E+15	.716E+15	.539E+15	.401E+15	.297E+15	.219E+15	.163E+15	
8	35.00	.867E+15	.837E+15	.754E+15	.636E+15	.513E+15	.401E+15	.308E+15	.234E+15	.177E+15	.134E+15	
9	40.00	.569E+15	.553E+15	.516E+15	.439E+15	.366E+15	.295E+15	.233E+15	.182E+15	.141E+15	.109E+15	
10	45.00	.381E+15	.371E+15	.344E+15	.305E+15	.260E+15	.216E+15	.175E+15	.139E+15	.110E+15	.864E+14	
11	50.00	.258E+15	.253E+15	.237E+15	.213E+15	.186E+15	.157E+15	.130E+15	.106E+15	.850E+14	.678E+14	
12	55.00	.177E+15	.174E+15	.164E+15	.150E+15	.133E+15	.114E+15	.962E+14	.796E+14	.651E+14	.527E+14	
13	60.00	.123E+15	.121E+15	.115E+15	.106E+15	.950E+14	.830E+14	.710E+14	.597E+14	.495E+14	.406E+14	
14	65.00	.861E+14	.849E+14	.811E+14	.754E+14	.683E+14	.604E+14	.523E+14	.446E+14	.375E+14	.311E+14	
15	70.00	.608E+14	.603E+14	.576E+14	.539E+14	.492E+14	.440E+14	.386E+14	.333E+14	.223E+14	.238E+14	
16	75.00	.432E+14	.427E+14	.412E+14	.387E+14	.357E+14	.322E+14	.285E+14	.248E+14	.214E+14	.182E+14	
17	80.00	.310E+14	.307E+14	.297E+14	.281E+14	.260E+14	.236E+14	.211E+14	.186E+14	.161E+14	.138E+14	
18	85.00	.225E+14	.223E+14	.216E+14	.205E+14	.191E+14	.175E+14	.158E+14	.140E+14	.122E+14	.106E+14	
19	90.00	.166E+14	.164E+14	.160E+14	.152E+14	.142E+14	.131E+14	.119E+14	.106E+14	.935E+13	.815E+13	
20	95.00	.125E+14	.124E+14	.121E+14	.116E+14	.109E+14	.100E+14	.914E+13	.821E+13	.727E+13	.636E+13	
21	100.00	.988E+13	.979E+13	.953E+13	.912E+13	.858E+13	.795E+13	.726E+13	.654E+13	.583E+13	.511E+13	
22	105.00	0.	0.	0.	0.	0.	0.	0.	0.	0.	0.	
23	110.00	0.	0.	0.	0.	0.	0.	0.	0.	0.	0.	
24	115.00	0.	0.	0.	0.	0.	0.	0.	0.	0.	0.	
25	120.00	0.	0.	0.	0.	0.	0.	0.	0.	0.	0.	
26	125.00	.527E+13	.523E+13	.511E+13	.491E+13	.465E+13	.431E+13	.392E+13	.347E+13	.298E+13	.245E+13	
27	130.00	.385E+13	.382E+13	.374E+13	.359E+13	.340E+13	.316E+13	.287E+13	.255E+13	.220E+13	.183E+13	
28	135.00	.284E+13	.281E+13	.275E+13	.265E+13	.251E+13	.233E+13	.212E+13	.189E+13	.164E+13	.139E+13	
29	140.00	.211E+13	.210E+13	.205E+13	.197E+13	.187E+13	.174E+13	.159E+13	.142E+13	.124E+13	.106E+13	
30	145.00	.161E+13	.160E+13	.156E+13	.150E+13	.143E+13	.133E+13	.122E+13	.109E+13	.959E+12	.827E+12	
31	150.00	.127E+13	.126E+13	.123E+13	.119E+13	.113E+13	.105E+13	.961E+12	.864E+12	.761E+12	.658E+12	

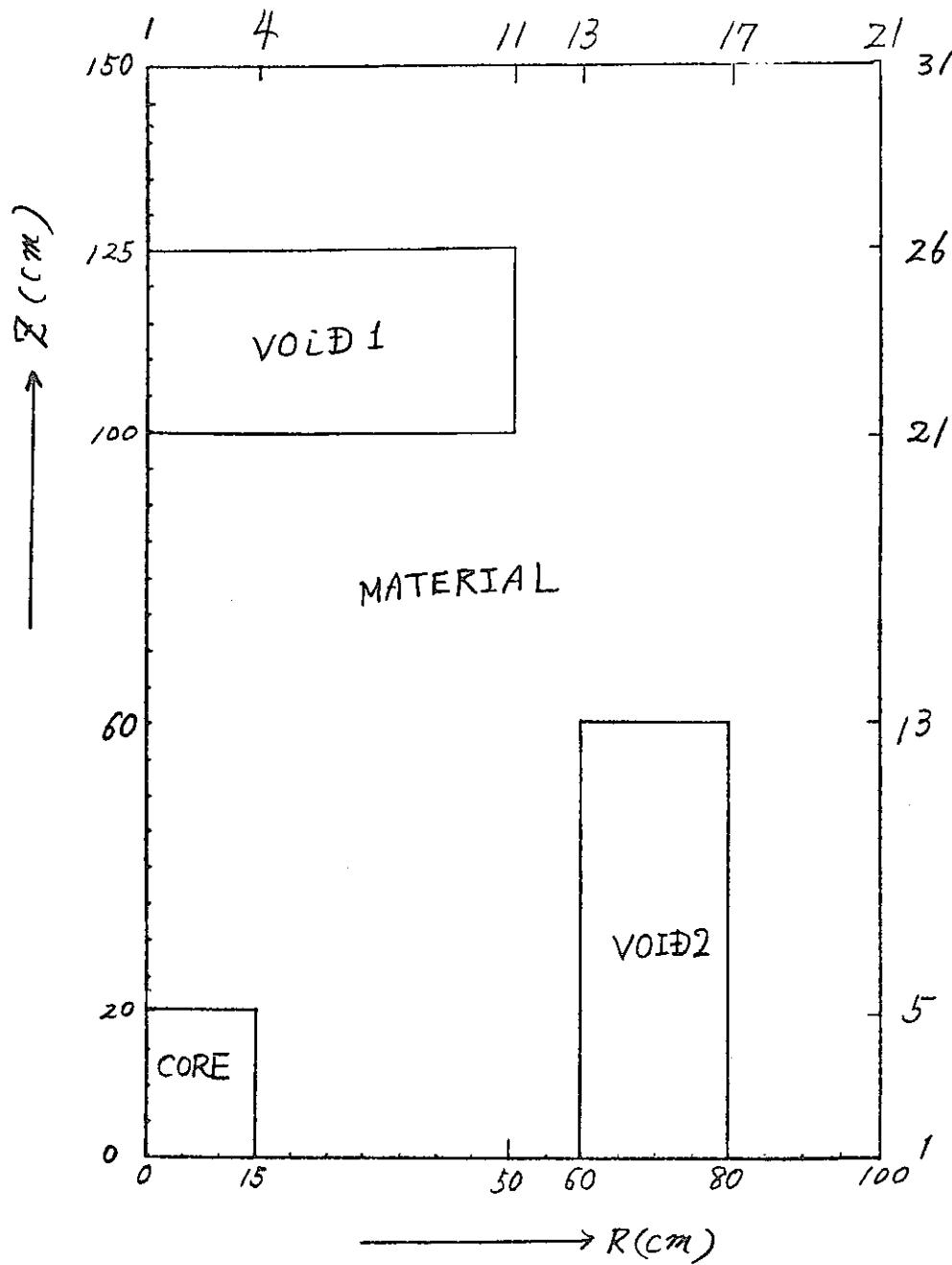
Table (VII-4) Example of DIAC Output (14)

FLUXES IN GROUP 1 FOR COLUMNS 11 TO 20

Z / R	50.00	55.00	60.00	65.00	70.00	75.00	80.00	85.00	90.00	95.00
1 0.00	.205E+15	.148E+15	.105E+15	0.	0.	0.	.515E+14	.404E+14	.312E+14	.245E+14
2 5.00	.203E+15	.148E+15	.113E+15	0.	0.	0.	.560E+14	.413E+14	.313E+14	.245E+14
3 10.00	.195E+15	.144E+15	.112E+15	0.	0.	0.	.563E+14	.413E+14	.310E+14	.242E+14
4 15.00	.181E+15	.135E+15	.136E+15	0.	0.	0.	.547E+14	.402E+14	.302E+14	.235E+14
5 20.00	.163E+15	.123E+15	.967E+14	0.	0.	0.	.519E+14	.383E+14	.288E+14	.225E+14
6 25.00	.143E+15	.109E+15	.860E+14	0.	0.	0.	.482E+14	.357E+14	.270E+14	.211E+14
7 30.00	.123E+15	.942E+14	.746E+14	0.	0.	0.	.438E+14	.327E+14	.248E+14	.194E+14
8 35.00	.103E+15	.796E+14	.633E+14	0.	0.	0.	.391E+14	.293E+14	.223E+14	.175E+14
9 40.00	.942E+14	.650E+14	.527E+14	0.	0.	0.	.342E+14	.257E+14	.197E+14	.155E+14
10 45.00	.679E+14	.538E+14	.431E+14	0.	0.	0.	.292E+14	.221E+14	.175E+14	.134E+14
11 50.00	.540E+14	.432E+14	.348E+14	0.	0.	0.	.241E+14	.184E+14	.143E+14	.113E+14
12 55.00	.425E+14	.342E+14	.277E+14	0.	0.	0.	.191E+14	.148E+14	.117E+14	.934E+13
13 60.00	.331E+14	.266E+14	.209E+14	.168E+14	.154E+14	.146E+14	.139E+14	.114E+14	.925E+13	.750E+13
14 65.00	.256E+14	.209E+14	.170E+14	.141E+14	.124E+14	.113E+14	.102E+14	.867E+13	.720E+13	.591E+13
15 70.00	.198E+14	.164E+14	.136E+14	.114E+14	.990E+13	.874E+13	.771E+13	.662E+13	.557E+13	.441E+13
16 75.00	.153E+14	.128E+14	.118E+14	.913E+13	.783E+13	.681E+13	.593E+13	.509E+13	.430E+13	.354E+13
17 80.00	.114E+14	.998E+13	.844E+13	.718E+13	.616E+13	.532E+13	.459E+13	.394E+13	.333E+13	.273E+13
18 85.00	.908E+13	.775E+13	.660E+13	.564E+13	.483E+13	.416E+13	.357E+13	.305E+13	.258E+13	.216E+13
19 90.00	.703E+13	.603E+13	.516E+13	.442E+13	.379E+13	.325E+13	.278E+13	.238E+13	.201E+13	.168E+13
20 95.00	.547E+13	.473E+13	.404E+13	.347E+13	.297E+13	.255E+13	.218E+13	.186E+13	.157E+13	.131E+13
21 100.00	.424E+13	.369E+13	.315E+13	.273E+13	.234E+13	.200E+13	.171E+13	.145E+13	.123E+13	.102E+13
22 105.00	.336E+13	.293E+13	.253E+13	.216E+13	.185E+13	.158E+13	.135E+13	.114E+13	.964E+12	.803E+12
23 110.00	.226E+13	.241E+13	.204E+13	.174E+13	.148E+13	.126E+13	.107E+13	.904E+12	.761E+12	.633E+12
24 115.00	.246E+13	.262E+13	.168E+13	.141E+13	.119E+13	.101E+13	.851E+12	.718E+12	.603E+12	.501E+12
25 120.00	.216E+13	.171E+13	.139E+13	.115E+13	.963E+12	.809E+12	.682E+12	.574E+12	.460E+12	.399E+12
26 125.00	.187E+13	.144E+13	.116E+13	.945E+12	.783E+12	.654E+12	.548E+12	.460E+12	.385E+12	.319E+12
27 130.00	.147E+13	.117E+13	.943E+12	.770E+12	.636E+12	.530E+12	.443E+12	.371E+12	.309E+12	.256E+12
28 135.00	.114E+13	.933E+12	.761E+12	.624E+12	.516E+12	.429E+12	.358E+12	.299E+12	.250E+12	.207E+12
29 140.00	.892E+12	.739E+12	.610E+12	.504E+12	.418E+12	.347E+12	.290E+12	.242E+12	.202E+12	.167E+12
30 145.00	.701E+12	.587E+12	.438E+12	.406E+12	.337E+12	.281E+12	.235E+12	.196E+12	.164E+12	.136E+12
31 150.00	.560E+12	.471E+12	.393E+12	.327E+12	.272E+12	.227E+12	.190E+12	.159E+12	.132E+12	.110E+12

Table (VII-4) Example of DIAC Output (5)

Z / R	1.00.00
1 0.00	201E+14
2 5.00	200E+14
3 10.00	198E+14
4 15.00	193E+14
5 20.00	184E+14
6 25.00	173E+14
7 30.00	159E+14
8 35.00	143E+14
9 40.00	127E+14
10 45.00	110E+14
11 50.00	929E+13
12 55.00	755E+13
13 60.00	614E+13
14 65.00	484E+13
15 70.00	378E+13
16 75.00	293E+13
17 80.00	227E+13
18 85.00	176E+13
19 90.00	137E+13
20 95.00	107E+13
21 100.00	937E+12
22 105.00	657E+12
23 110.00	518E+12
24 115.00	410E+12
25 120.00	327E+12
26 125.00	261E+12
27 130.00	210E+12
28 135.00	169E+12
29 140.00	137E+12
30 145.00	111E+12
31 150.00	898E+11



	Diffusion Constant	Absorbtion Cross-section	Slowing-Down Cross-section
CORE	3.82	0.0743	0.0153
MATERIAL	11.57	0.0344	0.0155

Fig VII-2 Geometry of DiAC Test Calculation